Engineering Statement

1 Introduction

Intelsat License LLC, as debtor in possession ("Intelsat"), seeks authority in this application to launch and operate the Galaxy 32 satellite at the 91.0° W.L orbital location. The characteristics of the satellite, as well as its compliance with the various provisions of Part 25 of the Federal Communication Commission's ("FCC or "Commission") rules,[1] are provided in the remainder of this Engineering Statement.

2 Spacecraft Overview

Galaxy 32 is capable of operating in the C- and Ku-band frequencies listed below.

Direction	Frequency
Uplink	5925-6425 MHz
	13750-14000 MHz
Downlink	3700-4200 MHz[2]
	11450-11701 MHz

The spacecraft provides the following coverage:

Frequency band	Beam	Coverage
C-band	Wide fixed	United States including Alaska and Hawaii; Central America; and the Caribbean
Ku-band	Fixed	Contiguous United States

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1 Unless otherwise stated, all references to rule sections in this document refer to sections in Title 47 of the Code of Federal Regulations.

2 With the exception of tracking, telemetry, and command frequencies ("TT&C"), Intelsat is only seeking authority to operate in the 4000-4200 MHz downlink band in the contiguous United States. Outside of the contiguous United States, Intelsat is seeking to operate in the entire 3700-4200 MHz band. *See Expanding Flexible Use of the 3.7 to 4.2 GHz Band*, Report and Order and Order of Proposed Modification, FCC 20-22, 35 FCC Rcd 2343, ¶ 152 (2020); 47 C.F.R. § 2.106, NG182.

2.1 Spacecraft Characteristics

Galaxy 32 is a Maxar FS 1300-122 three-axis stabilized type spacecraft that has a rectangular outer body structure. Galaxy 32 utilizes two deployable solar array wings and deployable and fixed antennas.

The Galaxy 32 satellite is composed of the following subsystems:

- Thermal;
- Power;
- Attitude Control;
- Propulsion;
- Telemetry, Command and Ranging ("TC&R");
- Uplink Power Control ("ULPC"); and
- Communications

These subsystems maintain the correct position and attitude of the spacecraft; ensure that all internal units are maintained within the required temperature range; and ensure that the spacecraft can be commanded and controlled with a high level of reliability from launch to the end of its useful life. The spacecraft design incorporates redundancy in all of the various subsystems in order to avoid single-point failures.

The structural design of Galaxy 32 provides mechanical support for all subsystems. The structure supports the communication antennas, solar arrays, and thrusters. It also provides a stable platform for preserving the alignment of critical elements of the spacecraft.

2.2 Communication Subsystem

The Galaxy 32 satellite employs channels with bandwidth of 36 MHz, 43 MHz, and 126 MHz for the C-band payload and 36 MHz for the Ku-band payload. The C- and Ku-band frequencies, polarization, and channel plan as well as the coverage contours and performance characteristics for the satellite beams are provided in Schedule S. Exhibits 1 and 2 provide the beam parameters for the uplink and downlink beams, respectively, and Exhibit 4 provides the Schedule S beam designations.

2.3 Telemetry, Command and Ranging Subsystem

The TC&R subsystem provides the following functions:

1) Acquisition, processing and transmission of spacecraft telemetry data;

2) Reception and retransmission of ground station generated ranging signals; and

3) Reception, processing and distribution of telecommands.

Galaxy 32's C-band TT&C frequencies are tunable and selectable via ground command in 100 kHz steps.

The satellite's command and telemetry channel frequencies and performance are shown in Exhibit 3. The coverage patterns of the command and telemetry beams have gain contours that vary by less than 8 dB across the surface of the Earth, and accordingly the gain at 8 dB below the peak falls beyond the edge of the Earth. Therefore, pursuant to Section 25.114(c)(4)(vi)(A), contours for these beams are not required to be provided and the associated GXT files have not been included in Schedule S.

2.4 Uplink Power Control Subsystem

Galaxy 32 utilizes one C-band and one Ku-band channel for ULPC, antenna tracking, and ranging. The coverage patterns of the ULPC beams have gain contours that vary by less than 8 dB across the surface of the Earth, and accordingly the gain at 8 dB below the peak falls beyond the edge of the Earth. Therefore, pursuant to Section 25.114(c)(4)(vi)(A), contours for these beams are not required to be provided and the associated GXT files have not been included in Schedule S. Galaxy 32's ULPC frequencies and subsystem performance are summarized in Exhibit 3.

2.5 Satellite Station-Keeping

The satellite will be maintained within 0.05° of its nominal longitudinal position in the east-west direction in compliance with Section 25.210(j).

The attitude of the spacecraft will be maintained with accuracy consistent with the achievement of the specified communications performance, after taking into account all error sources (i.e., attitude perturbations, thermal distortions, misalignments, orbital tolerances, and thruster perturbations, etc.).

3 Services

Galaxy 32 is a general-purpose communications satellite and has been designed to support various services offered within the Intelsat satellite system. Depending upon the needs of the users, the transponders on Galaxy 32 can provide a range of communications services, including compressed digital video, high speed digital data and digital single channel per carrier data channels.

4 Power Flux Density

The power flux density limits for space stations operating in the 3700-4200 MHz and 11450-11701 MHz band are specified in Section 25.208. As provided in Schedule S, the Galaxy 32 transmissions do not exceed these limits.

5 Emission Compliance

Section 25.202(e) requires that the carrier frequency of each space station transmitter be maintained within 0.002% of the reference frequency. Galaxy 32 is designed to be compliant with the provisions of this rule.

Galaxy 32 will comply with the provisions of Section 25.202(f) with regard to its emissions.

6 Orbital Location

Intelsat requests that Galaxy 32 be assigned the 91.0° W.L. orbital location as replacement satellite for Galaxy 17, which is currently licensed and operating at 91.0°W.L.

7 Coordination Statement and Certifications

The downlink effective isotropic radiated power ("EIRP") density of the satellite's transmissions in the conventional C-band will not exceed 3 dBW/4kHz for digital transmissions or 8 dBW/4kHz for analog transmissions and in the extended Ku-band the downlink EIRP density will not exceed 14 dBW/4kHz for digital transmissions or 17 dBW/4kHz for analog transmissions. The associated uplink transmissions will not exceed the applicable EIRP density envelopes in Sections 25.218 unless the non-routine uplink and/or downlink operation is coordinated with operators of authorized co-frequency space stations at assigned locations within six degrees of Galaxy 32 at 91.0°W.L.

8 Orbital Debris Mitigation Plan

Intelsat is proactive in ensuring safe operation and disposal of this and all spacecraft under its control. The four elements of debris mitigation are addressed below.

8.1 Spacecraft Hardware Design

Galaxy 32 is designed such that no debris will be released during normal operations. Intelsat has assessed the probability of collision with meteoroids and other debris. In order to limit the effects of such unlikely collisions critical satellite components are located inside the protective body of the satellite and are properly shielded. The satellite does not use any subsystems for end-of-life disposal that are not used for normal operations.

8.2 Minimizing Accidental Explosions

Intelsat has assessed the probability of accidental explosions during and after completion of mission operations. The satellite is designed in a manner to minimize the potential for such explosions. Propellant tanks and thrusters are isolated using redundant valves and electrical power systems are shielded in accordance with standard industry practices. At the completion of the mission and upon disposal of the spacecraft, Intelsat will ensure the removal of all stored energy on the spacecraft by depleting all propellant tanks, venting all pressurized systems and by leaving the batteries in a permanent discharge state.

8.3 Safe Flight Profiles

Intelsat has assessed and limited the probability of the space station becoming a source of debris as a result of collisions with large debris or other operational space stations.

Galaxy 32 will replace Galaxy 17 at 91.0° W.L. and these satellites may be nominally collocated during transfer of traffic. Additionally, Intelsat is planning to launch Intelsat 40e that will be co-located with Galaxy 32 at the nominal 91.0° W.L. orbital location. For both of these cases Intelsat will ensure that sufficient spatial separation is achieved between these satellites through the use of orbit eccentricity and inclination offsets and thus minimize the risk of collision.

Intelsat is not aware of any other FCC licensed system, or any other system applied for and under consideration by the FCC, having an overlapping station-keeping volume with Galaxy 32. Intelsat is also not aware of any system with an overlapping station-keeping volume with Galaxy 32 that is the subject of an ITU filing and that is either in orbit or progressing towards launch.

8.4 Post Mission Disposal

At the end of the mission, Intelsat will dispose of the spacecraft by moving it to an altitude of at least 263.5 kilometers above the geostationary arc. Intelsat has reserved 55.0 kilograms of bi-propellant for that purpose.

In calculating the disposal orbit, Intelsat has used simplifying assumptions as permitted under the Commission's Orbital Debris Report and Order.[3] The effective area to mass ratio (Cr*A/M) of the satellite is 0.026 m^2/kg, resulting in a minimum perigee disposal altitude under the Inter-Agency Space Debris Coordination Committee formula of 263.5 kilometers above the geostationary arc. Accordingly, Galaxy 32's planned disposal orbit complies with the FCC's rules.

The reserved propellant figure is an estimate. This figure is calculated taking into account the expected mass of the satellite at the end of life and the required delta-velocity to achieve the desired orbit.

9 TC&R Control Earth Stations

Intelsat will conduct TC&R operations through earth stations at one or more of the following teleports: Napa, California; Hagerstown, Maryland; Fillmore, California; Riverside, California; Castle Rock, Colorado; Paumalu, Hawaii; Brewster, Washington; Andover, Maine; and Ellenwood, Georgia. Additionally, Intelsat is capable of remotely controlling Galaxy 32 from its facilities in McLean, Virginia or Long Beach, California.

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[3] *Mitigation of Orbital Debris*, Second Report and Order, 19 FCC Rcd 11567 (2004).

Certification Statement

I hereby certify that I am a technically qualified person and am familiar with Part 25 of the Commission's rules. The contents of this engineering statement were prepared by me or under my direct supervision and to the best of my knowledge are complete and accurate.

/s/ Giselle Creeser	January 7, 2020
Giselle Creeser Intelsat Director, Spectrum Policy and Engineering	Date

EXHIBIT 1

COMMUNICATION SUBSYSTEM UPLINK BEAM PARAMETERS

Beam Name	C-band Americas	C-band Americas
Schedule S Beam ID	CAHU	CAVU
Frequency Band (MHz)	5925 - 6425	5925 - 6425
Polarization	Horizontal	Vertical
G/T (dB/K)	4.5	4.5
Minimum SFD-- (dBW/m^2)	-102.0	-102.0
Maximum SFD-- (dBW/m^2)	-77.0	-77.0

Beam Name	Ku-band CONUS	Ku-band CONUS
Schedule S Beam ID	KUHU	KUVU
Frequency Band (MHz)	13750 - 14000	13750 - 14000
Polarization	Horizontal	Vertical
G/T (dB/K)	5.0	5.0
Minimum SFD-- (dBW/m^2)	-95.0	-95.0
Maximum SFD-- (dBW/m^2)	-70.0	-70.0

EXHIBIT 2

COMMUNICATION SUBSYSTEM DOWNLINK BEAM PARAMETERS

Beam Name	C-band Americas	C-band Americas
Schedule S Beam ID	CAHD	CAVD
Frequency Band (MHz)	3700 - 4200	3700 - 4200
Polarization	Horizontal	Vertical
Maximum Beam Peak EIRP (dBW)	47.0	47.0
Maximum Beam Peak EIRP Density (dBW/4kHz)	7.5	7.5
Maximum Beam Peak EIRP Density (dBW/Hz)	-28.6	-28.6

Beam Name	Ku-band CONUS	Ku-band CONUS
Schedule S Beam ID	KUHD	KUVD
Frequency Band (MHz)	11450 - 11701	11450 - 11701
Polarization	Horizontal	Vertical
Maximum Beam Peak EIRP (dBW)	52.5	52.5
Maximum Beam Peak EIRP Density (dBW/4kHz)	13.0	13.0
Maximum Beam Peak EIRP Density (dBW/Hz)	-23.1	-23.1

EXHIBIT 3

TC&R SUBSYSTEM CHARACTERISTICS

Beam Name	Command 1 Dish	Command 1 Omni	Command 2 Dish	Command 2 Omni
Schedule S Beam ID	CMD1	CMD2	CMD3	CMD4
Frequencies (MHz)	6422.0*	6422.0*	6424.5*	6424.5*
Polarization	Horizontal	RHCP	Horizontal	RHCP
Peak Flux Density at Command Threshold (dBW/m^2-Hz)	-80.0	-80.0	-80.0	-80.0

*Tunable with 100 KHz step (5925-6425 MHz)

Beam Name	Telemetry 1 Dish	Telemetry 1 Omni	Telemetry 2 Dish	Telemetry 2 Omni	ULPC Ku-band	ULPC C-band
Schedule S Beam ID	TLMD	TLMO	TLMP	TLMM	UPC2	UPC1
Frequencies (MHz) -1.	4198.25**	4198.25**	4198.75**	4198.75**	11700.25	4199.95
Polarization	Vertical	LHCP	Vertical	LHCP	RHCP	RHCP
Maximum Channel EIRP (dBW)	7.5	7.5	7.5	7.5	11.5	7.5
Maximum Beam Peak EIRP Density (dBW/4kHz)	-13.5	-13.5	-13.5	-13.5	3.0	-0.5
Maximum Beam Peak EIRP Density (dBW/Hz)	-49.5	-49.5	-49.5	-49.5	-33.0	-36.5

**Tunable with 100 KHz step (3700-4200 MHz)

Note: *RHCP: Right Hand Circular Polarization, LHCP: Left Hand Circular Polarization*

EXHIBIT 4
Beam Polarizations and GXT File Names

	Schedule S Beam Names							
	Linear Polarization				**Circular Polarization**			
Beam Designation	**Uplink**	**Uplink**	**Downlink**	**Downlink**	**Uplink**	**Uplink**	**Downlink**	**Downlink**
	(H-Pol.)	**(V-Pol.)**	**(H-Pol.)**	**(V-Pol.)**	**(LHCP)**	**(RHCP)**	**(LHCP)**	**(RHCP)**
C-band Beams								
C-band North America, Alaska, Hawaii & Caribbean	CAHU	CAVU	CAHD	CAVD	----	----	----	----
Telemetry 1	----	----	----	TLMD*	----	----	TLMO*	----
Telemetry 2	----	----	----	TLMP*	----	----	TLMM*	----
Command 1	CMD1*	----	----	----	----	CMD2*	----	----
Command 2	CMD3*	----	----	----	----	CMD4*	----	----
ULPC	----	----	----	----	----	----	----	UPC1*
Ku-band Beams								
Ku-band CONUS	KUHU	KUVU	KUHD	KUVD	----	----	----	----
ULPC	----	----	----	----	----	----	----	UPC2*

** GXT files are not provided for the indicated beams because their -8 dB gain contours extend beyond the edge of the Earth*