FEDERAL COMMUNICATIONS COMMISSION SATELLITE SPACE STATION AUTHORIZATIONS (Technical and Operational Description)

S1. GENERAL INFORMATION Complete for all satellite applications.

a. Space Station or Satellite Network Name:	e. Estimated Date of Placement into Service:	i Will the space station(s) operate on a Common Carrier Basis:			
XM-5	3/1/2010	N			
b. Construction Commencement Date: 6/3/2005	f. Estimated Lifetime of Satellite(s): 15 Years	j. Number of transponders offered on a common carrier basis:			
c. Construction Completion Date:	g. Total Number of Transponders:	k. Total Common Carrier Transponder Bandwidth:			
2/1/2009	2	MHz			
d1. Est Launch Date Begin:d2. Est Launch Date End:11/15/20092/1/2010	h. Total Transponder Bandwidth (no. transponders x Bandwidth) 3.68 MHz	I. Orbit Type: Mark all boxes that apply:			

S2. OPERATING FREQUENCY BANDS Identify the frequency range and transmit/receive mode for all frequency bands in which this station will oper Also indicate the nature of service(s) for each frequency band.

	Frequency Band Limits									
Lower Frequency (_Hz)	Upper Frequency (_	_Hz)	e. T/R Mode	f. Nature of Service(s): List all that apply to this band					
a. Numeric	b. Unit (K/M/G)	c. Numeric	d. Unit (K/M/G)							
2332.5	М	2345.0	М	Т	Satellite Digital Audio Radio Service					
7025	М	7075	М	R	Satellite Digital Audio Radio Service					
2320.0	М	2332.5	М	Т	Satellite Digital Audio Radio Service					

S3. ORBITAL INFORMATION FOR GEOSTATIONARY SATELLITES ONLY:

a. Nominal Orbital Longitude (Degrees E/W):	b. Alternate Orbital Longit	ude (Degrees E/W):			c. Reason for orbital location selection:				
85.2 W				Optimal Look Angle for CONUS/CANADA. Colocation						
Longitudinal Tolerance or E/W			Range of orbital are in which	adequate serv	ice can be	with Primary Satellite XM Repeaters fed from 85.083				
d. Toward West:	0.05 Degrees	N/S Station-Keeping Tolerance:	provided (Optional):	Degrees	E/W	location of XM-3.				
e. Toward East:	0.05 Degrees	0.05 Degrees	g. Westernmost: h. Easternmost:							
i. Reason for service are s	selection (Optional)	:								

Page 2: NGSO Orbits

S4. ORBITAL INFORMATION FOR NON-GEOSTATIONARY SATELLITES ONLY

S4a. Total Number of Satellites in Network or System:

S4b. Total Number of Orbital Planes in Network or System:

S4c. Celestial Reference Body (Earth, Sun, Moon, etc.):

S4d. Orbit Epoch Date:

For each Orbital Plane Provide:

(e) Orbital	(f) No. of	(g) Inclination	(h) Orbital	(i) Apogee (km)	(j) Perigee (km)	(k) Right Ascension	(I) Argument of	Active Service Arc Range (Degrees)		
Plane No.	Satellites in	Angle (degrees)	Period			of the Ascending	Perigee	(m) Begin	(n) End	(o) Other
	Plane		(Seconds)			Node (Deg.)	(Degrees)	Angle	Angle	

S5. INITIAL SATELLITE PHASE ANGLE For each satellite in each orbital plane, provide the intital phase angle.

(b) Satellite	(c) Initial
Number	Phase Angle
	(Degrees)
	· · /

NO NGSO DATA FILED

S6. SERVICE AREA CHARACTERISTICS for each service area provide:

(a) Service Area ID	(b) Type of Associated Station (Earth or Space)	(d) Service Area Description. Provide list of geographic areas (state postal codes or ITU 3-ltr codes), satellites or Figure No. of Service Area Diagram.
USA	S	USA
CAN	S	CAN

Page 3: Service Areas

Page 4: Antenna Beams

S7. SPACE STATION ANTENNA BEAM CHARACTERIS	TICS For each antenna beam provide:

(a)	(b)	Isotropic	Antenna	(e)	(f)	(g) Min.	(h) Polar-	(i) Polarization	(j) Service		Transmit				Receive		
Beam	T/R	Ga	ain	Pointing	Rotational	Cross-	ization	Alignment Rel.	Area ID	(k)	(I) Effective	(m)	(n)	(o) G/T	(p) Min.	Input Atten	uator (dB)
ID	Mode	(c) Peak	(d) Edge	Error	Error	Polar Iso-	Switch-	Equatorial		Input	Output	Max.	System	Max.	Saturation	(q) Max.	(r) Step
		(dBi)	(dBi)	(Degrees)	(Degrees)	lation (dB)		Plane (Degrees)		Losses	Power (W)	EIRP			Flux Density	Value	Size
							(Y/N)			(dB)		(dBW)	Temp (k)	(db/K)	(dBW/m2)	, aluo	0.20
XM2T	Т	34.6	27	0.1	0.1	17	Y		USA	1		71					
XM2	R	18.61	18	0.1	0.1	30	Y		USA				509	-5.3	-92	17	0.5
CMD	R	21.45	20.5	0.1	0.1	30	N		USA						-105		
TLM	Т	34.62	14.62	0.1	0.1	20	Ν		USA	5.6	3.2	32					

(a) (b) (c) Co-or (d) GSO e) NGSO Antenna Gain (f) GSO Antenna Max. Power Flux Density (dBW/M2/Hz) Beam Ť/Ŕ Cross Ref. Contour Description Gain Contour Data At Angle of Arrival above horizontal (for emission with highest PFD) ID Mode Polar Orbital (Figure/Table/ Exhibit) (GXT File) (g) 5 Deg (h) 10 Deg (i) 15 Deg (j) 20 Deg (k) 25 Deg Mode ("C" Longitude (Deg. E/W) or" X") XM2T С -85.2 5083WL_Gain Conto -126 -126 -122 -119 -119 XM2 R С -85.2 3WL_Receive Gain C CMD С -85.2 mmand_copol_85W.g R TLM -85.2 ost_copol_85W.gxt С -125 -125 -124 -123 -119

S8. ANTENNA BEAM DIAGRAMS For each beam pattern provide the reference to the graphic image and numerical data: Also provide the power flux density levels in each beam that result from the emission with the highest power flux density. Page 5: Beam Diagrams

Page 6: Channels and Transponders

(a) Channel No.	(B) Assigned Bandwidth (kHz)	(c) T/R Mode	(d) Center Frequency (MHz)	(e) Polarization (H, V, L, R)	(f) TTC or Comm Channel (T or C)
1	1840	Т	2333.465	L	С
2	1840	Т	2335.305	L	С
3	1840	Т	2342.205	L	С
4	1840	Т	2344.045	L	С
1R	1840	R	7063.993	R	С
2R	1840	R	7061.561	R	С
3R	1840	R	7065.965	R	С
4R	1840	R	7068.397	R	С
CMD1	800	R	7043.0	L	Т
CMD2	800	R	7074.0	L	Т
TLM1	100	Т	2336.7	R	Т
TLM2	100	Т	2334.0	R	Т
TLM3	100	Т	2337.2	R	Т
TLM4	100	Т	2342.5	R	Т
5	1840	Т	2338.75	L	С
5R	1840	R	7058.521	R	С
S1	4500	Т	2322.29	L	С
S2	4500	Т	2330.21	L	С
S1R	4500	R	7062.29	R	С
S2R	4500	R	7058.5	R	С

S9. SPACE STATION CHANNELS For each frequency channel provide: S10. SPACE STATION TRANSPONDERS For each transponder provide:

(a)	(b)	Receive	Band	Transm	it Band
Transponder ID	Transponder Gain (dB)	(c) Channel No.	(d) Beam ID	(e) Channel No.	(f) Beam ID
1	150	1R	XM2R	1	XM2T
2	150	2R	XM2R	2	XM2T
3	150	3R	XM2R	3	XM2T
4	150	4R	XM2R	4	XM2T
CMD1		CMD1	CMD		
CMD2		CMD2	CMD		
TLM1				TLM1	TLM
5	150	5R	XM2R	5	XM2T
TLM2				TLM2	TLM
TLM3				TLM3	TLM
TLM4				TLM4	TLM
S1	140	S1R	XM2R	S1	XM2T
S2	140	S2R	XM2R	S2	XM2T

S11. DIGITAL MODULATION PARAMETERS For each digital emission provide:

(a) Digital Mod. ID	(b) Emission Designator	(c) Assigned Bandwidth (kHz)	(d) No. of Phases	(e)Uncoded Data Rate (kbps)	(f) FEC Error Correction Coding Rate	(g) CDMA Processing Gain (dB)	(h) Total C/N Performance Objective (dB)	(i) Single Entry C/I Objective (dB)
TDM	1M84G1W	1840	4	2048	0.75		69.1	85
CMD	800KF2D	800	1	0.25				
TLM	100KG2D	100	2	4.8			48	
TDMH	1M84G1W	1840	12				70.6	85
SIR	4M50G7E	4500	4	7515	0.6667		4	

Page 7: Digital Modulation

Page 8: Analog Modulation

S12. ANALOG MODULATION PARAMETERS For each analog emission provide:

(a)	(b) Emission	(c)	(d) Signal	(e)		Multi-channel Telephony				(k) Video	(I) Video	()	(n) Total C/N	(-) - 5 -
Analog Mod. ID		Assigned Bandwidth (kHz)	Туре	Channels per Carrier	(f) Ave. Companded Talker Level (dBm0)	(g) Bottom Baseband Freq. (MHz)	(h) Top Baseband Freq. (MHz)	(i) RMS Modulation Index	Standard NTSC, PAL, etc.	Noise- Weighting (dB)	and SCPC/FM Modulation Index	Compander, Preemphasis, and Noise Weighting (dB)	Performance Objective (dB)	Entry C/I Objective (dB)

Page 9: Typical Emissions

S13. TYPICAL EMISSIONS For each planned type of emission provide:

Associated Transponder ID Range (a) Start (b) End		Modulation ID		(e) Carriers (f) Carrie		(0)	(h) Energy	Receive Band (Assoc. Transmit Stn)			Transmit Band (This Space Station)			
		(C) Digital (d) Analog (Table (Table S12)	per Spacing Transponder (kHz)	Spacing (kHz)	Reference (Table No.)	Dispersal Bandwidth (kHz)	(i)Assoc. Stn. Max.	Assoc. Station Transmit Power (dBW)		EIRP (dBW)		Power Flux	(o)Assoc. Stn	
(d) etait	(2) 2.10	S11)	S11)		(1112)	Antenna Gain (dBi)	(j) Min.	(k) Max.	(I) Min.	(m) Max.	Density (dBW/m2/Hz)	Rec. G/T (dB/K)		
1	5	TDM		1			0	52.3	41	64	60	71	-118.58	-20
CMD1	CMD2	CMD		1			0	52.4	64	78				
TLM1	TLM1	TLM		1							18	34	-150	21.8
S1	S2	SIR		1				50.5	40	60	60	71	-118.58	-20

S14. Is the space station(s) controlled and monitored remotely? If Yes, provide the location and telephone number of the TT and C control point(s): No

Remote Control (TT C) Location(s):

S14a: Street Address: 2875 Fork Creek Church Rd.								
S14b. City: Ellenwood	, , , , , , , , , , , , , , , , , , , ,		S14d. State/Country GA	S14e. Zip Code: 30294				
S14f. Telephone Number: 4043812000		S14g. Call Sign of C E040204	S14g. Call Sign of Control Station (if appropriate): E040204					
Remote Control (TT C) Location	on(s):							
S14a: Street Address: 1500 Eckington Place NE								
S14b. City: Washington	S14c. County:		S14d. State/Country DC	S14e. Zip Code: 20002				
S14f. Telephone Number: 2023804000		S14g. Call Sign of Control Station (if appropriate): E000158						
Remote Control (TT C) Location	on(s):							
S14a: Street Address: 1780 Centre Ave NE								
S14b. City: Calgary	S14c. County: Alberta		S14d. State/Country	S14e. Zip Code: T2E 0A6				
S14f. Telephone Number: 4032355751		S14g. Call Sign of C	Control Station (if appropriate):					
Remote Control (TT C) Location	on(s):							
S14a: Street Address: 133438 Allan Park Rd								
S14b. City: Allan Park	, , , , , , , , , , , , , , , , , , , ,		S14d. State/Country S14e. Zip Co N4N 3B8					
S14f. Telephone Number: 5193641013	I	S14g. Call Sign of C	Control Station (if appropriate):					

Remote Control (TT C) Location(s):					
S14a: Street Address: 24 Vernon Crossing Road					
S14b. City: Vernon	S14c. County: Sussex		S14d. State/Country NJ	S14e. Zip Code: 07462	
S14f. Telephone Number: 9737644021		S14g. Call Sign of Control Sta E040363	ition (if appropriate):	i	

Page 11: Characteristics and Certifications

S15. SPACECRAFT PHYSICAL CHARACTERISTICS:

S15a: Mass of spacecraft without fuel (kg): 2889	Spacecraft Dimensions (meters)	Probability of Survival to End of Life (0.0 - 1.0)
S15b. Mass of fuel and disposables at launch (kg): 3064		
S15c. Mass of spacecraft and fuel at launch (kg): 5953	S15f. Length (m): 8.5	S15i. Payload: 0.9093
S15d. Mass of fuel, in orbit, at beginning of life (kg): 350	S15g. Width (m): 27.3	S15j. Bus: 0.8401
S15e. Deployed Area of Solar Array (square meters): 89	S15h. Height (m): 32.4	S15k. Total: 0.7638

S16. SPACECRAFT ELECTRICAL CHARACTERISTICS:

Spacecraft Subsystem	Electrical Pov Beginnir	ver (Watts) At ng of Life	Electrical Power (Watts) At End of Life				
	At Equinox	At Solstice	At Equinox	At Solstice			
Payload (Watts):	^{(a):} 13250	^{(f):} 13250	^{(k):} 13250	^{(p):} 13250			
Bus (Watts):	^{(b):} 1890	^{(g):} 1890	^{(I):} 1890	^{(q):} 1890			
Total (Watts):	^{(c):} 15140	^{(h):} 15140	^(m) 15140	^{(r):} 15140			
Solar Array (Watts):	^{(d):} 21732	^{(i):} 19370	^{(n):} 16839	^{(s):} 16822			
Depth of Battery Discharge (%):	^(e) 65.1 %	^(j) 0 %	⁽⁰⁾ 68.7 %	^(t) 0 %			

S17. CERTIFICATIONS:

a. Are the power flux density limits of § 25.208 met?:	X	YES		NO		N/A			
b. Are the appropriate service area coverage requirements of § 25.143(b)(ii) and (iii), or § 25.145(c)(1) and (2) met		YES		NO	X	N/A			
c. Are the frequency tolerances of § 25.202(e) and the out-of-band emission limits of § 25.202(f)(1), (2) and (3) met	? X	YES		NO		N/A			
In addition to the information required in this Form, the space station applicant is required to provide all the information specified in Section 25.114 of the									
Commission's rules, 47 C.F.R § 25.114.									