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FEDERAL COMMUNICATIONS COMMISSION  
OFFICE OF SECRETARY

Before the  
FEDERAL COMMUNICATIONS COMMISSION  
Washington, D.C. 20554

Received

JAN 16 1996

In the Matter of )  
)  
Application of )  
HUGHES COMMUNICATIONS GALAXY, INC. )  
)  
For Authority to Construct, Launch )  
and Operate GALAXY/SPACEWAY, a )  
Global Interconnected System of )  
Geostationary Ka-band Fixed-Service )  
and Ku-band Broadcast )  
Communications Satellites )

Satellite Engineering Branch  
Int'l

File Nos. 174-181-SAT-P/LA-95  
3/4-DSS-P/LA-94  
CSS-94-021-25

**COMMENTS OF ORION ASIA PACIFIC CORPORATION**

Orion Asia Pacific Corporation ("Orion"), an applicant for authority to construct, launch and operate a proposed hybrid Ku/Ka-band international separate satellite space station at 126° E.L. (File Nos. CSS-94-009; 206-SAT-AMEND-95), hereby submits its comments on the above-captioned application of Hughes Communications ("Hughes"), as amended September 29, 1995, for authority to construct, launch, and operate a global, interconnected system of hybrid geostationary satellites.<sup>1/</sup> The Hughes application includes a proposed space station with a Ka-band payload to be located at 125° E.L. Orion's comments are directed at that portion of the Hughes system application.

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<sup>1/</sup> The Orion and Hughes applications were accepted for filing on Nov. 1, 1995. See Public Notice, Report No. SPB-29, DA 95-2273. Pursuant to that notice, comments or petitions on these applications may be filed on or before December 15, 1995.

On October 26, 1995, all of the Ka-band satellite applicants in the current processing round were assembled by the Commission's International Bureau staff for a "Status Conference." The applicants were informed that the generic Appendix-4 ("AP-4") materials for Ka-band satellite systems to be submitted by the Commission before the close of the 1995 World Radiocommunications Conference to the International Telecommunications Union ("ITU") Radiocommunications Bureau must, among other things, conform to 2° spacing, e.g., the Commission would not submit AP-4 materials for a Ka-band satellite at both the 126° E.L. and 125° E.L. locations. Subsequently, the Ka-band applicants worked cooperatively to develop a joint proposal for the Commission's AP-4 submission which was submitted to the Commission on November 9, 1995.<sup>2/</sup>

On October 27, 1995, Dr. Robert Sorbello, Orion's Director of Systems Engineering, met with Mr. Vu Phang, Systems Engineer with Hughes' Galaxy Systems Engineering, to discuss the apparent conflict in the requested orbital locations of Hughes at 125° E.L. and of Orion at 126° E.L. Orion and Hughes discussed the possibility of simultaneous operation of both the Hughes and Orion space stations from the proposed orbital locations with 1° spacing, provided that suitable spectrum sharing arrangements could be made. Both Orion and Hughes agreed to collaborate to achieve this type of accommodation. However, in order to comply with the Commission's instructions regarding the need to maintain 2° spacing in the AP-4 submissions, Orion and Hughes each agreed to a

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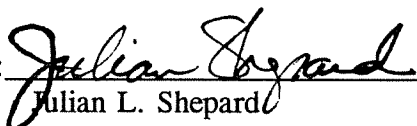
<sup>2/</sup> See Letter of Raymond G. Bender, Jr., counsel for Lockheed Martin Corporation, to the Chief of the Commission's Satellite Engineering Branch, enclosing a proposed "generic" AP-4 for submission by the Commission to the ITU's Radiocommunications Bureau.

generic AP-4 submission that provided orbital locations adjusted by  $.5^\circ$  in opposite directions (copies attached as Appendix "A" hereto), i.e.,  $124.5^\circ$  E.L. and  $126.5^\circ$  E.L.

In the event that Orion and Hughes cannot achieve suitable spectrum-sharing arrangements to permit one-degree spacing from the orbital locations proposed in their respective applications, Orion recommends that the Commission adjust the orbital locations specified in the Hughes and Orion applications to conform with the orbital locations specified in the generic AP-4 materials.

Respectfully submitted,

ORION ASIA PACIFIC CORPORATION

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April McClain-Delaney, Esq.  
Director of Regulatory Affairs  
Orion Asia Pacific Corporation  
2440 Research Boulevard, Suite 4000  
Rockville, Maryland 20850

December 15, 1995

## **APPENDIX A**

DATE (Day/Month/Year) <input type="text" value="01"/> <input type="text" value="11"/> <input type="text" value="95"/>	<b>FORM OF NOTICE</b> <b>SATELLITE NETWORK</b> (APPENDIX 4)	PAGE 1 OF <input type="text" value="31"/>	<b>AP4</b>
Administration Serial Number <input type="text" value=""/> <input type="text" value=""/> <input type="text" value=""/> <input type="text" value=""/> <input type="text" value=""/> <input type="text" value=""/>	NOTIFYING ADMINISTRATION <input type="text" value="USA"/> <input type="text" value=""/> <input type="text" value=""/> <input type="text" value=""/> <input type="text" value=""/> <input type="text" value=""/>		NOTIFICATION INTENDED FOR ADD <input checked="" type="checkbox"/> MOD <input type="checkbox"/> SUP <input type="checkbox"/>
RR1042 Advance Publication <input checked="" type="checkbox"/>	RR1047A Request for Assistance of the IFRB <input type="checkbox"/>	IFRB IDENTIFICATION NO OF NETWORK TO BE MODIFIED/SUPPRESSED <input type="text" value=""/> <input type="text" value=""/> <input type="text" value=""/> <input type="text" value=""/> <input type="text" value=""/> <input type="text" value=""/> <input type="text" value=""/> <input type="text" value=""/> <input type="text" value=""/> <input type="text" value=""/>	

## B: CHARACTERISTICS OF THE NETWORK

1 NAME OF THE SPACE STATION <input type="text" value="USASAT-33F"/>																																						
2 DATE OF BRINGING INTO USE Day Month Year <input type="text" value="01"/> <input type="text" value="11"/> <input type="text" value="2001"/>	REFERENCE TO PREVIOUS SPECIAL SECTION NUMBER (if network modified) <input type="text" value="AR111A"/> <input type="text" value=""/> <input type="text" value=""/> <input type="text" value=""/> Number																																					
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<b>4 ORBITAL INFORMATION</b>																																						
a. FOR GEOSTATIONARY SATELLITES																																						
1. NOMINAL ORBIT LONGITUDE <table border="1" style="width:100%; text-align: center;"><tr><th colspan="2">Degrees</th></tr><tr><td><input type="text" value="126"/> <input type="text" value="50"/> E</td><td><input type="text" value=""/></td></tr></table>	Degrees		<input type="text" value="126"/> <input type="text" value="50"/> E	<input type="text" value=""/>	2. LONGITUDINAL TOLERANCE <table border="1" style="width:100%; text-align: center;"><tr><th colspan="2">Degrees</th></tr><tr><td>To West <input type="text" value="0"/> <input type="text" value="05"/> <input type="text" value="0"/> <input type="text" value="05"/></td><td>To East <input type="text" value="0"/> <input type="text" value="05"/> <input type="text" value="0"/> <input type="text" value="05"/></td></tr></table>	Degrees		To West <input type="text" value="0"/> <input type="text" value="05"/> <input type="text" value="0"/> <input type="text" value="05"/>	To East <input type="text" value="0"/> <input type="text" value="05"/> <input type="text" value="0"/> <input type="text" value="05"/>	3. INCLINATION EXCURSION <table border="1" style="width:100%; text-align: center;"><tr><th>Degrees</th></tr><tr><td><input type="text" value="0"/> <input type="text" value="05"/></td></tr></table>	Degrees	<input type="text" value="0"/> <input type="text" value="05"/>	4. VISIBILITY ARC <table border="1" style="width:100%; text-align: center;"><tr><th colspan="4">Degrees</th></tr><tr><th>From W</th><th>EW</th><th>To E</th><th>EW</th></tr><tr><td><input type="text" value="051"/> E</td><td><input type="text" value="158"/> W</td><td><input type="text" value="158"/> W</td><td><input type="text" value="051"/> E</td></tr></table>	Degrees				From W	EW	To E	EW	<input type="text" value="051"/> E	<input type="text" value="158"/> W	<input type="text" value="158"/> W	<input type="text" value="051"/> E	5. SERVICE ARC <table border="1" style="width:100%; text-align: center;"><tr><th colspan="4">Degrees</th></tr><tr><th>From W</th><th>EW</th><th>To E</th><th>EW</th></tr><tr><td><input type="text" value="051"/> E</td><td><input type="text" value="158"/> W</td><td><input type="text" value="158"/> W</td><td><input type="text" value="051"/> E</td></tr></table>	Degrees				From W	EW	To E	EW	<input type="text" value="051"/> E	<input type="text" value="158"/> W	<input type="text" value="158"/> W	<input type="text" value="051"/> E
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b. FOR NON-GEOSTATIONARY SATELLITES																																						
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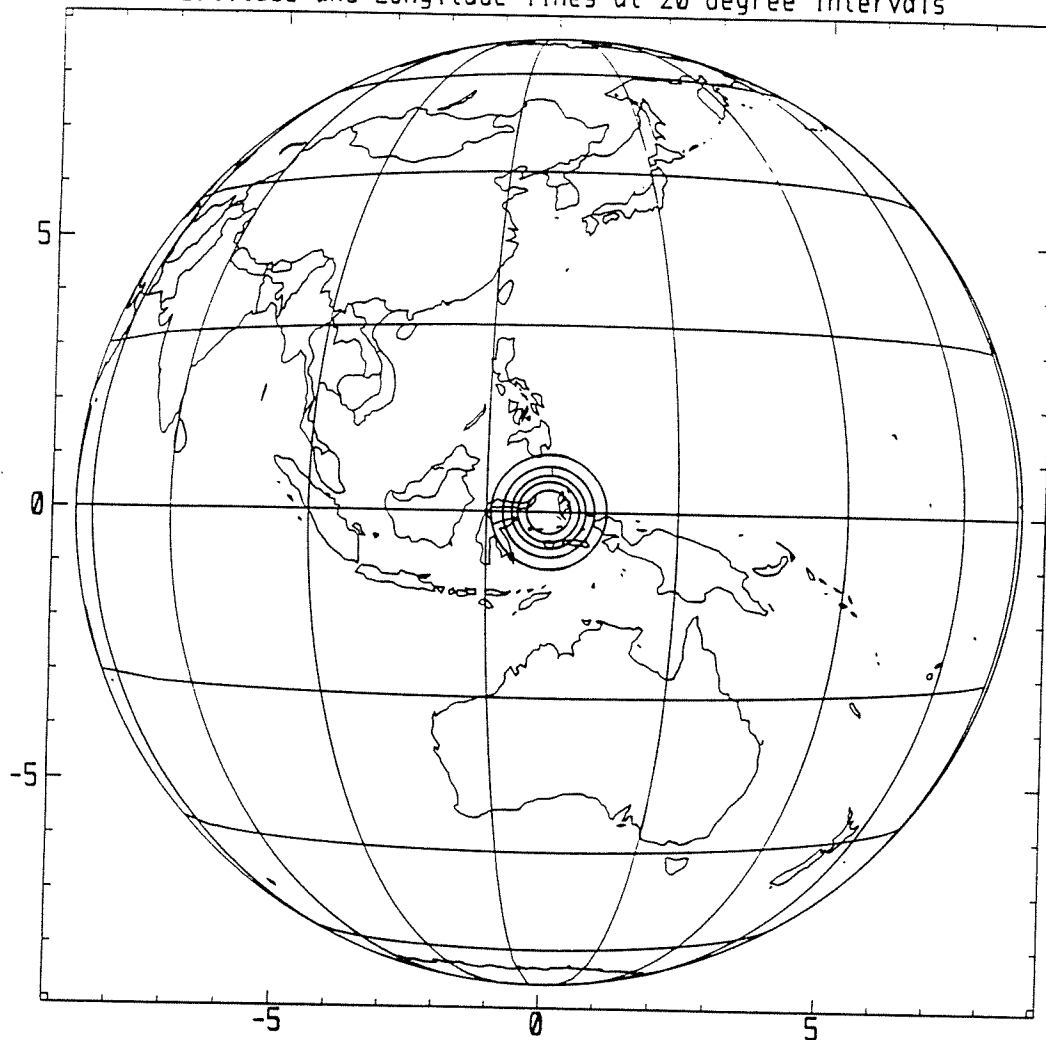
SEE NOTE 1.

### GENERAL NOTES :

- i. This form of notice consists of four parts - 1, 2, 3, and 4. In each part, each information item/data field includes a number in its label. This number is the same as that used for the same item in Appendix 4 (ORB-88) within the same part. For example, on the page labelled "Form AP4 - 2" (at the bottom), the field "4a1. Maximum power density" is the first item in section (a) of the paragraph numbered 4 in Part C. The items from parts F and G of Appendix 4 have been included in the parts C and D referred to above. The items from these parts have the letters F and G (correspondingly) preceding the number that is included in their labels.
- ii. Data items that are related are grouped together in a box. For example, the page labelled "Form AP4 - 2" (at the bottom) contains a box titled "Emissions and power characteristics". It is possible to specify 6 different emissions with the associated power and power density information in this box. If there are more emissions, use another page of the same type to provide additional data, after checking ( X ) the field labelled "More emissions on next page" on the preceding page. In all cases where there is more information than can fit in a box, follow this procedure.
- iii. This form can be used to add to, modify or suppress an existing station, by checking the corresponding box at the top right-hand corner of this page in the area titled "Notification intended for". In the case of a modification of an existing station, where certain data fields are to be added, modified or suppressed, provide ALL the data in the particular box as they would look after the change. In addition, indicate that the corresponding beam, associated station or frequency range value is being modified by entering M in the field that has been provided for this purpose at these levels.
- iv. Certain fields in this notice form have a superscript "1" as part of their labels. This has the following meaning :  
1 - This information is to be provided only if available.

FIGURE 01  
SPACE STATION RECEIVING AND TRANSMITTING  
ANTENNA GAIN CONTOURS FOR BEAM "K1R"

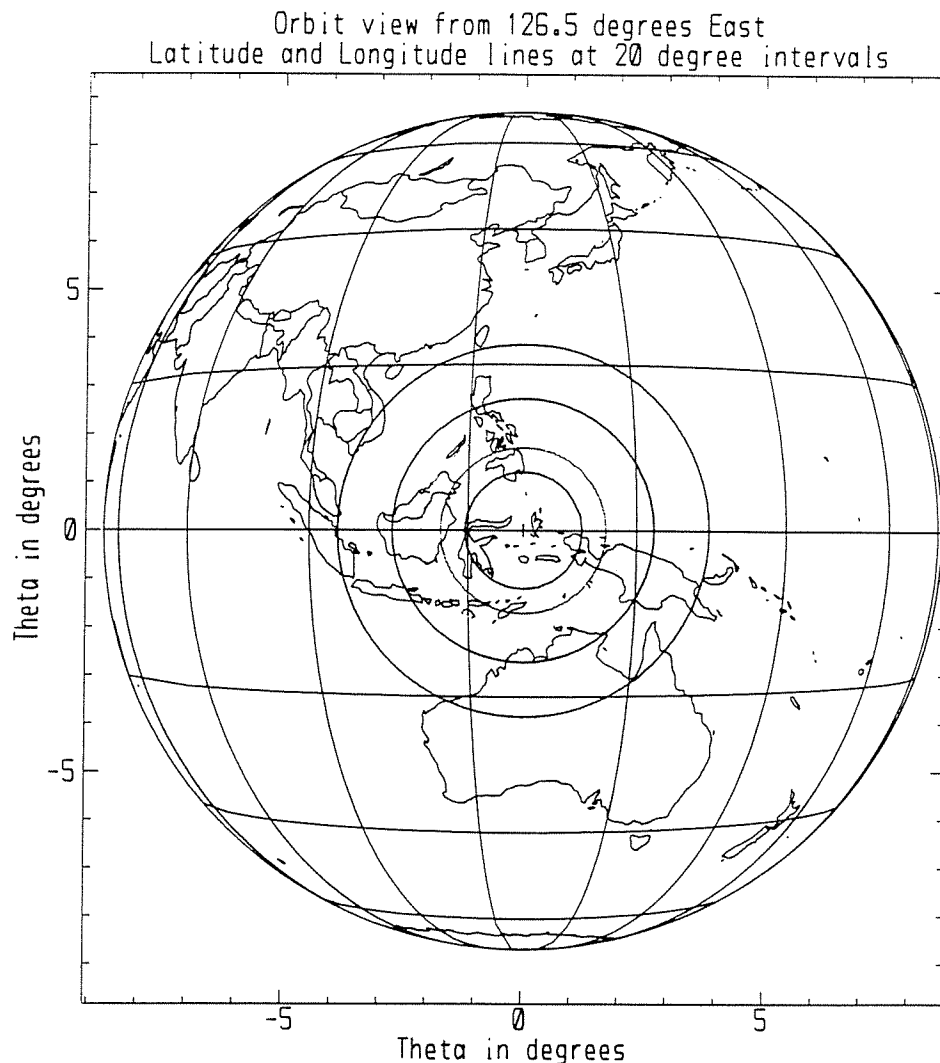
Orbit view from 126.5 degrees East  
Latitude and Longitude lines at 20 degree intervals



**Notes:**

1. The space station antenna coverage actually consists of multiple overlapping spot-beams, capable of being pointed to any part of the visible Earth's surface. Only one such typical spot-beam is shown as an example. All beams may operate at all associated assigned frequencies.
2. Service area consists of all visible areas of the Earth.
3. Maximum isotropic gain is +45.0 dBi.
4. Contours shown are -2, -4, -6, -10 and -20 dB relative to maximum gain.
5. Does not include antenna pointing error which is  $\pm 0.15^\circ$  worst case.
6. Power flux density at the Earth's surface will be controlled in order to be compliant with the limits specified in Article 28 of the Radio Regulations.

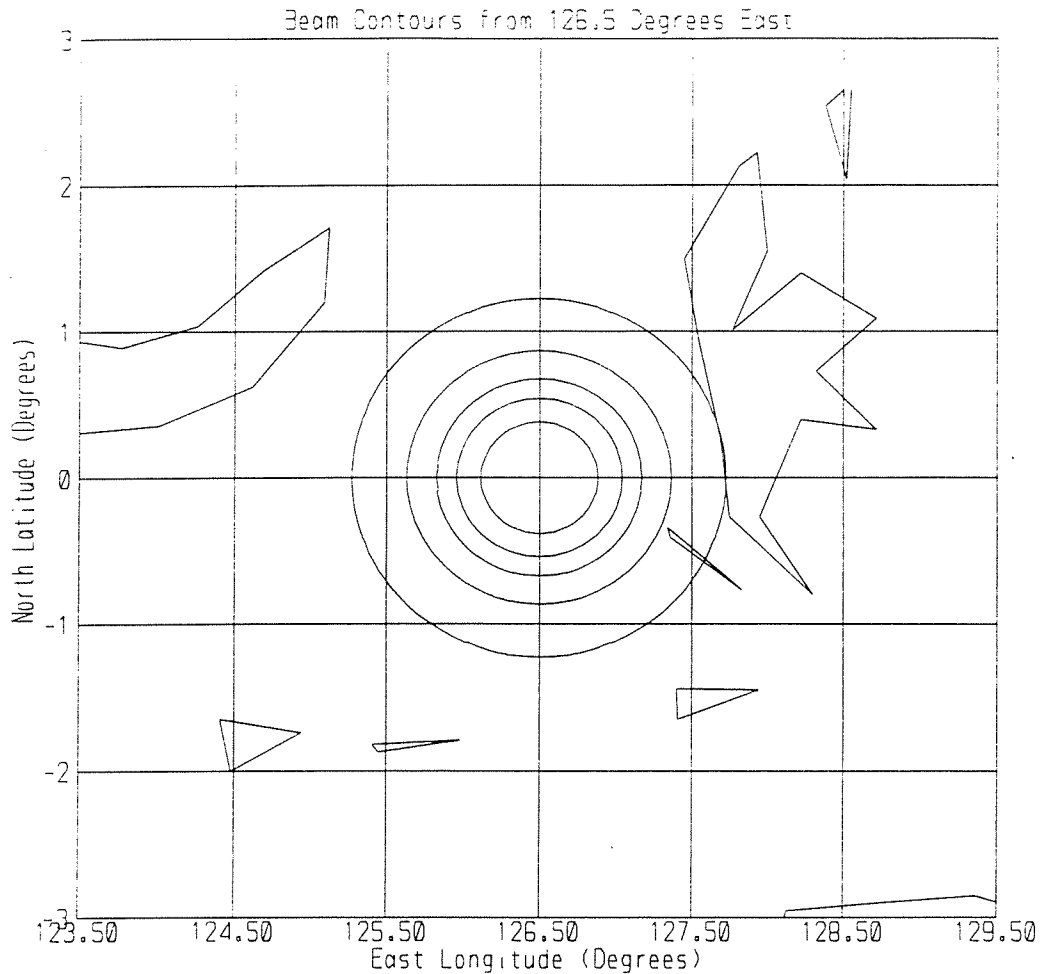
FIGURE 02  
SPACE STATION RECEIVING AND TRANSMITTING  
ANTENNA GAIN CONTOURS FOR BEAM "K2R"



**Notes:**

1. The space station antenna coverage actually consists of multiple overlapping spot-beams, capable of being pointed to any part of the visible Earth's surface. Only one such typical spot-beam is shown as an example. All beams may operate at all associated assigned frequencies.
2. Service area consists of all visible areas of the Earth.
3. Maximum isotropic gain is +35.0 dBi.
4. Contours shown are -2, -4, -6, -10 and -20 dB relative to maximum gain.
5. Does not include antenna pointing error which is  $\pm 0.15^\circ$  worst case.
6. Power flux density at the Earth's surface will be controlled in order to be compliant with the limits specified in Article 28 of the Radio Regulations.

**FIGURE 03  
SPACE STATION RECEIVING AND TRANSMITTING  
ANTENNA GAIN CONTOURS FOR BEAM "K3R"**



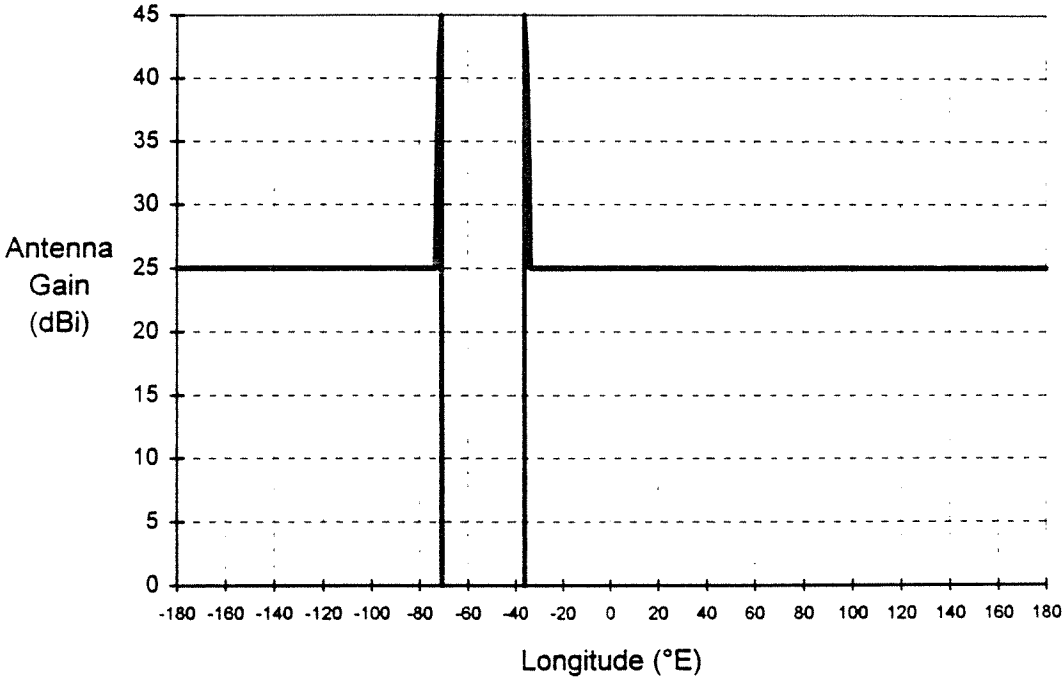
**Notes:**

1. The space station antenna coverage actually consists of multiple overlapping spot-beams, capable of being pointed to any part of the visible Earth's surface. Only one such typical spot-beam is shown as an example. All beams may operate at all associated assigned frequencies.
2. Service area consists of all visible areas of the Earth.
3. Maximum isotropic gain is +60.0 dBi.
4. Contours shown are -2, -4, -6, -10 and -20 dB relative to maximum gain.
5. Does not include antenna pointing error which is  $\pm 0.15^\circ$  worst case.
6. Power flux density at the Earth's surface will be controlled in order to be compliant with the limits specified in Article 28 of the Radio Regulations.



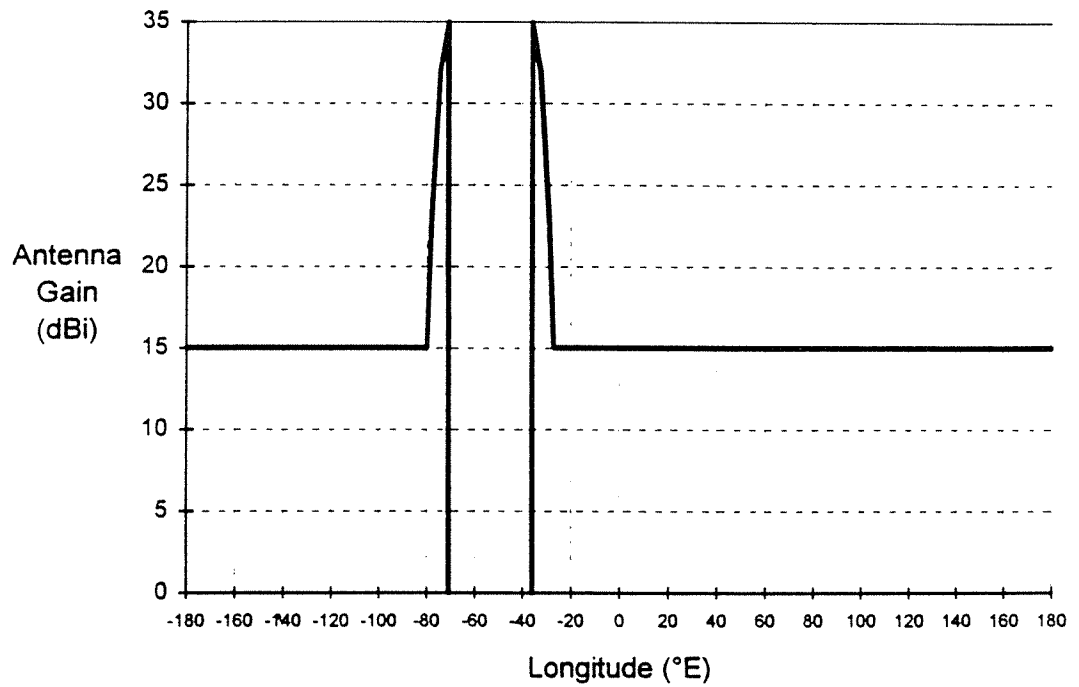
**FIGURE 04**  
**GAIN TOWARDS GEOSTATIONARY SATELLITE ORBIT**  
**FOR BEAM "K1R"**

Orbit Location 126.5E



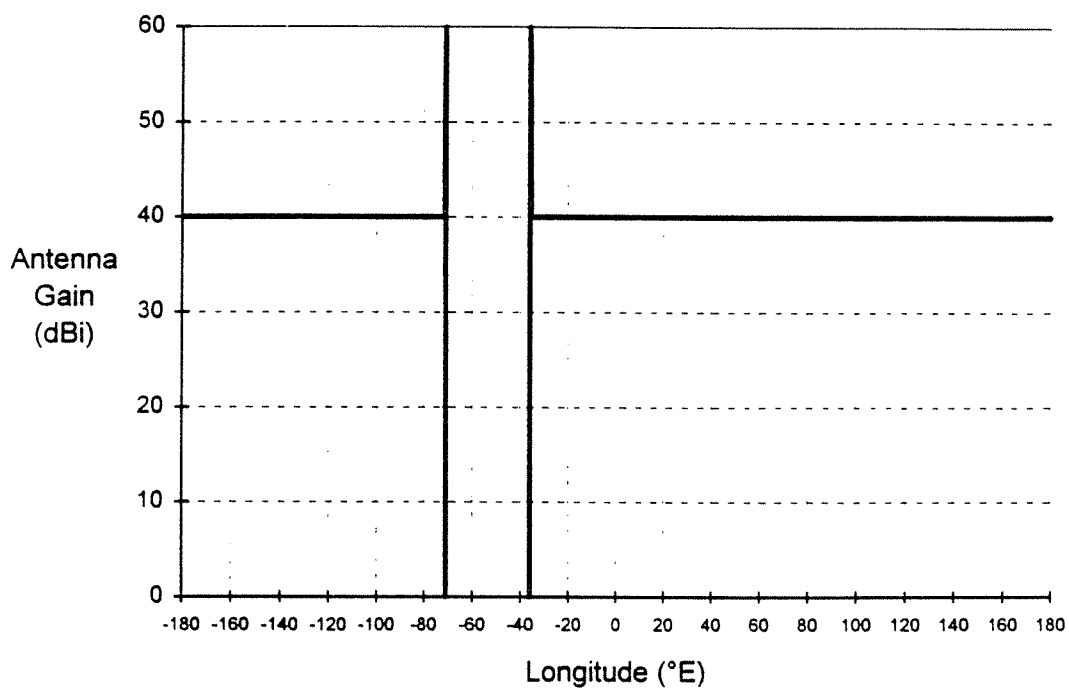
**FIGURE 05**  
**GAIN TOWARDS GEOSTATIONARY SATELLITE ORBIT**  
**FOR BEAM "K2R"**

Orbit Location 126.5E



**FIGURE 06**  
**GAIN TOWARDS GEOSTATIONARY SATELLITE ORBIT**  
**FOR BEAM "K3R"**

Orbit Location 126.5E



DATE (Day/Month/Year) <input type="text" value="9"/> <input type="text" value="5"/>	<b>FORM OF NOTICE</b> <b>SATELLITE NETWORK</b> (APPENDIX 4)	PAGE 1 OF <input type="text" value="3"/> <input type="text" value="1"/>	<b>AP4</b>
Administration Serial Number <input type="text"/>	RR1042 Advance Publication <input checked="" type="checkbox"/>	RR1047A Request for Assistance of the IFRB <input type="checkbox"/>	NOTIFICATION INTENDED FOR ADD <input checked="" type="checkbox"/> MOD <input type="checkbox"/> SUP <input type="checkbox"/>
NOTIFYING ADMINISTRATION <input type="text" value="USA"/>			IFRB IDENTIFICATION NO OF NETWORK TO BE MODIFIED/SUPPRESSED <input type="text"/>

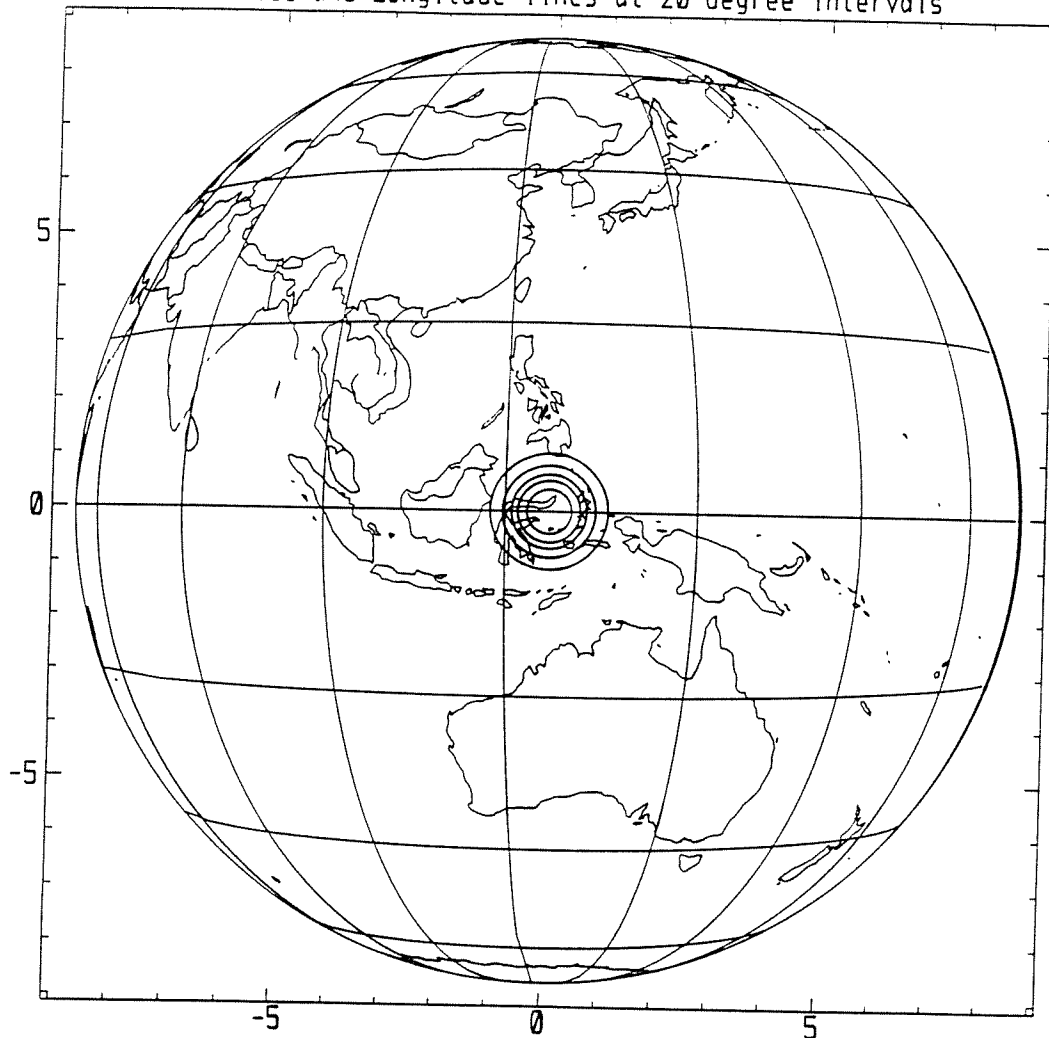
## B: CHARACTERISTICS OF THE NETWORK

1 NAME OF THE SPACE STATION <input type="text" value="USASAT-33E"/>	
2 DATE OF BRINGING INTO USE Day Month Year <input type="text" value="01"/> <input type="text" value="11"/> <input type="text" value="20"/> <input type="text" value="01"/>	REFERENCE TO PREVIOUS SPECIAL SECTION NUMBER (if network modified) <input type="text" value="AR11"/> <input checked="" type="checkbox"/> <input type="text" value="A"/> <input checked="" type="checkbox"/> <input type="text"/>
3a ADMINISTRATIONS IN GROUP <input type="text"/>	3b OPERATING AGENCY OR COMPANY <input type="text" value="120"/>
	3c ADMINISTRATION RESPONSIBLE FOR THE STATION <input type="text" value="A"/>
<b>4 ORBITAL INFORMATION</b>	
a. FOR GEOSTATIONARY SATELLITES	
1. NOMINAL ORBIT LONGITUDE Degrees EW <input type="text" value="124"/> <input checked="" type="checkbox"/> <input type="text" value="50"/> <input type="text" value="E"/>	2. LONGITUDINAL TOLERANCE Degrees To West To East <input type="text" value="0"/> <input checked="" type="checkbox"/> <input type="text" value="05"/> <input type="checkbox"/> <input checked="" type="checkbox"/> <input type="text" value="05"/>
3. INCLINATION EXCURSION Degrees <input type="text" value="0"/> <input checked="" type="checkbox"/> <input type="text" value="05"/>	4. VISIBILITY ARC Degrees From W EW To E EW <input type="text" value="0"/> <input type="text" value="49"/> <input type="text" value="E"/> <input type="text" value="1"/> <input type="text" value="60"/> <input type="text" value="W"/>
	5. SERVICE ARC Degrees From W EW To E EW <input type="text" value="0"/> <input type="text" value="49"/> <input type="text" value="E"/> <input type="text" value="1"/> <input type="text" value="60"/> <input type="text" value="W"/>
6. REASON FOR SERVICE ARC < VISIBILITY ARC ATTACHED <input type="text"/>	
b. FOR NON-GEOSTATIONARY SATELLITES	
1. INCLINATION ANGLE Degrees <input type="text"/> <input checked="" type="checkbox"/>	2. PERIOD Days D Hours Hours H Min. <input type="text"/> <input type="text"/> <input type="text"/> <input type="text"/>
3. APOGEE (km) <input type="text"/>	4. PERIGEE (km) <input type="text"/>
5. CELESTIAL BODY <input type="text"/>	6. NUMBER OF SATS. <input type="text"/>
SEE NOTE 1.	
GENERAL NOTES :	

- i. This form of notice consists of four parts - 1, 2, 3, and 4. In each part, each information item/data field includes a number in its label. This number is the same as that used for the same item in Appendix 4 (ORB-88) within the same part. For example, on the page labelled "Form AP4 - 2" (at the bottom), the field "4a1. Maximum power density" is the first item in section (a) of the paragraph numbered 4 in Part C. The items from parts F and G of Appendix 4 have been included in the parts C and D referred to above. The items from these parts have the letters F and G (correspondingly) preceding the number that is included in their labels.
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- iv. Certain fields in this notice form have a superscript "1" as part of their labels. This has the following meaning :  
1 - This information is to be provided only if available.

FIGURE 01  
SPACE STATION RECEIVING AND TRANSMITTING  
ANTENNA GAIN CONTOURS FOR BEAM "K1R"

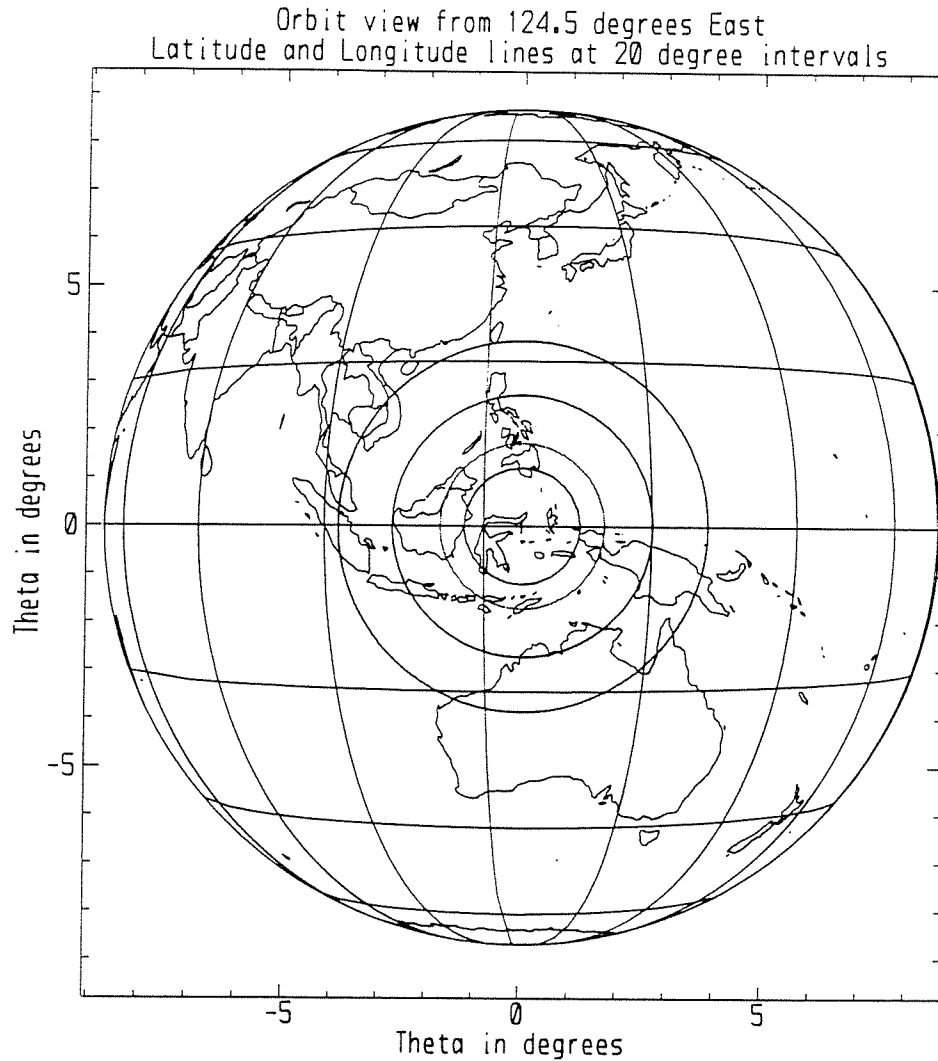
Orbit view from 124.5 degrees East  
Latitude and Longitude lines at 20 degree intervals



**Notes:**

1. The space station antenna coverage actually consists of multiple overlapping spot-beams, capable of being pointed to any part of the visible Earth's surface. Only one such typical spot-beam is shown as an example. All beams may operate at all associated assigned frequencies.
2. Service area consists of all visible areas of the Earth.
3. Maximum isotropic gain is +45.0 dBi.
4. Contours shown are -2, -4, -6, -10 and -20 dB relative to maximum gain.
5. Does not include antenna pointing error which is  $\pm 0.15^\circ$  worst case.
6. Power flux density at the Earth's surface will be controlled in order to be compliant with the limits specified in Article 28 of the Radio Regulations.

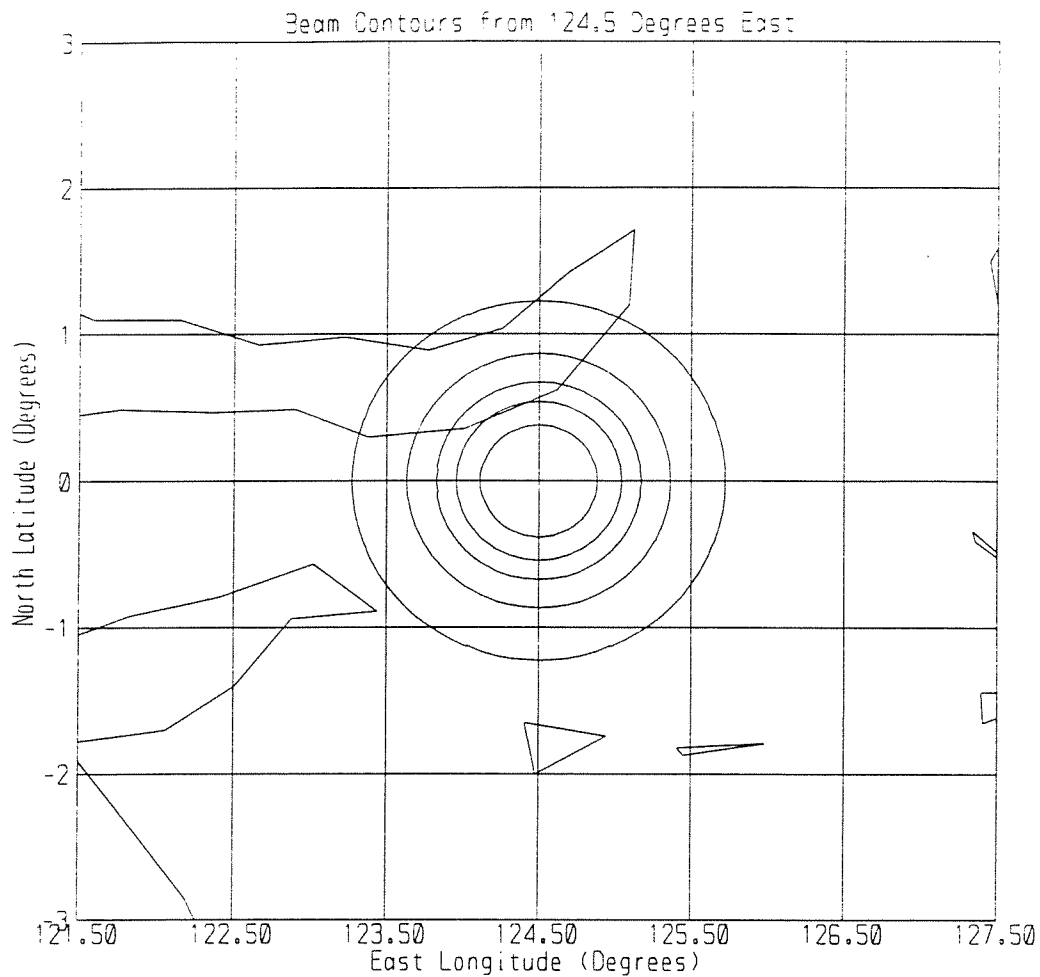
FIGURE 02  
SPACE STATION RECEIVING AND TRANSMITTING  
ANTENNA GAIN CONTOURS FOR BEAM "K2R"



**Notes:**

1. The space station antenna coverage actually consists of multiple overlapping spot-beams, capable of being pointed to any part of the visible Earth's surface. Only one such typical spot-beam is shown as an example. All beams may operate at all associated assigned frequencies.
2. Service area consists of all visible areas of the Earth.
3. Maximum isotropic gain is +35.0 dBi.
4. Contours shown are -2, -4, -6, -10 and -20 dB relative to maximum gain.
5. Does not include antenna pointing error which is  $\pm 0.15^\circ$  worst case.
6. Power flux density at the Earth's surface will be controlled in order to be compliant with the limits specified in Article 28 of the Radio Regulations.

**FIGURE 03  
SPACE STATION RECEIVING AND TRANSMITTING  
ANTENNA GAIN CONTOURS FOR BEAM "K3R"**

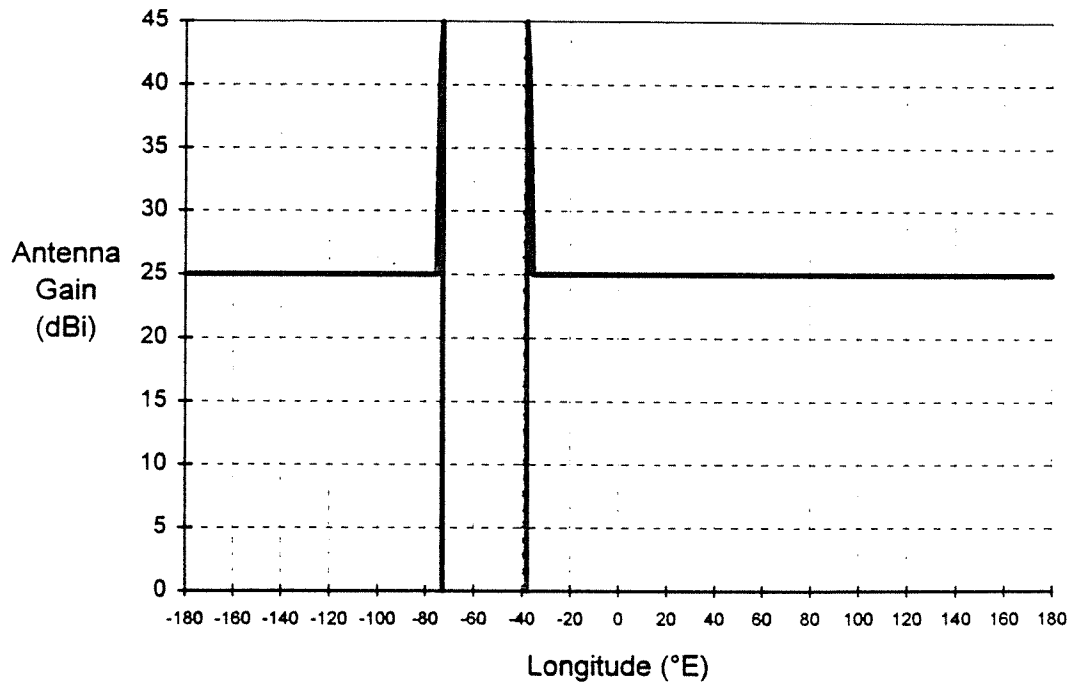


**Notes:**

1. The space station antenna coverage actually consists of multiple overlapping spot-beams, capable of being pointed to any part of the visible Earth's surface. Only one such typical spot-beam is shown as an example. All beams may operate at all associated assigned frequencies.
2. Service area consists of all visible areas of the Earth.
3. Maximum isotropic gain is +60.0 dBi.
4. Contours shown are -2, -4, -6, -10 and -20 dB relative to maximum gain.
5. Does not include antenna pointing error which is  $\pm 0.15^\circ$  worst case.
6. Power flux density at the Earth's surface will be controlled in order to be compliant with the limits specified in Article 28 of the Radio Regulations.

**FIGURE 04**  
**GAIN TOWARDS GEOSTATIONARY SATELLITE ORBIT**  
**FOR BEAM "K1R"**

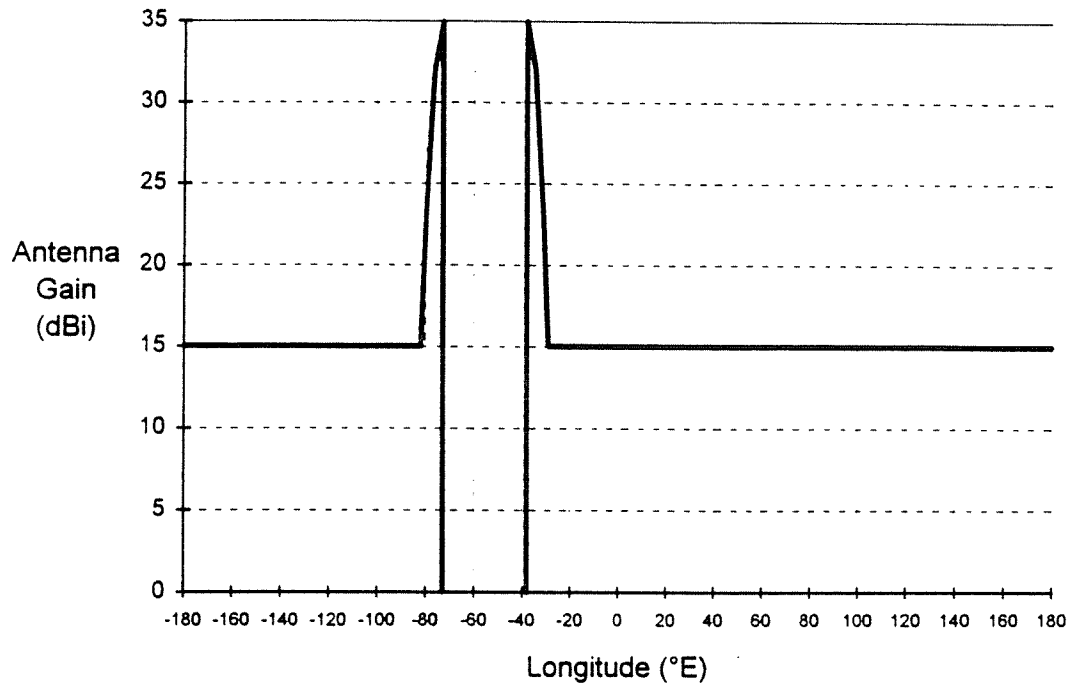
Orbit Location 124.5E





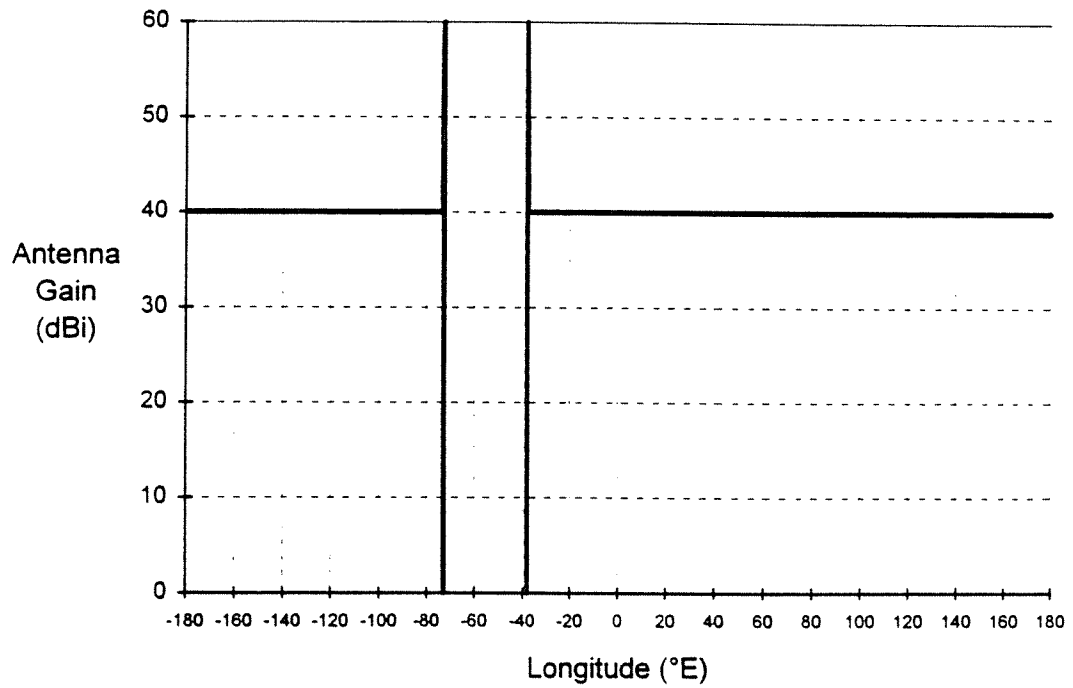
**FIGURE 05**  
**GAIN TOWARDS GEOSTATIONARY SATELLITE ORBIT**  
**FOR BEAM "K2R"**

Orbit Location 124.5E



**FIGURE 06**  
**GAIN TOWARDS GEOSTATIONARY SATELLITE ORBIT**  
**FOR BEAM "K3R"**

Orbit Location 124.5E



CERTIFICATE OF SERVICE

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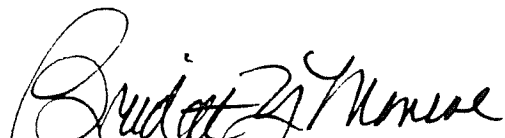
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