FCC	312	
Sche	dule	S

FEDERAL COMMUNICATIONS COMMISSION SATELLITE SPACE STATION AUTHORIZATIONS (Technical and Operational Description)

Page 1: General, Frequency Bands, and GSO Orbit

a. Space Station or Satellite N DIRECTV 13	etwork Name:	e. Estimated Date of Placement into Service:		i Will the space station(s) operate on a Common Carrier Basis: N		
b. Construction Commenceme	ent Date:	f. Estimated Lifetime of Satellite(s): 15	Years	j. Number of transponders offered on a com 0	nmon carrier basis:	
c. Construction Completion Date:		g. Total Number of Transponders:		k. Total Common Carrier Transponder Bandwidth: 0 MHz		
d1. Est Launch Date Begin:	d2. Est Launch Date End:	h. Total Transponder Bandwidth (no. transponde 82	ers x Bandwidth) MHz	I. Orbit Type: Mark all boxes that apply: X GSO	NGSO	

S2. OPERATING FREQUENCY BANDS Identify the frequency range and transmit/receive mode for all frequency bands in which this station will oper Also indicate the nature of service(s) for each frequency band.

	Frequency	Band Limits						
Lower Frequency ((_Hz) Upper Frequency (_Hz)		_Hz)	e. T/R Mode	f. Nature of Service(s): List all that apply to this band			
a. Numeric	b. Unit (K/M/G)	c. Numeric	d. Unit (K/M/G)					
17.3	G	17.8	G	R	Fixed Satellite Service			
12.2	G	12.7	G	Т	Broadcasting Satellite Service - Video			

S3. ORBITAL INFORMATION FOR GEOSTATIONARY SATELLITES ONLY:

a. Nominal Orbital Longitude (Degrees E/W):		b. Alternate Orbital Longitude (Degrees E/W):				c. Reason for orbital location selection:
109.8 W						
Longitudinal Tolerance or E/\	W Station-Keeping:	f. Inclination Excursion or	Range of orbital are in which	ch adequate serv	rice can be	
d. Toward West: e. Toward East:	0.05 Degrees 0.05 Degrees	N/S Station-Keeping Tolerance: 0.05 Degrees	provided (Optional): g. Westernmost: h. Easternmost:	Degrees	E/W	
i. Reason for service are	selection (Optional):		•			

Page 2: NGSO Orbits

FCC Form 312 - Schedule S: (Technical and Operational Description)

S4. ORBITAL INFORMATION FOR NON-GEOSTATIONARY SATELLITES ONLY

S4a. Total Number of Satellites in Network or System:

S4c. Celestial Reference Body (Earth, Sun, Moon, etc.):

S4b. Total Number of Orbital Planes in Network or System: S4d. Orbit Epoch Date:

For each Orbital Plane Provide:

(e) Orbital	(f) No. of	(g) Inclination	(h) Orbital	(i) Apogee (km)	(j) Perigee (km)	(k) Right Ascension	(I) Argument of	Active Se	rvice Arc Rang	e (Degrees)
Plane No.	Satellites in	Angle (degrees)	Period			of the Ascending	Perigee	(m) Begin	(n) End	(o) Other
	Plane		(Seconds)			Node (Deg.)	(Degrees)	Angle	Ångle	,
			((13 111)	7g.c	79.0	

S5. INITIAL SATELLITE PHASE ANGLE For each satellite in each orbital plane, provide the intital phase angle.

(a) Orbital Plane No.	(b) Satellite Number	(c) Initial Phase Angle
		(Degrees)

NO NGSO DATA FILED

FCC Form 312 - Schedule S: (Technical and Operational Description)

S6. SERVICE AREA CHARACTERISTICS for each service area provide:

(a) Service Area ID	(b) Type of Associated Station (Earth or		(d) Service Area Description. Provide list of geographic areas (state postal codes or ITU 3-ltr codes), satellites or Figure No. of Service Area Diagram.
	Space)	(2.1. 1.1.)	
CONUS1	S		CONUS
LACR1	S		Los Angeles, CA+Castle Rock, CO
AKHI	S		Alaska+Hawaii

Page 3: Service Areas

Page 4: Antenna Beams

FCC Form 312 - Schedule S: (Technical and Operational Description)

S7. SPACE STATION ANTENNA BEAM CHARACTERISTICS For each antenna beam provide:

(a)	(b)	Isotropic	Antenna	(e)	(f)	(g) Min.	(h) Polar-	(i) Polarization	(j) Service		Transmit				Receive		
Beam	T/R	Ga	ain	3	Rotational	Cross-	ization	Alignment Rel.	Area ID	(k)	(I) Effective	(m)	(n)	(o) G/T	(p) Min.	Input Attenu	uator (dB)
ID	Mode		(d) Edge	Error		Polar Iso-	Switch-	Equatorial		Input	Output	Max.	System	Max.	Saturation	(q) Max.	(r) Step
		(dBi)	(dBi)	(Degrees)	(Degrees)	lation (dB)	able? (Y/N)	Plane (Degrees)		Losses	Power (W)	EIRP			Flux Density	Value	Size
							(Y/IN)			(dB)		(dBW)	Temp (k)	(db/K)	(dBW/m2)		
DL_1	Т	36.4	30.4	0.2		30	Ν		CONUS1	2.7	600	61.5					
UL1	R	35	31	0.2		30	N		LACR1				900	5.5	-93.5	20	1
DL2	Т	36	30	0.2		30	N		AKHI	3	56.2	50.5					

FCC Form 312 - Schedule S: (Technical and Operational Description)

S8. ANTENNA BEAM DIAGRAMS For each beam pattern provide the reference to the graphic image and numerical data:
Also provide the power flux density levels in each beam that result from the emission with the highest power flux density.

(a)	(b)	(c) Co-or	` '	(e) NGSO Antenna Gain	(f) GSO Antenna Gain Contour Data			lux Density (dB		
Beam	T/R	Cross	Ref.	Contour Description		At Angle of	Arrival above he	orizontal (for em	ission with hig	hest PFD)
ID	Mode	Polar Mode ("C" or" X")	Orbital Longitude (Deg. E/W)	(Figure/Table/ Exhibit)	(GXT File)	(g) 5 Deg	(h) 10 Deg	(i) 15 Deg	(j) 20 Deg	(k) 25 Deg
DL_1	Т	С	-101		DPTX_co.gxt					
DL2	Т	С	-101		AHTX_co.gxt					
UL1	R	С	-101		DPRX_co.gxt					

Page 5: Beam Diagrams

Page 6: Channels and Transponders

FCC Form 312 - Schedule S: (Technical and Operational Description)

S9. SPACE STATION CHANNELS For each frequency channel provide: S10. SPACE STATION TRANSPONDERS For each transponder provide:

(a) Channel No.	(B) Assigned Bandwidth (kHz)	(c) T/R Mode	(d) Center Frequency (MHz)	(e) Polarization (H, V, L, R)	(f) TTC or Comm Channel (T or C)
R028	24000	R	17717.66	L	С
R030	24000	R	17746.82	L	С
R032	24000	R	17775.98	L	С
RWB1A	36000	R	17724	L	С
RWB1B	36000	R	17770	L	С
RWB2A	82000	R	17746.82	L	С
T028	24000	T	12617.66	L	С
T030	24000	Т	12646.82	L	С
T032	24000	Т	12675.98	L	С
TWB1A	36000	Т	12624	L	С
TWB1B	36000	T	12670	L	С
TWB2A	82000	Т	12646.82	L	С
CMD1	1000	R	17305	L	T
CMD2	1000	R	17307	L	Т
TLM1	1000	Т	12203	L	Т
TLM2	1000	Т	12204	L	Т

(a)	(b)	Receive	Band	Transmi	t Band
Transponder ID	Transponder Gain (dB)	(c) Channel No.	(d) Beam ID	(e) Channel No.	(f) Beam ID
T0001	133.6	R028	UL1	T028	DL_1
T0002	133.6	R030	UL1	T030	DL_1
T0003	133.6	R032	UL1	T032	DL_1
T0004	133.6	RWB1A	UL1	TWB1A	DL_1
T0005	133.6	RWB1B	UL1	TWB1B	DL_1
T0006	133.6	RWB2A	UL1	TWB2A	DL_1
T0007	133.6	R028	UL1	T028	DL2
T0008	133.6	R030	UL1	T030	DL2
T0009	133.6	R032	UL1	T032	DL2
T0010	133.6	RWB1A	UL1	TWB1A	DL2
T0011	133.6	RWB1B	UL1	TWB1B	DL2
T0012	133.6	RWB2A	UL1	TWB2A	DL2
TLMA				TLM1	DL_1
TLMB				TLM2	DL_1
CMDA		CMD1	UL1		
CMDB	_	CMD2	UL1		

Page 7: Digital Modulation

FCC Form 312 - Schedule S: (Technical and Operational Description)

S11. DIGITAL MODULATION PARAMETERS For each digital emission provide:

(a) Digital Mod. ID	(b) Emission Designator	(c) Assigned Bandwidth (kHz)	(d) No. of Phases	(e)Uncoded Data Rate (kbps)	(f) FEC Error Correction Coding Rate	(g) CDMA Processing Gain (dB)	(h) Total C/N Performance Objective (dB)	(i) Single Entry C/I Objective (dB)
DTH	24M0G7W	24000	4	40000	0.48		7.6	28
DTH2	36M0G7W	36000	4	60000	0.48		7.6	28
DTH3	82M0G7W	82000	4	136700	0.48		7.6	28

Page 8: Analog Modulation

FCC Form 312 - Schedule S: (Technical and Operational Description)

S12. ANALOG MODULATION PARAMETERS For each analog emission provide:

` '	(b) Emission	(c)	(d) Signal	(e)	' '			(j) Video	(k) Video	(I) Video	(m) SCPC/FM	` '	` '	
Analog Mod. ID		Assigned Bandwidth (kHz)	Туре	Channels per Carrier	(f) Ave. Companded Talker Level (dBm0)	(g) Bottom Baseband Freq. (MHz)	(h) Top Baseband Freq. (MHz)	(i) RMS Modulation Index	Standard NTSC, PAL, etc.		and SCPC/FM Modulation Index	Compander, Preemphasis, and Noise Weighting (dB)	Performance Objective (dB)	Entry C/I Objective (dB)
TLM	1M00F9D	1000		1									7.6	28
CMD	1M00F9D	1000		1									7.6	28

Page 9: Typical Emissions

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S13. TYPICAL EMISSIONS For each planned type of emission provide:

Associated		Modulation ID		` '	(f) Carrier		(h) Energy	Receive Ba	and (Assoc. Tr	Transmit Band (This Space Station)				
	er ID Range	(C) Digital (d) F	(c) Digital (d) Analog (Table (Table S12)	(c) Digital (d) Analog Transp		Dispersal Bandwidth	(i)Assoc. Stn. Max.	Assoc. Station		EIRP	(dBW)	(n) Max. Power Flux	(o)Assoc. Stn	
(a) Start) Start (b) End S11)			(kHz)	Antenna Gain (dBi)	(j) Min.	(k) Max.	(I) Min.	(m) Max.	Density (dBW/m2/Hz)	Rec. G/T (dB/K)			
T0001	T0006	DTH		1				64.2	12.8	21.8	52	58		4.9
T0001	T0006	DTH2		1				64.2	14.6	23.6	53.8	59.8		4.9
T0001	T0006	DTH3		1				64.2	18.1	27.1	55.5	61.5		4.9
T0007	T0012	DTH		1				64.2	12.8	21.8	41	47		13
T0007	T0012	DTH2		1				64.2	14.6	23.6	42.8	48.8		13
T0007	T0012	DTH3		1				64.2	18.1	27.1	44.5	50.5		13
TLMA	TLMB		TLM	1							12	18.3		40
CMDA	CMDB		CMD	1				64.2	-3.6	48.4				

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Page 10: TT and C

FCC Form 312 - Schedule S: (Technical and Operational Description)

S14. Is the space station(s) controlled and monitored remotely? If Yes, provide the location and telephone number of the TT and C control point(s): No

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Page 11: Characteristics and Certifications

S15. SPACECRAFT PHYSICAL CHARACTERISTICS:

S15a: Mass of spacecraft without fuel (kg): 1480	Spacecraft Dimensions (meters)	Probability of Survival to End of Life (0.0 - 1.0)
S15b. Mass of fuel and disposables at launch (kg): 2230		
S15c. Mass of spacecraft and fuel at launch (kg): 3710	S15f. Length (m): 31.3	S15i. Payload: 0.8845
S15d. Mass of fuel, in orbit, at beginning of life (kg): 2060	S15g. Width (m): 8.7	S15j. Bus: 0.9115
S15e. Deployed Area of Solar Array (square meters): 60.6	S15h. Height (m): 6.2	S15k. Total: 0.8062

S16. SPACECRAFT ELECTRICAL CHARACTERISTICS:

Spacecraft Subsystem	Electrical Pov Beginnir	ver (Watts) At ng of Life	Electrical Power (Watts) At End of Life				
	At Equinox	At Solstice	At Equinox	At Solstice			
Payload (Watts):	^{(a):} 5958	^{(f):} 5958	^{(k):} 5800	^{(p):} 5800			
Bus (Watts):	^{(b):} 1864	^{(g):} 955	^{(l):} 1900	(q): 900			
Total (Watts):	^{(c):} 7833	^{(h):} 6913	^(m) 7700	^{(r):} 6700			
Solar Array (Watts):	^{(d):} 9340	^{(i):} 8379	^{(n):} 8400	^{(s):} 7659			
Depth of Battery Discharge (%):	^(e) 74 %	^(j) 0 %	⁽⁰⁾ 72 %	^(t) 0 %			

S17. CERTIFICATIONS:

a. Are the power flux density limits of § 25.208 met?:	X YES	3	NO	N/A				
b. Are the appropriate service area coverage requirements of § 25.143(b)(ii) and (iii), or § 25.145(c)(1) and (2) met?	YES	3	NO	X N/A				
c. Are the frequency tolerances of § 25.202(e) and the out-of-band emission limits of § 25.202(f)(1), (2) and (3) met	? X YES	3	NO	N/A				
In addition to the information required in this Form, the space station applicant is required to provide all the information specified in Section 25.114 of the								

Commission's rules, 47 C.F.R § 25.114.