#### FCC 312 Schedule S

# FEDERAL COMMUNICATIONS COMMISSION SATELLITE SPACE STATION AUTHORIZATIONS (Technical and Operational Description)

Page 1: General, Frequency Bands, and GSO Orbit

S1. GENERAL INFORMATION Complete for all satellite applications.

Space Station or Satellite Notice     FLOCK 1	etwork Name:	e. Estimated Date of Placement into Service: 12/15/2013	i Will the space station(s) operate on a Common Carrier Basis:			
b. Construction Commenceme 3/1/2013	nt Date:	f. Estimated Lifetime of Satellite(s):  1 Years	Number of transponders offered on a common carrier basis: 0			
c. Construction Completion Date: 9/1/2013		g. Total Number of Transponders: 0	k. Total Common Carrier Transponder Bandwidth:  0 MHz			
d1. Est Launch Date Begin: 12/1/2013	d2. Est Launch Date End: 3/1/2013	h. Total Transponder Bandwidth (no. transponders x Bandwidth) 0 MHz	I. Orbit Type: Mark all boxes that apply:  GSO X NGSO			

S2. OPERATING FREQUENCY BANDS Identify the frequency range and transmit/receive mode for all frequency bands in which this station will oper Also indicate the nature of service(s) for each frequency band.

	Frequency Band Limits								
Lower Frequency (_Hz) Upper Frequency (_Hz)		_Hz)	e. T/R Mode	Nature of Service(s): List all that apply to this band					
a. Numeric	b. Unit (K/M/G)	c. Numeric	d. Unit (K/M/G)		()				
8025	M	8400	M	Т	Earth Exploration Satellite Service				
2025	М	2110	М	R	Earth Exploration Satellite Service				
401	М	402	M	Т	Space Operations Service				
449.75	М	450.25	M	R	Space Operations Service				

S3. ORBITAL INFORMATION FOR GEOSTATIONARY SATELLITES ONLY:

FCC Form 312 - Schedule S: (Technical and Operational Description)

#### S4. ORBITAL INFORMATION FOR NON-GEOSTATIONARY SATELLITES ONLY

S4a. Total Number of Satellites in Network or System: 28 S4c. Celestial Reference Body (Earth, Sun, Moon, etc.): E

S4b. Total Number of Orbital Planes in Network or System: 1 S4d. Orbit Epoch Date: 12/15/2013

#### For each Orbital Plane Provide:

(e) Orbital Plane No.	(f) No. of Satellites in Plane	(g) Inclination Angle (degrees)	(h) Orbital Period (Seconds)	(i) Apogee (km)	(j) Perigee (km)	(k) Right Ascension of the Ascending Node (Deg.)	(I) Argument of Perigee (Degrees)	(m) Begin	rvice Arc Range (n) End Angle	- ( -3/
1	28	51.6	,	410	410	280.5		Angle	360	

#### S5. INITIAL SATELLITE PHASE ANGLE For each satellite in each orbital plane, provide the intital phase angle.

(a) Orbital Plane No.	(b) Satellite Number	(c) Initial Phase Angle (Degrees)
1	1	0
1	2	12.86
1	3	25.71
1	4	38.57
1	5	51.43
1	6	64.29
1	7	77.14
1	8	90
1	9	102.86
1	10	115.71
1	11	128.57
1	12	141.43
1	13	154.29
1	14	167.14
1	15	180
1	16	192.86
1	17	205.71
1	18	218.57
1	19	231.43
1	20	244.29
1	21	257.14

Page 2: NGSO Orbits

1	22	270
1	23	282.26
1	24	295.71
1	25	308.57
1	26	321.43
1	27	334.29
1	28	347.14

FCC Form 312 - Schedule S: (Technical and Operational Description)

### S6. SERVICE AREA CHARACTERISTICS for each service area provide:

	(a) Service Area ID	(b) Type of Associated Station (Earth or Space)	(d) Service Area Description. Provide list of geographic areas (state postal codes or ITU 3-ltr codes), satellites or Figure No. of Service Area Diagram.
١	/ISEARTH	E	XVE

Page 3: Service Areas

Page 4: Antenna Beams

FCC Form 312 - Schedule S: (Technical and Operational Description)

### S7. SPACE STATION ANTENNA BEAM CHARACTERISTICS For each antenna beam provide:

(a)	(b)	Isotropic	Antenna	(e)	(f)	(g) Min.	(h) Polar-	(i) Polarization	(j) Service		Transmit				Receive		
Beam	T/R		ain	0	Rotational	Cross-	ization	Alignment Rel.	Area ID	(k) Input	(I) Effective	(m)		(o) G/T	(p) Min.	Input Atten	uator (dB)
ID	Mode	(c) Peak		Error	Error	Polar Iso-	Switch- able?	Equatorial		Losses	Output	Max.	System	Max.	Saturation	(q) Max.	(r) Step
		(dBi)	(dBi)	(Degrees)	(Degrees)	iation (db)	(Y/N)	Plane (Degrees)		(dB)	Power (W)	EIRP (dBW)	Noice Temp (k)		Flux Density (dBW/m2)	Value	Size
XDLH	Т	12.3	12.3	0.5	0	15	N	0	VISEART	1	2.9	15.92					
XDL	T	5	-2	0.5	0	9	N		VISEART	1	2.9	9.12					
SUL	R	5	-1	0.5	0	9	N		VISEART				100	-15	0		
UDL	T	2.19	-2	0.5	0	20	N		VISEART	1	0.7	0.19					
UUL	R	2.16	-2.5	0.5	0	20	N		VISEART				100	-17.84	0		

FCC Form 312 - Schedule S: (Technical and Operational Description)

S8. ANTENNA BEAM DIAGRAMS For each beam pattern provide the reference to the graphic image and numerical data:

Also provide the power flux density levels in each beam that result from the emission with the highest power flux density.

(a)	(b)	(c) Co-or	(d) GSO	(e) NGSO Antenna Gain	(f) GSO Antenna		Max. Power F	lux Density (dB	W/M2/Hz)		
Beam			Cross Ref.	Contour Description	Gain Contour Data	At Angle of Arrival above horizontal (for emission with highest PFD)					
ID	Mode	Polar Mode ("C"	Orbital Longitude	(Figure/Table/ Exhibit)	(GXT File)	(g) 5 Deg	(h) 10 Deg	(i) 15 Deg	(j) 20 Deg	(k) 25 Deg	
		or" X")	(Deg. E/W)								
XDLH	T	С		x-band-helical.pdf		-159.07	-157.13	-155.39	-153.86	-152.53	
XDLH	T	Χ		x-band-helical.pdf							
XDL	T	С		x-band-patch.pdf		-158.88	-156.94	-155.2	-153.67	-152.34	
XDL	T	X		x-band-patch.pdf							
SUL	R	С		s-band-patch.pdf							
SUL	R	Χ		s-band-patch.pdf							
UDL	Т	С		uhf-antenna.pdf		-152.09	-150.16	-148.41	-146.88	-145.55	
UUL	R	С		uhf-antenna.pdf							

Page 5: Beam Diagrams

Page 6: Channels and Transponders

FCC Form 312 - Schedule S: (Technical and Operational Description)

S9. SPACE STATION CHANNELS For each frequency channel provide: S10. SF

(a) Channel No.	(B) Assigned Bandwidth (kHz)	(c) T/R Mode	(d) Center Frequency (MHz)	(e) Polarization (H, V, L, R)	(f) TTC or Comm Channel (T or C)
XCH1	66830	T	8133	R	С
XCH2	66830	Т	8200	R	С
SCH1	1310	R	2056	R	T
UDCH1	60	Т	401	Н	Т
UUCH1	60	R	450	Н	Т

S10. SPACE STATION TRANSPONDERS For each	h transponder provide:
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(a)	(b)	Receive	Band	Transmit Band		
Transponder ID	Transponder Gain (dB)	(c) Channel No.	(d) Beam ID	(e) Channel No.	(f) Beam ID	
NONE	1	SCH1	SUL	XCH1	XDLH	

Page 7: Digital Modulation

FCC Form 312 - Schedule S: (Technical and Operational Description)

### S11. DIGITAL MODULATION PARAMETERS For each digital emission provide:

(a) Digital Mod. ID	(b) Emission Designator	(c) Assigned Bandwidth (kHz)	(d) No. of Phases	(e)Uncoded Data Rate (kbps)	(f) FEC Error Correction Coding Rate	(g) CDMA Processing Gain (dB)	(h) Total C/N Performance Objective (dB)	(i) Single Entry C/I Objective (dB)
XCH1	66M8G1D	66830	8	120000	0.89	0	11.54	22.84
XCH2	66M8G1D	66830	8	120000	0.89	0	11.54	22.84
SCH1	1M31F1D	1310	2	250	0.5	0	6.81	16.01
UDCH1	60K0F1D	60	2	2.4	0.5	0	0.02	11.32
UUCH1	60K0F1D	60	2	2.4	0.5	0	0.02	11.32

Page 8: Analog Modulation

FCC Form 312 - Schedule S: (Technical and Operational Description)

### S12. ANALOG MODULATION PARAMETERS For each analog emission provide:

(a)	(b) Emission	(c)	(d) Signal	(e)	Multi-channel Telephony			(j) Video	(k) Video	(I) Video	(m) SCPC/FM	(n) Total C/N	(o) Single	
Analog Mod. ID		Assigned Bandwidth (kHz)	Туре	Channels per Carrier	Companded	(g) Bottom Baseband Freq. (MHz)	(h) Top Baseband Freq. (MHz)	(i) RMS Modulation Index	Standard NTSC, PAL, etc.	Noise- Weighting (dB)	and SCPC/FM Modulation Index	Compander, Preemphasis, and Noise Weighting (dB)	Performance Objective (dB)	Entry C/I Objective (dB)

Page 9: Typical Emissions

FCC Form 312 - Schedule S: (Technical and Operational Description)

### S13. TYPICAL EMISSIONS For each planned type of emission provide:

Associated				` '	( )	(0)	(h) Energy	Receive Band (Assoc. Transmit Stn)			Transmit Band (This Space Station)			tion)
	er ID Range	(c) Digital (Table S11)	(d) Analog (Table S12)	per Transponder	Spacing (kHz)	Reference (Table No.)	Dispersal Bandwidth	(i)Assoc. Stn. Max.	Assoc. Stati Power	on Transmit (dBW)	EIRP	(dBW)	(n) Max. Power Flux Density	(o)Assoc. Stn Rec.
(a) Start	(b) End	(13.312 211)	(**************************************				(kHz)	Antenna Gain (dBi)	(j) Min.	(k) Max.	(I) Min.	(m) Max.	(dBW/m2/Hz)	G/T (dB/K)

# FEDERAL COMMUNICATIONS COMMISSION SATELLITE SPACE STATION AUTHORIZATIONS FCC Form 312 - Schedule S: (Technical and Operational Description)

Page 10: TT and C

S14. Is the space station(s) controlled and monitored remotely? If Yes, provide the location and telephone number of the TT and C control point(s): Yes

### Remote Control (TT C) Location(s):

S14a: Street Address: 490 2nd St.							
S14b. City: San Francisco	S14c. County: San Francisco	S14d. State/Country CA	S14e. Zip Code: 94107				
5a ra	-	J	0				
S14f. Telephone Number:	S14g. Call Sign o	S14g. Call Sign of Control Station (if appropriate):					
415-317-6404	NONE	NONE					

### FEDERAL COMMUNICATIONS COMMISSION **SATELLITE SPACE STATION AUTHORIZATIONS** FCC Form 312 - Schedule S: (Technical and Operational Description)

Page 11: Characteristics and Certifications

#### S15. SPACECRAFT PHYSICAL CHARACTERISTICS:

S15a: Mass of spacecraft without fuel (kg): 4.3	Spacecraft Dimensions (meters)	Probability of Survival to End of Life (0.0 - 1.0)
S15b. Mass of fuel and disposables at launch (kg): 0		
S15c. Mass of spacecraft and fuel at launch (kg): 4.3	S15f. Length (m): 0.34	S15i. Payload: 0.8
S15d. Mass of fuel, in orbit, at beginning of life (kg): 0	S15g. Width (m): 0.62	S15j. Bus: 0.9
S15e. Deployed Area of Solar Array (square meters): 0.156	S15h. Height (m): 0.1	S15k. Total: 0.712

#### S16. SPACECRAFT ELECTRICAL CHARACTERISTICS:

Spacecraft Subsystem		ver (Watts) At ng of Life	Electrical Power (Watts) A				
	At Equinox	At Solstice	At Equinox	At Solstice			
Payload (Watts):	<sup>(a):</sup> 13.16	<sup>(f):</sup> 13.16	<sup>(k):</sup> 13.16	<sup>(p):</sup> 13.16			
Bus (Watts):	<sup>(b):</sup> 2.12	<sup>(g):</sup> 2.12	<sup>(l):</sup> 2.12	<sup>(q):</sup> 2.12			
Total (Watts):	<sup>(c):</sup> 15.28	<sup>(h):</sup> 15.28	<sup>(m)</sup> 15.28	<sup>(r):</sup> 15.28			
Solar Array (Watts):	<sup>(d):</sup> 19.1	<sup>(i):</sup> 19.1	<sup>(n):</sup> 18.9	<sup>(s):</sup> 18.9			
Depth of Battery Discharge (%):	<sup>(e)</sup> 30 %	<sup>(j)</sup> 30 %	<sup>(o)</sup> 30 %	<sup>(t)</sup> 30 %			

#### S17. CERTIFICATIONS:

a. Are the power flux density limits of § 25.208 met?:	YES	; [_]	NO	X N/A				
b. Are the appropriate service area coverage requirements of § 25.143(b)(ii) and (iii), or § 25.145(c)(1) and (2) met?	YES		NO	X N/A				
c. Are the frequency tolerances of § 25.202(e) and the out-of-band emission limits of § 25.202(f)(1), (2) and (3) met	X YES		NO	N/A				
In addition to the information required in this Form, the space station applicant is required to provide all the information specified in Section 25.114 of the								

Commission's rules, 47 C.F.R § 25.114.