FCC	312	
Sche	dule	S

# FEDERAL COMMUNICATIONS COMMISSION SATELLITE SPACE STATION AUTHORIZATIONS (Technical and Operational Description)

Page 1: General, Frequency Bands, and GSO Orbit

S1. GENERAL INFORMATION Complete for all satellite applicatio	S1.	GENERAL	INFORMATION	Complete for a	II satellite	application
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a. Space Station or Satellite N PROBA-V	etwork Name:	e. Estimated Date of Placement into Service: 11/7/2013		i Will the space station(s) operate on a Common Carrier Basis: N		
b. Construction Commencement 1/1/2011	ent Date:	f. Estimated Lifetime of Satellite(s): 5	Years	j. Number of transponders offered on a comm 0	non carrier basis:	
c. Construction Completion Date: 6/1/2012		g. Total Number of Transponders: 1		k. Total Common Carrier Transponder Bandwidth:  0 MHz		
d1. Est Launch Date Begin: 5/7/2013	d2. Est Launch Date End: 5/7/2013	h. Total Transponder Bandwidth (no. transponde 0	rs x Bandwidth) MHz	I. Orbit Type: Mark all boxes that apply:  GSO X No.	GSO	

S2. OPERATING FREQUENCY BANDS Identify the frequency range and transmit/receive mode for all frequency bands in which this station will oper Also indicate the nature of service(s) for each frequency band.

Frequency Band Limits					f.
Lower Frequency (	_Hz)	Upper Frequency (	_Hz)	e. T/R Mode	Nature of Service(s): List all that apply to this band
a. Numeric	b. Unit (K/M/G)	c. Numeric	d. Unit (K/M/G)		,, .,,
8040.0	M	8140.0	M	Т	Earth Exploration Satellite Service

S3. ORBITAL INFORMATION FOR GEOSTATIONARY SATELLITES ONLY:

Page 2: NGSO Orbits

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#### S4. ORBITAL INFORMATION FOR NON-GEOSTATIONARY SATELLITES ONLY

S4a. Total Number of Satellites in Network or System:

1 S4c. Celestial Reference Body (Earth, Sun, Moon, etc.): E

S4b. Total Number of Orbital Planes in Network or System: 1 S4d. Orbit Epoch Date: 9/8/2014

#### For each Orbital Plane Provide:

	(e) Orbital Plane No.	(f) No. of Satellites in Plane	(g) Inclination Angle (degrees)	(h) Orbital Period (Seconds)	(i) Apogee (km)	(j) Perigee (km)	(k) Right Ascension of the Ascending Node (Deg.)	(I) Argument of Perigee (Degrees)	rvice Arc Rang (n) End Angle	- ( -3/
[	1	1	98.7	6072	826	818	330.1	74.9		

S5. INITIAL SATELLITE PHASE ANGLE For each satellite in each orbital plane, provide the intital phase angle.

(a) Orbital Plane No.	(b) Satellite Number	(c) Initial Phase Angle (Degrees)
1		0

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### S6. SERVICE AREA CHARACTERISTICS for each service area provide:

(a) Service Area ID	(b) Type of Associated Station (Earth or Space)		(d) Service Area Description. Provide list of geographic areas (state postal codes or ITU 3-ltr codes), satellites or Figure No. of Service Area Diagram.
1	E	None	Entire planet science gatthering

Page 3: Service Areas

Page 4: Antenna Beams

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### S7. SPACE STATION ANTENNA BEAM CHARACTERISTICS For each antenna beam provide:

(a)	(b)	Isotropic	Antenna	(e)	(f)	(g) Min.	(h) Polar-	(i) Polarization	(j) Service		Transmit				Receive		
Beam		Ga	ain	Pointing	Rotational	Cross-		Alignment Rel.	Area ID	(k) Input	(I) Effective	(m)	(n)	(o) G/T	(p) Min.	Input Attenu	uator (dB)
ID	Mode	(c) Peak	(d) Edge	Error	Error	Polar Iso-	Switch-	Equatorial		Losses	Output	Max.	System	Max.	Saturation	(q) Max.	(r) Step
		(dBi)	(dBi)	(Degrees)	(Degrees)	lation (dB)		Plane (Degrees)		(dB)	Power (W)	EIRP			Flux Density	Value	Size
							(Y/N)					(dBW)	Temp (k)	(db/K)	(dBW/m2)	Value	CIZO
WBD	Т	3	-1	1	0	13	N	0	1	1	10	13					

Page 5: Beam Diagrams

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S8. ANTENNA BEAM DIAGRAMS For each beam pattern provide the reference to the graphic image and numerical data:

Also provide the power flux density levels in each beam that result from the emission with the highest power flux density.

(a)	(b)	(c) Co-or	( - )	(e) NGSO Antenna Gain	(f) GSO Antenna			lux Density (dB			
Beam	T/R	Cross	Ref.	Contour Description	Gain Contour Data	At Angle of Arrival above horizontal (for emission with highest PFD)					
ID	Mode	Polar Mode ("C" or" X")	Orbital Longitude (Deg. E/W)	(Figure/Table/ Exhibit)	(GXT File)	(g) 5 Deg	(h) 10 Deg	(i) 15 Deg	(j) 20 Deg	(k) 25 Deg	
WBD	Т			′ antenna pattern Exhit	None	-160.2	-160.2	-160.2	-160.2	-160.2	

Page 6: Channels and Transponders

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S9. SPACE STATION CHANNELS For each frequency channel provide: S10. SPACE STATION TRANSPONDERS For each transponder provide:

	(a) Channel No.	(B) Assigned Bandwidth (kHz)	(c) T/R Mode	(d) Center Frequency (MHz)	(e) Polarization (H, V, L, R)	(f) TTC or Comm Channel (T or C)
İ	WBD	100000	T	8090.0	R	Т

(a)	(b)	Receive	Band	Transmit Band		
Transponder ID	Transponder Gain (dB)	(c) Channel No.	(d) Beam ID	(e) Channel No.	(f) Beam ID	
A1	1	WBD	WBD	WBD	WBD	

Page 7: Digital Modulation

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### S11. DIGITAL MODULATION PARAMETERS For each digital emission provide:

(a) Digital Mod. ID	(b) Emission Designator	(c) Assigned Bandwidth (kHz)	(d) No. of Phases	(e)Uncoded Data Rate (kbps)	(f) FEC Error Correction Coding Rate	Processing	(h) Total C/N Performance Objective (dB)	(i) Single Entry C/I Objective (dB)
WBD	100MG2D	100000	4	84448	1	0	0	0

Page 8: Analog Modulation

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### S12. ANALOG MODULATION PARAMETERS For each analog emission provide:

(a)	(b) Emission	(c)	(d) Signal	(e)	Multi-channel Telephony			(j) Video	(k) Video	(I) Video	(m) SCPC/FM	(n) Total C/N	(o) Single	
Analog Mod. ID		Assigned Bandwidth (kHz)	Туре	Channels per Carrier	Companded	(g) Bottom Baseband Freq. (MHz)	(h) Top Baseband Freq. (MHz)	(i) RMS Modulation Index	Standard NTSC, PAL, etc.	Noise- Weighting (dB)	and SCPC/FM Modulation Index	Compander, Preemphasis, and Noise Weighting (dB)	Performance Objective (dB)	Entry C/I Objective (dB)

Page 9: Typical Emissions

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### S13. TYPICAL EMISSIONS For each planned type of emission provide:

Associated				` '	` '	(g)Noise Budget	(h) Energy	Receive Band (Assoc. Transmit Stn)			Tra	nsmit Band	(This Space Station)	
Transpond	er ID Range	(c) Digital (Table S11)	(d) Analog (Table S12)	per Transponder	Spacing (kHz)	Reference (Table No.)	Dispersal Bandwidth	(i)Assoc. Stn. Max.	Assoc. Station		EIRP	(dBW)	(n) Max. Power Flux Density	(o)Assoc. Stn Rec.
(a) Start	(b) End	(Table 511) (Table 512)					(kHz)	Antenna		(- /			(dBW/m2/Hz)	G/T
								Gain (dBi)	(j) Min.	(k) Max.	(I) Min.	(m) Max.	(== :::::=,	(dB/K)
A1	A1	WBD									9	13	-160.2	37.7

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Page 10: TT and C

S14. Is the space station(s) controlled and monitored remotely? If Yes, provide the location and telephone number of the TT and C control point(s): No

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Page 11: Characteristics and Certifications

#### S15. SPACECRAFT PHYSICAL CHARACTERISTICS:

S15a: Mass of spacecraft without fuel (kg): 138	Spacecraft Dimensions (meters)	Probability of Survival to End of Life (0.0 - 1.0)
S15b. Mass of fuel and disposables at launch (kg): 0		
S15c. Mass of spacecraft and fuel at launch (kg): 138	S15f. Length (m): 0.8	S15i. Payload: 0.9
S15d. Mass of fuel, in orbit, at beginning of life (kg): 0	S15g. Width (m): 0.8	S15j. Bus: 0.9
S15e. Deployed Area of Solar Array (square meters): 1.9	S15h. Height (m):	S15k. Total: 0.9

#### S16. SPACECRAFT ELECTRICAL CHARACTERISTICS:

Spacecraft Subsystem		ver (Watts) At ng of Life	Electrical Power (Watts) At End of Life			
	At Equinox	At Solstice	At Equinox	At Solstice		
Payload (Watts):	<sup>(a):</sup> 30	<sup>(f):</sup> 30	<sup>(k):</sup> 30	<sup>(p):</sup> 30		
Bus (Watts):	<sup>(b):</sup> 48	<sup>(g):</sup> 48	<sup>(l):</sup> 48	<sup>(q):</sup> 48		
Total (Watts):	<sup>(c):</sup> 78	<sup>(h):</sup> 78	<sup>(m)</sup> 78	<sup>(r):</sup> 78		
Solar Array (Watts):	<sup>(d):</sup> 150	<sup>(i):</sup> 143	<sup>(n):</sup> 140	<sup>(s):</sup> 133		
Depth of Battery Discharge (%):	<sup>(e)</sup> 35 %	<sup>(j)</sup> 35 %	<sup>(o)</sup> 35 %	<sup>(t)</sup> 35 %		

#### S17. CERTIFICATIONS:

a. Are the power flux density limits of § 25.208 met?:	YES	# NO	# N/A	
b. Are the appropriate service area coverage requirements of § 25.143(b)(ii) and (iii), or § 25.145(c)(1) and (2)	met? YES	# NO	# N/A	
c. Are the frequency tolerances of § 25.202(e) and the out-of-band emission limits of § 25.202(f)(1), (2) and (3	3) met? YES	# NO	# N/A	
In addition to the information required in this Form, the cases station applicant is required to provide	do all the informatio	n chacified in	Saction 25 114	of the

In addition to the information required in this Form, the space station applicant is required to provide all the information specified in Section 25.114 of the Commission's rules, 47 C.F.R § 25.114.