FCC 312	
Schedule	S

FEDERAL COMMUNICATIONS COMMISSION SATELLITE SPACE STATION AUTHORIZATIONS (Technical and Operational Description)

Page 1: General, Frequency Bands, and GSO Orbit

S1. GENERAL INFORMATION Complete for all satellite application	S1.	GENERAL	INFORMATION	Complete for a	all satellite	application
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a. Space Station or Satellite Ne GEOEYE-2	etwork Name:	e. Estimated Date of Placement into Service:	i Will the space station(s) operate on a Common Carrier Basis:
b. Construction Commencement	nt Date:	f. Estimated Lifetime of Satellite(s): 7 Years	j. Number of transponders offered on a common carrier basis:
c. Construction Completion Dat	te:	g. Total Number of Transponders: 4	k. Total Common Carrier Transponder Bandwidth: MHz
d1. Est Launch Date Begin:	d2. Est Launch Date End:	h. Total Transponder Bandwidth (no. transponders x Bandwidth) 750 MHz	I. Orbit Type: Mark all boxes that apply: GSO X NGSO

S2. OPERATING FREQUENCY BANDS Identify the frequency range and transmit/receive mode for all frequency bands in which this station will oper Also indicate the nature of service(s) for each frequency band.

	Frequency	Band Limits			f.				
Lower Frequency (_Hz)	Upper Frequency (_	_Hz)	e. T/R Mode	Nature of Service(s): List all that apply to this band				
a. Numeric	b. Unit (K/M/G)	c. Numeric	d. Unit (K/M/G)						
8025	M	8400	M	Т	Earth Exploration Satellite Service				
2025	М	2110	М	Т	Earth Exploration Satellite Service				

S3. ORBITAL INFORMATION FOR GEOSTATIONARY SATELLITES ONLY:

Page 2: NGSO Orbits

FCC Form 312 - Schedule S: (Technical and Operational Description)

S4. ORBITAL INFORMATION FOR NON-GEOSTATIONARY SATELLITES ONLY

S4a. Total Number of Satellites in Network or System: 2 S4c. Celestial Reference Body (Earth, Sun, Moon, etc.): E

S4b. Total Number of Orbital Planes in Network or System: 1 S4d. Orbit Epoch Date: 6/1/2013

For each Orbital Plane Provide:

(e) Orbital Plane No.	(f) No. of Satellites in Plane	(g) Inclination Angle (degrees)	(/	(i) Apogee (km)	(j) Perigee (km)	(k) Right Ascension of the Ascending Node (Deg.)	(I) Argument of Perigee (Degrees)		rvice Arc Range (n) End Angle	,
1	2	98.114	, ,	684	668			3 -	360	

S5. INITIAL SATELLITE PHASE ANGLE For each satellite in each orbital plane, provide the intital phase angle.

(a) Orbital Plane No.	(b) Satellite Number	(c) Initial Phase Angle (Degrees)
1	1	0
1	2	180

FCC Form 312 - Schedule S: (Technical and Operational Description)

S6. SERVICE AREA CHARACTERISTICS for each service area provide:

(a) Se	Service Area ID	(b) Type of Associated Station (Earth or Space)	(d) Service Area Description. Provide list of geographic areas (state postal codes or ITU 3-ltr codes), satellites or Figure No. of Service Area Diagram.
GLO	BAL	E	Global

Page 3: Service Areas

Page 4: Antenna Beams

FCC Form 312 - Schedule S: (Technical and Operational Description)

S7. SPACE STATION ANTENNA BEAM CHARACTERISTICS For each antenna beam provide:

(a)	(b)	Isotropic	Antenna	(e)	(f)	(g) Min.	(h) Polar-	(i) Polarization	(j) Service		Transmit				Receive		
Beam	T/R	Ga		0	Rotational	Cross-	ization	Alignment Rel.	Area ID	(k) Input	(I) Effective	(m)	(n)	(o) G/T	(p) Min.	Input Atten	uator (dB)
ID	Mode			Error		Polar Iso-	Switch-	Equatorial		Losses	Output	Max.	System	Max.	Saturation	(q) Max.	(r) Step
		(dBi)	(dBi)	(Degrees)	(Degrees)	lation (dB)	able? (Y/N)	Plane (Degrees)		(dB)	Power (W)	EIRP			Flux Density	Value	Size
							` '					(dBW)	Temp (k)	(db/K)	(dBW/m2)		
B1	Т	27.9	3	1		23	N		GLOBAL	9.5	0.84	26.7					
B2	Т	27.9	3	1		23	Ν		GLOBAL	9.5	0.84	26.7					
В3	Т	0	-7	0		0.01	N		GLOBAL	8.1	0.54	-10					
B4	R	2	-7	0		0.01	N		GLOBAL				290	-38.2	-68.1		

FEDERAL COMMUNICATIONS COMMISSION SATELLITE SPACE STATION AUTHORIZATIONS FCC Form 312 - Schedule S: (Technical and Operational Description)

Page 5: Beam Diagrams

S8. ANTENNA BEAM DIAGRAMS For each beam pattern provide the reference to the graphic image and numerical data:

Also provide the power flux density levels in each beam that result from the emission with the highest power flux density.

(a) Beam	(b) T/R	(c) Co-or Cross	(d) GSO Ref.	(e) NGSO Antenna Gain Contour Description	(f) GSO Antenna	At Angle of		lux Density (dB		neet DED)
ID	Mode	Polar Mode ("C" or" X")	Orbital Longitude (Deg. E/W)	(Figure/Table/ Exhibit)	Gain Contour Data (GXT File)	(g) 5 Deg	Arrival above ho (h) 10 Deg	(i) 15 Deg	(j) 20 Deg	(k) 25 Deg
B1	Т	Х				-157	-155	-154	-152	-151
B2	Т	Х				-157	-155	-154	-152	-151
В3	Т	С				-163	-161	-160	-158	-157

Page 6: Channels and Transponders

FCC Form 312 - Schedule S: (Technical and Operational Description)

S9. SPACE STATION CHANNELS For each frequency channel provide: S10. SPACE STATION TRANSPONDERS For each transponder provide:

(a) Channel No.	(B) Assigned Bandwidth (kHz)	(c) T/R Mode	(d) Center Frequency (MHz)	(e) Polarization (H, V, L, R)	(f) TTC or Comm Channel (T or C)
CH1	375000	T	8185	L	С
CH2	375000	Т	8185	R	С
CH3	120	T	8386	R	Т
CH4	128	R	2052	R	Т

(a)	(b)	Receive	Band	Transmit Band			
Transponder ID	Transponder Gain (dB)	(c) Channel No.	(d) Beam ID	(e) Channel No.	(f) Beam ID		
T1				CH1	B1		
T2				CH3	B3		
R1		CH4	B4				
T3				CH2	B2		

Page 7: Digital Modulation

FCC Form 312 - Schedule S: (Technical and Operational Description)

S11. DIGITAL MODULATION PARAMETERS For each digital emission provide:

(a) Digital Mod. ID	(b) Emission Designator	(c) Assigned Bandwidth (kHz)	(d) No. of Phases	(e)Uncoded Data Rate (kbps)	(f) FEC Error Correction Coding Rate	(g) CDMA Processing Gain (dB)	(h) Total C/N Performance Objective (dB)	(i) Single Entry C/I Objective (dB)
DM1	375MG1D	375000	4	348387	0.87			9.8
DM2	375MG1D	375000	4	348387	0.87			9.8
DM3	64K7G1D	128	4	59.7				9.8
DM4	120KG1D	120	2	105	0.87			9.8

Page 8: Analog Modulation

FCC Form 312 - Schedule S: (Technical and Operational Description)

S12. ANALOG MODULATION PARAMETERS For each analog emission provide:

(a)	(b) Emission	(c)	(d) Signal	(e)	Multi-channel Telephony				(j) Video	(k) Video	(I) Video	(m) SCPC/FM	(n) Total C/N	(o) Single
Analog Mod. II		Assigned Bandwidth (kHz)	21 -	Channels per Carrier	(f) Ave. Companded Talker Level (dBm0)	(g) Bottom Baseband Freq. (MHz)	(h) Top Baseband Freq. (MHz)	(i) RMS Modulation Index	Standard NTSC, PAL, etc.	Noise- Weighting (dB)	and SCPC/FM Modulation Index	Compander, Preemphasis, and Noise Weighting (dB)	Performance Objective (dB)	Entry C/I Objective (dB)

Page 9: Typical Emissions

FCC Form 312 - Schedule S: (Technical and Operational Description)

S13. TYPICAL EMISSIONS For each planned type of emission provide:

	Associated		Modulation ID			(g)Noise Budget	(h) Energy	Receive Band (Assoc. Transmit Stn)			Transmit Band (This Space Station)			
Transponder ID Range (a) Start (b) End		(c) Digital (d) Analog (Table S11) (Table S12)	per Spacing Transponder (kHz)	Reference (Table No.)	Dispersal Bandwidth (kHz)	(i)Assoc. Stn. Max. Antenna	Assoc. Stati Power	on Transmit (dBW)	EIRP	(dBW)	(n) Max. Power Flux Density (dBW/m2/Hz)	(o)Assoc. Stn Rec. G/T		
								Gain (dBi)	(j) Min.	(k) Max.	(I) Min.	(m) Max.	(UBVV/IIIZ/FIZ)	(dB/K)
T1	T1	DM1		1							25	27.6	-146.9	29.9
T2	T2	DM3		1							-13.3	-10	-151.5	30
T3	T3	DM2		1							25	27.6	-146.9	29.9

Page 10: TT and C

FCC Form 312 - Schedule S: (Technical and Operational Description)

S14. Is the space station(s) controlled and monitored remotely? If Yes, provide the location and telephone number of the TT and C control point(s): Yes

Remote Control (TT C) Locat	ion(s):							
S14a: Street Address: 2325 Dulles Corner								
S14b. City: S14c. County: Fairfax			S14d. State/Country VA	S14e. Zip Code: 20171				
S14f. Telephone Number: 703-480-7500		S14g. Call Sign of Co E980375	S14g. Call Sign of Control Station (if appropriate): E980375					
Remote Control (TT C) Locat	ion(s):							
S14a: Street Address: 1625 Richardson Highway								
North Pole S14c. County:			S14d. State/Country AK	S14e. Zip Code: 99705				
S14f. Telephone Number: 703-480-7500		S14g. Call Sign of Co E980376	S14g. Call Sign of Control Station (if appropriate): E980376					

FEDERAL COMMUNICATIONS COMMISSION **SATELLITE SPACE STATION AUTHORIZATIONS** FCC Form 312 - Schedule S: (Technical and Operational Description)

Page 11: Characteristics and Certifications

S15. SPACECRAFT PHYSICAL CHARACTERISTICS:

S15a: Mass of spacecraft without fuel (kg): 2030.9	Spacecraft Dimensions (meters)	Probability of Survival to End of Life (0.0 - 1.0)
S15b. Mass of fuel and disposables at launch (kg): 454.4		
S15c. Mass of spacecraft and fuel at launch (kg): 2485.3	S15f. Length (m): 7.73	S15i. Payload: 0.9227
S15d. Mass of fuel, in orbit, at beginning of life (kg): 454.4	S15g. Width (m): 7.73	S15j. Bus: 0.5972
S15e. Deployed Area of Solar Array (square meters): 16	S15h. Height (m): 5.33	S15k. Total: 0.551

S16. SPACECRAFT ELECTRICAL CHARACTERISTICS:

Spacecraft Subsystem		ver (Watts) At ng of Life	Electrical Power (Watts) At End of Life				
	At Equinox	At Solstice	At Equinox	At Solstice			
Payload (Watts):	^{(a):} 297	^{(f):} 297	^{(k):} 297	^{(p):} 297			
Bus (Watts):	^{(b):} 1065	^{(g):} 1065	^{(l):} 1065	^{(q):} 106			
Total (Watts):	^{(c):} 1362	^{(h):} 1362	^(m) 1362	^{(r):} 1362			
Solar Array (Watts):	^{(d):} 3670	^{(i):} 3817	^{(n):} 3494	^{(s):} 3641			
Depth of Battery Discharge (%):	^(e) 14 %	^(j) 14 %	⁽⁰⁾ 17 %	^(t) 17 %			

S17. CERTIFICATIONS:

a. Are the power flux density limits of § 25.208 met?:	X YES	NO	N/A					
b. Are the appropriate service area coverage requirements of § 25.143(b)(ii) and (iii), or § 25.145(c)(1) and (2) met?	YES	NO	X N/A					
c. Are the frequency tolerances of § 25.202(e) and the out-of-band emission limits of § 25.202(f)(1), (2) and (3) met	YES	NO	X N/A					
In addition to the information required in this Form, the space station applicant is required to provide all the information specified in Section 25.114 of the								

Commission's rules, 47 C.F.R § 25.114.