FCC	312	
Sche	dule	S

# FEDERAL COMMUNICATIONS COMMISSION SATELLITE SPACE STATION AUTHORIZATIONS (Technical and Operational Description)

Page 1: General, Frequency Bands, and GSO Orbit

S1. GENERAL INFORMATION Complete for all satellite a	١.	GENERAL	INFORMATION	Complete for all	l satellite application	ıs.
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a. Space Station or Satellite No USASAT 30A	etwork Name:	e. Estimated Date of Placement into Service: 1/3/2008	i Will the space station(s) operate on a Common Carrier Basis:  N
b. Construction Commenceme 8/1/2003	nt Date:	f. Estimated Lifetime of Satellite(s): 7 Years	<ul><li>j. Number of transponders offered on a common carrier basis:</li><li>0</li></ul>
c. Construction Completion Date: 7/20/2007		g. Total Number of Transponders: 0	k. Total Common Carrier Transponder Bandwidth:  0 MHz
d1. Est Launch Date Begin: 9/18/2007	d2. Est Launch Date End: 9/18/2007	h. Total Transponder Bandwidth (no. transponders x Bandw 0 MHz	vidth)  I. Orbit Type: Mark all boxes that apply:  GSO  X  NGSO

S2. OPERATING FREQUENCY BANDS Identify the frequency range and transmit/receive mode for all frequency bands in which this station will oper Also indicate the nature of service(s) for each frequency band.

	Frequency Band Limits							
Lower Frequency (_Hz) Upper Frequency (_Hz)		_Hz)	e. T/R Mode	Nature of Service(s): List all that apply to this band				
a. Numeric	b. Unit (K/M/G)	c. Numeric	d. Unit (K/M/G)		,, .,,			
8025	M	8400	M	Т	Earth Exploration Satellite Service			
2025	М	2110	М	R	Earth Exploration Satellite Service			

S3. ORBITAL INFORMATION FOR GEOSTATIONARY SATELLITES ONLY:

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#### S4. ORBITAL INFORMATION FOR NON-GEOSTATIONARY SATELLITES ONLY

S4a. Total Number of Satellites in Network or System: 3 S4c. Celestial Reference Body (Earth, Sun, Moon, etc.): E

S4b. Total Number of Orbital Planes in Network or System: 3 S4d. Orbit Epoch Date: 3/1/2014

#### For each Orbital Plane Provide:

(e) Orbital Plane No.	(f) No. of Satellites in Plane	(g) Inclination Angle (degrees)	(h) Orbital Period (Seconds)	(i) Apogee (km)	(j) Perigee (km)	(k) Right Ascension of the Ascending Node (Deg.)	(I) Argument of Perigee (Degrees)	rvice Arc Rang (n) End Angle	
1	1	97.36	5670	521	506	135.035	90		

S5. INITIAL SATELLITE PHASE ANGLE For each satellite in each orbital plane, provide the intital phase angle.

(a) Orbital Plane No.	(b) Satellite Number	(c) Initial Phase Angle (Degrees)
1	1	0

Page 2: NGSO Orbits

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### S6. SERVICE AREA CHARACTERISTICS for each service area provide:

	(a) Service Area ID	(b) Type of Associated Station (Earth or Space)		(d) Service Area Description. Provide list of geographic areas (state postal codes or ITU 3-ltr codes), satellites or Figure No. of Service Area Diagram.
1	0	E	not applicable.gxt	N/A

Page 3: Service Areas

Page 4: Antenna Beams

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### S7. SPACE STATION ANTENNA BEAM CHARACTERISTICS For each antenna beam provide:

(a)	(b)	Isotropic	Antenna	(e)	(f)	(g) Min.	(h) Polar-	(i) Polarization	(j) Service		Transmit				Receive		
Beam	T/R		ain		Rotational	Cross-	ization	Alignment Rel.	Area ID	(k) Input	(I) Effective	(m)		(o) G/T	(p) Min.	Input Attenu	uator (dB)
ID	Mode	(c) Peak (dBi)	(d) Edge (dBi)	Error (Degrees)	Error (Degrees)	Polar Iso- lation (dB)	Switch- able? (Y/N)	Equatorial Plane (Degrees)		Losses (dB)	Output Power (W)	Max. EIRP (dBW)	System Noice Temp (k)	Max. Gain Pt. (db/K)	Saturation Flux Density (dBW/m2)	(q) Max. Value	(r) Step Size
WB-L	Т	29	28	1	0	30	Υ			7	2.4	31.8					
WB-	T	29	28	1	0	30	Υ			7	2.4	31.8					
NB	T	7	-9	0	0		N			2	3	11.8					
CMD	R	4	-10	0	0		Ν						580	-23.6			

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S8. ANTENNA BEAM DIAGRAMS For each beam pattern provide the reference to the graphic image and numerical data:

Also provide the power flux density levels in each beam that result from the emission with the highest power flux density.

(a)	(b)	(c) Co-or	(d) GSO	(e) NGSO Antenna Gain	(f) GSO Antenna	Max. Power Flux Density (dBW/M2/Hz)				
Beam	T/R	Cross	Ref.	Contour Description	Gain Contour Data	At Angle of	Arrival above he	orizontal (for em	ission with higl	hest PFD)
ID	Mode	Polar	Orbital	(Figure/Table/ Exhibit)	(GXT File)	(g) 5 Deg	(h) 10 Deg	(i) 15 Deg	(j) 20 Deg	(k) 25 Deg
		Mode ("C" or" X")	Longitude (Dea. E/W)			(0)	, ,	.,	<i>3,</i>	, ,
		OI A)	(Deg. E/W)							

Page 5: Beam Diagrams

Page 6: Channels and Transponders

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S9. SPACE STATION CHANNELS For each frequency channel provide: S10. SPA

(a) Channel No.	(B) Assigned Bandwidth (kHz)	(c) T/R Mode	(d) Center Frequency (MHz)	(e) Polarization (H, V, L, R)	(f) TTC or Comm Channel (T or C)
WB-L	375000	T	8185	L	С
WB-R	375000	T	8185	R	С
NB-I	2695	T	8380	L	Т
NB-Q	5800	Т	8380	L	Т
CMD	1316	R	2085.6875	R	Т

S10. SPACE STATION TRANSPONDERS For each transponder provide	de:
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(a) (b)	Receive	Band	Transmit Band		
Transponder   Transponder   Gain (dB)	(c) Channel No.	(d) Beam ID	(e) Channel No.	(f) Beam ID	

Page 7: Digital Modulation

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### S11. DIGITAL MODULATION PARAMETERS For each digital emission provide:

(a) Digital Mod. ID	(b) Emission Designator	(c) Assigned Bandwidth (kHz)	(d) No. of Phases	(e)Uncoded Data Rate (kbps)	(f) FEC Error Correction Coding Rate	(g) CDMA Processing Gain (dB)	(h) Total C/N Performance Objective (dB)	(i) Single Entry C/I Objective (dB)
WB-L	375MG7D	375000	4	348387.2	0.871	0	19.9	32.5
WB-R	375MG7D	375000	4	348387.2	0.871	0	19.9	32.5
NB-I	2M70G7D	2695	2	32.768	0.5	0	19.9	32.5
NB-Q	5M80G7D	5800	2	524.288	0.5	0	11.2	32.5
CMD	1M32G1D	1316	2	64	1	0	18.8	

Page 8: Analog Modulation

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### S12. ANALOG MODULATION PARAMETERS For each analog emission provide:

(a)	(b) Emission	(c)	(d) Signal	(e)		Multi-channel	Telephony		(j) Video	(k) Video	(I) Video	(m) SCPC/FM	(n) Total C/N	(o) Single
Analog Mod. ID		Assigned Bandwidth (kHz)	Туре	Channels per Carrier	Companded	(g) Bottom Baseband Freq. (MHz)	(h) Top Baseband Freq. (MHz)	(i) RMS Modulation Index	Standard NTSC, PAL, etc.	Noise- Weighting (dB)	and SCPC/FM Modulation Index	Compander, Preemphasis, and Noise Weighting (dB)	Performance Objective (dB)	Entry C/I Objective (dB)

Page 9: Typical Emissions

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### S13. TYPICAL EMISSIONS For each planned type of emission provide:

Assoc			lation ID	(e) Carriers	` '	(0)	(h) Energy	Receive Ba	and (Assoc. T	ransmit Stn)	Tra	nsmit Band	(This Space Stat	tion)
	er ID Range	(c) Digital (Table S11)	(d) Analog (Table S12)	per Transponder	Spacing (kHz)	Reference (Table No.)	Dispersal Bandwidth	(i)Assoc. Stn. Max.	Assoc. Stati Power	on Transmit (dBW)	EIRP	(dBW)	(n) Max. Power Flux Density	(o)Assoc. Stn Rec.
(a) Start	(b) End	(13.312 211)	(**************************************				(kHz)	Antenna Gain (dBi)	(j) Min.	(k) Max.	(I) Min.	(m) Max.	(dBW/m2/Hz)	G/T (dB/K)

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Page 10: TT and C

S14. Is the space station(s) controlled and monitored remotely? If Yes, provide the location and telephone number of the TT and C control point(s): Yes

### Remote Control (TT C) Location(s):

S14a: Street Address: 1601 Dry Creek Dr.				
S14b. City: Longmont	S14c. County: Boulder		S14d. State/Country CO	S14e. Zip Code: 80503
S14f. Telephone Number: 303-684-4589		S14g. Call Sign of Control Stat	ion (if appropriate):	

### FEDERAL COMMUNICATIONS COMMISSION **SATELLITE SPACE STATION AUTHORIZATIONS** FCC Form 312 - Schedule S: (Technical and Operational Description)

Page 11: Characteristics and Certifications

#### S15. SPACECRAFT PHYSICAL CHARACTERISTICS:

S15a: Mass of spacecraft without fuel (kg): 2050	Spacecraft Dimensions (meters)	Probability of Survival to End of Life (0.0 - 1.0)
S15b. Mass of fuel and disposables at launch (kg): 383	1	
S15c. Mass of spacecraft and fuel at launch (kg): 2433	S15f. Length (m): 2.54	S15i. Payload: 0.9735
S15d. Mass of fuel, in orbit, at beginning of life (kg): 383	S15g. Width (m): 2.54	S15j. Bus: 0.866
S15e. Deployed Area of Solar Array (square meters): 13.1	S15h. Height (m): 3.94	S15k. Total: 0.8435

#### S16. SPACECRAFT ELECTRICAL CHARACTERISTICS:

Spacecraft Subsystem		ver (Watts) At ng of Life	Electrical Power (Watts) End of Life				
	At Equinox	At Solstice	At Equinox	At Solstice			
Payload (Watts):	<sup>(a):</sup> 56	<sup>(f):</sup> 56	<sup>(k):</sup> 56	<sup>(p):</sup> 56			
Bus (Watts):	<sup>(b):</sup> 1954	<sup>(g):</sup> 1954	<sup>(l):</sup> 1954	<sup>(q):</sup> 1954			
Total (Watts):	<sup>(c):</sup> 2010	<sup>(h):</sup> 2010	<sup>(m)</sup> 2010	<sup>(r):</sup> 2010			
Solar Array (Watts):	<sup>(d):</sup> 3574	<sup>(i):</sup> 3440	<sup>(n):</sup> 2929	<sup>(s):</sup> 2819			
Depth of Battery Discharge (%):	<sup>(e)</sup> 35 %	<sup>(j)</sup> 35 %	<sup>(o)</sup> 35 %	<sup>(t)</sup> 35 %			

#### S17. CERTIFICATIONS:

a. Are the power flux density limits of § 25.208 met?:	YES	; [_]	NO	X N/A				
b. Are the appropriate service area coverage requirements of § 25.143(b)(ii) and (iii), or § 25.145(c)(1) and (2) met?	YES		NO	X N/A				
c. Are the frequency tolerances of § 25.202(e) and the out-of-band emission limits of § 25.202(f)(1), (2) and (3) met	X YES		NO	N/A				
In addition to the information required in this Form, the space station applicant is required to provide all the information specified in Section 25.114 of the								

Commission's rules, 47 C.F.R § 25.114.