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Federal Republic of Germany

Bundesamt für Seeschifffahrt und Hydrographie
Federal Maritime and Hydrographic Agency



BUNDESAMT FÜR
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Conformance test report of a

**GPS receiver modul
integrated into an AIS transponder**

Equipment under test: **Kongsberg Seatex AIS class A**
Type: **AIS 300**
Applying test standards: **IEC 61108-1:2003**

Test Report No.: **BSH/4543/001/4142912/15**

Applicant: **Kongsberg Seatex AS
Pirsenteret
7462 Trondheim
Norway**

Hamburg, 1st April 2015

For the Federal Maritime and Hydrographic Agency

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nach DIN EN 17025
akkreditiertes Prüflaboratorium



D-PL-12084-01-01



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Registrierungsnummer der Urkunde: **D-PL-12084-01-01**

Frankfurt am Main, 08.03.2013

Siehe Hinweise auf der Rückseite



Im Auftrag Dipl.-Ing. (FH) Ralf Egnor
Leiter Abteilung 2



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1 General

Applicant: Kongsberg Seatex AS
Pirsenteret
7462 Trondheim
Norway

Equipment under test: AIS class A transponder
Type: AIS 300
Manufacturer: Same as applicant

Place of test: BSH test laboratory Hamburg,
Room 908/015

Start of test: 4th January 2015
End of test: 1st April 2015

1.1 Summary

Test standard: IEC 61108-1 : 2003 – GPS equipment

Test No.	Reference	Section	Result (passed/ not passed / not applicable / not tested)
1	IEC 61108-1	4.1 Object compliance with IEC 61162-1:2010 Interface output compliance with IEC 60945:2002 General requirements	Not tested Not tested
2/16	IEC 61108-1	4.2 GPS receiver equipment	Passed
3/17	IEC 61108-1	4.3.1 General	Passed
4/18	IEC 61108-1	4.3.2 Equipment output	Not tested
5/19-23	IEC 61108-1	4.3.3 Accuracy	Passed
6/24-27	IEC 61108-1	4.3.4 Acquisition	Passed
7/28	IEC 61108-1	4.3.5 Protection	Passed
8/29	IEC 61108-1	4.3.6 Antenna design	Passed
9/30-31	IEC 61108-1	4.3.7 Dynamic range	Passed
10/32-33	IEC 61108-1	4.3.8 Effects of specific interfering signals	Passed
11/34-35	IEC 61108-1	4.3.9 Position update	Passed
12/36	IEC 61108-1	4.3.10 Differential GPS input	Passed
13/37-40	IEC 61108-1	4.3.11 Failure warnings and status indications	Passed
14/41-42	IEC 61108-1	4.3.12 Output of COG, SOG and UTC	Passed
15/43-44	IEC 61108-1	4.3.13 Typical interference conditions	Passed

Note

EUT is an AIS Class A transponder with integrated GPS module



1.2 Equipment history

<u>Main Unit and MKD</u>				
Type	AIS class A		Part No.:	A300-01
Delivery date	2nd February 2015		Serial number:	AIS300-141001
HW Version:				
	Delivery date	2nd February 2015	Version no	---
	Installation date	2nd February 2015		
HW Version:				
	Delivery date	---	Version no	---
	Installation date	---		
SW Version:				
	Delivery date	2nd February 2015	Version no	1.00.01.b46
	Installation date	2nd February 2015		
SW Version:				
	Delivery date	31st March 2015	Version no	1.00.01.b47
	Installation date	31st March 2015		
SW Version:				
	Delivery date	---	Version no	---
	Installation date	---		

<u>GPS Antenna Unit</u>				
Type	Procom GPS Ant.		Part No.:	GPS 4-TNC
Delivery date	2nd February 2015		Serial number:	031004042
HW Version:				
	Delivery date	2nd February 2015	Version no	Ver.9
	Installation date	2nd February 2015		



1.3 Test environment

Documentation of equipment tests and dates of tests.

Test environment is completely equipped as described in Annex A.

Room	BSH room 908 / 015
Test engineer	T. Ehlers (S3301)
Location	BSH, Hamburg (Germany)

Equipment no	Start of test	End of test	Test engineer
1	2nd February 2015	1st April 2015	T. Ehlers (S3301)
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1.4 Legend

Result marking (in the "result" column)²:

Passed	Item was OK, test successful No colour marking
Not passed	Test of a required item was not successful, change required
N/T	Not Tested
N/A	Not Applicable

Specific remarks (in the "remark" column, marked "bold italic"):

REC	recommendation (in terms of IEC17025 "opinion"); an improvement or change is recommended
Note	Note or comment (in terms of IEC17025 "interpretation"); rationale for specific results or interpretation of requirements as appropriate

1.5 General observations

General observations unrelated to any paragraphs of applied test standards.

² Test items maybe colour marked in draft versions of the report as follows:

Passed	no colour marking
Not passed	yellow
N/T	blue
N/A	no colour marking
REC	green

2 Functional Tests

2.1 IEC 61108-1 – GPS testing

No. of test	IEC 61108-1	Requirement/Condition	Remark	Result
	4	Minimum Performance Standards		
1	4.1	Object compliance with IEC 61162-1 Interface output compliance with IEC 60945 General requirements		N/T N/T
2	4.2	GPS Receiver equipment		
	4.2.1	(M.112/A2.1) The words “GPS receiver equipment” as used in this performance standard include all the components and units necessary for the system to properly perform its intended functions. The equipment shall include the following minimum facilities: a) antenna capable of receiving GPS signals; b) GPS receiver and processor; c) means of accessing the computed latitude/longitude position; d) data control and interface; and e) position display and, if required, other form of output.	Note EUT is a GPS receiver integrated into an AIS Transponder	Passed
	4.2.2	The equipment may be supplied in one of the several configurations to provide the necessary position information. Examples are: stand-alone receiver with means of accessing computed position via a keyboard with the position information suitably displayed; GPS black box receiver fed with operational parameters from external devices/remote locations and feeding an integrated system with means of access to the computed position via an appropriate interface, and the positional information available to at least one remote location. The above examples should not be implied as limiting the scope of future development.	Note EUT is a GPS receiver integrated into an AIS Transponder	Passed



No. of test	IEC 61108-1	Requirement/Condition	Remark	Result
	4.3	Performance standard for GPS receiver equipment		
3	4.3.1	<p>General (M.112/A3.1) The GPS receiver equipment shall be capable of receiving and processing the Standard Positioning Service (SPS) and provide position information in latitude and longitude World Geodetic System (WGS 84) co-ordinates in degrees, minutes and thousandths of minutes and time of solution referenced to UTC (USNO). Means may be provided to transform the computed position based upon WGS-84 into data compatible with the datum of the navigational chart in use. Where this facility exists, the display shall indicate that co-ordinate conversion is being performed and shall identify the co-ordinate system in which the position is expressed. (M.112/A3.2) The GPS receiver equipment shall operate on the L1 signal and C/A code.</p>	<p>Note EUT is a GPS receiver integrated into an AIS Transponder capable of receiving GPS navigational signals</p>	Passed



No. of test	IEC 61108-1	Requirement/Condition	Remark	Result
4	4.3.2	<p>Equipment output (M.112/A3.3) The GPS receiver equipment shall be provided with at least one output from which position information can be supplied to other equipment. The output of position information based upon WGS-84 shall be in accordance with International Standards - IEC 61162 The position information output shall be in accordance with IEC 61162 as follows: For positioning reporting purposes the following sentences shall be available in any combination. DMT – Datum reference GBS – GNSS satellite fault detection GGA – GPS fix data GNS – GNSS fix data RMC – Recommended minimum specific GNSS data VTG – Course over ground and ground speed ZDA – Time and date If a sentences uses a datum other than WGS-84 then the DTM sentence must be used in compliance with IEC 61162. In addition, for integrating with other navigational aids the following sentences may be available in any combination. GRS – GNSS range residuals GSA – GNSS DOP and active satellites GST – GNSS pseudorange error statistics GSV – GNSS satellites in view NOTE GRS, GSA, GST, GSV are required to support external integrity checking. They are to be synchronized with corresponding fix data (GGA or GNS).</p>	<p>Note EUT is a GPS receiver integrated into an AIS Transponder, NMEA AIS data output available</p>	N/T



No. of test	IEC 61108-1	Requirement/Condition	Remark	Result
5	4.3.3	Accuracy		
	4.3.3.1	Static Accuracy (M.112/A3.4) The GPS receiver equipment shall have static accuracy such that the position of the antenna is determined to 100 m (95 %) with horizontal dilution of position (HDOP)≤4 (or PDOP≤6). Since Selective Availability has been set to zero, the static accuracy has been determined to be within 13 m (95 %) as specified by the GPS SPS Performance Standards of October 2001.	See test results under test no. 19 – 21.	Passed
	4.3.3.2	Dynamic Accuracy (M.112/A3.5) The GPS receiver equipment shall have dynamic accuracy such that the position of the antenna is determined to within an accuracy of 100 m (95 %) with HDOP ≤ 4 (or PDOP ≤ 6) under the conditions of sea state and ship's motion likely to be experienced in ships (see IMO Resolution A.694, IEC 60721-3-6 and IEC 60945). Since Selective Availability has been set to zero, the dynamic accuracy has been determined to be within 13 m (95 %) as specified by the GPS SPS Performance Standards of October 2001.	See test results under test no. 22 and 23.	Passed

No. of test	IEC 61108-1	Requirement/Condition	Remark	Result
6	4.3.4	<p>Acquisition</p> <p>(M.112/A3.6) The GPS receiver equipment shall be capable of selecting automatically the appropriate satellite transmitted signals for determination of the ship's position with the required accuracy and update rate.</p> <p>(M.112/A3.8) The GPS receiver equipment shall be capable of acquiring position to the required accuracy, within 30 min, when there is no valid almanac data.</p> <p>(M.112/A3.9) The GPS receiver equipment shall be capable of acquiring position to the required accuracy, within 5 min, when there is valid almanac data.</p> <p>(M.112/A3.10) The GPS receiver equipment shall be capable of re-acquiring position to the required accuracy, within 5 min, when the GPS signals are interrupted for a period of at least 24 h, but there is no loss of power.</p> <p>(M.112/A3.11) The GPS receiver equipment shall be capable of re-acquiring position to the required accuracy, within 2 min, when subjected to a power interruption of 60 s.</p> <p>Acquisition is defined as the processing of GPS satellite signals to obtain a position fix within the required accuracies.</p> <p>Four conditions of the GPS receiver equipment are set out under which the minimum performance standards shall be met.</p> <p>Condition A Initialization - the equipment has been transported over large distances (>1 000 km to <10 000 km) without power or GPS signals or by the deletion of the current almanac; or not been powered for >7 days</p> <p>Condition B Power outage: under normal operation the equipment losses power for at least 24 h.</p> <p>Condition C Interruption of GPS signal reception - under normal operation the GPS signal reception is interrupted for at least 24 h, but there is no loss of power.</p> <p>Condition D Brief interruption of GPS signals for 60 s.</p> <p>No user action other than applying power and providing a clear view from the antenna for the GPS signals, shall be necessary, from any of the</p>	See test results under test no. 24 - 27	Passed



		initial conditions above, in order to achieve the required acquisition time limits in Table 1: A: 30 minutes B: 5 minutes C: 5 minutes D: 2 minutes		
7	4.3.5	Protection		
	4.3.5.1	Antenna and input/output connections (M.112/A4) Precautions shall be taken to ensure that no permanent damage can result from an accidental short circuit or grounding of the antenna or any of its input or output connections or any of the GPS receiver equipment inputs or outputs for a duration of 5 min.		Passed
8	4.3.6	Antenna design (M.112/A2.2) The antenna design shall be suitable for fitting at a position on the ship which ensures a clear view of the satellite constellation.		Passed
9	4.3.7	Dynamic range (M.112/A3.7) The GPS receiver equipment shall be capable of acquiring satellite signals with input signals having carrier levels in the range of –130 dBm to –120 dBm as measured at the output of a 3 dBi linear polarized receiving antenna. Once the satellite signals have been acquired the equipment shall continue to operate satisfactorily with satellite signals having carrier levels down to –133 dBm as measured at the output of a 3 dBi linear polarized receiving antenna.	See test results under test no. 30 and 31.	Passed

No. of test	IEC 61108-1	Requirement/Condition	Remark	Result
10	4.3.8	<p>Effects of specific interfering signals The GPS receiver equipment shall meet the following requirements:</p> <p>a) In a normal operating mode, i.e. switched on and with antenna attached, it is subject to radiation of 3 W/m² at a frequency of 1 636.5 MHz for 10 min. When the unwanted signal is removed and the GPS receiver antenna is exposed to the normal GPS satellite signals, the GPS receiver equipment shall calculate valid position fixes within 5 min without further operator intervention.</p> <p>b) In a normal operating mode, i.e. switched on and with antenna attached, it is subject to radiation consisting of a burst of 10 pulses, each 1.0 μs to 1.5 μs long on a duty cycle of 1600:1 at a frequency lying between 2.9 GHz and 3.1 GHz at power density of about 7.5 kW/m². The condition shall be maintained for 10 min with the bursts of pulses repeated every 3 s. When the unwanted signal is removed and the GPS receiver antenna is exposed to the normal GPS satellite signals, the receiver shall calculate valid position fixes within 5 min without further operator intervention.</p> <p>Advice shall be given in the manual for adequate installation of the antenna unit, to minimise interference with other radio equipment such as marine radars, Inmarsat SES's, etc.</p>	<p>See test results under test no. 32 and 33.</p> <p>NOTE (Condition B)</p> <p>This condition is approximately equivalent to exposing the antenna to radiation from a 60kW 'S' band marine radar operating at a nominal 1,2μs pulse width at 600 pulses/s using a 4m slot antenna rotating at 20r/min with the GPS antenna placed in the plane of the bore site of the radar antenna at a distance of 10m from the centre of rotation.</p>	<p>Passed</p> <p>Passed</p>
11	4.3.9	<p>Position update (M.112/A3.12) The GPS receiver equipment shall generate and output to a display and digital interface a new position solution at least once every 1 s. (M.112/A3.13) The minimum resolution of position i.e. latitude and longitude shall be 0.001 min.</p>	<p>NOTE For craft meeting the HSC code, a new position solution at least every 0.5 s is recommended.</p> <p>See test results under test no. 34 and 35.</p>	Passed

No. of test	IEC 61108-1	Requirement/Condition	Remark	Result
12	4.3.10	<p>Differential GPS input (M.112/A3.15) The GPS receiver equipment shall have the facilities to process differential GPS (DGPS) data fed to it in accordance with the standards of Recommendation ITU-R M.823 and an appropriate RTCM standard. When a GPS receiver is equipped with a differential receiver, performance standards for static and dynamic accuracy (M.112/A3.4 and A3.5) shall be 10 m (95 %) together with integrity monitoring. An integrated DGPS receiver shall have an ITU-R M823 compliant data output port for testing or alternatively, a possibility to display Word Error Rate (WER) on the integrated equipment. The WER is the number of incorrect ITU-R M.823 words in relation to total number of words received.</p>	<p>Note EUT is an AIS class A with integrated GPS module. Differential corrections are supplied by AIS message 17</p>	Passed
13	4.3.11	<p>Failure warnings and status indications (M.112/A5.1) The equipment shall provide an indication if the position calculated is likely to be outside of the requirements of these performance standards;</p>	See test results under test No. 37 - 40	Passed
	4.3.11.1	<p>General (M112/A52) The GPS receiver equipment shall provide as a minimum: a) (M.112/A5.2.1) an indication within 5 s if either: 1) the specified HDOP has been exceeded; or 2) a new position has not been calculated for more than 1 s; 3) under such conditions the last known position and the time of the last valid fix, with explicit indication of this state, so that no ambiguity can exist, shall be output until normal operation is resumed; b) (M.112/A5.2.2) a warning of loss of position; and c) (M.112/A5.2.3) differential GPS status indication of: 1) the receipt of DGPS signals; and 2) whether DGPS corrections are being applied to the indicated ship's position. d) (M112/A5.2.5) DGPS text message display. The GPS receiver either shall have as a minimum the capability of displaying appropriate DGPS text messages or forwarding those messages to for display on a remote system.</p>	<p>NOTE a2) For craft meeting the HSC code, a new position solution at least every 0.5 s is recommended.</p>	Passed



No. of test	IEC 61108-1	Requirement/Condition	Remark	Result
	4.3.11.2	<p>Integrity using RAIM</p> <p>The GPS receiver equipment shall incorporate integrity monitoring using fault detection, for example receiver autonomous integrity monitoring (RAIM), or similar means to determine if accuracy is within the performance standards and provide an integrity indication. An integrity indication shall be used to present the result of the integrity calculation with respect to the selected accuracy level appropriate for vessels operational mode. According to IMO Resolution A.815 these accuracy levels shall be user selectable for 10 m and 100 m. Additional accuracy levels for user selection may be provided.</p> <p>The integrity indication for different position accuracy levels shall be expressed in three states: “safe” “caution”, and “unsafe” for the currently selected accuracy level with a 95 % confidence level.</p> <p>The integrity status shall be continuously displayed along with an indication of the accuracy level selected. The integrity status and the accuracy level selected, shall be provided to other equipment in accordance with the equipment output requirements in 4.3.2.</p> <p>The manufacturer may use colours for integrity indication and if so the following colours shall be used: “safe” shall be green, “caution shall be yellow, and “unsafe” shall be red.</p> <p>The maximum delay for reaction of the integrity calculation by means of RAIM due to negative changes affecting the integrity status is 10 s. The integrity status shall be provided to other equipment in accordance with the equipment output requirements in 4.3.2. For receiver equipment which do not provide information by a dedicated display, the provision of the integrity indication status and the selected accuracy level with the appropriate output interface is mandatory.</p>	See test results under test no. 39 and 40 .	Passed

No. of test	IEC 61108-1	Requirement/Condition	Remark	Result
	4.3.11.3	<p>GPS integrity status using DGPS (M.112/A5.2) The GPS receiver equipment shall provide as a minimum GPS integrity status using DGPS. If the range-rate correction or the pseudo range correction of a satellite is out of tolerance, the binary code in the ITU-R M.823-2 types 1, 9, 31 and 34 messages will cause the GPS receiver not to use that satellite.</p>		Passed
	4.3.11.4	<p>DGPS integrity status and alarm (M.112/A5.2.4) The GPS receiver equipment shall provide as a minimum DGPS integrity status and alarm. The following functions shall be performed in either an integrated DGPS receiver or an associated GPS receiver connected to a DGPS radio beacon receiver. When in differential mode, the GPS receiver shall present a DGPS integrity indication on a display, or forward those messages for display on a remote system: if no DGPS message is received within 10 s; while in manual station selection mode and the selected station is unhealthy, unmonitored, or signal quality is below threshold; while in automatic station selection mode and the only available station is unhealthy, unmonitored, or signal quality is below threshold.</p>		Passed
14	4.3.12	<p>Output of COG, SOG and UTC (M.112/A3.14) The GPS receiver equipment shall generate and output to the digital interface (conforming to the IEC 61162 series) course over ground (COG), speed over ground (SOG) and universal time coordinated (UTC). Such outputs shall have a validity mark aligned with that on the position output. The accuracy requirement for COG and SOG shall no be inferior to the relevant performance standards for heading (Resolution A.424(XI)) and SDME (Resolution A.824/19)), within the limitations of GPS measurements provided by one antenna, compared to the requirements of those standards. Generation and output of COG and SOG are not intended to satisfy the carriage requirements of SOLAS, Chapter V for Heading Devices and SDME by GPS receivers. GPS receivers of this standard have limitations in COG accuracy under high dynamic movement.</p>		Passed

		Such limitations shall be described in the manufacturer's operating manual as shown in Table 2.		
	4.3.12.1	<p>Accuracy of COG information The error in the COG (the path of the antenna position over ground) due to the actual ship's speed over ground shall not exceed the following values:</p> <p>Table 2: Speed range (knots) Accuracy of COG output to user 0 to ≤1 knot Unreliable or not available >1 to ≤17 knots ±3 >17 knots ±1</p> <p>Due to the limitations of GPS receivers of this standard, it is not appropriate to include requirements for COG errors attributed to high dynamic movement. Such limitations shall be in the manufacturer's operation manual.</p>	See test results under test no. 41 .	Passed
	4.3.12.2	<p>Accuracy of SOG information Errors in the SOG (velocity of the antenna position over ground) shall not exceed 2 % of the actual speed or 0.2 knots, whichever is greater.</p>	See test results under test no. 42	Passed
	4.3.12.3	<p>Availability and validity of time information The GPS receiver equipment shall provide UTC with resolution of 0.01 s on the digital interface. The validity mark of the digital interface for position contained in GGA message of IEC 61162 shall be used for interpretation of validity of digital interface for UTC contained in ZDA message of IEC 61162.</p>		Passed
15	4.3.13	<p>Typical interference conditions (M.112/A3.16) The GPS receiver equipment shall be capable of operating satisfactorily in typical interference conditions. For clarification of this requirement see 5.7.1 and for the associated tests see 5.7.2.</p>	See test results under test no. 43 and 44.	Passed
	5.6	Methods of test and required test results	<p>NOTE The number in brackets is the sub-clause of the relevant performance standard.</p>	
16	5.6.1 (4.2)	<p>GPS receiver equipment The equipment under test (EUT) shall be checked for composition by inspection of the equipment and the manufacturer's documentation.</p>		Passed

No. of test	IEC 61108-1	Requirement/Condition	Remark	Result
17	5.6.2 (4.3.1)	Position output The EUT shall be checked for the form of the position output by inspection of the manufacturer's documentation.		Passed
18	5.6.3 (4.3.2)	Equipment output The EUT shall be checked for conformity to IEC 61162-1 by inspection of the manufacturer's documentation and protocol tests.		Passed
	5.6.4 (4.3.3)	Accuracy		
	5.6.4.1 (4.3.3.1)	Static accuracy		
19	5.6.4.1.1	GPS Position fix measurements shall be taken over a period of not <24 h. The absolute horizontal accuracy shall be within 13 m (95 %), having discarded measurements taken in conditions of HDOP \geq 4 and PDOP \geq 6.		Passed
20	5.6.4.1.2	Differential GPS Position fix measurements shall be taken once per second over a period of not <24 h. The distribution of the horizontal error shall be within 10 m (95 %). The horizontal position of the antenna shall be known to within 0.1 m in the datum used for the generation of the corrections. The corrections shall be provided by an actual DGPS broadcast in accordance with ITU-R M.823.		Passed
21	5.6.4.2	Angular movement of the antenna The static tests specified in 5.6.4.1.1 and 5.6.4.1.2 shall be repeated with the antenna performing an angular displacement of ± 22.5 (simulating roll) in a period of about 8 s (see IEC 60721-3-6) during the duration of the tests.		Passed

No. of test	IEC 61108-1	Requirement/Condition	Remark	Result
	5.6.4.3 (4.3.3.2)	Dynamic accuracy		
22	5.6.4.3.1	<p>GPS</p> <p>The tests for dynamic accuracy are a practical interpretation of the conditions set out in IEC 60721-3-6, Table V, item e), X-direction (surge) and Y-direction (sway). These are stated as surge 5 m/s² and sway 6 m/s² for all classes of environment. When using a simulator, the simulator characteristics shall accurately represent the signals required.</p> <p>The results of the test performed by simulation facilities shall be identical with those in a) and b) below.</p> <p>Alternatively to the use of a simulator, an example of applying these accelerations is given below:</p> <p>a) a fully locked and settled EUT travelling in a straight line at 48 knots ± 2 knots for a minimum of 1.2 min which is reduced to 0 knots in the same straight line in 5 s, shall not indicate a positional offset >±13 m from the final position 10 s after coming to rest;</p> <p>a fully locked and settled EUT travelling at least 100 m at 24 knots ± 1 knot in a straight line then subjected, for at least 2 min, to smooth deviations either side of the straight line of approximately 2 m at a period of 11 s to 12 s shall remain in lock and follow the actual position to within an lane of 30 m wide centred on the mean direction of motion.</p> <p>For all methods above, the rest position shall be established by one of the following methods: providing a stationary receiver identical to the EUT alongside the rest point and comparing indicated output positions; or providing the reference inputs from the simulator.</p>		Passed

No. of test	IEC 61108-1	Requirement/Condition	Remark	Result
23	5.6.4.3.2	<p>Differential GPS</p> <p>The tests for dynamic accuracy are a practical interpretation of the conditions set out in IEC 60721-3-6, Table V, item e), X-direction (surge) and Y-direction (sway). These are stated as surge 5 m/s² and sway 6 m/s² for all classes of environment.</p> <p>When using a simulator, the simulator characteristics shall accurately represent the signals required.</p> <p>The results of the test performed by simulation facilities shall be identical with those in a) and b) below.</p> <p>Alternatively to the use of a simulator, an example of applying these accelerations is given below:</p> <p>a fully locked and settled EUT travelling in a straight line at 48 knots \pm 2 knots for a minimum of 1.2 min which is reduced to 0 knots in the same straight line in 5 s, shall not indicate a positional offset $>\pm 10$ m from the true position at rest and the indicated position shall settle to within ± 2 m of the rest position indication within 10 s of coming to rest;</p> <p>a fully locked and settled EUT travelling at least 100 m at 24 knots \pm 1 knot in a straight line then subjected, for at least 2 min, to smooth deviations either side of the straight line of approximately 2 m at a period of 11 s to 12 s shall remain in lock and follow the actual position to within an lane of 30 m wide centred on the mean direction of motion.</p> <p>For the methods above, the true and rest positions shall be established by one of the following methods:</p> <p>a) for method a) above, the rest position indication shall be determined by averaging the 15 consecutive position indications recorded following the 10 s settling period and the true position at rest shall be measured to an accuracy of 1 m;</p> <p>b) providing the reference inputs from a simulator within 1 m.</p>		Passed

No. of test	IEC 61108-1	Requirement/Condition	Remark	Result
	5.6.5 (4.3.4)	Acquisition		
24	5.6.5.1	<p>Condition A - Initialisation The EUT shall be either: a) initialised to a false position at least 1 000 km and not greater than 10 000 km from the test position, or alternatively, by deletion of the current almanac; or b) isolated from a power source and GPS signals for >7 days. A performance check shall be carried out after the time limit contained in Table 1.</p>	<p>Note Every simulton scenario forces almanach reset, EUT locked on to the simulated signal within less than five minutes, see Annex B of this report</p>	Passed
25	5.6.5.2	<p>Condition B - Power outage The EUT shall be isolated from the power source for a period within 24 h to 25 h. At the end of the period, a performance check shall be carried out after the time limit contained in Table 1.</p>	<p>Note GPS cold start takes less than 5 Minutes, see any simulator test in Annex B</p>	Passed
26	5.6.5.3	<p>Condition C - Interruption of GPS signals During normal operation of the EUT, the antenna shall be completely masked for a period within 24 h to 25 h. At the end of the period, a performance check shall be carried out after the time limit contained in Table 1.</p>		Passed
27	5.6.5.4	<p>Condition D – Brief interruption of power During normal operation of the EUT, the power shall be removed for a period of 60 s. At the end of this period, the power shall be restored. A performance check shall be carried out after the time limit contained in Table 1.</p>		Passed

No. of test	IEC 61108-1	Requirement/Condition	Remark	Result
	5.6.6 (4.3.5)	Protection		
28	5.6.6.1 (4.3.5.1)	Antenna and input/output connections The antenna input of the receiver, if provided, shall be connected to ground for 5 min. After completion of the test and reset of the EUT, if required, the antenna or input/output connections shall be connected normally, and a performance check shall be carried out to ensure that no permanent damage has resulted.		Passed
29	5.6.7 (4.3.6)	Antenna design The antenna of the EUT shall be checked by inspection of the documentation provided by the manufacturer, to confirm that it is suitable for shipborne installation to ensure a clear view of the satellite constellation.		Passed
	5.6.8 (4.3.7)	Sensitivity and dynamic range		
30	5.6.8.1	Acquisition This is tested by using a simulator. Method: Transmit the simulator signal over a suitable antenna. Adjust the signal power by use of a calibrated test receiver to $-125 \text{ dBm} \pm 5 \text{ dBm}$. Replace the antenna of the calibrated test receiver by the receiving unit of the EUT. A performance check shall be carried out. Required result: The EUT shall meet the requirements of this check, with this signal range.	EUT tracked GPS L1 signal at -130 dBm This test was performed by using a simulator	Passed

No. of test	IEC 61108-1	Requirement/Condition	Remark	Result
31	5.6.8.2	<p>Tracking</p> <p>The received satellite signals shall be monitored by a suitable test receiver. These signals shall be attenuated down to -133 dBm. Under these conditions the performance requirements of a performance check shall be met. This is tested by using a simulator.</p> <p>Method:</p> <p>Transmit the simulator signal over a suitable antenna. Adjust the signal power by use of a calibrated test receiver to $-125 \text{ dBm} \pm 5 \text{ dBm}$. Replace the antenna of the calibrated test receiver by the receiving unit of the EUT. After the start of transmission and tracking with the nominal transmission level condition, gradually reduce transmission level down to -133 dBm.</p> <p>Required result:</p> <p>The EUT shall continue tracking at least one satellite.</p>	<p>EUT tracked GPS L1 signal at -133 dBm and a position is calculated properly</p> <p>This test was performed by using a simulator</p>	Passed
	5.6.9 (4.3.8)	Effects of specific interfering signals		
32	5.6.9.1 (4.3.8 a)	<p>L Band Interference</p> <p>In a normal operating mode, using an appropriate signal source, the EUT shall be subjected to radiation of 3 W/m^2 at a frequency of 1 636.5 MHz for 10 min. The signal shall be removed and a successful performance check shall be carried out within 5 min.</p>	For test results see Annex B of this report	Passed
33	5.6.9.2 (4.3.8 b)	<p>S Band Interference</p> <p>In a normal operating mode, using an appropriate signal source, the EUT shall be subjected to radiation consisting of a burst of 10 pulses, each $1.0 \mu\text{s}$ to $1.5 \mu\text{s}$ long on a duty cycle of 1600:1 at a frequency in the range of 2.9 GHz to 3.1 GHz at power density of approximately 7.5 kW/m^2. This condition shall be maintained for 10 min with the bursts of pulses repeated every 3 s.</p> <p>NOTE The peak power density is 7.5 kW/m^2 to be measured at the EUT, this is approximately 4.7 W/m^2 average power at a fixed transmitting antenna.</p> <p>The signal shall be removed and a successful performance check shall be carried out within 5 min.</p>	For test results see Annex B of this report	Passed

No. of test	IEC 61108-1	Requirement/Condition	Remark	Result
	5.6.10 (4.3.9)	Position update		
34	5.6.10.1	<p>Slow speed update rate The EUT shall be placed upon a platform, moving in approximately a straight line, at a speed of 5 knots \pm 1 knot. The position output of the EUT shall be checked at intervals of 10 s, over a period of 10 min. The output position shall be observed to be updated on each occasion. This test may be carried out by using a simulator.</p>	(see Annex B).	Passed
35	5.6.10.2	<p>High speed update rate The EUT shall be placed upon a platform, moving in approximately a straight line, at a speed of 50 knots \pm 5 knots. The position output of the EUT shall be checked at intervals of 1 s, over a period of 10 min. The output position shall be observed to be updated on each occasion. This test may be carried out by using a simulator with a speed of 70 knots at intervals of 0.5 s.</p> <p>The minimum resolution of position, i.e. latitude and longitude shall be checked by observation during 5.6.10.1 and 5.6.10.2 above. Record the IEC 61162 output of the EUT during this test and confirm that received positions at the end of each interval are in compliance with the real or simulated reference position.</p>	(see Annex B).	Passed
36	5.6.11 (4.3.10)	<p>Differential GPS input The manufacturer's documentation shall be inspected to:</p> <p>a) verify that the EUT will correctly process the message protocol of</p> <ol style="list-style-type: none"> 1) the RTCM recommended standards for differential NAVSTAR GPS service; or 2) in the case where maritime radio beacons are used as the means of communication of the differential corrections, the standard contained in ITU-R M.823, and <p>b) confirm that</p> <ol style="list-style-type: none"> 1) receipt of DGPS signals will be indicated; 2) that the application of DGPS signals to the output ship's position is indicated; and 3) the WER information is provided on an output port or at the display. 		Passed

No. of test	IEC 61108-1	Requirement/Condition	Remark	Result
	5.6.12 (4.3.11)	Failure warnings and status indications		
	5.6.12.1	General alarm tests		
37	5.6.12.1.1 (4.3.11.1a 4.3.11.1b)	Position/HDOP alarm test Set up the EUT in a simulation environment with an HDOP <4. Select a specific EUT HDOP value as an indication threshold >4. Modify the simulator output until its HDOP is greater than the EUT specified HDOP threshold. Observe that an indication is given at the EUT within 5 s. Modify the simulator output until HDOP <4 and observe that the indication is removed. Switch off transmission of simulated signals and observe that the EUT releases an appropriate indication within 5 s. Verify that the last known position and ist time stamp are being displayed indicating the "loss of position" condition. Verify that this mode is provided constantly on display and output interface until removal of the error condition at the simulation environment. Switch on transmission of simulated signals and observe that the EUT resumes normal operation.	Note EUT is an integrated GPS modul in an AIS transceiver, no HDOP output available, Position quality information coded in PA and RAIM flag of AIS message	Passed
38	5.6.12.1.2 (4.3.11.1c)	Differential GPS status indication test Set up the EUT in a simulation environment providing with an HDOP <4. Observe that the status of the EUT operation is GPS without using DGPS corrections. Set the EUT differential correction age mask to 30 s (if available). Start transmission of ITU-R M.823 differential corrections. Observe that the indication for DGPS status of EUT operation is given within 40 s. Stop transmission of ITU-R M.823 differential corrections. Observe that the status of EUT operation resumes to GPS without using DGPS corrections within 40 s	Note EUT is a GPS receiver integrated into an AIS Transponder, Position status information provided by PA and RAIM flag of AIS msg. 1	Passed
	5.6.12.2 (4.3.11.2)	Test of integrity monitoring using RAIM For the purpose of testing the RAIM functionality, it is recommended that means are provided for real-time display of the actual position error with reference to the simulated position.		Passed



No. of test	IEC 61108-1	Requirement/Condition	Remark	Result
39	5.6.12.2.1	<p>Testing of “safe” and “caution” status The EUT shall be set up under simulated conditions, providing 8 “healthy” satellites available, acquired and tracked.</p> <p>a) Select an accuracy level of 100 m.</p> <p>b) Observe that</p> <ol style="list-style-type: none"> 1) RAIM is indicated as “in operation”, and 2) the “safe” status is indicated. <p>c) Consecutively reduce the number of “healthy” satellites until the “caution” state is raised. Observe that</p> <ol style="list-style-type: none"> 1) RAIM is still indicated as “in operation”, and 2) the status indication switched to “caution” within 10 s of the satellite change that caused it. <p>d) Increase the number of “healthy” satellites until the RAIM state returns to “safe” state. Observe that</p> <ol style="list-style-type: none"> 1) RAIM is still indicated as “in operation”, and 2) the status indication switches to “safe” within 2 min of the satellite change that prompted it. <p>For each step of the above test sequence observe if the appropriate interface output is provided. Repeat the above test sequence for a selected accuracy level of 10 m and, if provided, for another accuracy level.</p>	(see Annex B).	Passed



No. of test	IEC 61108-1	Requirement/Condition	Remark	Result
40	5.6.12.2.2	<p>Testing of “unsafe” status</p> <p>The EUT shall be set up under simulated conditions, providing 8 “healthy” satellites available, acquired and tracked. Select an accuracy level of 100 m. Observe that RAIM is indicated as “in operation”, and The “safe” status is indicated. Change the behaviour of at least 1 satellite by varying the satellite clocks with the result that the position accuracy gradually degrades until it will no longer be inside the selected accuracy level with 95 % confidence level. Observe that RAIM is still indicated as “in operation”, and the status indication switches to “unsafe” within 10 s if the actual position error exceeding the selected accuracy level. Change the behaviour of the satellites back to regular behaviour with the result that the position accuracy will be again inside of the selected accuracy level within 95 % confidence level. Observe that RAIM is still indicates as “in operation”, and The status indication switches to “safe” within 2 min. For each step of the above test sequence observe if the appropriate interface output is provided. Repeat the above test sequence for a selected accuracy level of 10 m and, if provided, for another accuracy level.</p>	(see Annex B).	Passed

No. of test	IEC 61108-1	Requirement/Condition	Remark	Result
41	5.6.13 (4.3.12)	<p>Accuracy of COG and SOG Methods of test The EUT shall be set up on an appropriate mobile unit or simulator and all outputs indicating course over ground shall be monitored. At a constant forward direction, the forward speed shall be within 0 knots to 1 knot. Ten seconds after being in the range, measurements shall be made for a duration of 2 min. This cycle shall be repeated for all speed ranges of the Table 2.</p> <p>Required results The test results shall be observed on the display and the approved interface. For SOG tests, no reading of the speed indicator shall differ from the constant speed being applied at the time by more than 2 % of that speed or 0.2 knots, whichever is the greater. For COG tests, the differences between the reference direction and the measured course over ground in each test cycle shall not exceed the limits of Table 2.</p> <p>Validity of COG and SOG information The quality indicator of the GGA and VTG message of IEC 61162 shall be used for interpretation of validity of COG and SOG.</p> <p>Methods of testing Check of digital interface with IEC 61162. With the EUT normally operating, preclude invalid position data by reducing the number of received satellites. Investigate the content of the resultant GGA and VTG.</p> <p>Required result Observe that the quality indicator of GGA and VTG messages of IEC 61162 turn to invalid. Observe that the COG and SOG information contained in VTG message of IEC 61162 is replaced by null fields.</p>	(see Annex B).	Passed
42	5.6.14 (4.3.12)	<p>Output of UTC - Method of testing Check of digital interface with IEC 61162. While the EUT is navigating, provoke an invalid position by reducing the number of received satellites to two. Investigate the content of the GGA and ZDA messages provided.</p> <p>Required results Observe that the resolution of UTC information contained in the ZDA message is according to</p>		Passed

		IEC 61162 requirements. Observe that the validity flag of GGA message of IEC 61162 turns to invalid. Observe that the ZDA message remains transmitted carrying complete UTC information.		
	5.7	Typical interference conditions		
	5.7.1	Requirements		
	5.7.1.1	<p>Typical interference conditions The GPS receiver equipment shall be capable of operating in typical interference condition. Operational situations include static accuracy and reacquisition within 30 s after satellite signals have been masked for 60 s or less by an obstruction, for example a bridge. Typical GPS interference effects can be characterised as being broadband noise-like interference, Continuous Wave Interference (CWI), or pulsed interference. Much work has been done in the aviation community to define interference levels in these three categories as reported in the Minimum Operational Performance Standards</p> <p>(MOPS) for Global Positioning System/Wide Area Augmentation System (GPS/WAAS) Airborne Equipment (RTCA/DO-229B October 6, 1999). The levels defined in this subclause are based upon the interference masks developed within RTCA. These masks are also described in ITU-R Recommendation M.1477.</p>	(see Annex B).	For results see test No. 43
	5.7.1.2	<p>Broadband interference levels The interference mask for broadband noise-like interference varies as a function of the bandwidth of the interfering signal. This interference effect can be represented by broadband noise centred at 1575.42 MHz. The bandwidth dependent interference mask can be seen in Figure 1.</p>		For results see test No. 43
	5.7.1.3	<p>Continuous wave interference (CWI) Continuous wave interference interacts with the individual C/A code spectral lines found in the GPS signal structure. GPS receivers are typically more susceptible to CWI than to any other type of interference. The CWI mask can be seen in Figure 2.</p>		For results see test No. 43



No. of test	IEC 61108-1	Requirement/Condition	Remark	Result
	5.7.1.4	Pulsed interference Pulsed interference can occur due to proximity to radars or other RF devices using pulsed waveforms. GPS receivers typically are fairly robust when exposed to low duty cycle pulsed interference. The interference mask for pulsed interference will consist of a pulse modulated carrier (CW) at 1575.42 MHz, with peak carrier level of -20 dBm and duty factor of 10 % while using a 1 ms pulse width.		For results see test No. 43
	5.7.2	Testing		
	5.7.2.1	The interference test procedures presented in this sub-clause follow closely the procedures used by aviation receiver manufacturers in the self-certification process used to show compliance with RTCA/DO-229B. The procedures have been adapted as necessary to meet the requirements of the IMO GPS requirements.		For results see test No. 43
	5.7.2.2	Simulator conditions The simulator conditions are as follows: five GPS satellites; one satellite at a maximum level of -120 dBm plus antenna gain at 90 elevation; one satellite at a minimum level of -130 dBm plus antenna gain at 5 elevation; three satellites at a level of -127 dBm plus antenna gain at 45 elevation.		For results see test No. 43



No. of test	IEC 61108-1	Requirement/Condition	Remark	Result
	5.7.2.3	<p>Navigation solution accuracy test The normalised error associated with the navigation solution, which will be compared with the 10 m, 95 % horizontal accuracy requirement shall be computed using the formula shown below: $NE = [4(d_i)] / [HDOP_i]$ where NE is the normalised error;</p> <p>d_i is the instantaneous 2-D horizontal position error (meters); $HDOP_i$ is the instantaneous horizontal dilution of precision. Scaling the instantaneous 2-dimensional position error (d_i) by $4/HDOP_i$ provides a means of normalising the tests to a constant $HDOP = 4$ and accounts for fluctuation in the satellite coverage due to changing geometry. $HDOP_i$ may be obtained from the receiver under test or calculated. Only those satellites used in the position solution shall be included in the $HDOP_i$ calculation.</p>		For results see test No. 43
	5.7.2.4	Navigation solution accuracy test procedures		
	5.7.2.4.1	<p>Interference conditions Interference conditions, including broadband noise centred at 1575.42 MHz, continuous wave interference (CWI), and pulsed interference shall be simulated. For the pulsed interference tests, a pulse-modulated carrier (CW) with peak carrier level of -20 dBm and duty factor of 10 % shall be used. The interference values are shown in the Three tables below.</p> <p>Broadband interference values Noise bandwidth: 1 MHz Total RMS power: -110.5 dBm</p> <p>Pulsed interference values Frequency: 1575.42 MHz Pulse width: 1 ms</p> <p>Continuous wave interference (CWI) values Frequency: 1575.42 MHz Power: -120.5 dBm Frequency: 1626.0 MHz Power: +8.0 dBm</p>		For results see test No. 43

No. of test	IEC 61108-1	Requirement/Condition	Remark	Result
	5.7.2.4.2	<p>Test procedures</p> <p>The EUT is subjected to one of the interference sources.</p> <p>The simulator scenario shall be engaged and the satellite signals turned on.</p> <p>The EUT shall be powered and initialised.</p> <p>While the EUT is providing position solutions, the interference shall be applied to the EUT, and the level of the interference shall be adjusted to the required value.</p> <p>When steady-state accuracy is reached, record a minimum of 100 position and HDOP values as reported by the EUT at a rate of one sample every 2 min.</p> <p>Repeat this cycle for any remaining interference source.</p>		For results see test No. 43
43	5.7.2.4.3	<p>Required results</p> <p>Pass/fail determination</p> <p>If the EUT reports a position with a normalised error greater than 10 m or fails to report a position in more than 5 % of the samples, a test failure is declared.</p>	(see Annex B)	Passed
	5.7.2.5	<p>Reacquisition test</p> <p>Method of test</p> <p>The reacquisition test is designed to simulate a temporary loss of signal, such as passing under a bridge. To determine the re-acquisition pass/fail criteria, consider a single trial where the EUT provides a valid position fix that is within required accuracy at 30 s from restoration of the satellite signals, and maintains a tracking status for at least the next 60 s. This unit is considered to have Passed one trial.</p>		For results see test No. 44
	5.7.2.5.1	Re-acquisition test procedures		
	5.7.2.5.2	<p>Interference conditions</p> <p>The interference condition to be tested is shown below. This is a broadband noise value centred at 1575.42 MHz.</p> <p>Noise bandwidth: 1 MHz</p> <p>Total RMS power: -110.5 dBm</p>		For results see test No. 44



No. of test	IEC 61108-1	Requirement/Condition	Remark	Result
	5.7.2.6	Re-acquisition scenarios		
	5.7.2.6.1	<p>Test procedures</p> <p>The EUT is subjected to the broadband interference source.</p> <p>The simulator scenario shall be engaged and the satellite signals turned on.</p> <p>The EUT shall be powered and initialised.</p> <p>The EUT shall be allowed to reach steady-state accuracy before the satellites are to be switched off. The simulator RF output shall be removed for 30 S. The simulator RF output shall be restored to the EUT.</p> <p>After 30 s record a position and HDOP value as reported by the EUT. If after 30 s, no position report has been sent from the receiver, record a trial failure and go to step i).</p> <p>Ensure that the receiver continues position reporting for the next 60 s.</p> <p>Go to step d) and repeat as required (note that if the simulator scenario is reset, some receiver may require purging of all previous data to enable proper operation. This is due to the persistence of time data in the receiver and the inability of the receiver's software to deal with a backward transition in time).</p>		For results see test No. 44
44	5.7.2.6.2	<p>Required results</p> <p>Pass/fail criteria</p> <p>A failure by the EUT to provide a position output after 30 s, reporting a position with normalised error greater than 10 m, or failing to continue position reporting for 60 s after sampling indicated a failure mode, and results in declaring a trial failure. To determine the reacquisition time pass/fail criteria, the test disposition table shall be used.</p>	(see Annex B)	Passed
45	5.8	<p>Performance checks under IEC 60945 conditions</p> <p>Environmental requirements of IEC 60945 appropriate to its category, i.e. "protected" and "exposed", shall be carried out. The manufacturer shall declare any pre-conditioning required before environmental checks.</p> <p>Performance checks shall be performed for initial (cold) start; acquisition; tracking (navigation)</p>	<p>NOTE</p> <p>For marine equipment environmental testing has to be carried out for granted type approval. 60945 testing is not part of this test report. See 60945 test for EUT.</p>	N/T

Annex A Test equipment

A.1 Test equipment summary

<i>Model / Program</i>	<i>Serial No. / Version No.</i>	<i>Calibrated / Function test</i>	<i>Used for</i>
Reference position roof of BSH building		Lat: 53 32.8136481666' Lon:9 58. 1016981666'	All test using real satellite signals
GNSS Simulation Unit	SPIRENT Communications Hardware:Typ: GSS8000, S/N: 8628/9 Software: PosApp Ver. 3.5	Calibration date 2012/10/22 Function tests performed successfully according documented test procedures before performance of tests	All GPS testing, unless stated otherwise
Furuno FA-100 AIS cl.A	FA-100	Function tests performed successfully	Generation of AIS message 17 for differential GPS corrections
MiniCircuits RF- Amplifier	ZHL-5W-2G-S+	Function tests performed successfully	L-Band interference
MiniCircuits RF- Amplifier	ZHL-16W-43S	Function tests performed successfully	S-Band interference
Signal Generator R&S SMJ100	S/N: 100858	2013/10/21	Interference tests IEC 61108- 1 Ed.2, §5.6.9; §5.7
Agilent spectral analyzer E4440A	S/N: MY44022884	2014/07/14	Calibration of GPS measurement inside RF- chamber
Narda Broadband Field Meter	NBM550	2012/06/06	Induced Power of L/S-Band
Horn Antenna Schwarzbeck BBHA 9120A	BBHA 9120A 535	2009/11/26	Calibration of GPS measurement inside RF- chamber



Reference position

Made by FREIE UND HANSESTADT HAMBURG
 Vermessungsamt –VA311-

Description of point	geocentrically co-ordinates (WGS84)	geodetical geographical co-ordinates (WGS84)	Gauß-Krüger (Bessel)
North	x(m) 3740601.680	N 53°32' 49".49049	x(m) 5935502.790
	y(m) 657439.492	E 9°58' 6".10408	y(m) 3564257.804
	z(m) 5107029.673	Height over Ellipsoid 95.900 m	Altitude above sea level 55.969 m
South	x(m) 3740618.106	N 53°32' 48".81889	x(m) 5935482.027
	y(m) 657442.338	E 9°58' 6".10189	y(m) 3564258.046
	z(m) 5107017.296	Height over Ellipsoid 95.849 m	Altitude above sea level 55.917

Accuracy of survey = 0.02 m - last survey dated 2009-05-04

A.2 Documentation of test equipment

A.2.1 L-Band interference signal amplifier

RF-power amplifier for L-Band interference simulation

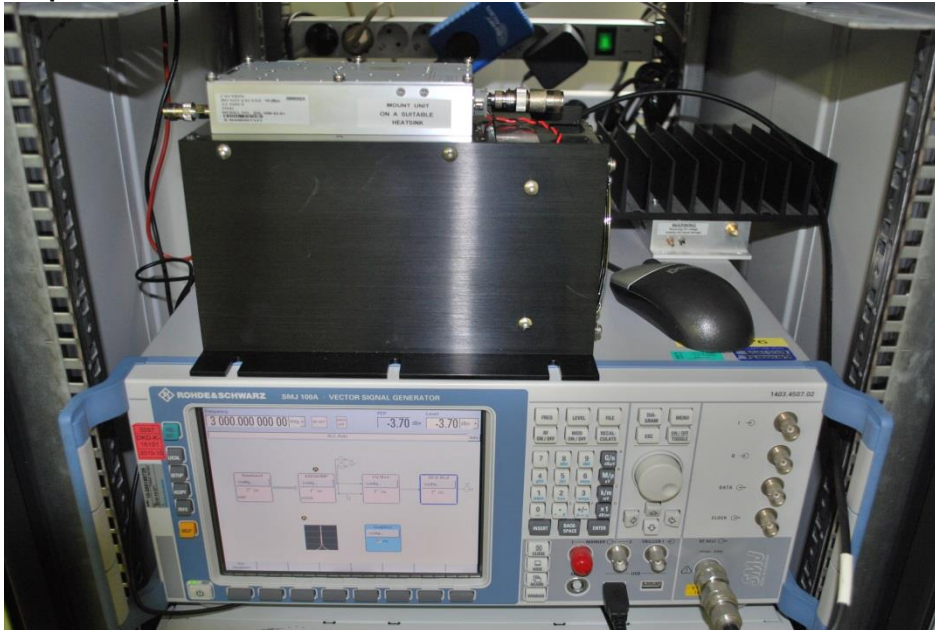


Signal generation for high power L-Band signals



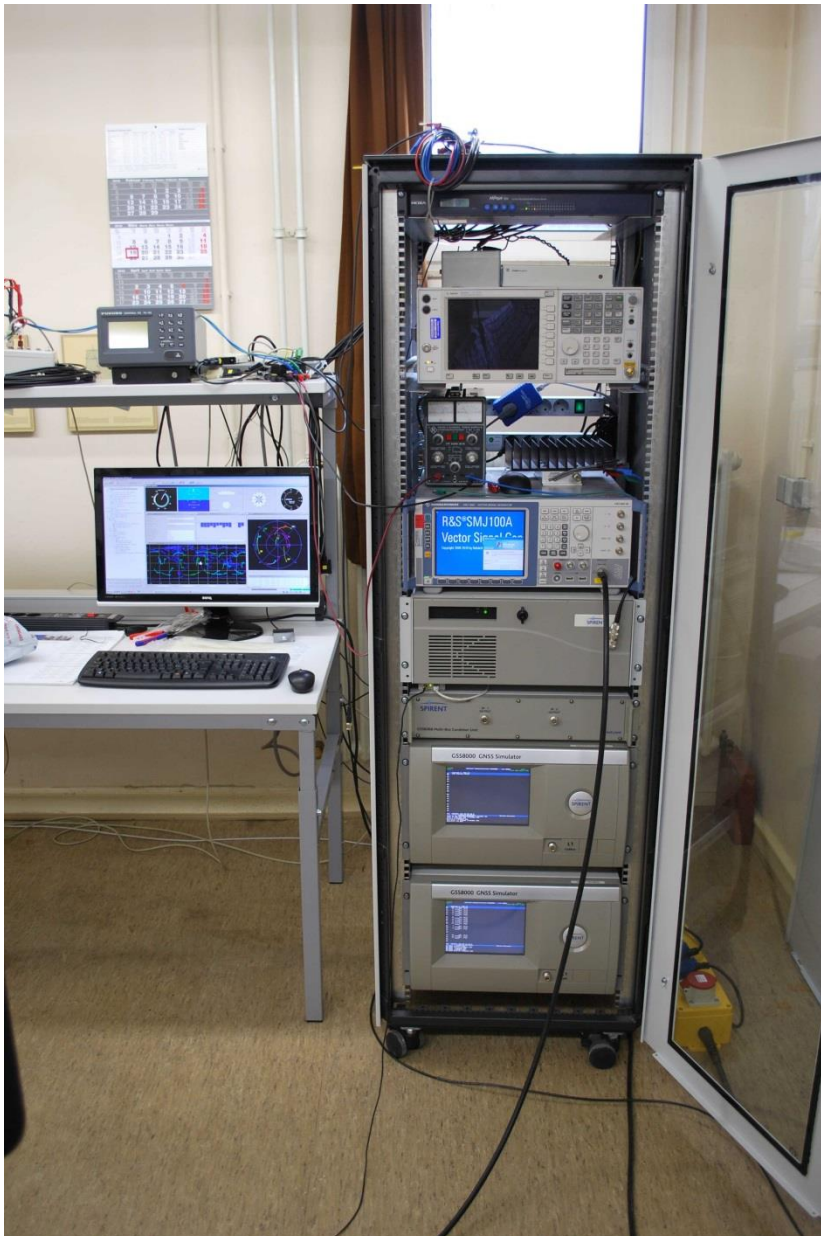
A.2.2 S-Band signal generation

Signal generation for high power S-Band signals
RF-power amplifier for S-Band interference simulation



A.2.3 GNSS Simulation

GNSS Simulation at BSH



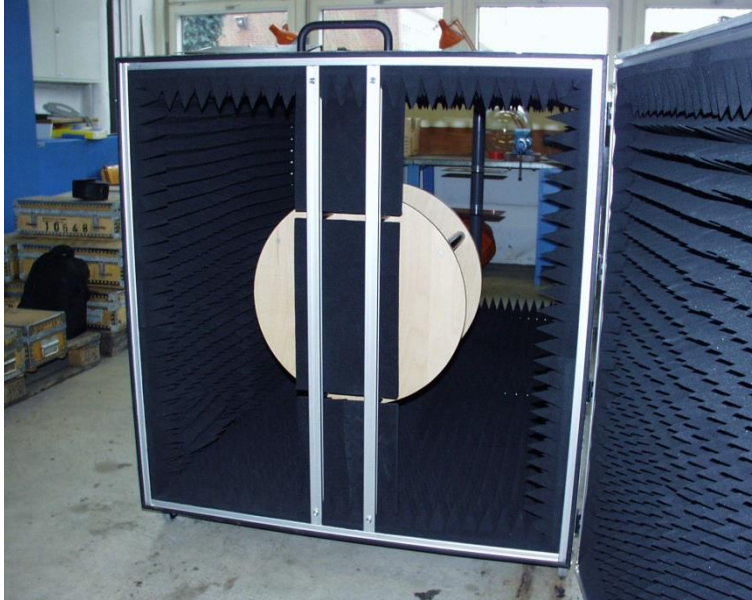
Arrangement of GPS- and noise- interference transmitting antennas



GPS test box, exterior view



GPS test box, interior view





A.2.4 Calibration protocol of RF- Chamber

Calibration protocol

Date	2nd March
Test eng.	Ehlers
Place of test	BSH, Room 015

Equipment

R&S SMJ100A Signalgenerator
 Helixantenna H1116R6, No. 1 and No. 2
 Schwarzbeck BBHA 9120 Hornantenna
 Agilent spectral analyser E4440A
 Spirent GSS8000

Pdef.	-130 dBm	ICD GPS200 defines -130dBm as minimum received power at 3dBi antenna IEC61108-1:2003 defines -125dBm for typ. interference testing up to -120dBm
G trans. Ant.	4,5 dBi	Gain of GPS- transmitting antenna
Prec.	-125,32 dBm	Needed received power @ Schwarzbeck BBHA9120
Grec.ant.	9 dBi	Gain of calibrated Schwarzbeck BBHA 9120 @ 1575MHz
Adapt. Factor	5 dB	Adaption factor of Schwarzbeck BBHA9120 vs. 3dB antenna incl. Cable loss
Attenuation	16,1 dB	attenuation needed for adjusted power level



Calibration of interference

Date	2 nd March 2015
Test eng.	Ehlers
Place of test	BSH, Room 015

Equipment

R&S SMJ100A Signalgenerator
 Helixantenna H1116R6
 Schwarzbeck BBHA 9120 Hornantenna
 GPS- transmitting antenna 2G1216P - Antcom Corp.
 Agilent spectral analyser E4440A
 Spirent GSS8000

Noise 1MHz -110,5dBm

Pdef.	-110,5 dBm	Needed interference power
Prec	-105,5 dBm	Needed measured power at Schwarzbeck BBHA9120 Antenna
G trans. Ant.	7 dBi	Gain of interference trans. ant.
Adapt. Factor	5 dB	Adaption factor of Schwarzbeck BBHA9120 vs. 3dB antenna incl. Cable loss
Power adj.	-76,5 dB	adjusted power level at SMJ100A

Pulsed -20dBm

Pdef.	-20 dBm	Needed interference power
Prec	-15 dBm	Needed measured power at Schwarzbeck BBHA9120 Antenna
G trans. Ant.	7 dBi	Gain of interference trans. ant.
Adapt. Factor	5 dB	Adaption factor of Schwarzbeck BBHA9120 vs. 3dB antenna incl. Cable loss
Power adj.	14 dB	adjusted power level at SMJ100A



CW @ 1575,42; -120,5dBm

Pdef.	-120,5 dBm	Needed interference power
Prec	-115,5 dBm	Needed measured power at Schwarzbeck BBHA9120 Antenna
G trans. Ant.	7 dBi	Gain of interference trans. ant.
Adapt. Factor	5 dB	Adaption factor of Schwarzbeck BBHA9120 vs. 3dB antenna incl. Cable loss
Power adj.	-86 dB	adjusted power level at SMJ100A

CW @ 1626,0; +8,0dBm alternativ CW @ 1596,0MHz; -80dBm

Pdef.	-80 dBm	Needed interference power
Prec	-75 dBm	Needed measured power at Schwarzbeck BBHA9120 Antenna
G trans. Ant.	7 dBi	Gain of interference trans. ant.
Adapt. Factor	5 dB	Adaption factor of Schwarzbeck BBHA9120 vs. 3dB antenna incl. Cable loss
Power adj.	-45,2 dB	adjusted power level at SMJ100A

Annex B GPS test diagrams

B.1 § 5.6.4.1 Static accuracy

B.1.1 § 5.6.4.1.1 Static accuracy – GPS

Position fix measurements shall be taken over a period of not <24 h. The absolute horizontal position accuracy shall be within 13m (95 %), having discarded measurements taken in conditions of HDOP ≥ 4 and PDOP ≥ 6 .

Conditions of tests performed – Real signal

Period of position fix measurements: ~24 h
Position fix measurements : >87000
Accuracy: HDOP ≤ 4 (or PDOP ≤ 6)

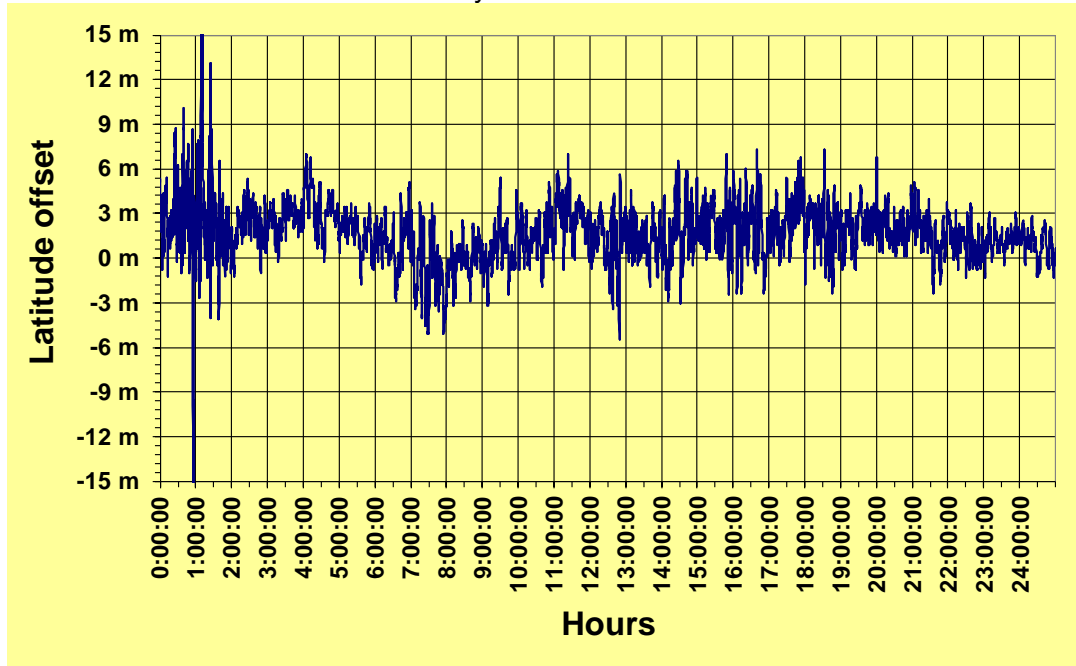
Test results

All deviations of measured positions from reference position are $< \pm 13$ m (95 %).
2 sigma value of position data: 6.62m.

Test result: Passed

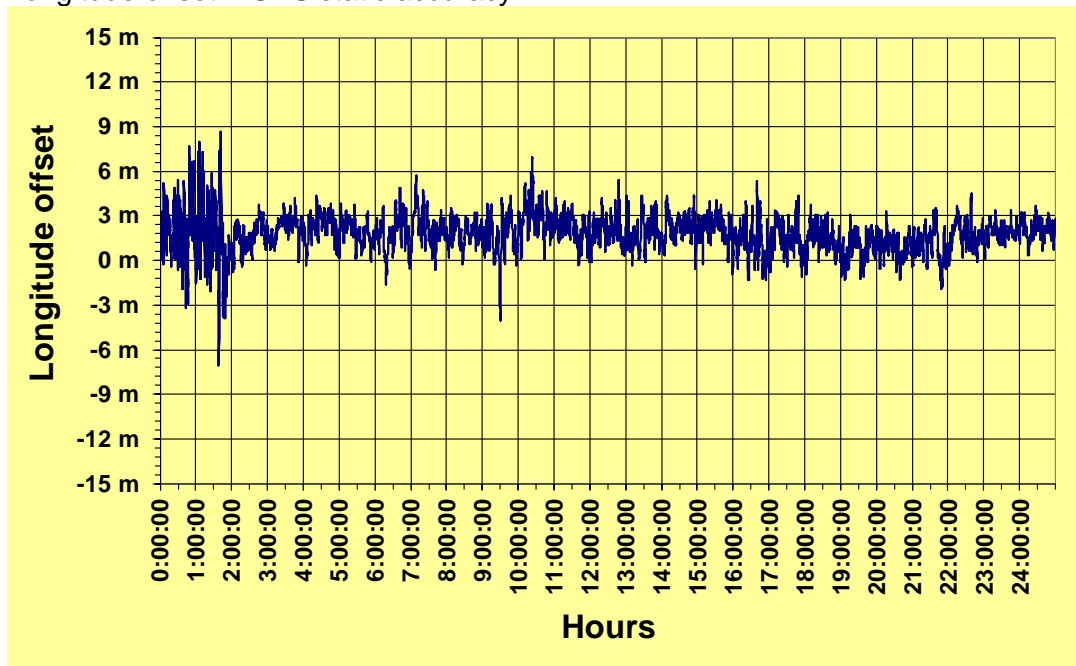
For details of validation of recorded data see the following pages.

Latitude offset – GPS static accuracy



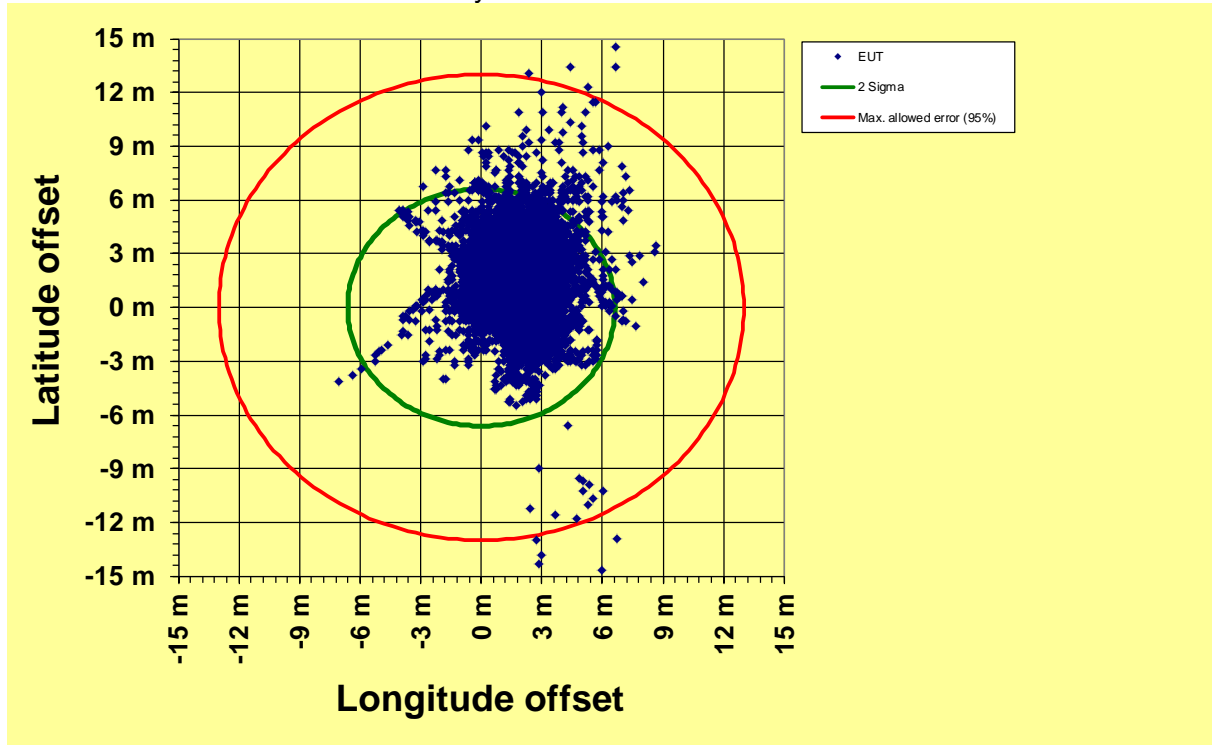
Pass/ fail criteria: 95% of all measurements within +/- 13m with respect to reference position- **Passed**

Longitude offset – GPS static accuracy



Pass/ fail criteria: 95% of all measurements within +/- 13m with respect to reference position- **Passed**

Position offset – GPS static accuracy



Pass/ fail criteria: 95% of all measurements within +/- 13m with respect to reference position– **Passed**

B.1.2 §5.6.4.1.2 Static accuracy – differential GPS

Position fix measurements shall be taken once per second over a period of not <24h. The distribution of the horizontal error shall be within 10m (95 %). The horizontal position of the antenna shall be known to within 0.1m in the datum used for the generation of the corrections. The corrections shall be provided by an actual GPS broadcast in accordance with ITU-R M.823.

Conditions of tests performed – Real signal

Period of position fix measurements:	~24 h
Position fix measurements :	~86700
Accuracy:	HDOP \leq 4 (or PDOP \leq 6)

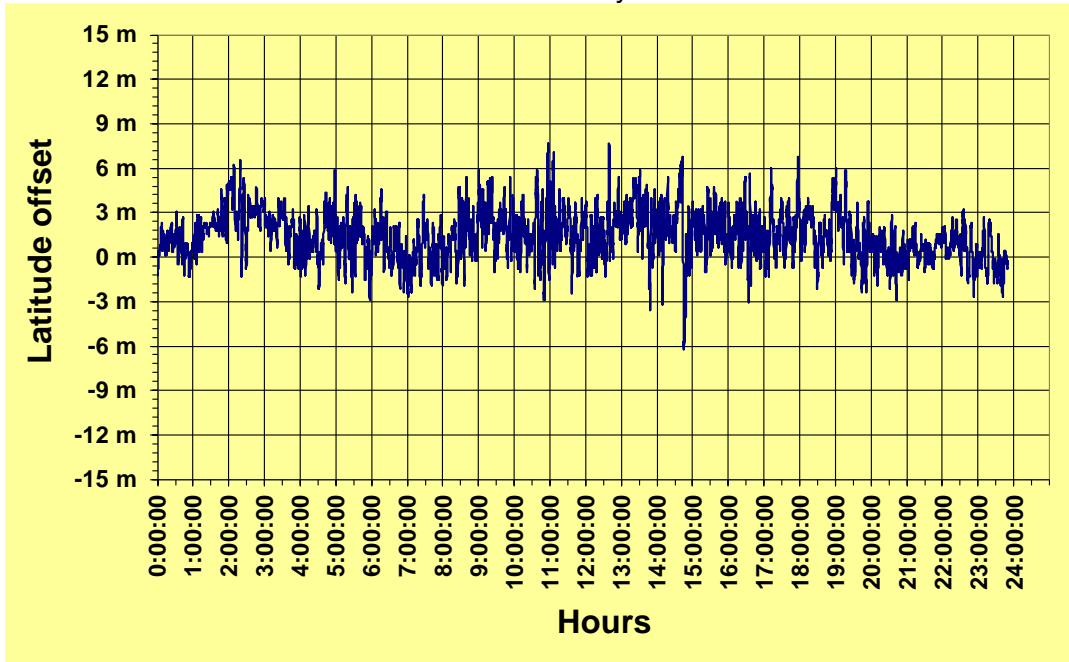
Test results

All deviations of measured positions from reference position are $<\pm 10$ m (95 %).
2 sigma value of position data: 6.16m.

Test result: Passed

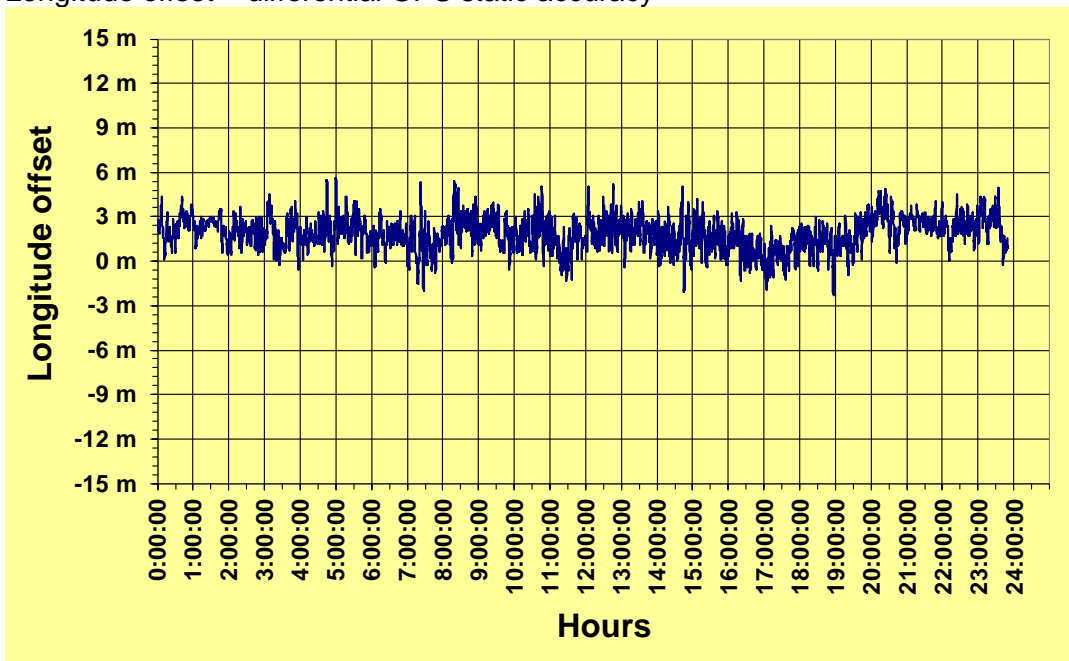
For details of validation of recorded data see the following pages.

Latitude offset – differential GPS static accuracy



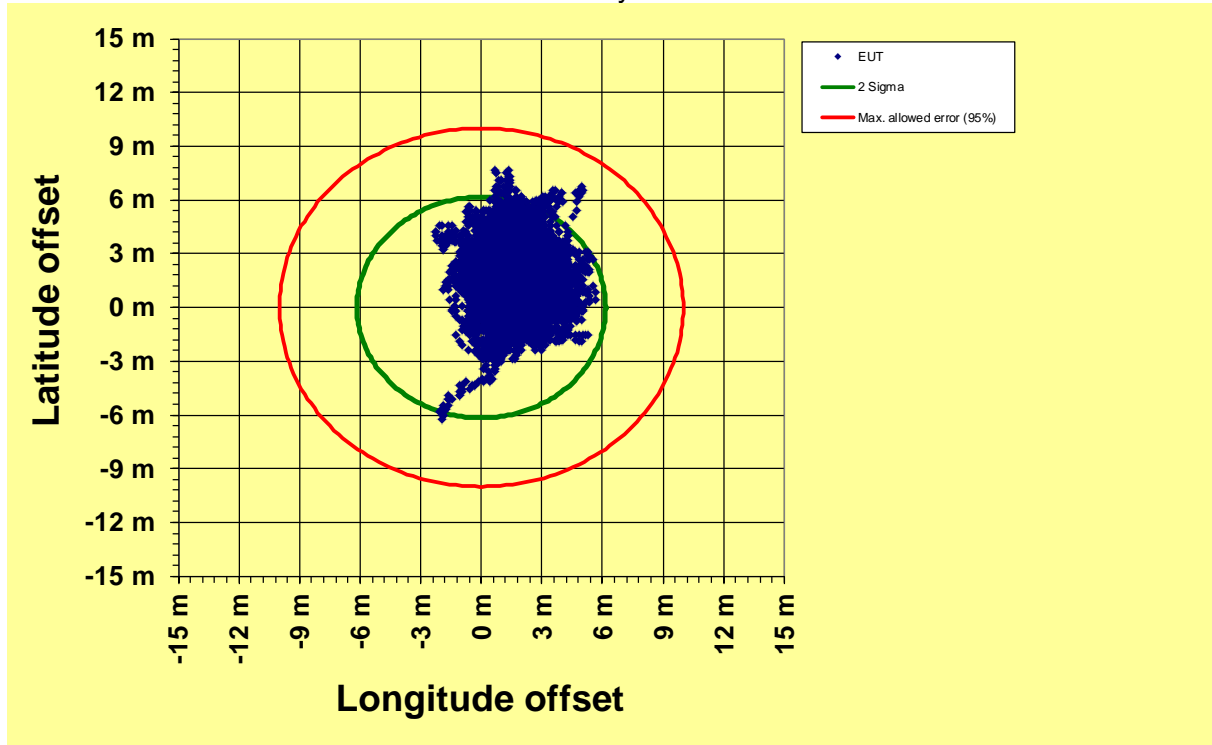
Pass/ fail criteria: 95% of all measurements within +/- 10m with respect to reference position- **Passed**

Longitude offset – differential GPS static accuracy



Pass/ fail criteria: 95% of all measurements within +/- 10m with respect to reference position- **Passed**

Position offset – differential GPS static accuracy



Pass/ fail criteria: 95% of all measurements within +/- 10m with respect to reference position- **Passed**

B.2 § 5.6.4.2 Angular movement of the antenna

B.2.1 § 5.6.4.2 Angular movement of the antenna – GPS

The static test(s) specified in 5.6.4.1.1 (and 5.6.4.1.2) shall be repeated with the antenna performing an angular displacement of $\pm 22.5^\circ$ (simulating roll) in a period of about 8s during the duration of the test.

Conditions of tests performed – Real signal

Period of position fix measurements:	~ 24 h
Position fix measurements :	~84600
Accuracy:	HDOP ≤ 4 (or PDOP ≤ 6)

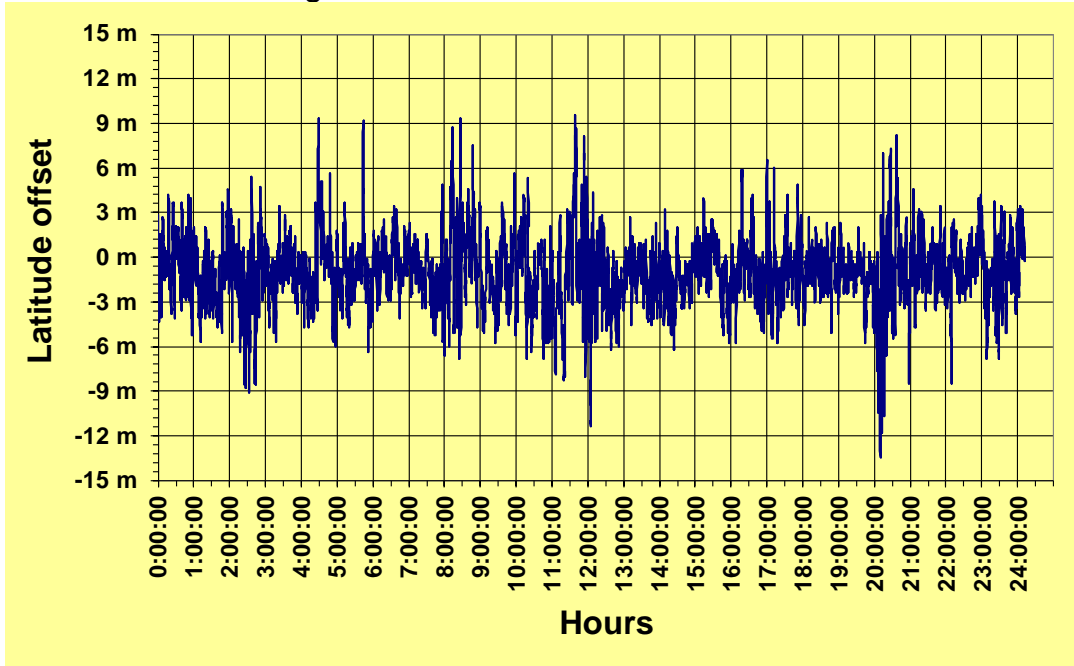
Test results for § 5.6.4.2.1

All deviations of measured positions from reference position are $< \pm 13$ m (95 %).
2 sigma value of position data: 6.03m.

Test result: Passed

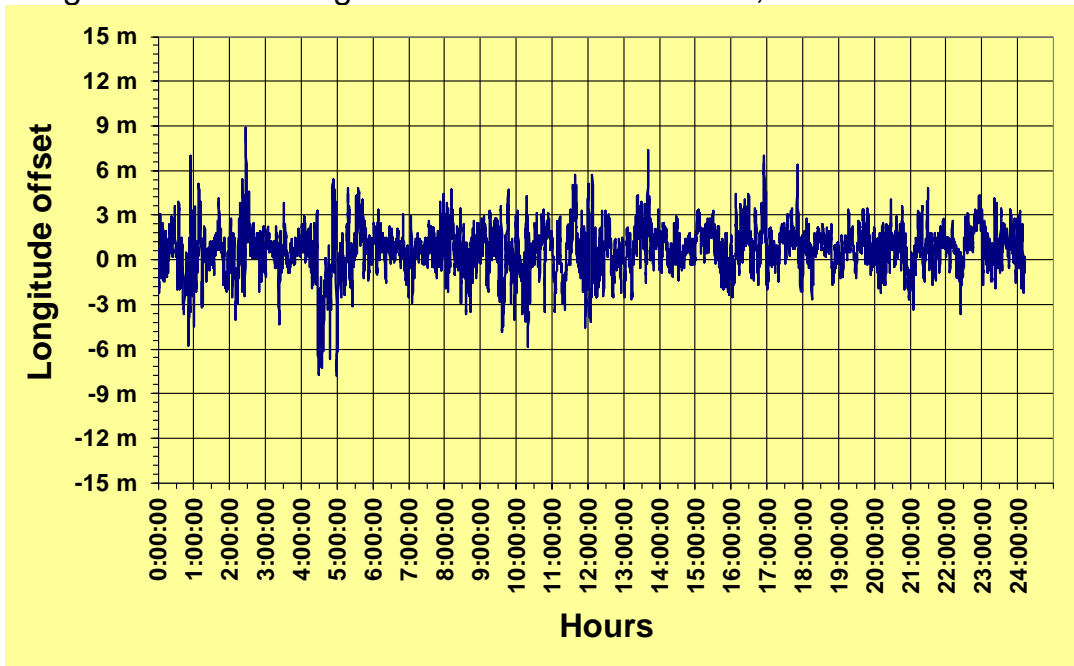
For details of validation of recorded data see the following pages.

Latitude offset – Angular movement of antenna, GPS



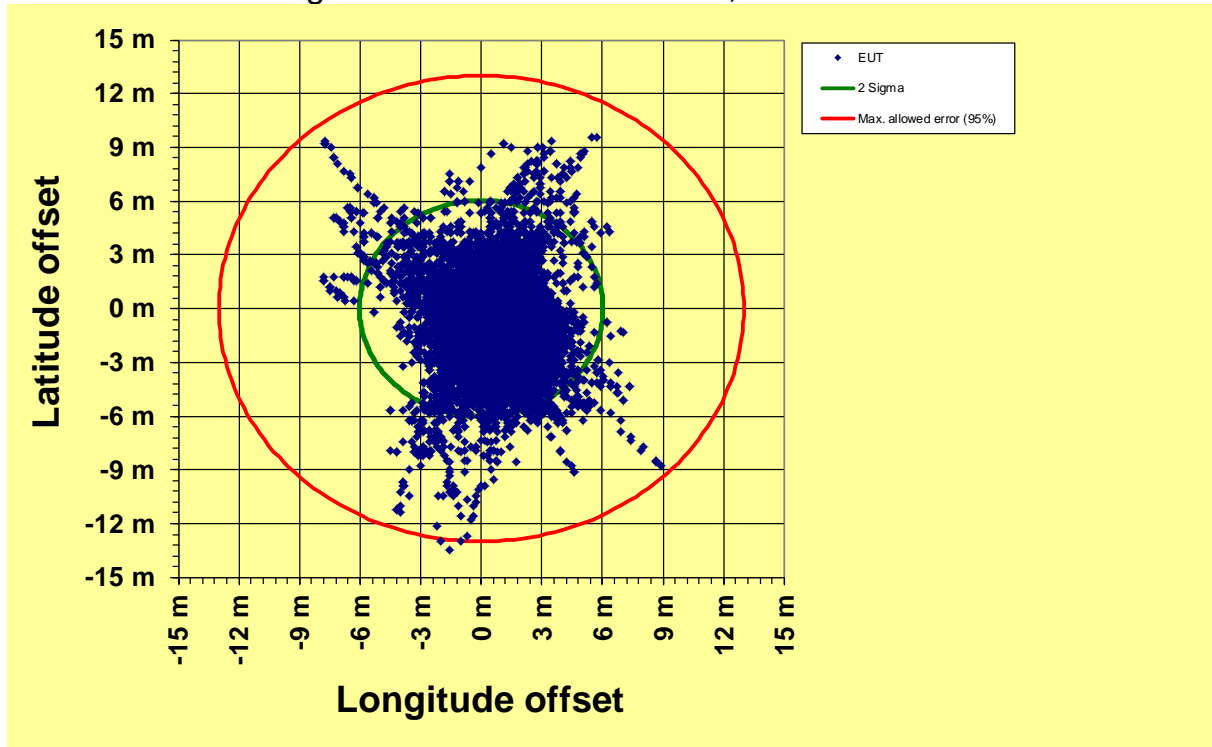
Pass/ fail criteria: 95% of all measurements within +/- 13m with respect to reference position- **Passed**

Longitude offset – Angular movement of antenna, GPS



Pass/ fail criteria: 95% of all measurements within +/- 13m with respect to reference position- **Passed**

Position offset – Angular movement of antenna, GPS



Pass/ fail criteria: 95% of all measurements within +/- 13m with respect to reference position- **Passed**

B.2.2 § 5.6.4.2 Angular movement of the antenna – DGPS

The static test(s) specified in 5.6.4.1.1 (and 5.6.4.1.2) shall be repeated with the antenna performing an angular displacement of $\pm 22.5^\circ$ (simulating roll) in a period of about 8s (see IEC60721-3-6) during the duration of the test.

Conditions of tests performed – Real signal

Antenna placed on a motor-driven socket simulating the angular displacement required for the test.

Period of position fix measurements:	~24 h
Position fix measurements :	~86400
Accuracy:	HDOP ≤ 4 (or PDOP ≤ 6)

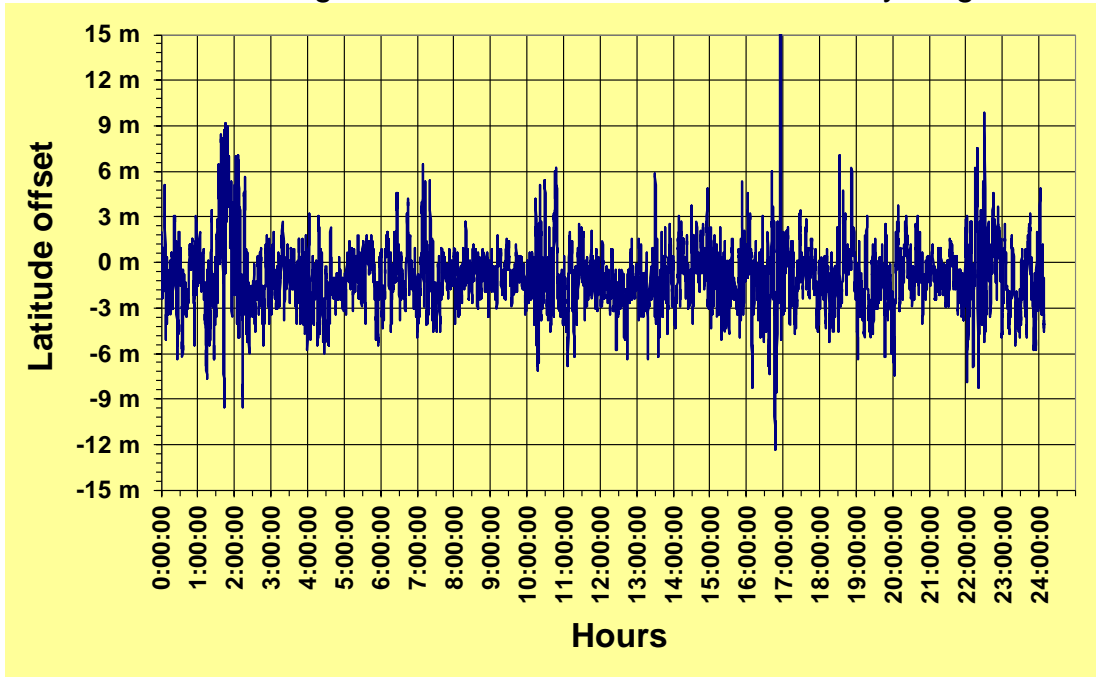
Test results for § 5.6.4.2.2

All deviations of measured positions from reference position are $< \pm 10$ m (95 %).
2 sigma value of position data: 5.91m.

Test result: Passed

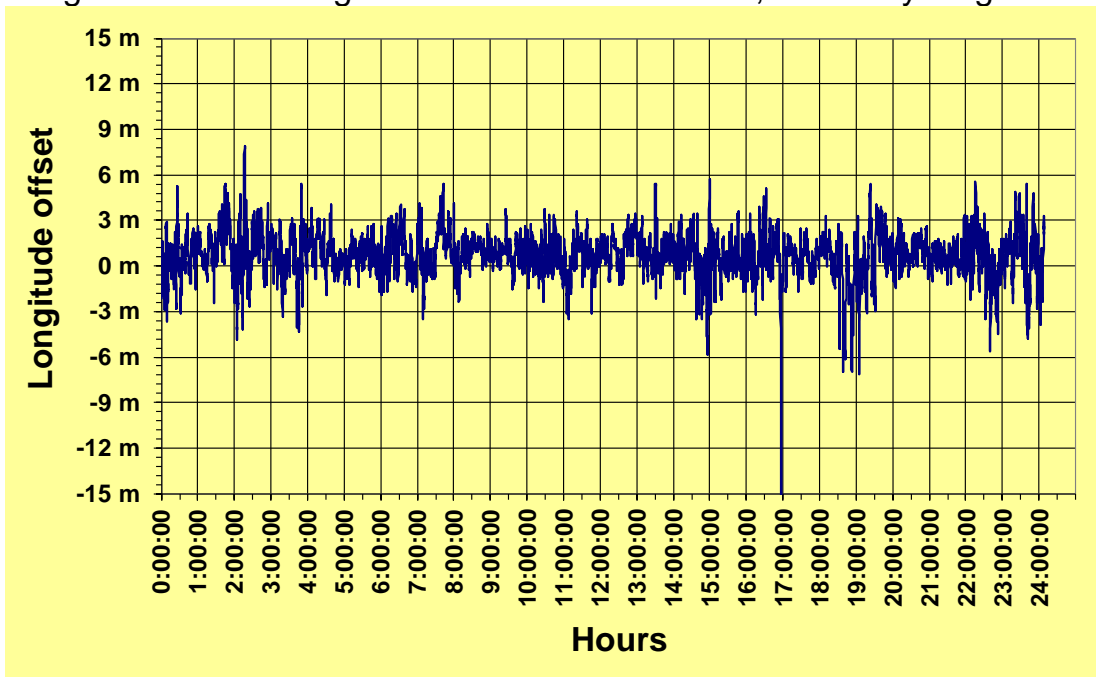
For details of validation of recorded data see the following pages.

Latitude offset – Angular movement of antenna, DGPS by msg. 17



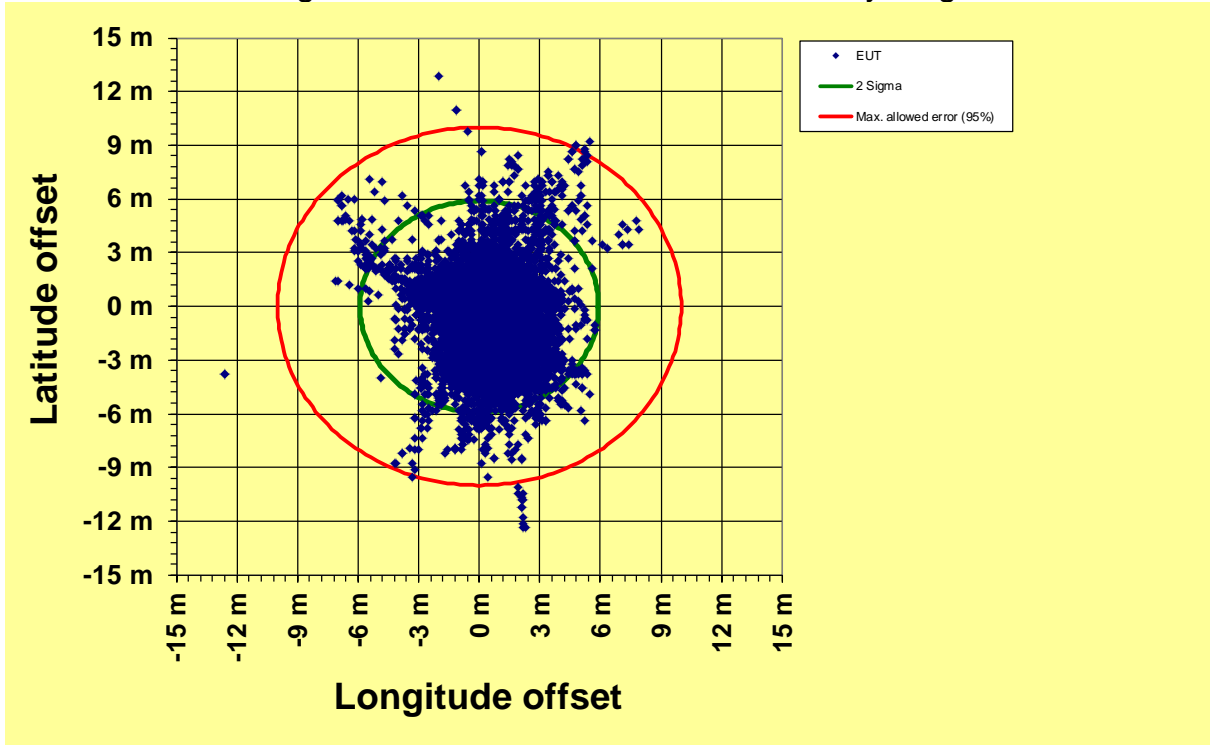
Pass/ fail criteria: 95% of all measurements within +/- 10m with respect to reference position- **Passed**

Longitude offset – Angular movement of antenna, DGPS by msg. 17



Pass/ fail criteria: 95% of all measurements within +/- 10m with respect to reference position- **Passed**

Positon offset – Angular movement of antenna, DGPS by msg. 17



Pass/ fail criteria: 95% of all measurements within +/- 10m with respect to reference position - **Passed**

B.3 § 5.6.4.3 Dynamic accuracy

Reference position:
By GPS-Simulator

	geocentrically co-ordinates (WGS84)
Latitude	0.000000000000 ° N
Longitude	0.000000000000 ° E

Accuracy of survey = ± 5 cm according to manufacturer documentation

B.3.1 § 5.6.4.3.1 GPS part a)

A fully locked and settled EUT travelling in a straight line at 48 kn \pm 2kn for a minimum of 1.2min which is reduced to 0 kn in the same straight line in 5 s, shall not indicate a position offset \pm 13m from the final position 10s after coming to rest.

Conditions of tests performed – Simulated signal

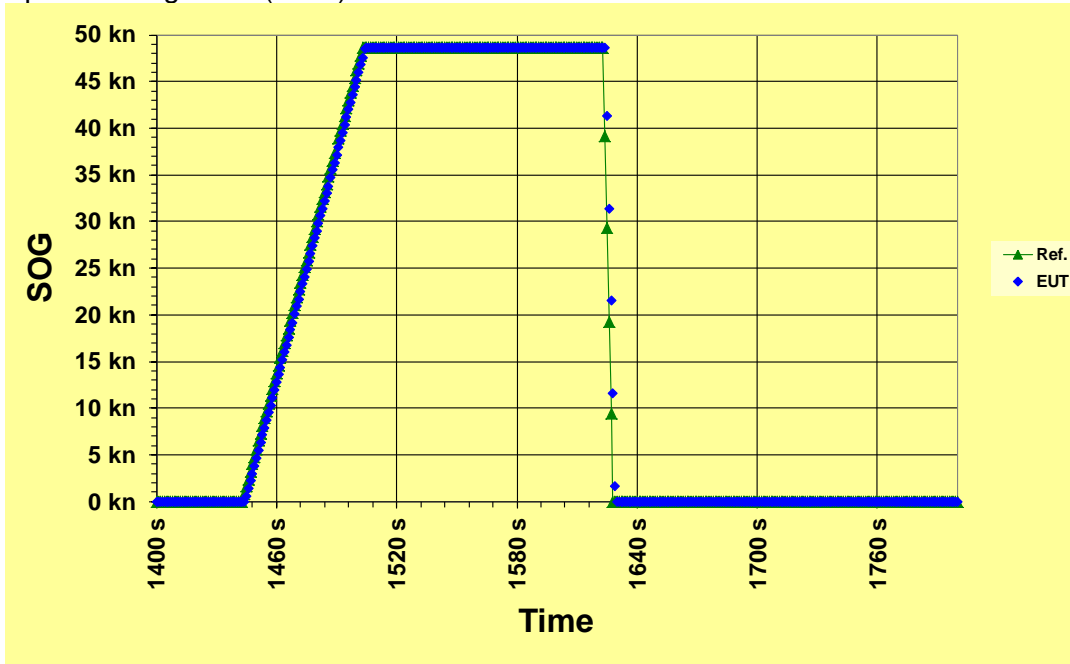
Test results

All positions offsets are $<\pm 13$ m.

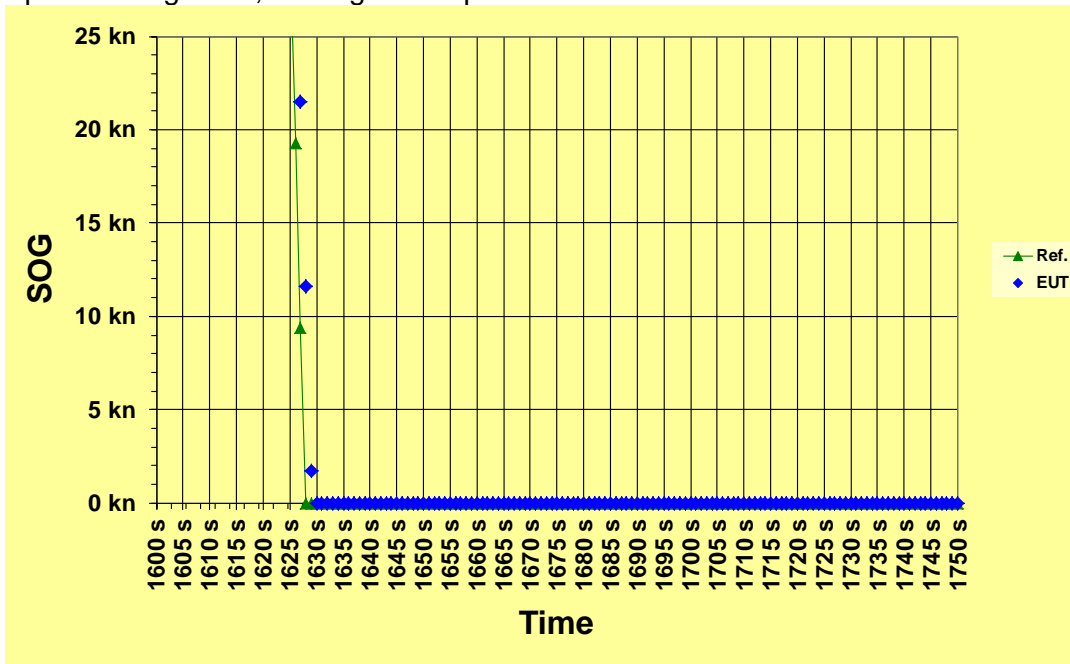
Test result: Passed

For details of validation of recorded data see the following pages.

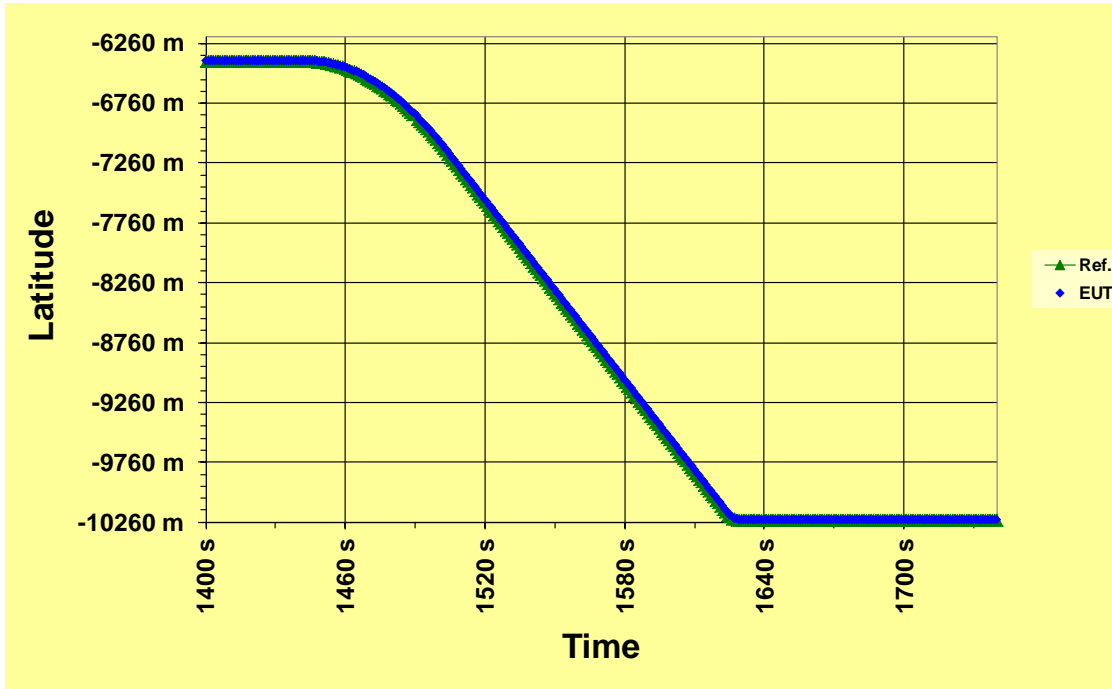
Speed over ground (SOG)



Speed over ground, coming to rest position

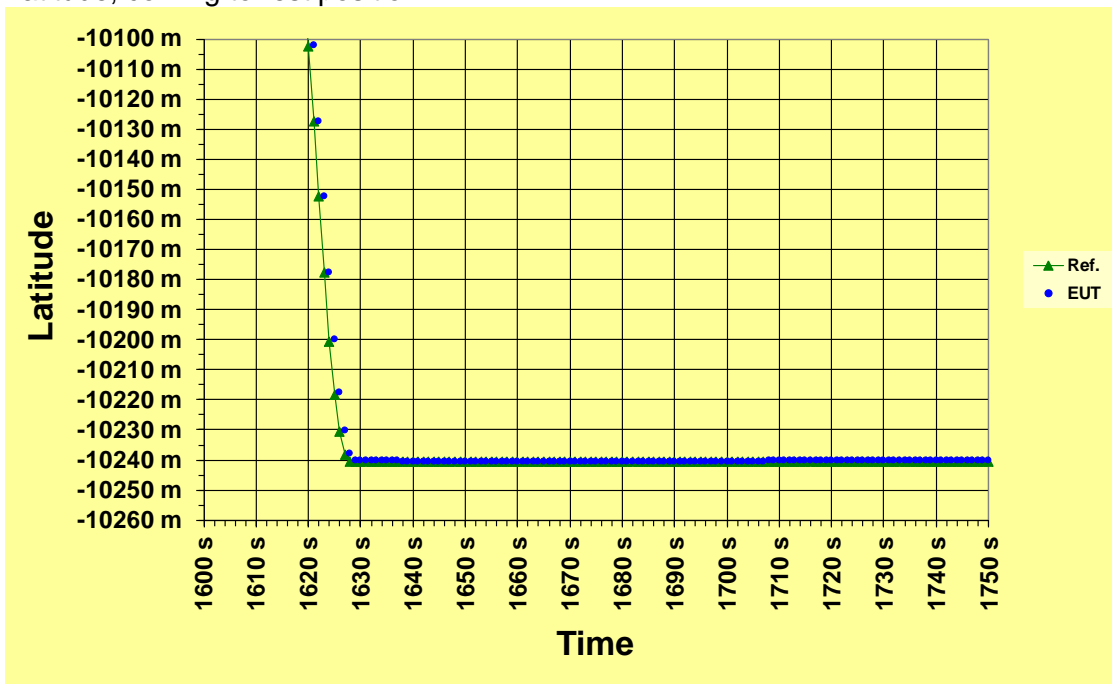


Latitude



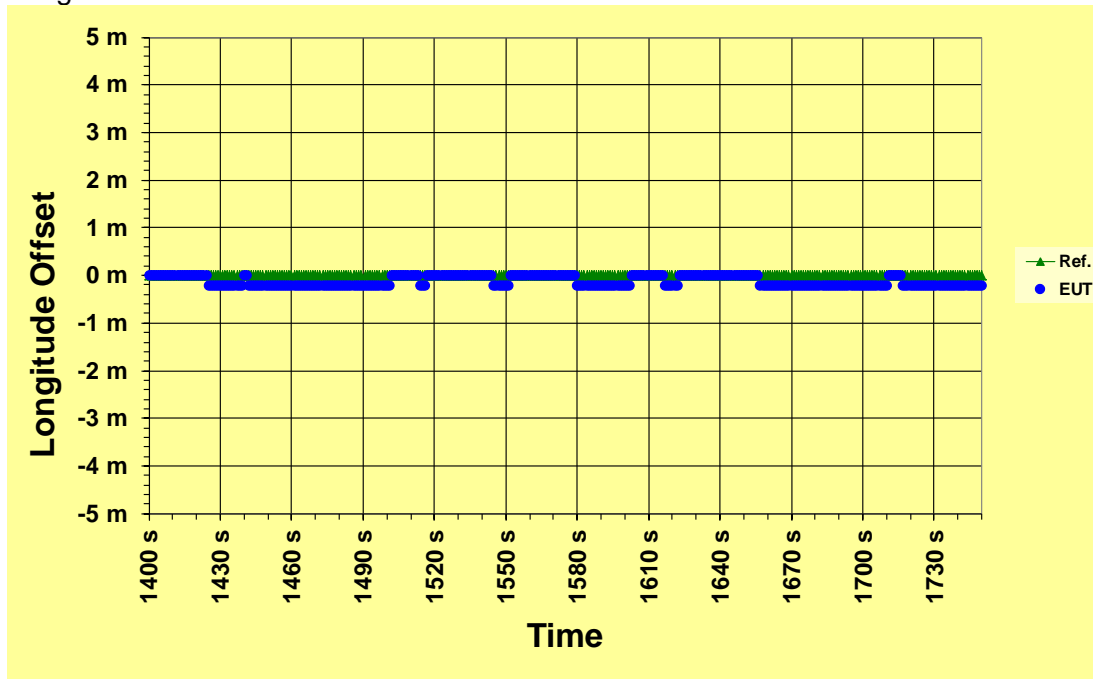
Pass/ fail criteria: 95% of all measurements within +/- 13m with respect to reference position - **Passed**

Latitude, coming to rest position



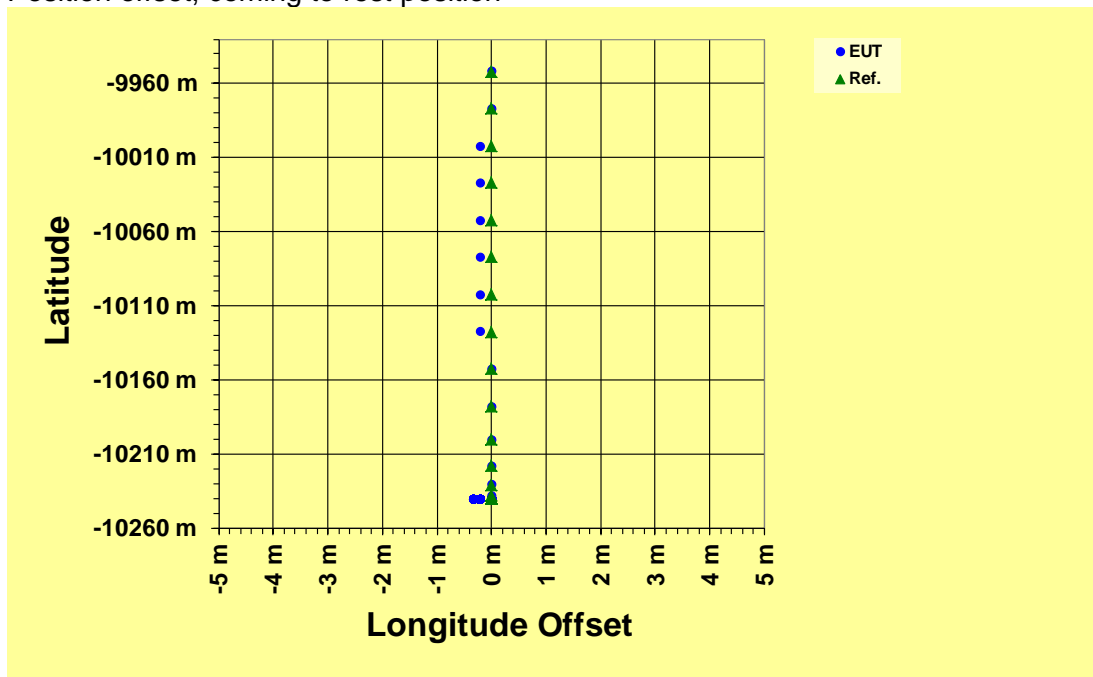
Pass/ fail criteria: 95% of all measurements within +/- 13m with respect to reference position - **Passed**

Longitude



Pass/ fail criteria: 95% of all measurements within +/- 13m with respect to reference position - **Passed**

Position offset, coming to rest position



Pass/ fail criteria: 95% of all measurements within +/- 13m with respect to reference position - **Passed**

B.3.2 §5.6.4.3.1 GPS part b)

A fully locked and settled EUT travelling at least 100m at 24kn \pm 1kn in a straight line then subjected, for at least 2 min, to smooth deviations either side of the straight line of approximately 2m at a period of 11s to 12s shall remain in lock and follow the actual position to within a lane of 30m wide centred on the mean direction of motion.

Conditions of tests performed – Simulated signal

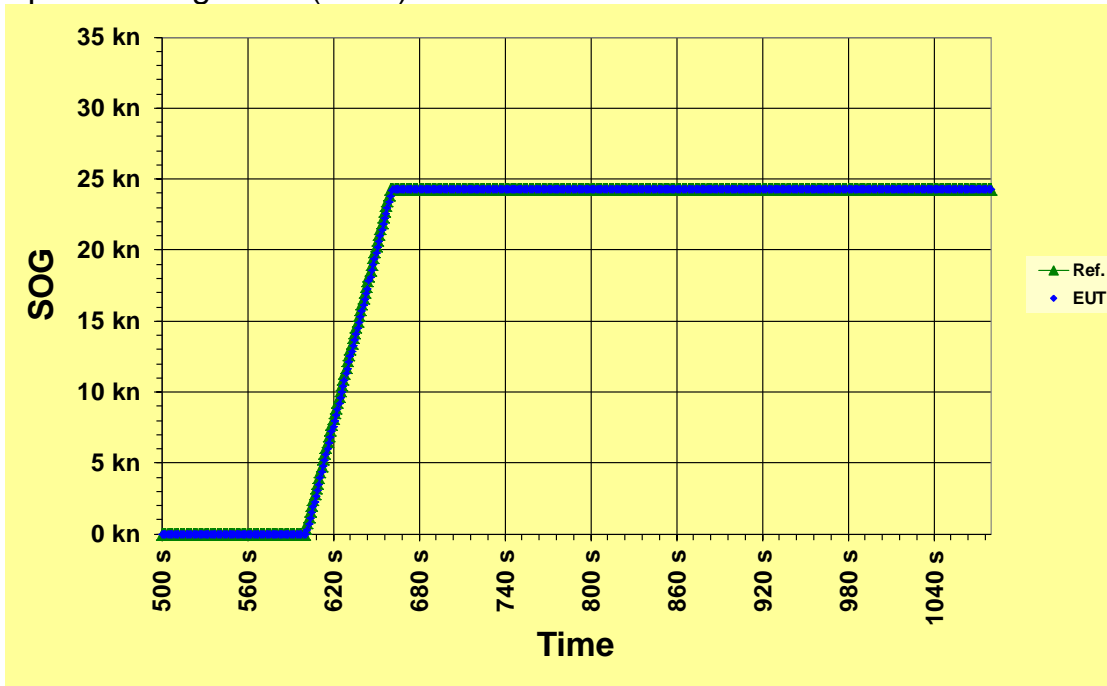
Test results

All positions offsets are within a lane of 30 m.

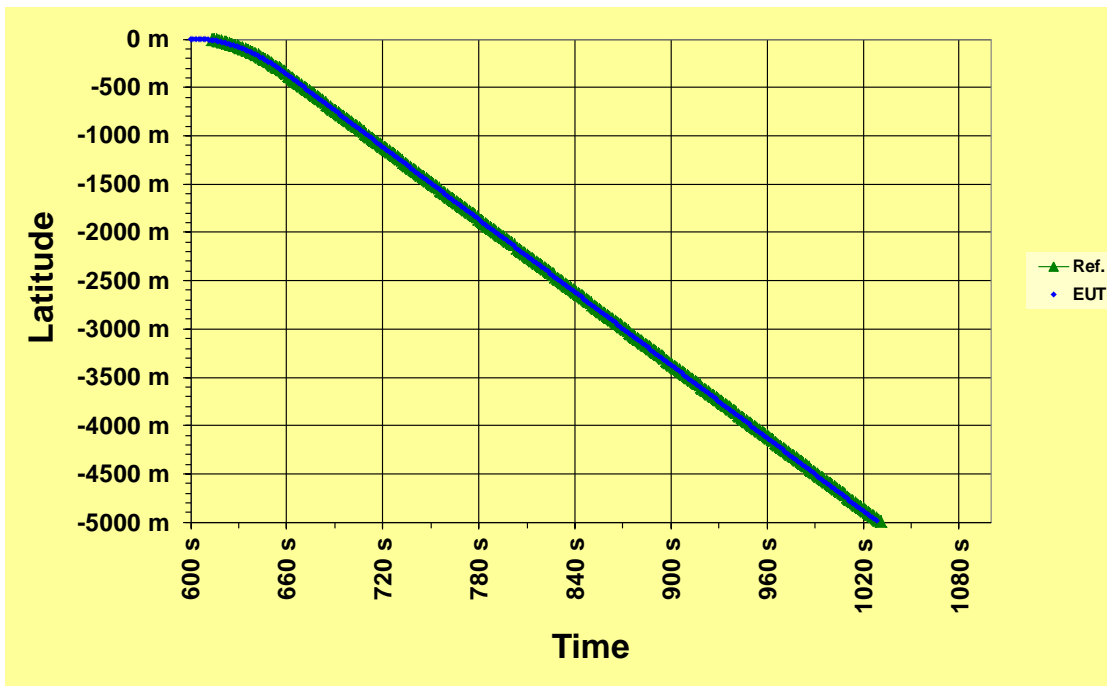
Test result: Passed

For details of validation of recorded data see the following pages.

Speed over ground (SOG) vs. time

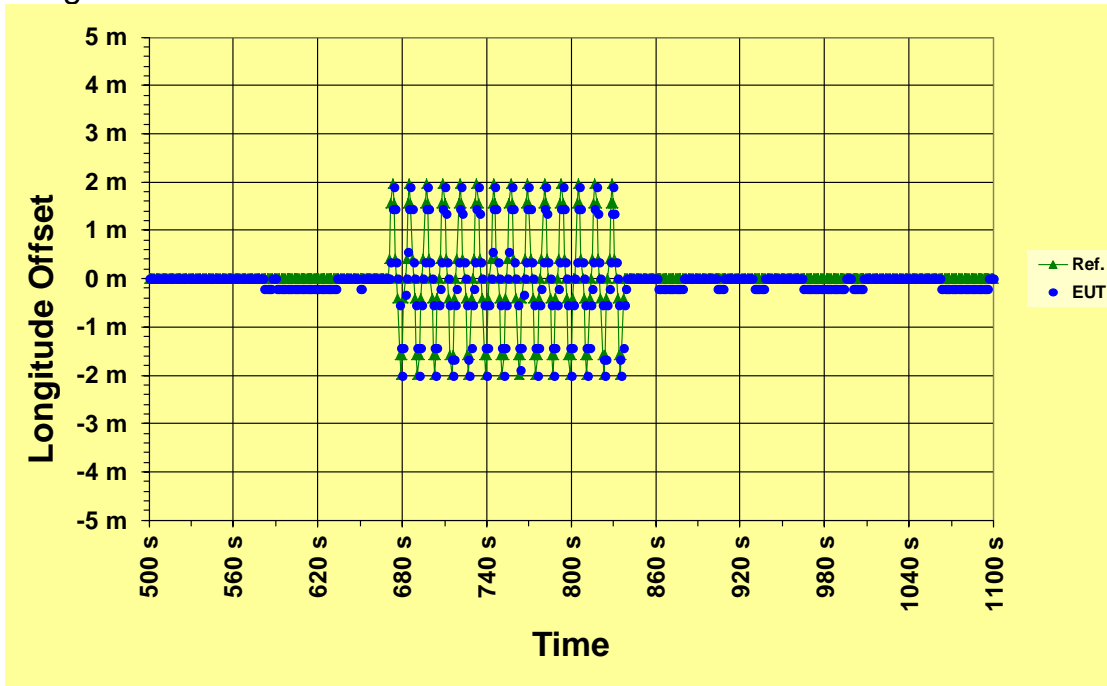


Latitude vs. time



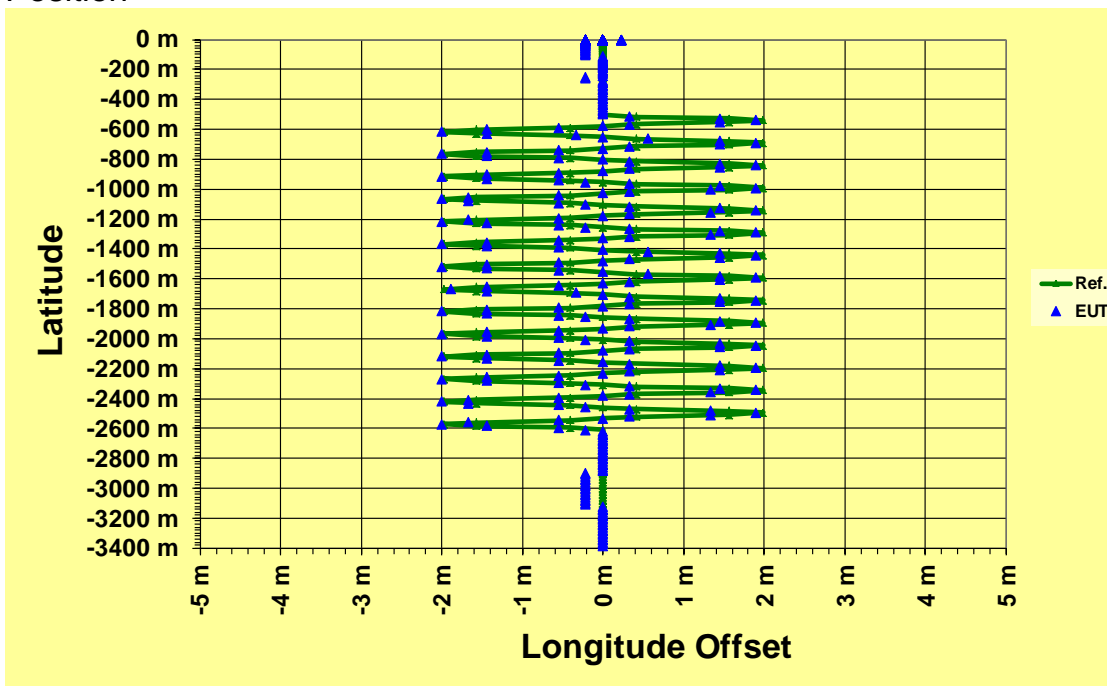
Pass/ fail criteria: All position offsets to be within a lane of 30 m with respect to reference - **Passed**

Longitude vs. time



Pass/ fail criteria: All position offsets to be within a lane of 30 m with respect to reference - **Passed**

Position



Pass/ fail criteria: All position offsets to be within a lane of 30 m with respect to reference - **Passed**

B.3.3 § 5.6.4.3.2 Differential GPS part a)

A fully locked and settled EUT travelling in a straight line at 48kn ± 2 kn for a minimum of 1.2min which is reduced to 0kn in the same straight line in 5s, shall not indicate a position offset ± 10 m from the true position at rest and the indicated position shall settle to within ± 2 m of the rest position indication within 10s of coming to rest.

Conditions of tests performed – Simulated signal

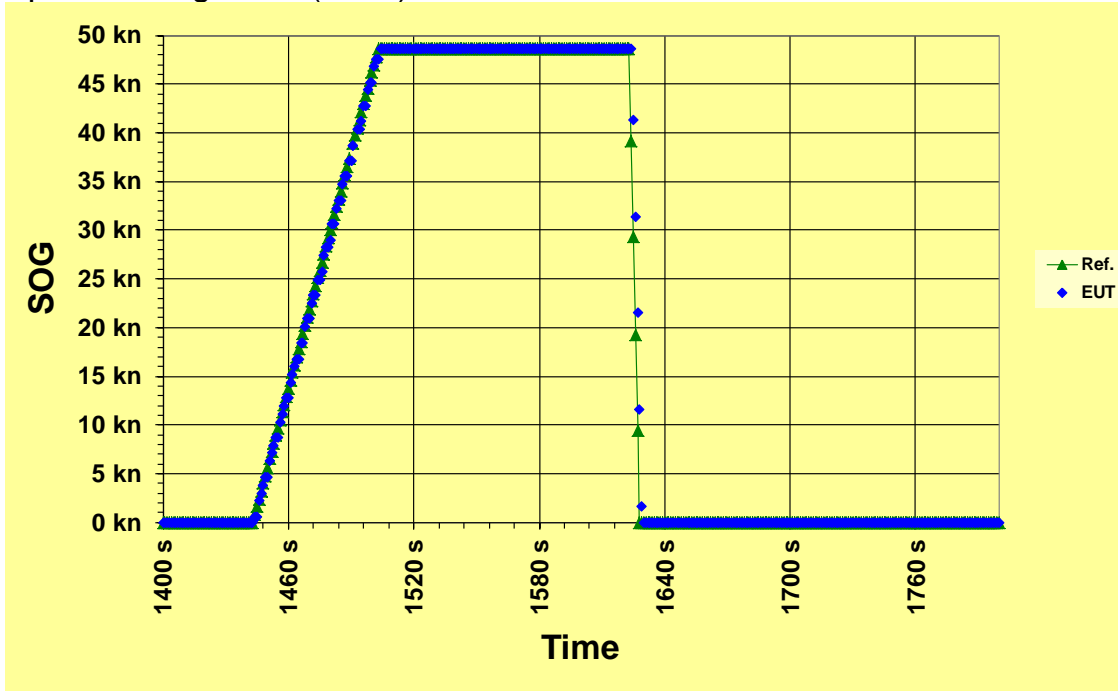
Test results

All positions offsets are $< \pm 10$ m and position variation within ± 2 m.

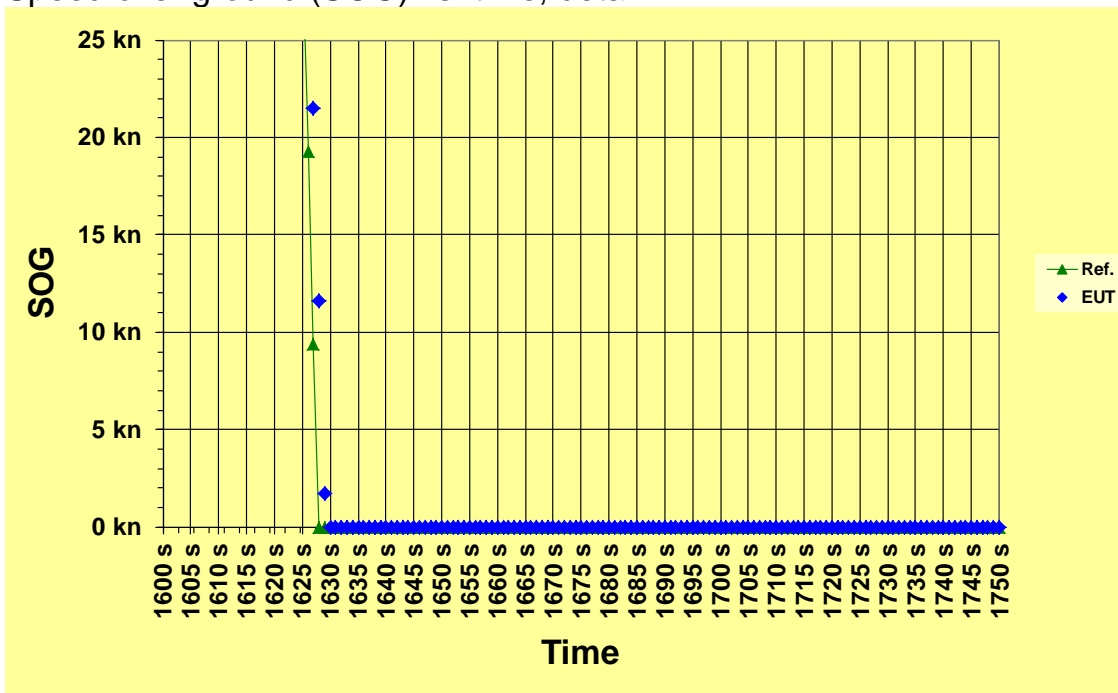
Test result: Passed

For details of validation of recorded data see the following pages.

Speed over ground (SOG) vs. time

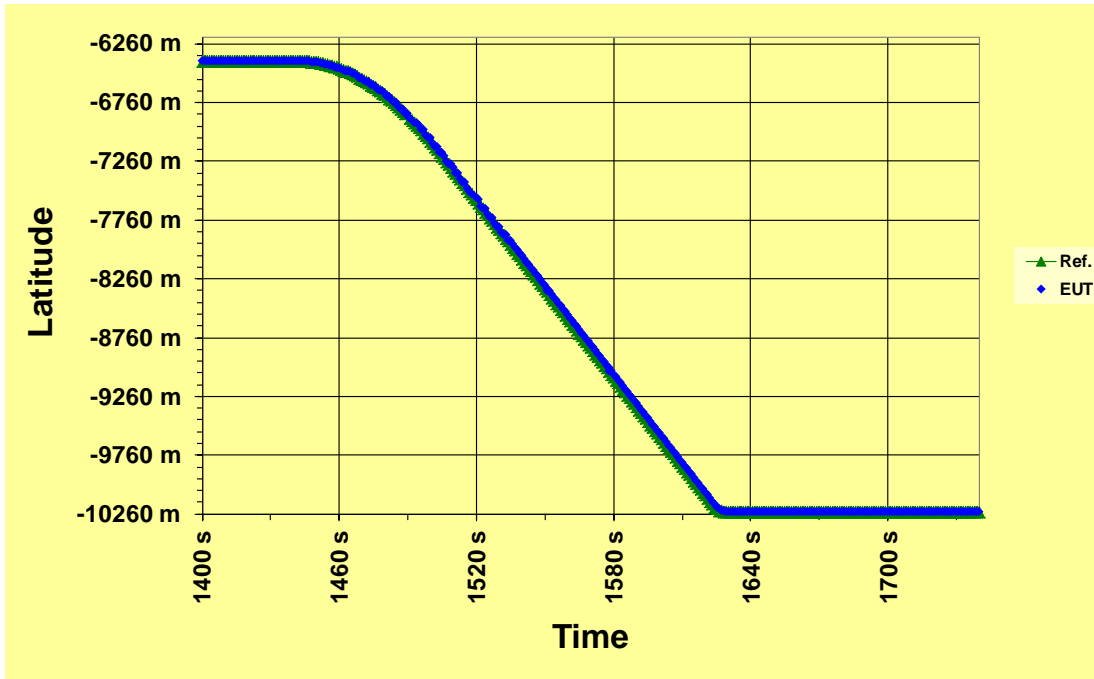


Speed over ground (SOG) vs. time, detail



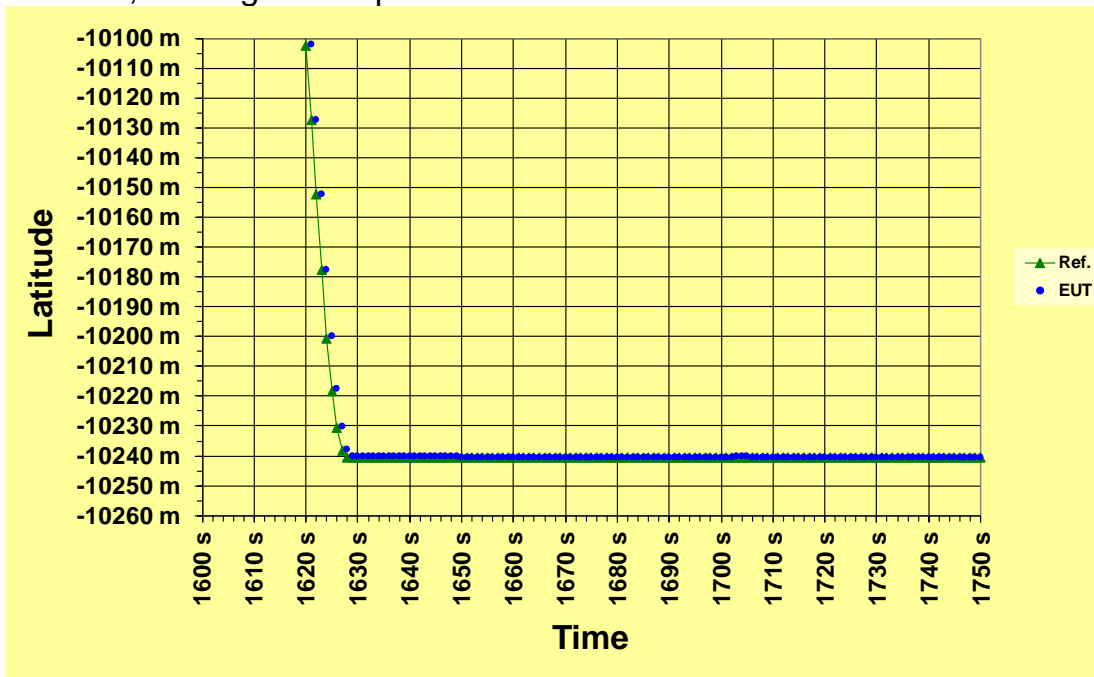
Pass/ fail criteria: 95% of all measurements within +/- 10m with respect to reference position - **Passed**

Latitude vs. time



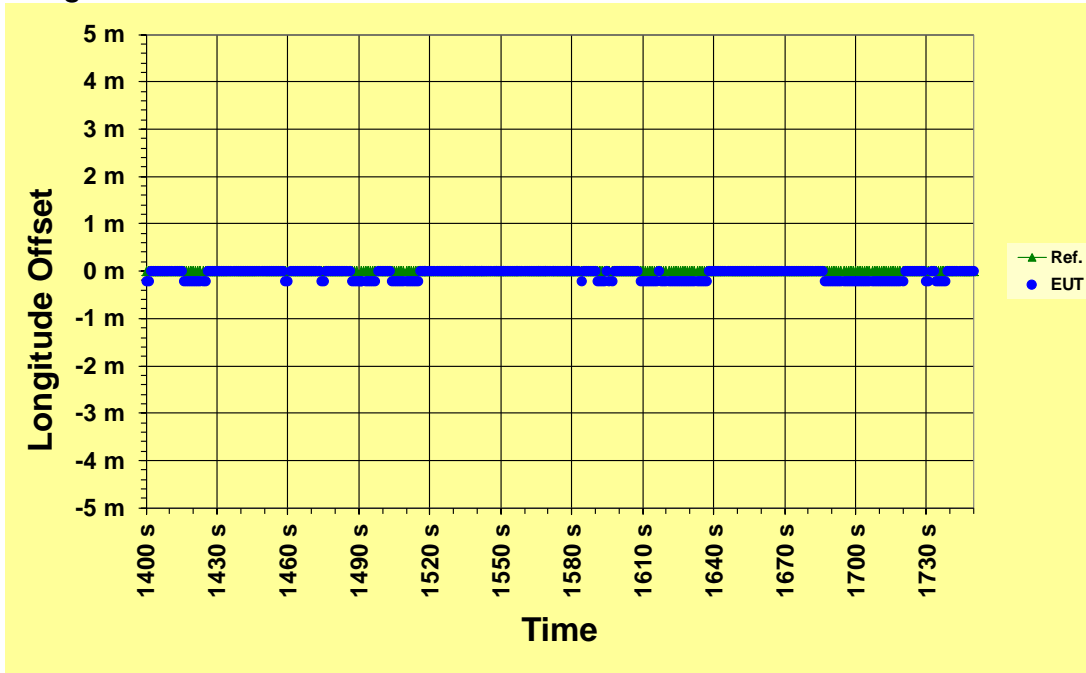
Pass/ fail criteria: 95% of all measurements within +/- 10m with respect to reference position - **Passed**

Latitude, coming to rest position



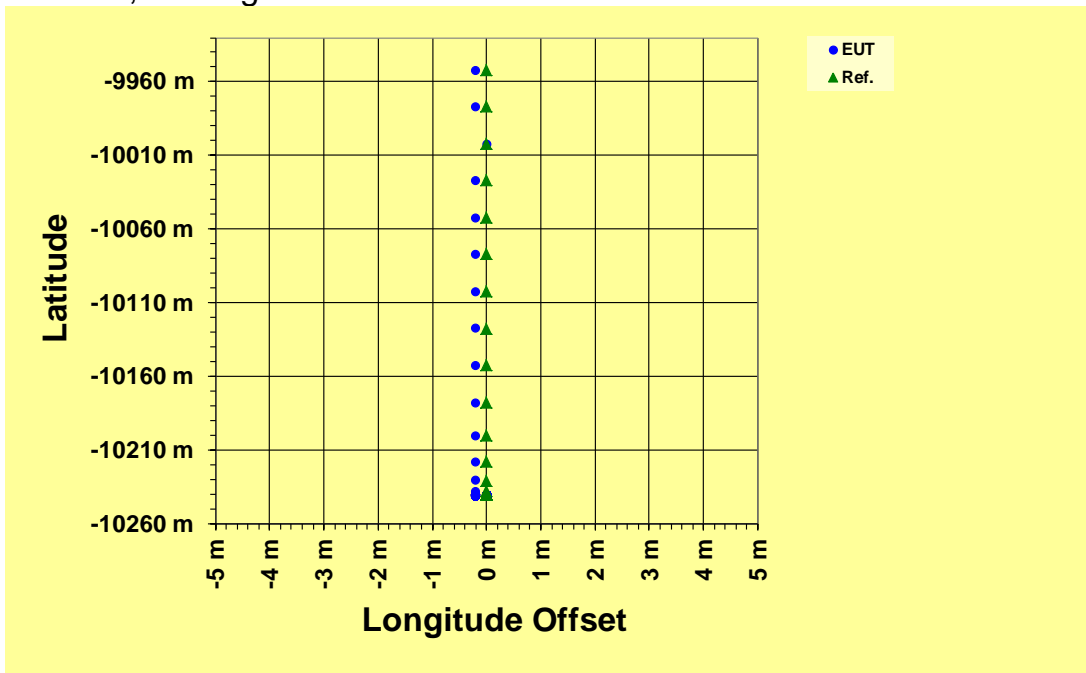
Pass/ fail criteria: 95% of all measurements within +/- 10m with respect to reference position - **Passed**

Longitude vs. time



Pass/ fail criteria: 95% of all measurements within +/- 10m with respect to reference position - **Passed**

Position, coming to rest



Pass/ fail criteria: 95% of all measurements within +/- 10m with respect to reference position - **Passed**

B.3.4 §5.6.4.3.2 Differential GPS part b)

A fully locked and settled EUT travelling at least 100m at 24kn \pm 1kn in a straight line then subjected, for at least 2 min, to smooth deviations either side of the straight line of approximately 2m at a period of 11s to 12s shall remain in lock and follow the actual position to within a lane of 30m wide centred on the mean direction of motion.

Conditions of tests performed

Tests performed by using a simulator

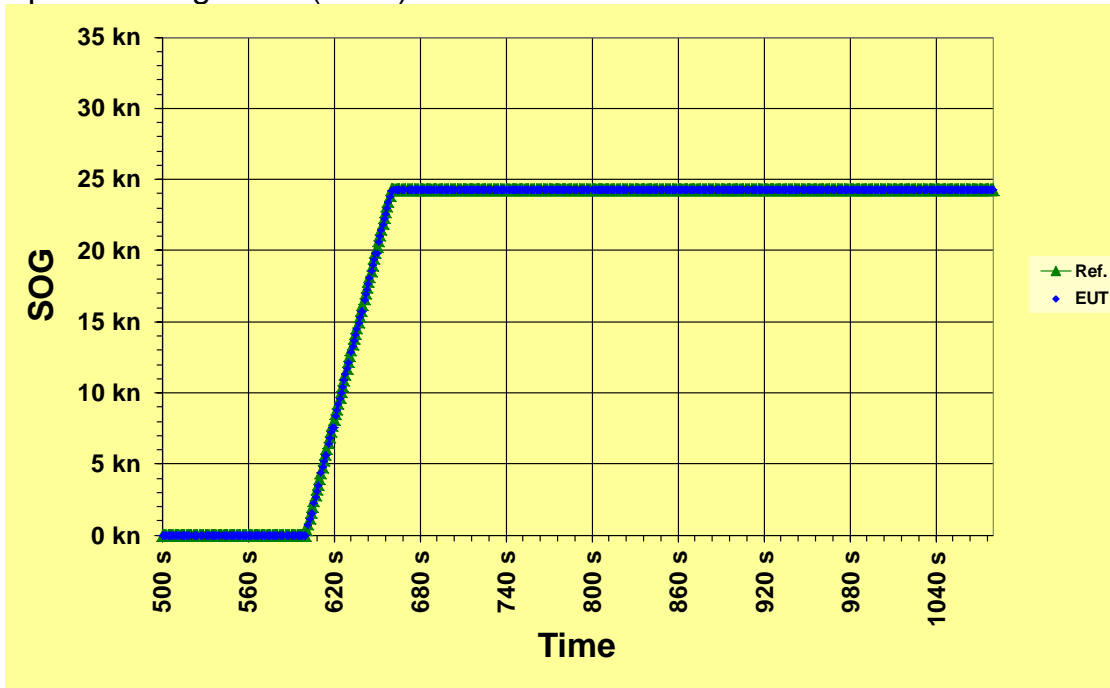
Test results

All positions offsets are within a lane of 30 m.

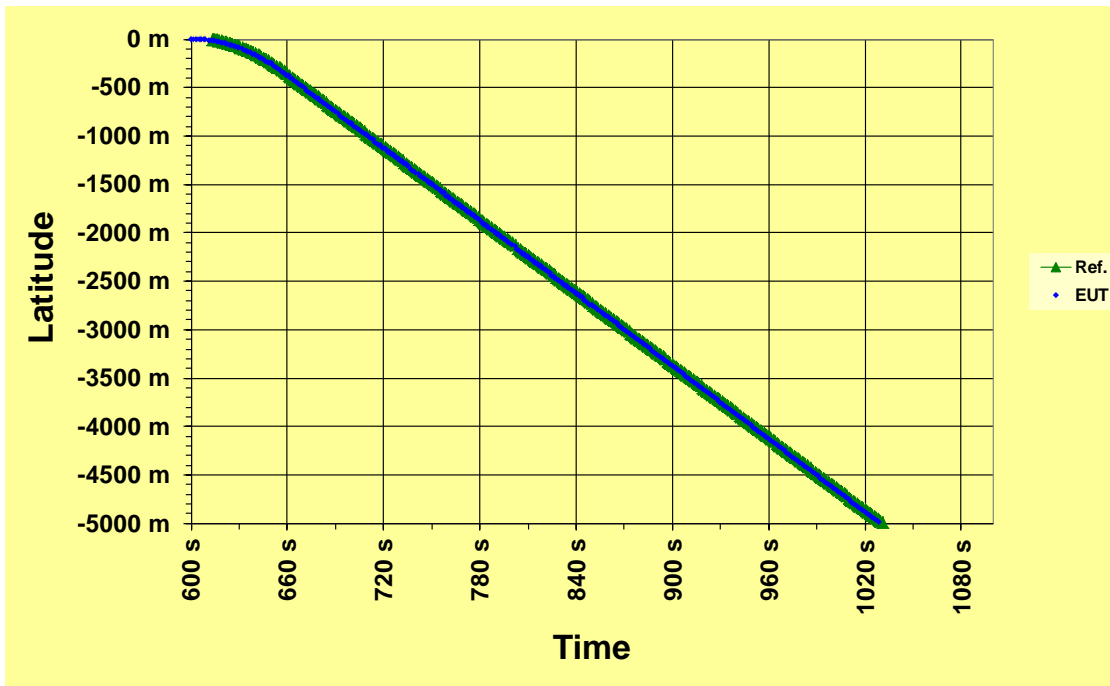
Test result: Passed

For details of validation of recorded data see the following pages.

Speed over ground (SOG) vs. time

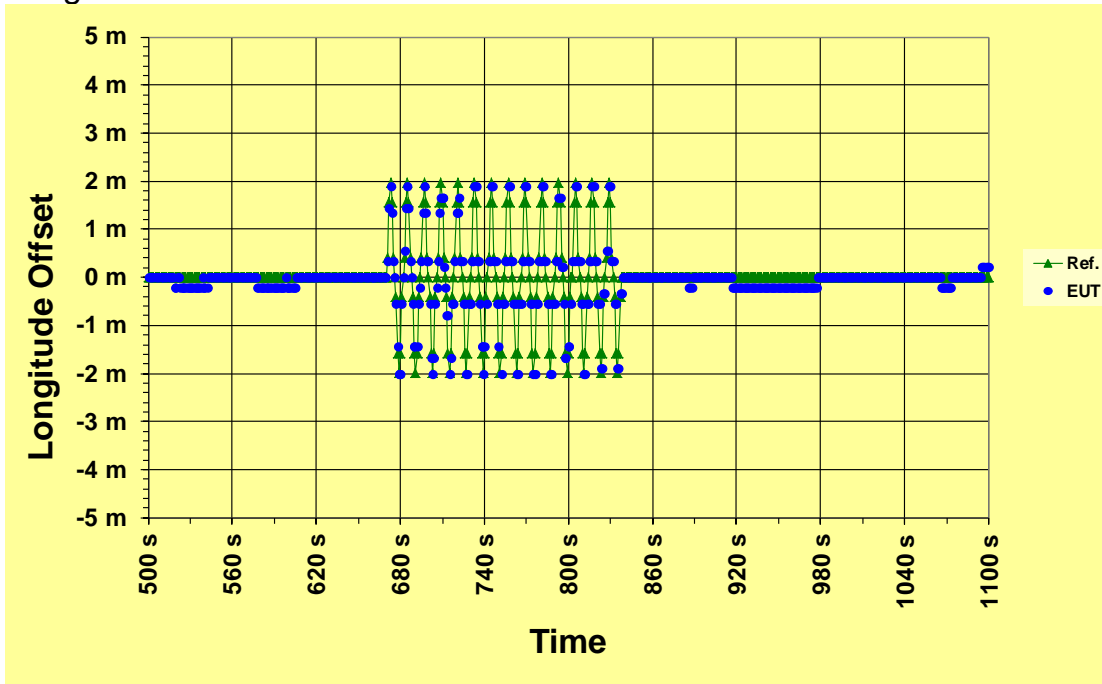


Latitude vs. time



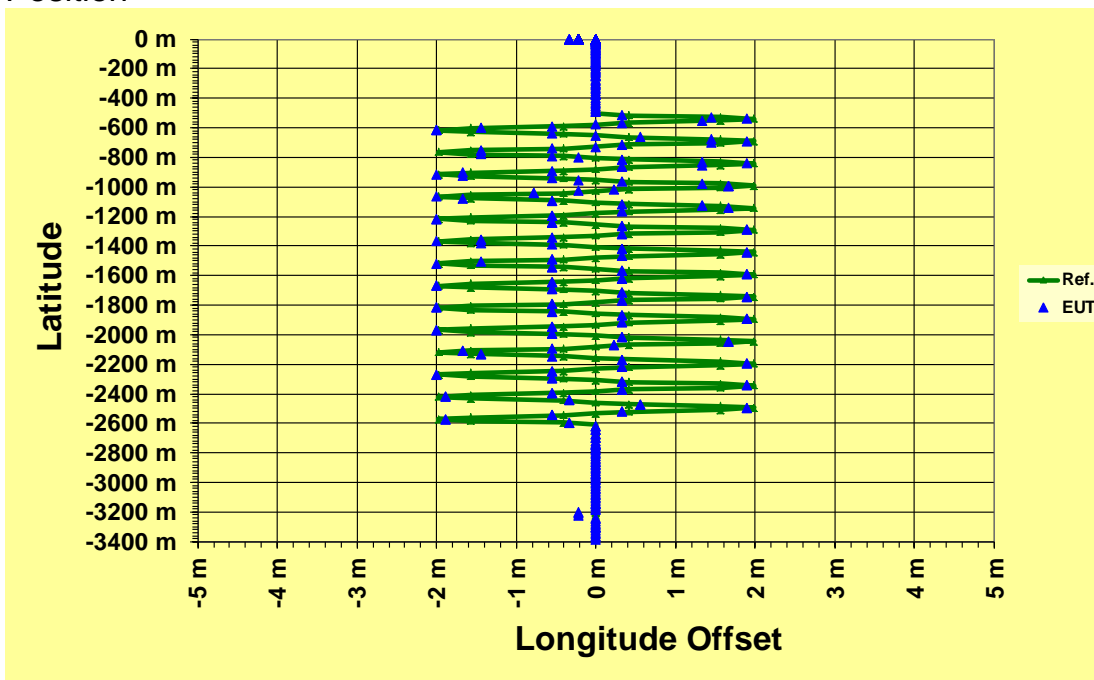
Pass/ fail criteria: All position offsets to be within a lane of 30 m with respect to reference - **Passed**

Longitude vs. time



Pass/ fail criteria: All position offsets to be within a lane of 30 m with respect to reference – **Passed**

Position



Pass/ fail criteria: All position offsets to be within a lane of 30 m with respect to reference - **Passed**

B.4 § 5.6.5 Aquisition

B.4.1 § 5.6.5.1 Condition A - Initialization

EUT shall be powered on without valid almanach data in memory.
A performance check shall be carried out after 30 minutes of operation.

Conditions of test performed

Every simulator test forces almanach reset, EUT locks on to the GPS transmissions within less than 30 minutes and provides a valid position.

Test result: Passed

B.4.2 § 5.6.5.2 Condition B – Power Outage

EUT shall be isolated from power source for 24 to 25 hours.
A performance check shall be carried out after 5 minutes of operation.

Conditions of test performed

Every simulator test forces almanach reset, EUT locks on to the GPS transmissions within less than 5 minutes and provides a valid position.

Test result: Passed

B.4.3 § 5.6.5.3 Condition C – Interruption of GPS signal

EUT antenna shall be completely masked for 24 to 25 hours.
A performance check shall be carried out after 5 minutes of operation.

Conditions of test performed – Simulated signal

EUT antenna was masked for approx. 24.5 hours.

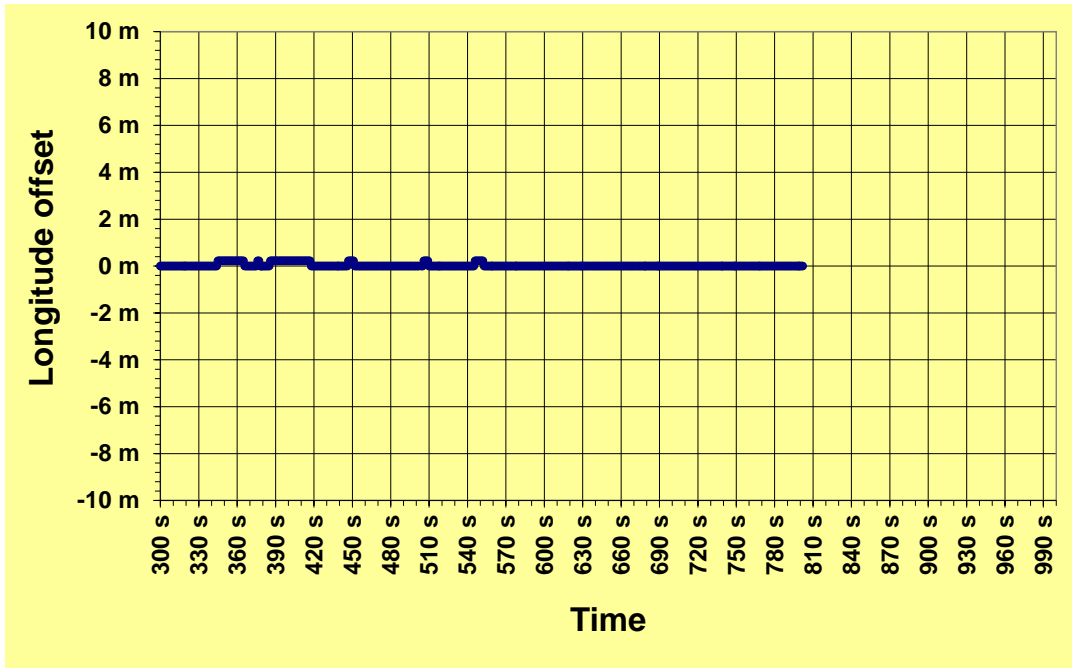
EUT locked on to GPS system within less than two minutes and provided a valid position fix.
A performance check was carried out after 5 minutes of operation.

2 Sigma value of performance check: 0.38m

Test result: Passed

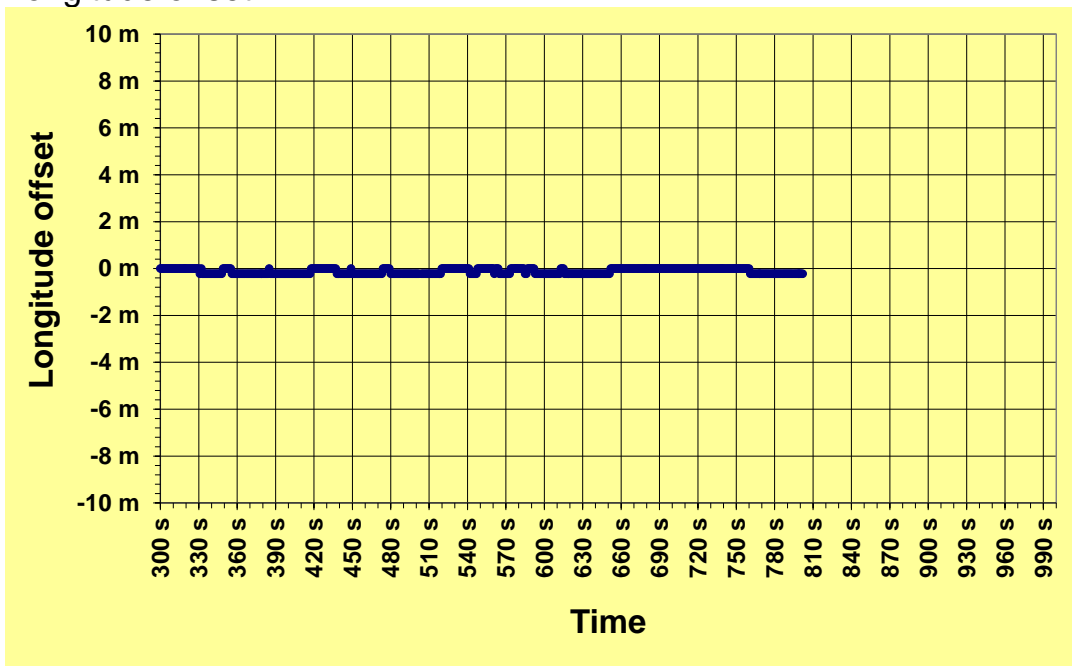
For details of validation of recorded data see the following pages.

Latitude offset



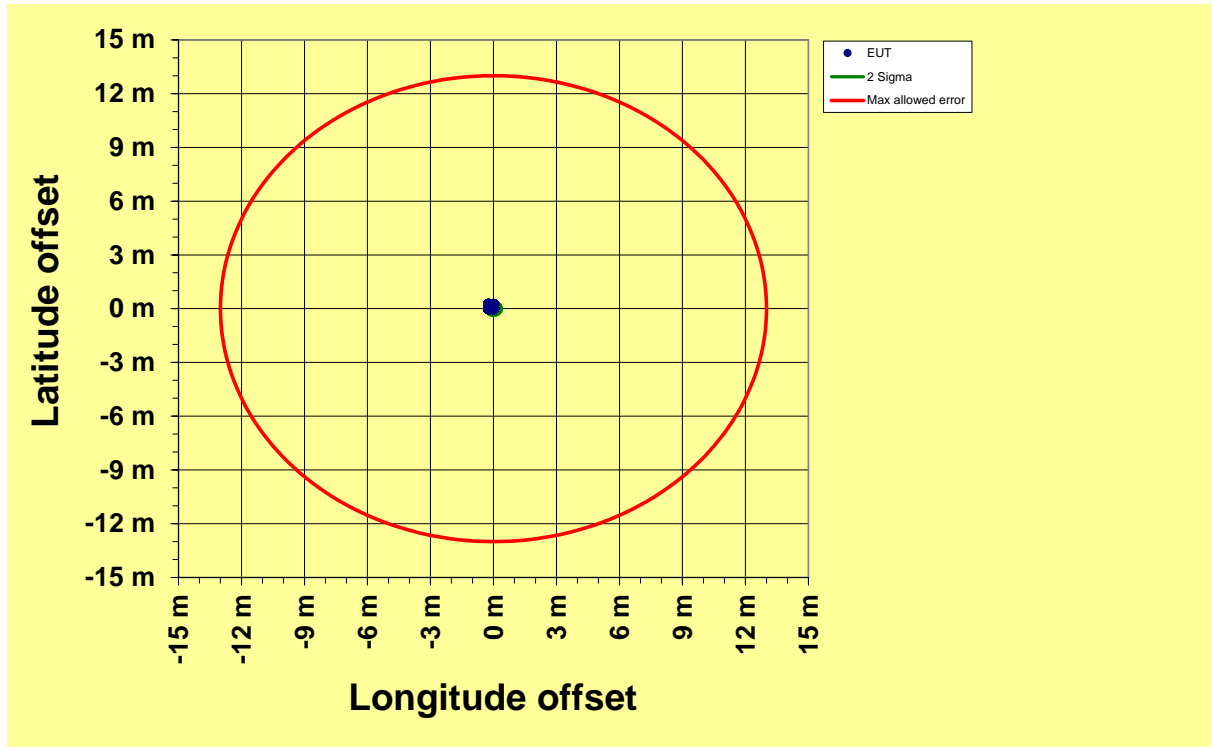
Pass/ fail criteria: All position offsets to be within +/-13m with respect to reference - **Passed**

Longitude offset



Pass/ fail criteria: All position offsets to be within +/-13m with respect to reference - **Passed**

Position offset



Pass/ fail criteria: All position offsets to be within +/-13m with respect to reference - **Passed**

B.4.4 § 5.6.5.4 Condition D – Brief interruption of power

EUT shall be isolated from power for 60s.
A performance check shall be carried out after 2 minutes of operation.

Conditions of test performed – Simulated signal

EUT was isolated from power for 60s.

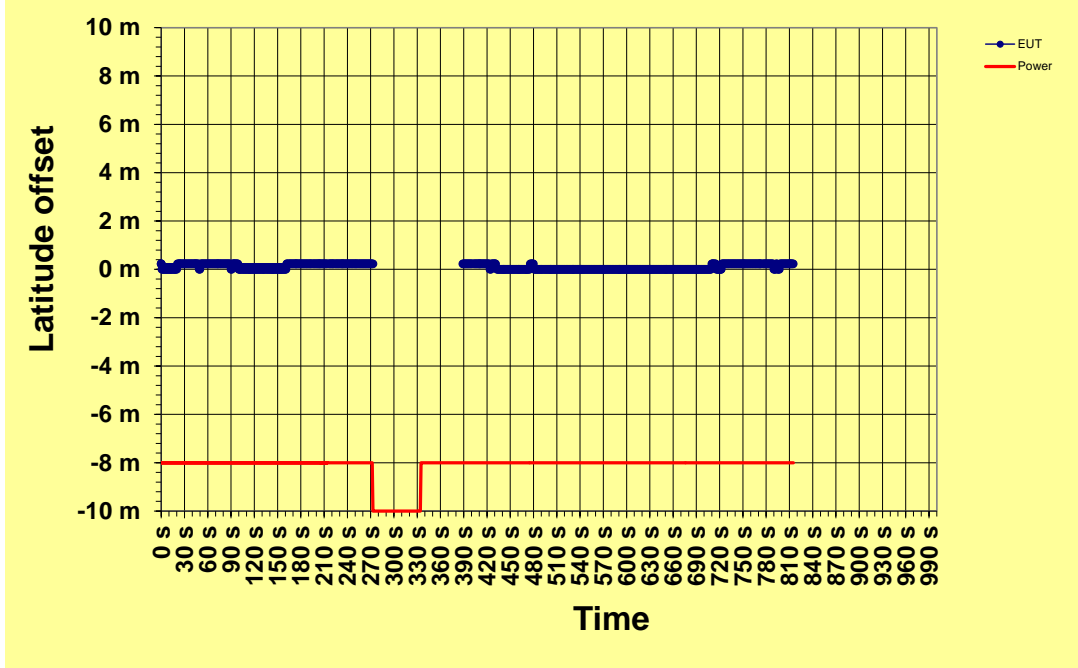
EUT locked on to GPS system within less than two minutes and provided a valid position fix.
A performance check was carried out after 2 minutes of operation.

2 Sigma value of performance check: 0.37m

Test result: Passed

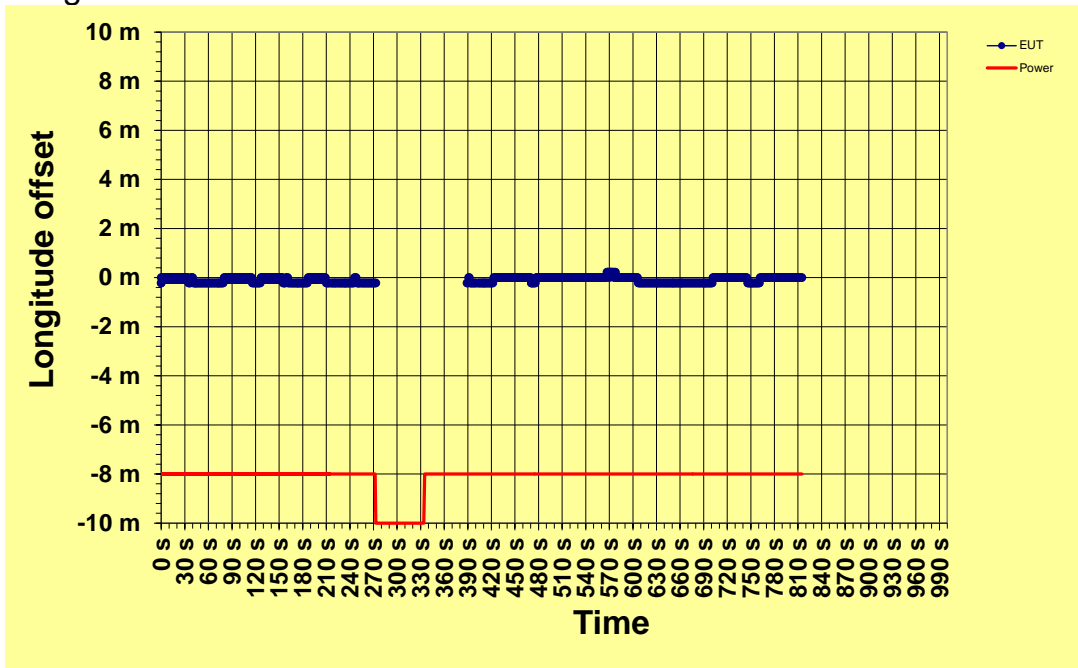
For details of validation of recorded data see the following pages.

Latitude offset



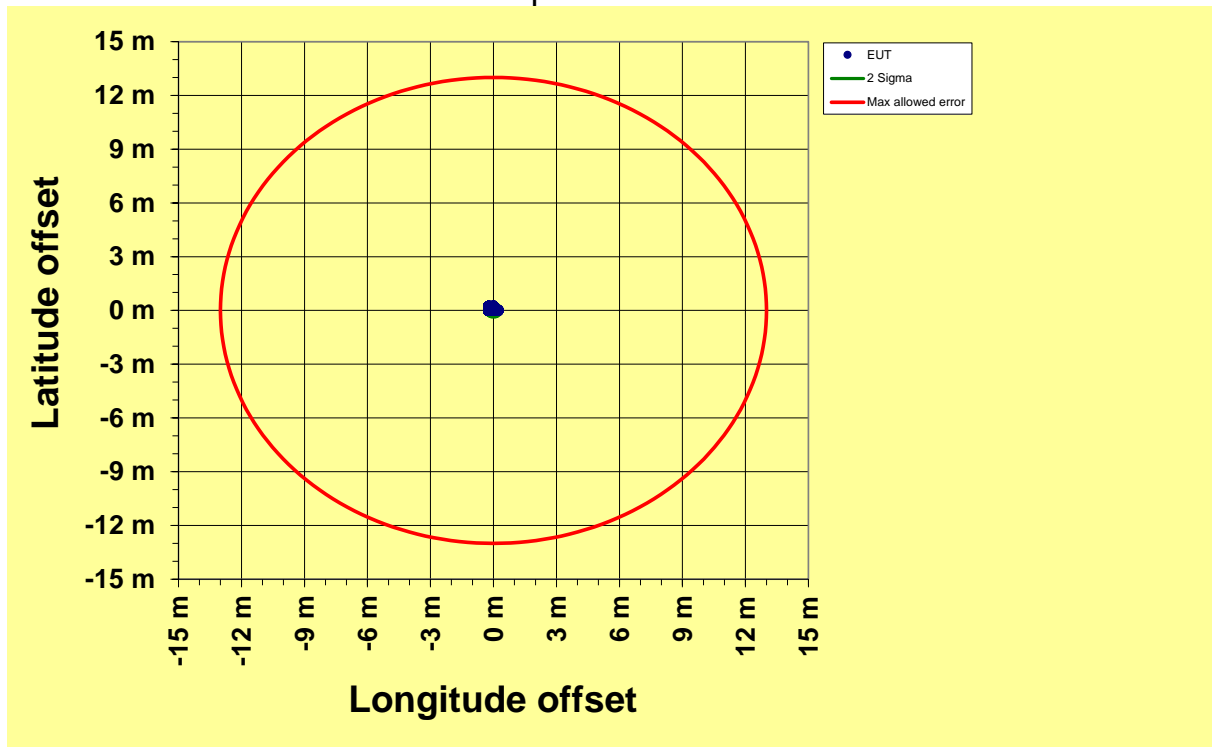
Pass/ fail criteria: All position offsets to be within +/-13m with respect to reference - **Passed**

Longitude offset



Pass/ fail criteria: All position offsets to be within +/-13m with respect to reference - **Passed**

Position offset after 2 Minutes of operation



Pass/ fail criteria: All position offsets to be within +/-13m with respect to reference - **Passed**

B.5 §5.6.9 Effects of specific interfering signals

Reference position

Made by FREIE UND HANSESTADT HAMBURG
 Vermessungsamt –VA311-

Description of point	geocentrically co-ordinates (WGS84)	geodetical geographical co-ordinates (WGS84)	Gauß-Krüger (Bessel)
North	x(m) 3740601.680	N 53° 32' 49".49049	x(m) 5935502.790
	y(m) 657439.492	E 9° 58' 6".10408	y(m) 564257.804
	z(m) 5107029.673	Height over Ellipsoid 95.900 m	Altitude above sea level 55.969 m
South	x(m) 3740618.106	N 53° 32' 48".81889	x(m) 5935482.027
	y(m) 657442.338	E 9° 58' 6".10189	y(m) 564258.046
	z(m) 5107017.296	Height over Ellipsoid 95.849 m	Altitude above sea level 55.917 m

Accuracy of survey = 0.02 m - last survey dated 2009-05-04

B.5.1 §5.6.9.1 L-Band interference

In a normal operating mode, using an appropriate signal source, the EUT shall be subjected to radiation of 3 W/m^2 at a frequency of 1636.5 MHz for 10 min. The signal shall be removed and a performance check shall be carried out.

Conditions of tests performed – Simulated signal

Frequency:	1636.5 MHz
Radiation:	3 W/m^2
Duration of test:	10 min

Test results

After removing the signal, the performance of the EUT was checked and found operating properly.

Position accuracy: 0.43m (2 sigma)

Test result: Passed

For details of validation of recorded data see the following pages.

B.5.2 §5.6.9.2 S- Band interference

In a normal operating mode, using an appropriate signal source, the EUT shall be subjected to radiation consisting of a burst of 10 pulses, each 1.0 to 1.5 μs long on a duty cycle of 1600:1 at a frequency in the range of 2.9 to 3.1 GHz at a power density of approximately 7.5 kW/m^2 . This condition shall be maintained for 10 min with the bursts of pulses repeated every 3 s. The signal shall be removed and a performance check shall be carried out.

Note:

The peak power density is 7.5 kW/m^2 to be measured at the EUT, this is approximately 4.7 W/m^2 average power at a fixed transmitting antenna.

The signal shall be removed and a successful performance check shall be carried out within 5 min.

Conditions of tests performed – Simulated signal

Frequency range:	3.0 GHz
Radiation:	4.7 W/m^2
Duration of test:	10 min

Test results

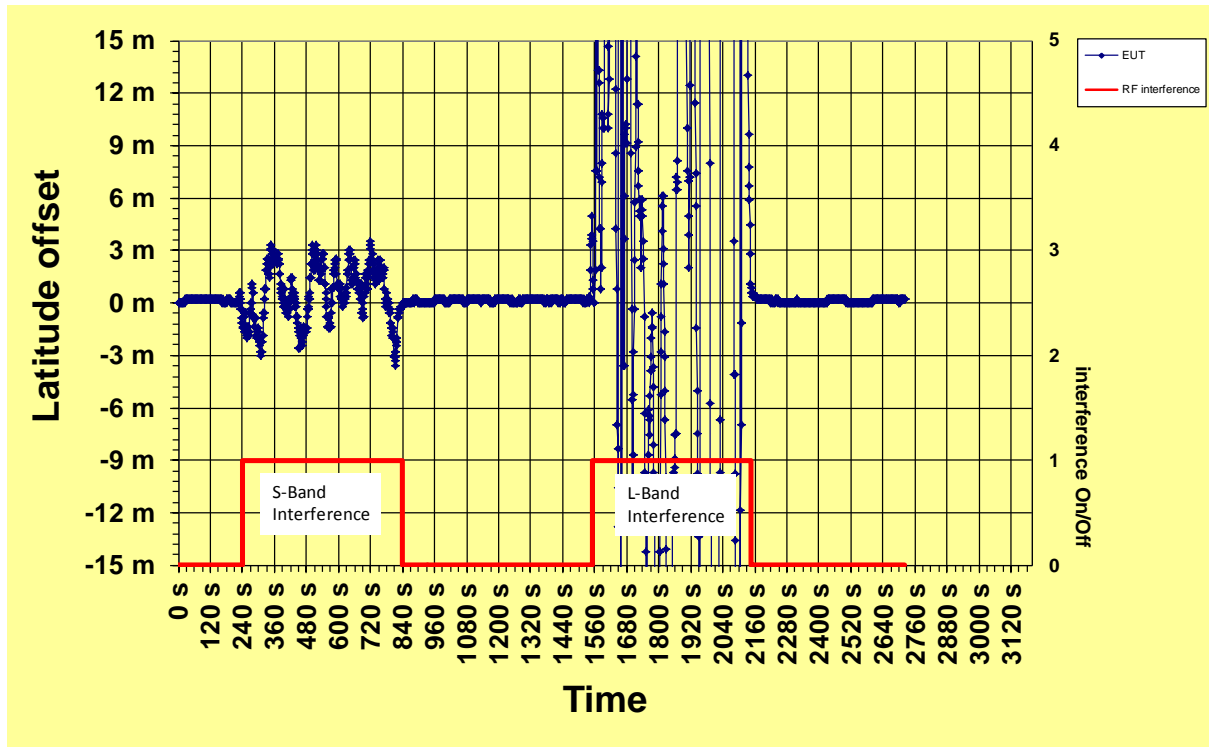
After removing the signal, the performance of the EUT was checked and found operating properly.

Position accuracy: 0.46m (2 sigma)

Test result: Passed

For details of validation of recorded data see the following pages.

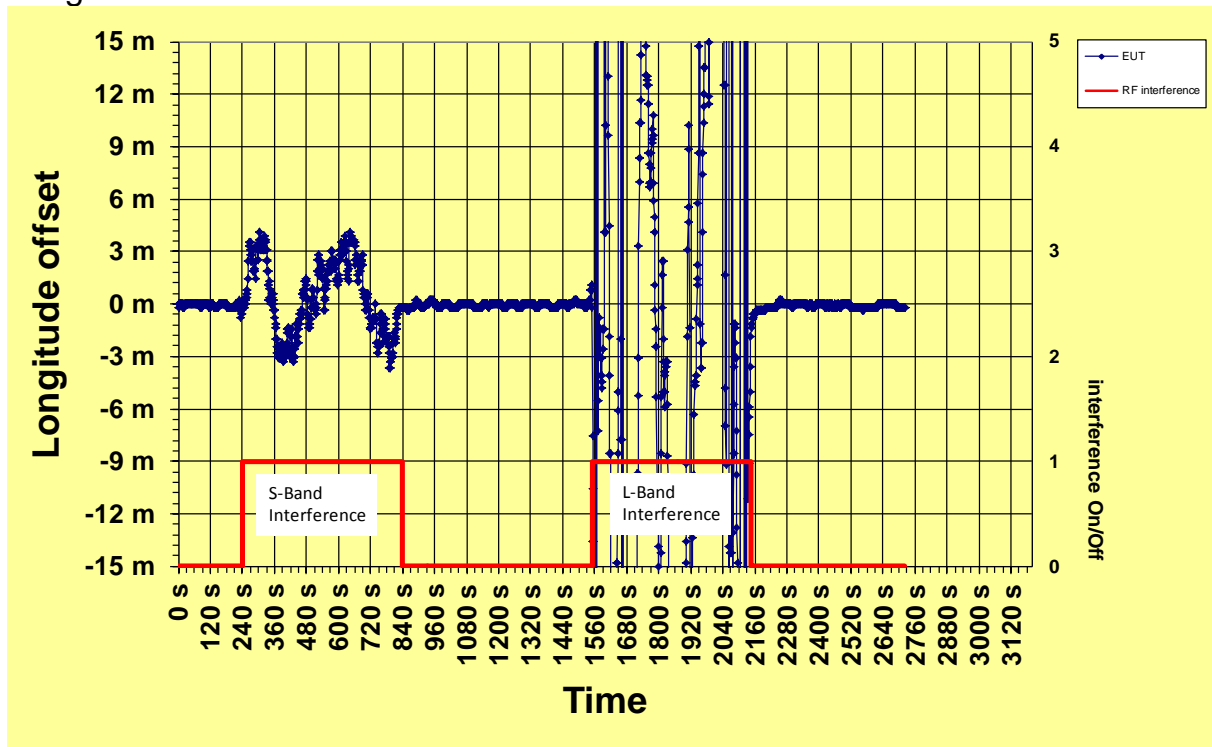
Latitude offset vs. time



Pass/ fail criteria:

All position offsets to be within +/-13m with respect to reference without interference – **Passed**

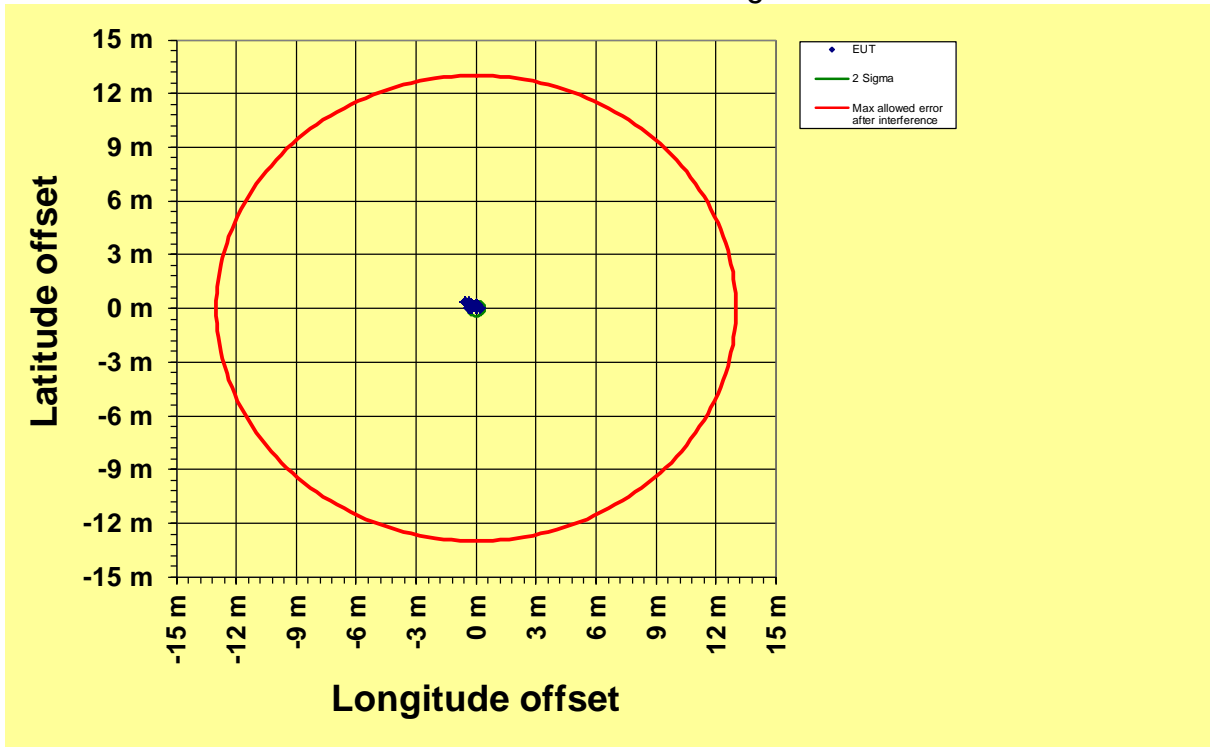
Longitude offset vs. time



Pass/ fail criteria:

All position offsets to be within +/-13m with respect to reference without interference – **Passed**

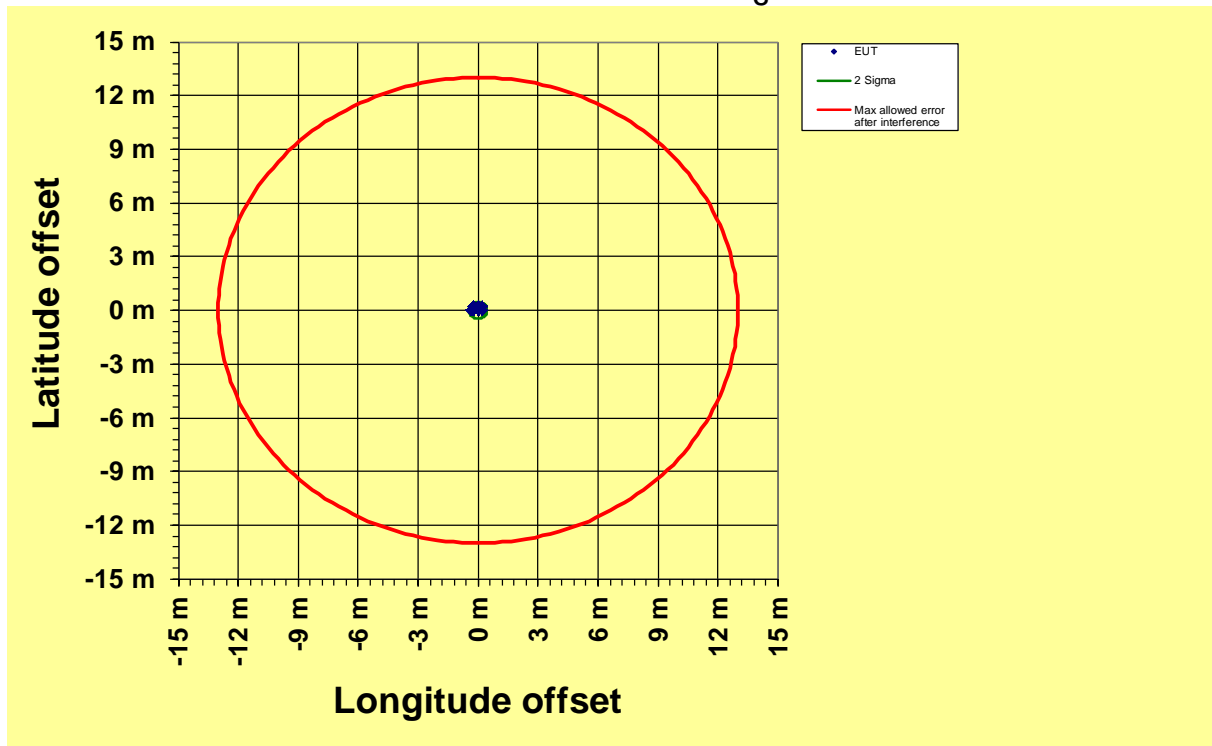
Position after L-Band interference switched off - §5.6.9.1



Pass/ fail criteria:

All position offsets to be within +/-13m with respect to reference without interference – **Passed**

Position after S-Band interference switched off - §5.6.9.2



Pass/ fail criteria:

All position offsets to be within +/-13m with respect to reference without interference – **Passed**

B.6 §5.6.10 Position update

B.6.1 § 5.6.10.1 Slow speed update rate

The EUT shall be placed upon a platform, moving in approximately a straight line, at a speed of 5knots \pm 1knots. The position output of the EUT shall be checked at intervals of 10s, over a period of 10min. The output position shall be observed to be updated on each occasion.

This test may be carried out by a simulator.

The minimum resolution of position, i.e. latitude and longitude shall be checked by observation during §5.6.10.1.

Record the IEC 61162 output of the EUT during this test and confirm that received positions at the end of each interval are in compliance with the real or simulated reference position.

Conditions of tests performed – Simulated signal

Test result: Passed

B.6.2 § 5.6.10.2 High speed update rate

The EUT shall be placed upon a platform, moving in approximately a straight line, at a speed of 50knots \pm 5knots. The position output of the EUT shall be checked at intervals of 10s, over a period of 10 min. The output position shall be observed to be updated on each occasion.

This test may be carried out by a simulator with a speed of 70knots at intervals of 0.5s.

The minimum resolution of position, i.e. latitude and longitude shall be checked by observation during 5.6.10.2.

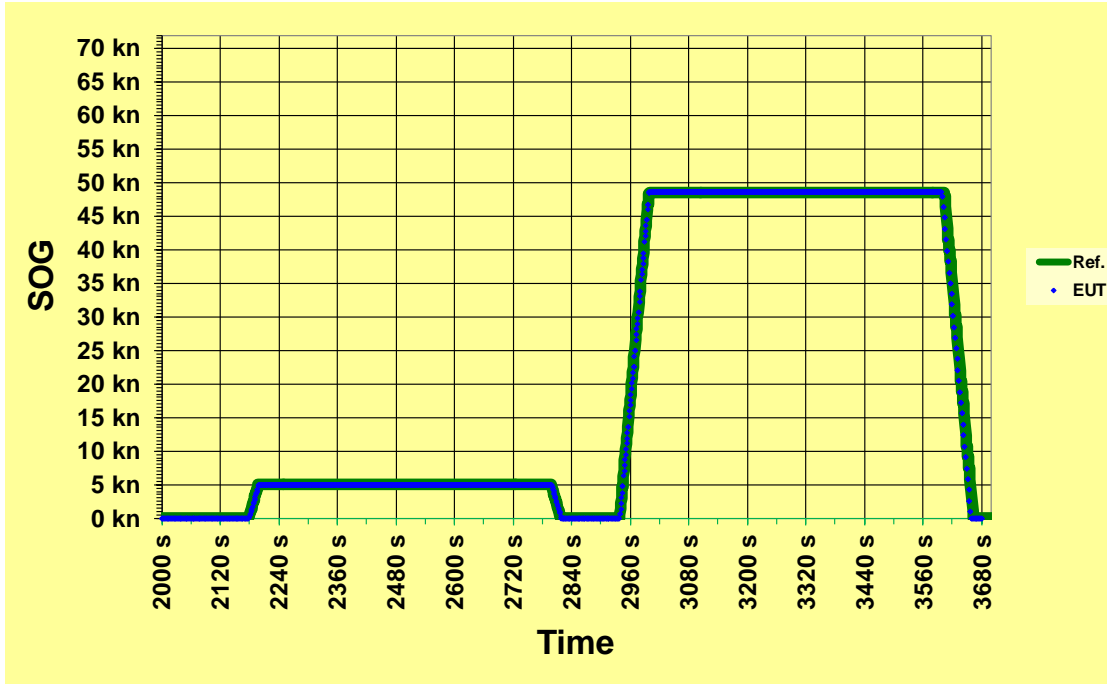
Record the IEC 61162 output of the EUT during this test and confirm that received positions at the end of each interval are in compliance with the real or simulated reference position.

Conditions of tests performed – Simulated signal

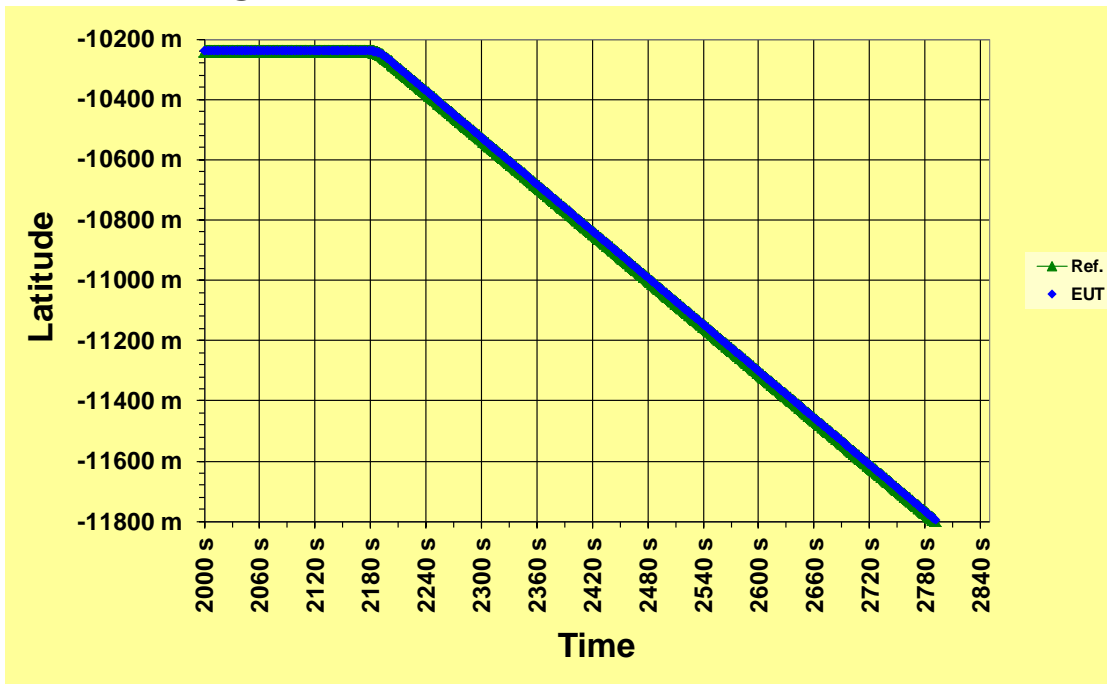
Test result: Passed

For details of validation of recorded data see the following pages.

SOG vs. time for 5knots and 48.6 knots



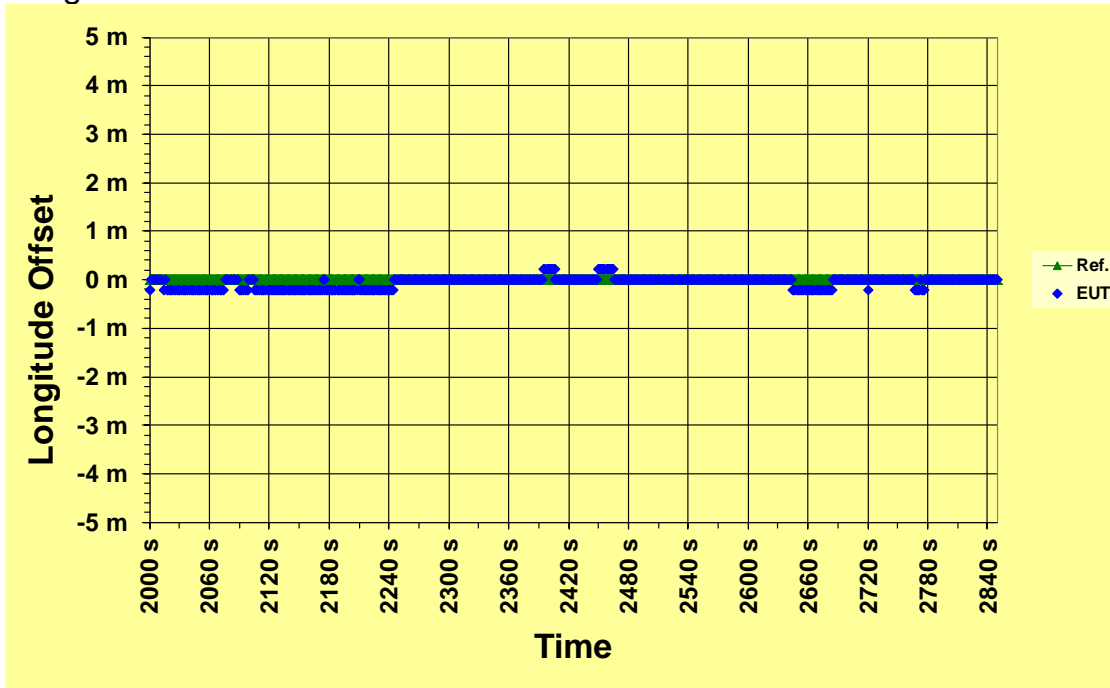
Latitude offset @ 5knots vs. time



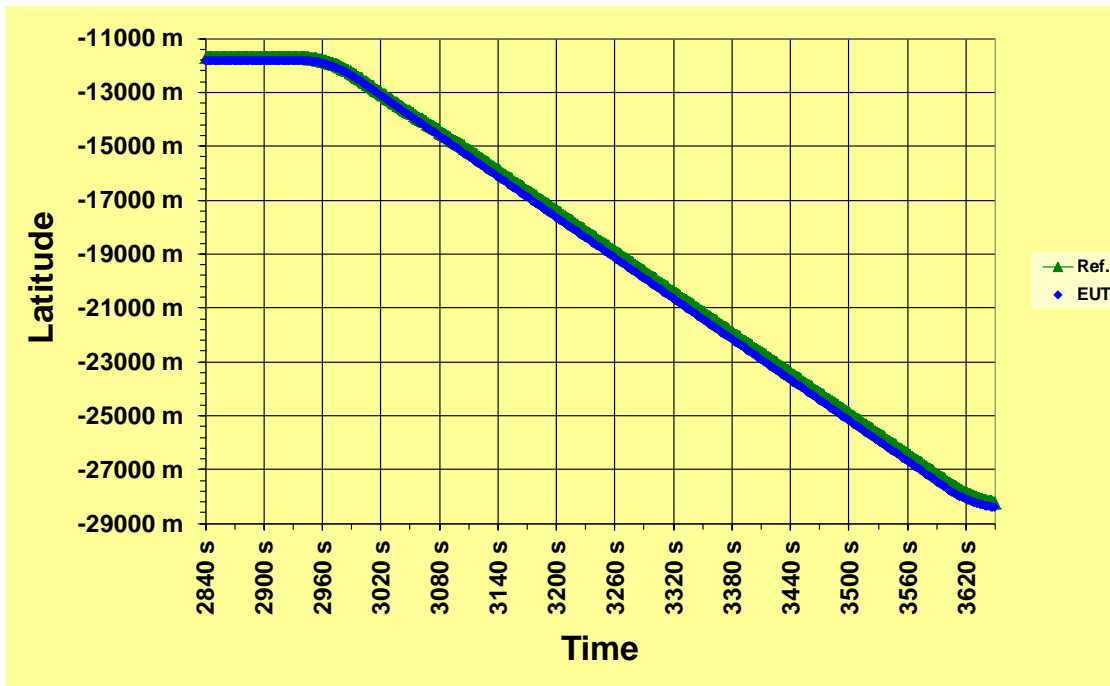
Pass/ fail criteria:

All position offsets to be within +/-13m with respect to reference and updated every second – **Passed**

Longitude offset@ 5knots



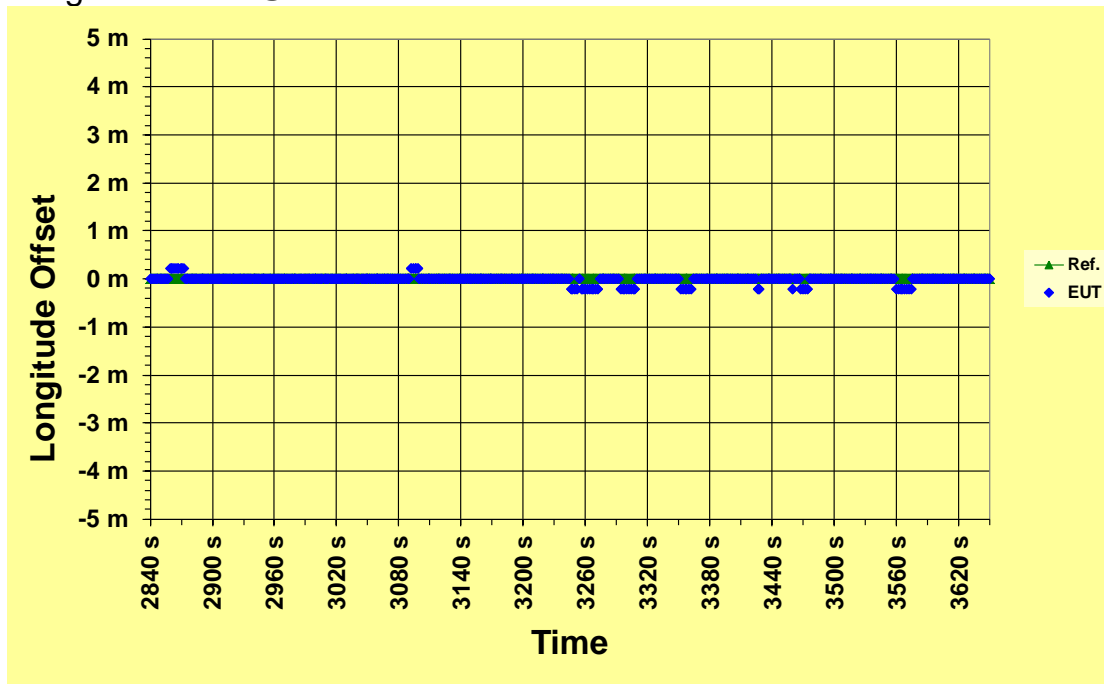
Latitude offset@ 48.6knots vs. time



Pass/ fail criteria:

All position offsets to be within +/-13m with respect to reference and updated every second – **Passed**

Longitude offset@ 48.6knots vs. time



B.7 §5.6.12 Failure warnings and status indication

B.7.1 Simulation scenarios for test of RAIM functionality

Reference position

Made by GPS-Simulator

	geocentrically co-ordinates (WGS84)
Latitude	0.00000000000 ° N
Longitude	0.00000000000 ° E

Accuracy of survey = ± 5 cm according to manufacturer documentation

RAIM scenario

Scenario: W224_1500_8_TO_4_SAT_RAMP_SHORT.SCEN;3

Start time: 20.12.2008, 15:00:00 Pseudorange - Ramp: 500 m
 4 min. Ramp up, 1 min. Hold, 4 min. Ramp down
 Sat.-Ids: 1, 2, 3, 11, 14, 16, 20, 31

8 Satellites		7 Satellites			6 Satellites			5 Satellites			4 Satellites															
00:00:00	00:15:00	00:19:00	00:20:00	00:24:00	00:27:00	00:28:00	00:32:00	00:33:00	00:37:00	00:40:00	00:41:00	00:45:00	00:46:00	00:50:00	00:53:00	00:54:00	00:58:00	00:59:00	01:03:00	01:06:00	01:07:00	01:11:00	01:12:00	01:16:00	01:19:00	
Sat. Off	Ramp up	Ramp hold	Ramp down	Ramp end	Sat. Off	Ramp up	Ramp hold	Ramp down	Ramp end	Sat. Off	Ramp up	Ramp hold	Ramp down	Ramp end	Sat. Off	Ramp up	Ramp hold	Ramp down	Ramp end	Sat. Off	Ramp up	Ramp hold	Ramp down	Ramp end	Ramp end	
13	14				14		2			2		1			1	31				31		20				
22																										
25																										
27																										
28																										

B.7.2 § 5.6.12.2 Test of integrity monitoring using RAIM

This test applies to 4.3.11.2.

For the purpose of testing of the RAIM functionality, it is recommended that means are provided for real-time display of the actual position error with reference to the simulated position.

§ 5.6.12.2.1 Testing of "safe" and "caution" status

The EUT shall be set up under simulated conditions, providing 8 "healthy" satellites available, acquired and tracked.

- a) Select an accuracy level of 100 m.
- b) Observe that
 - 1) RAIM is indicated as "in operation", and
 - 2) the "safe" status is indicated.
- c) Consecutively reduce the number of "healthy" satellites until the "caution" state is raised. Observe that
 - 1) RAIM is still indicated as "in operation", and
 - 2) that the status indication switched to "caution" within 10 s of the satellite change that caused it.
- d) Increase the number of "healthy" satellites until the RAIM state returns to "safe" state. Observe that
 - 1) RAIM is still indicated as "in operation", and
 - 2) the status indication switches to "safe" within 2 min of the satellite change that prompted it.

For each step of the above test sequence observe if the appropriate interface output is provided.

Repeat the above test sequence for a selected accuracy level of 10 m and, if provided for another accuracy level.

§ 5.6.12.2.2 Testing of "unsafe" status

The EUT shall be set up under simulated conditions, providing 8 "healthy" satellites available, acquired and tracked.

- a) Select an accuracy level of 100 m.
- b) Observe that
 - 1) RAIM is indicated as "in operation", and
 - 2) the "safe" status is indicated.
- c) Change the behaviour of at least 1 satellite by varying the satellite clocks with the result that the position accuracy gradually degrades until it will no longer be inside the selected accuracy level with 95 % confidence level. Observe that
 - 1) RAIM is still indicated as "in operation", and
 - 2) the status indication switches to "unsafe" within 10 s of the actual position error exceeding the selected accuracy level.
- d) Change the behaviour of the satellites back to regular behaviour with the result that the position accuracy will be again inside of the selected accuracy level with 95 % confidence level. Observe that
 - 1) RAIM is still indicated as "in operation", and
 - 2) the status indication switches to "safe" within 2 min.

For each step of the above test sequence observe if the appropriate interface output is provided.

Repeat the above test sequence for a selected accuracy level of 10 m and, if provided, for another accuracy level.

Conditions of tests performed

Tests performed by using a simulator.

Note:

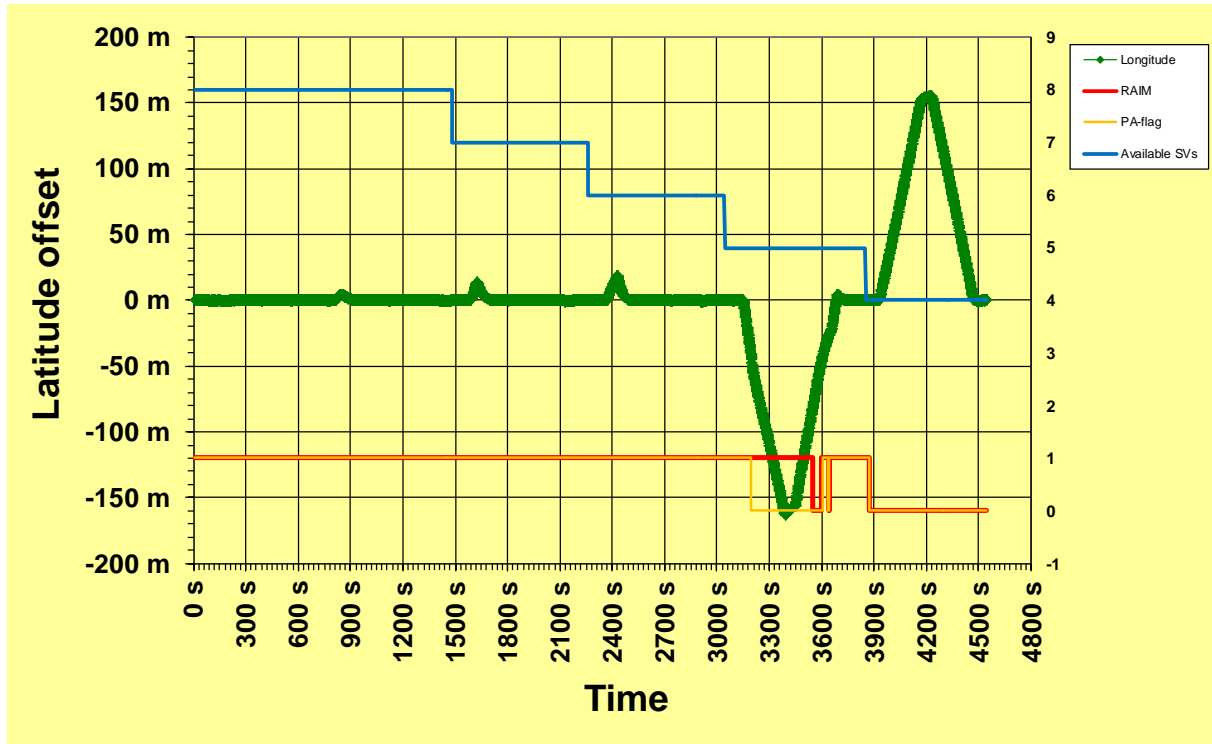
Evaluation of RAIM is done using the information provided by AIS- NMEA sentence
For AIS Transponders only the RAIM and PA flag are usable for RAIM status indication.

Test result: Not Passed

For details of validation of recorded data see the following pages.

B.7.3 RAIM test diagrams – 10Meter RAIM radius

Latitude offset vs. time



Note:

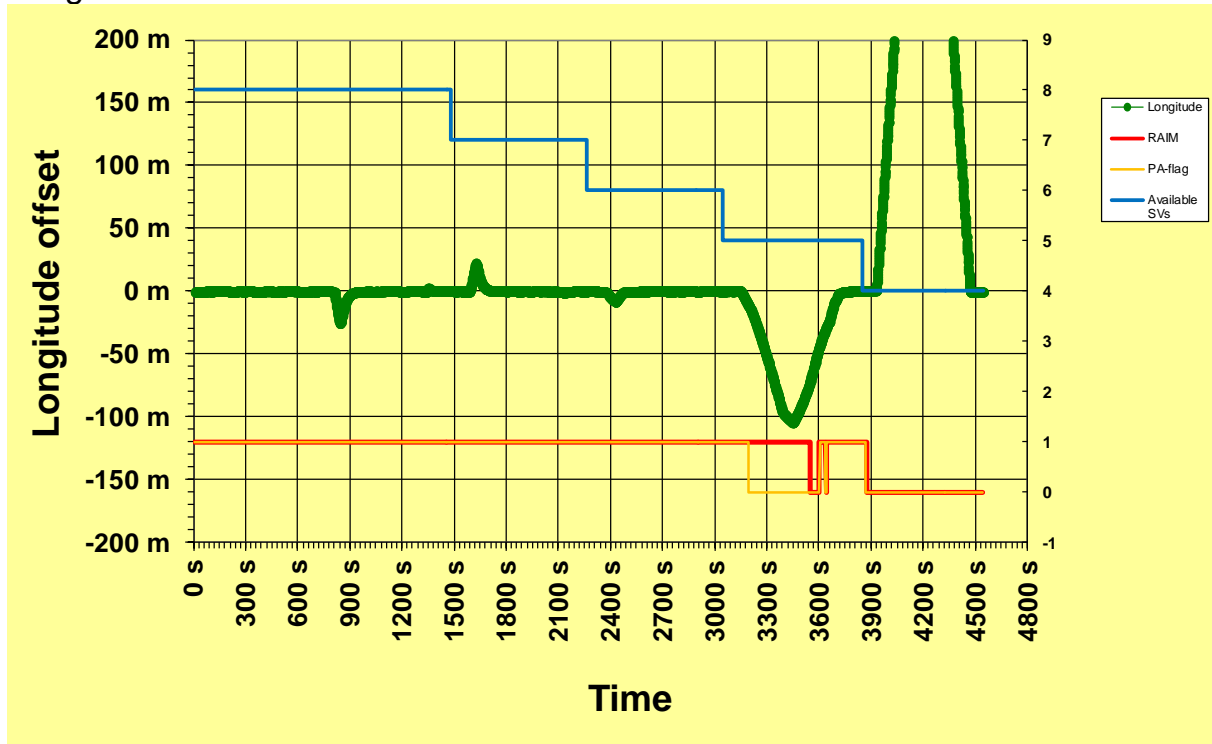
Evaluation of RAIM is done using the information provided by AIS- NMEA sentence
For AIS Transponders only the RAIM and PA flags are usable for RAIM status indication.

Pass/ fail criteria:

All position offsets to be within +/-10m (RAIM alert limit) with respect to reference position, otherwise indicated by RAIM flag and position accuracy flag after suitable alert raise time.

Result: Passed

Longitude offset vs. time



Note:

Evaluation of RAIM is done using the information provided by AIS- NMEA sentence
For AIS Transponders only the RAIM and PA flag are usable for RAIM status indication.

Pass/ fail criteria:

All position offsets to be within +/-10m (RAIM alert limit) with respect to reference position, otherwise indicated by RAIM flag and position accuracy flag after suitable alert raise time.

Result: Passed

B.8 § 5.6.13 Accuracy of COG and SOG

The EUT shall be set up on an appropriate mobile unit or simulator and all outputs indicating course over ground shall be monitored.

At a constant forward direction, the forward speed shall be within 0 knots to 1 knot. Ten seconds after being in the range, measurements shall be made for a duration of 2 min. This cycle shall be repeated for all speed ranges of the table above.

Table – Accuracy of COG

Speed range (knots)	Tested value	Accuracy of COG output to user
0 to ≤1 knot	(0.5 kn)	Unreliable or nor available
>1 to ≤17 knots	(15 kn)	± 3°
> 17 knots	(24 kn)	± 1°

Required results

The test results shall be observed the display and the approved interface.

For SOG tests, no reading of the speed indicator shall differ from the constant speed being applied at the time by more than 2 % of that speed or 0.2knots, whichever is the greater.

For COG tests, the difference between the reference direction and measured course over ground of in each test cycle shall not exceed the limits in the table.

Validity of COG and SOG information

The quality indicator of the GGA and the VTG message of IEC 61162 shall be used for interpretation of validity of COG and SOG.

Conditions of tests performed – Simulated signal

Test result: Passed

For details of validation of recorded data see the following pages.

Validity of COG and SOG information

Method of testing

Check digital interface with IEC 61162. With the EUT normally operating, preclude invalid position by reducing the number of received satellites. Investigate the content of the resultant GGA and VTG.

Required result

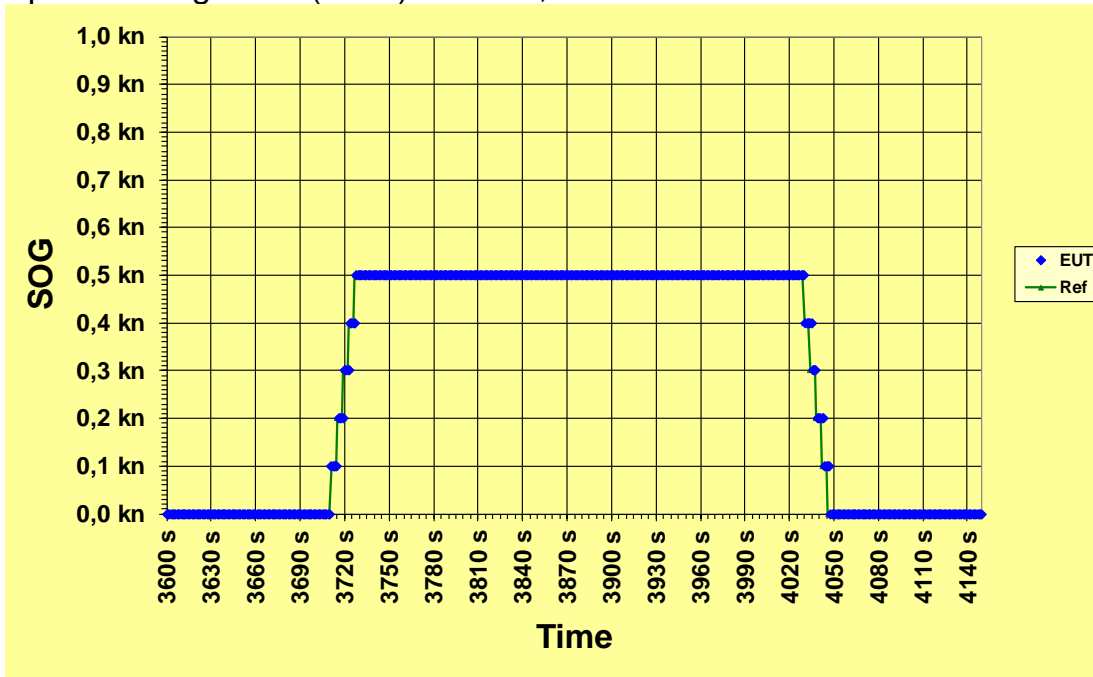
Observe that the quality indicators of GGA and VTG messages of IEC 61162 turn to invalid. Observe that the COG and SOG information contained in VTG message of IEC 61162 is replaced by null fields.

Test Results for Validity of COG and SOG information

Test result: Passed

For AIS Transponders COG and SOG are transmitted via the coded AIS sentences – Evaluation of AIS data showed validity of sentence usage.

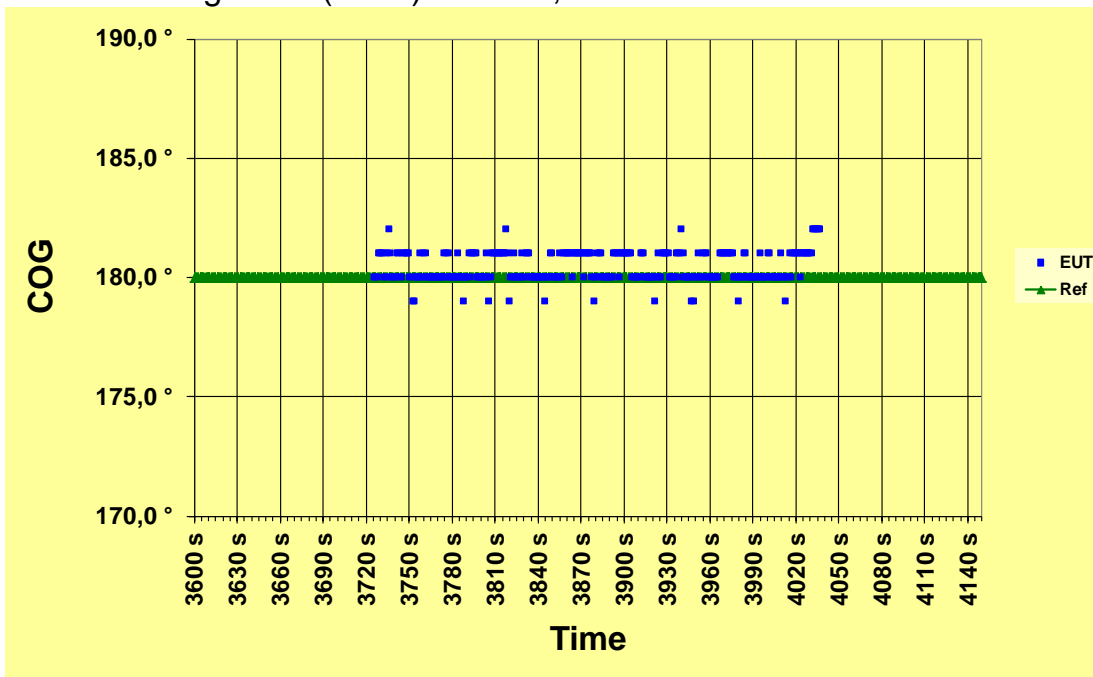
Speed over ground (SOG) vs. time, 0.5 Knots



Pass/ fail criteria:

All SOG measurements to be within 0.2knots with respect to reference SOG – **Passed**

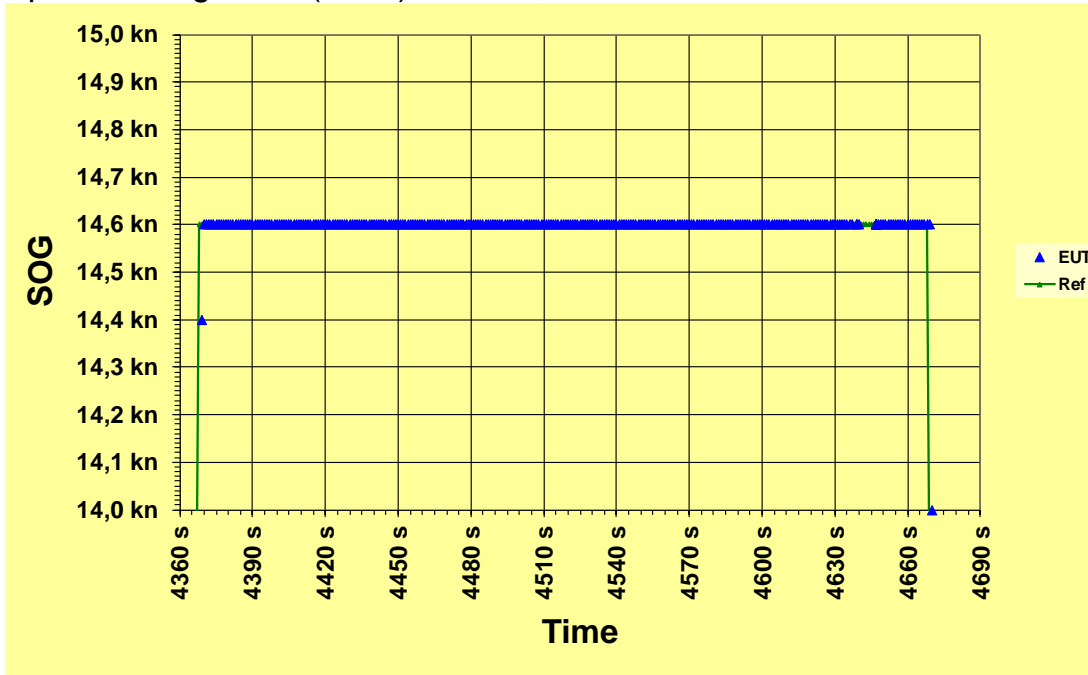
Course over ground (COG) vs. time, 0.5Knots



Pass/ fail criteria:

Reliable COG unavailable for less than 0.5knots

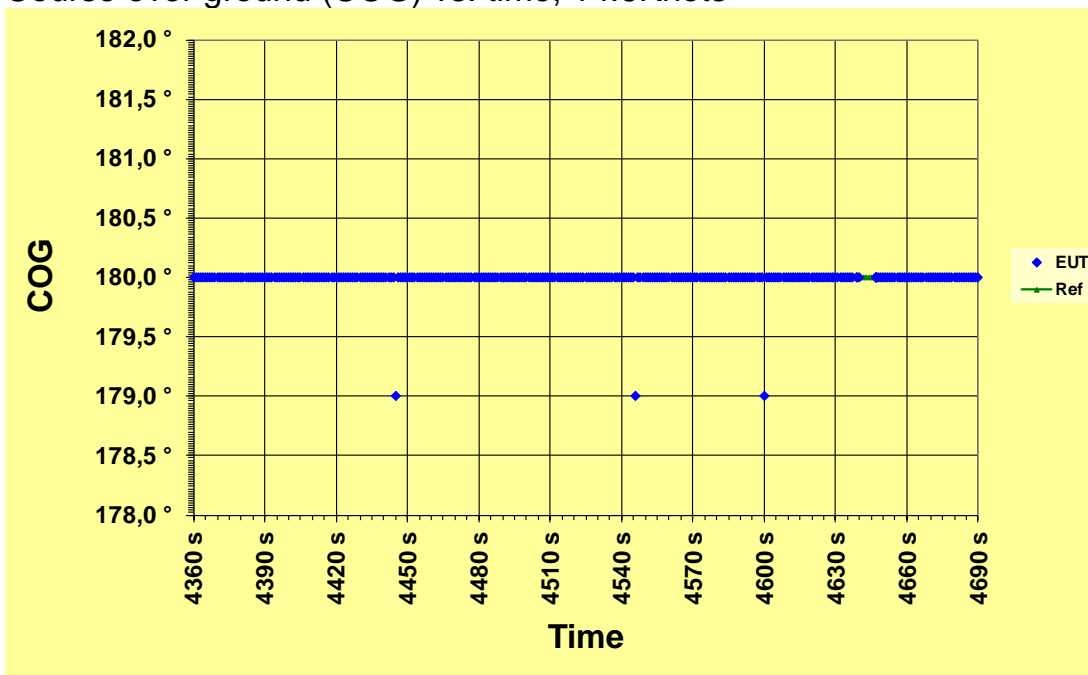
Speed over ground (SOG) vs. time, 14.6Knots



Pass/ fail criteria:

All SOG measurements to be within 0.2knots with respect to reference SOG – **Passed**

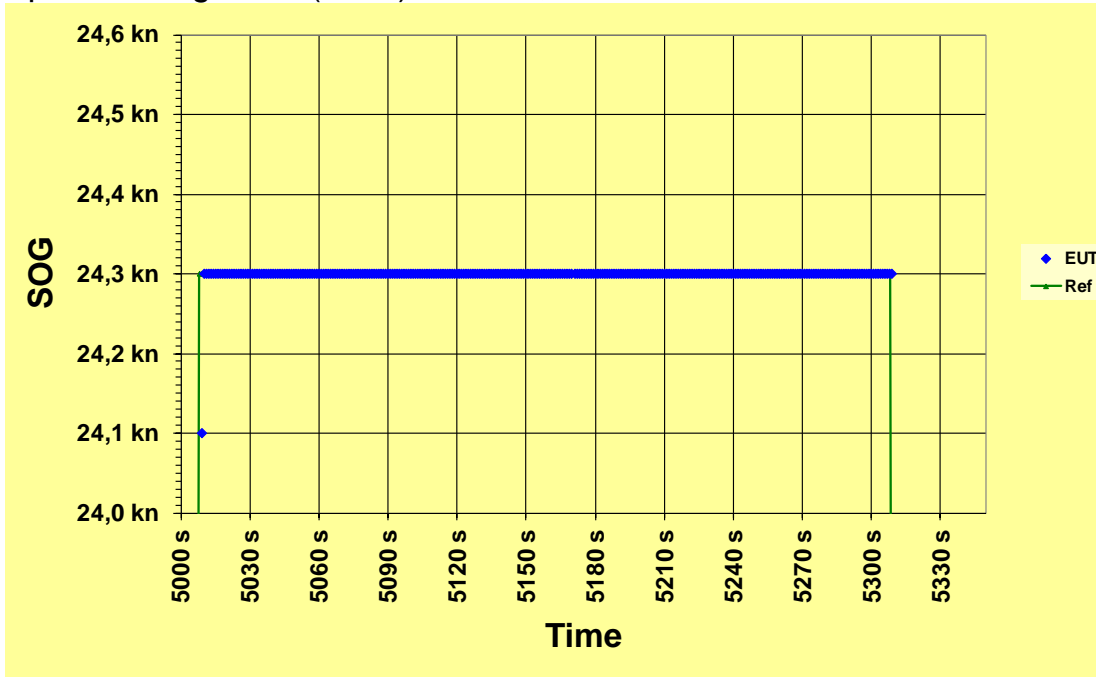
Course over ground (COG) vs. time, 14.6Knots



Pass/ fail criteria:

All COG measurements to be within +/-3° with respect to reference COG – **Passed**

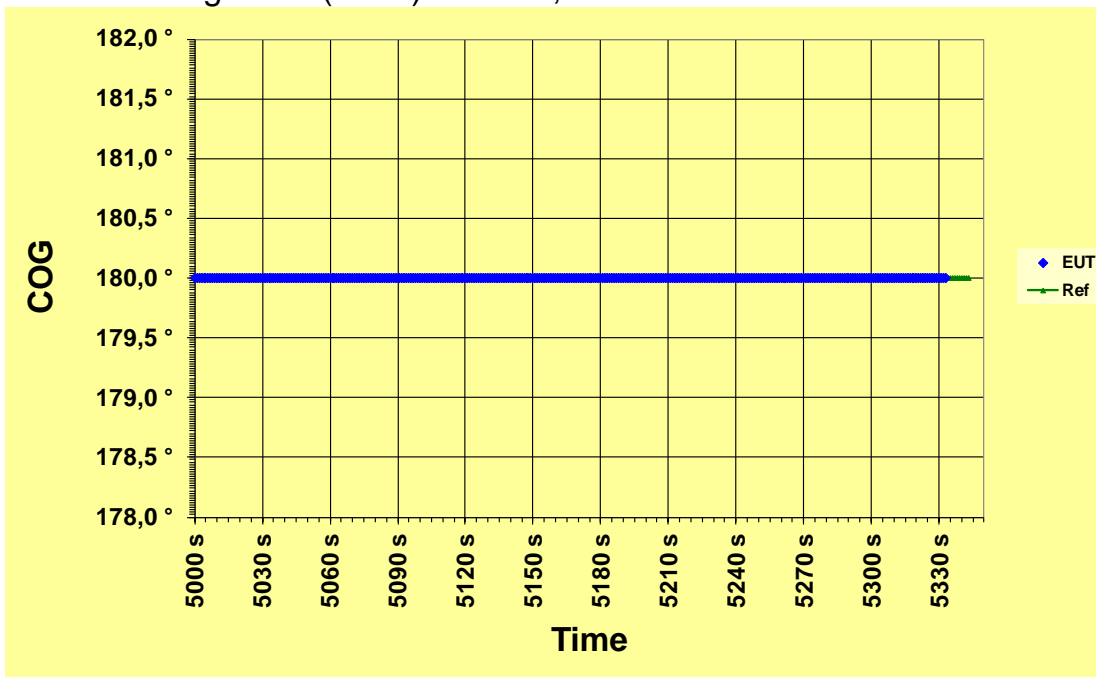
Speed over ground (SOG) vs. time, 24.3Knots



Pass/ fail criteria:

All SOG measurements to be within 0.2knots with respect to reference SOG – **Passed**

Course over ground (COG) vs. time, 24.3Knots



Pass/ fail criteria:

All COG measurements to be within +/-1° with respect to reference COG – **Passed**

B.9 § 5.7.2 Testing interference

B.9.1 § 5.7.2.4.1 Interference conditions

Interference conditions, including broadband noise centred at 1575.42 MHz, continuous wave interference (CWI), and pulsed interference shall be simulated. For the pulsed interference tests, a pulse-modulated carrier (CW) with peak carrier level of -20 dBm and duty factor of 10 % shall be used. The interference values are shown in the three tables below.

Broadband interference values		
Noise bandwidth (MHz)	Requested total RMS power (dBm)	Applied interference
1	-110.5	-110.5 dBm at 1MHz

Pulsed interference values		
Frequency (MHz)	Pulse width (ms)	Applied interference
1575.42	1	1 ms at 1575.42 MHz

Continuous wave interference (CWI) values		
Frequency (MHz)	Requested power (dBm)	Applied interference
1575.42	-120.5	-120 dBm at 1575.42 MHz
1626.0	+8.0	-80 dBm at 1596.0 MHz

§ 5.7.2.4.2 Test procedures

- a) The equipment under test is subjected to one of the interference sources.
- b) The simulator scenario shall be engaged and the satellite signals turned on.
- c) The equipment under test shall be powered and initialised.
- d) While the EUT is providing position solutions, the interference shall be applied to the equipment under test, and the level of the interference shall be adjusted to the required value.
- e) When steady-state accuracy is reached, record a minimum of 100 position and HDOP value as reported by the EUT at a rate of one sample every 2 min.
- f) Repeat this cycle for any remaining interference source.

§ 5.7.2.4.3 Required results

Pass/fail determination

If the EUT reports a position with a normalised error greater than 10 m or fails to report a position in more than 5 % of the samples, a test failure is declared.

Remark: Due to normalizing of the position error in reference to an HDOP of 4, values for normalized error can show wide ranges in case of measured HDOP values around 1.0 or even below. Test passed, if the calculated position is well within GPS position accuracy level defined by IEC61108-1 Ed. 2 (for standard GPS= 13.0m).

Conditions of tests performed – Simulated signal

B.9.2 §5.7.2.4.3 Required results

Broadband noise (-110.5dBm @ 1MHz bandwidth)

2 nd March 09:45	Test start
09:55:00	Interference start
10:05:00	Interference stop
2 nd March 10:55	Test end

Pulsed interference (-20 dBm at 1575.42 MHz with 1 ms pulse width)

2 nd March 09:45	Test start
10:10.00	Interference start
10:20:00	Interference stop
2 nd March 10:55	Test end

Continuous wave interference (-120.5 dBm at 1575.42 MHz)

2 nd March 09:45	Test start
10:25:00	Interference start
10:35:00	Interference stop
2 nd March 10:55	Test end

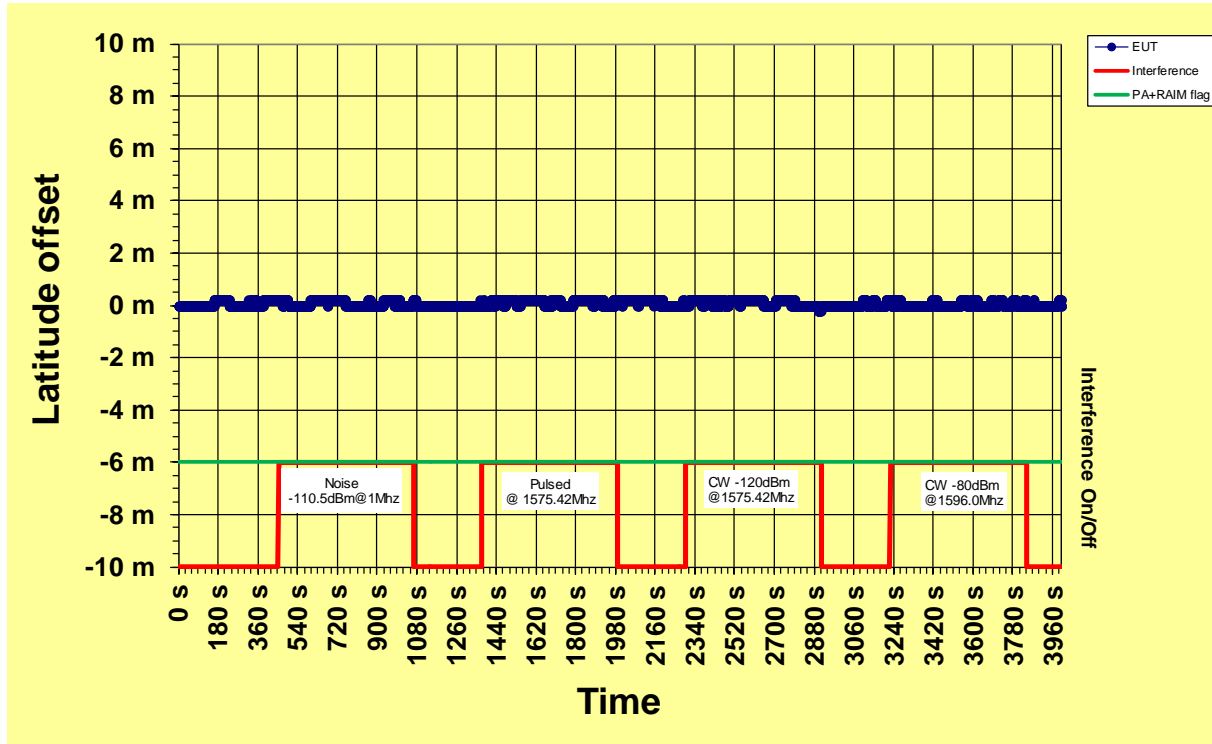
Continuous wave interference (-80 dBm at 1596 MHz)

2 nd March 09:45	Test start
10:40:00	Interference start
10:50:00	Interference stop
2 nd March 10:55	Test end

Test result: Test passed

For details of validation of recorded data see the following pages.

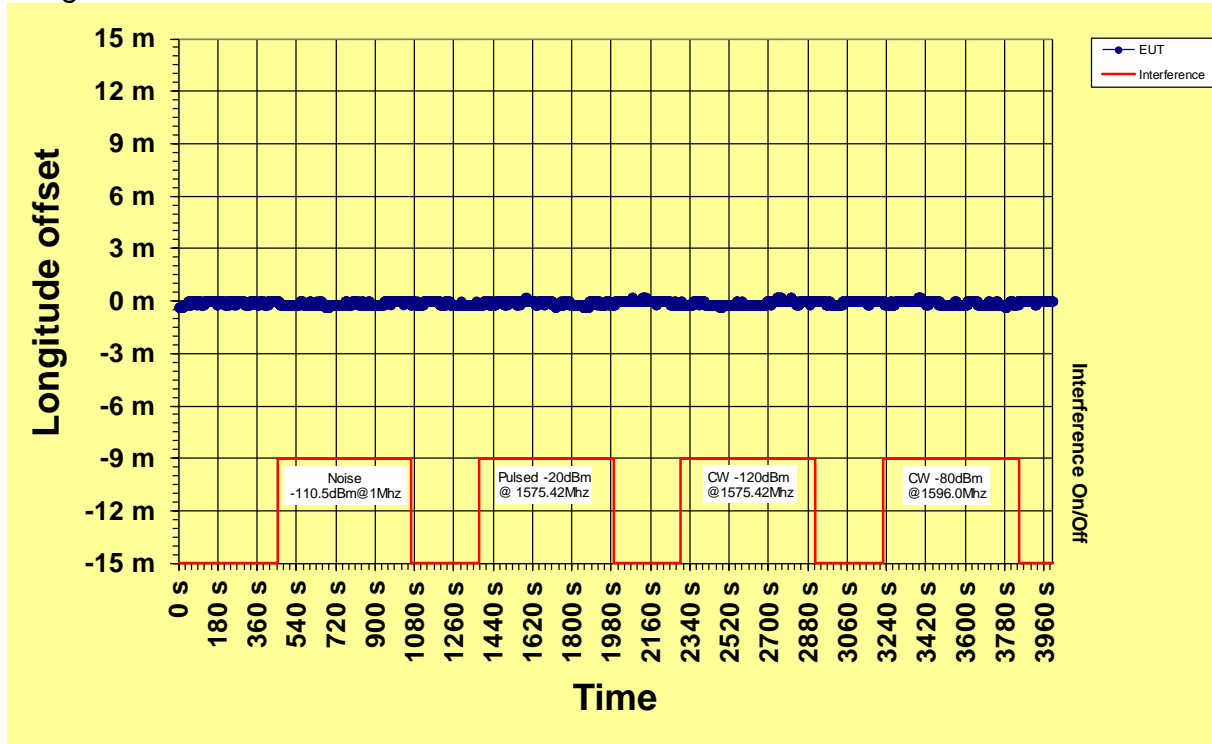
Latitude offset vs. time



Pass/ fail criteria:

95% of all position offsets to be within +/-13m with respect to reference position – **Passed**

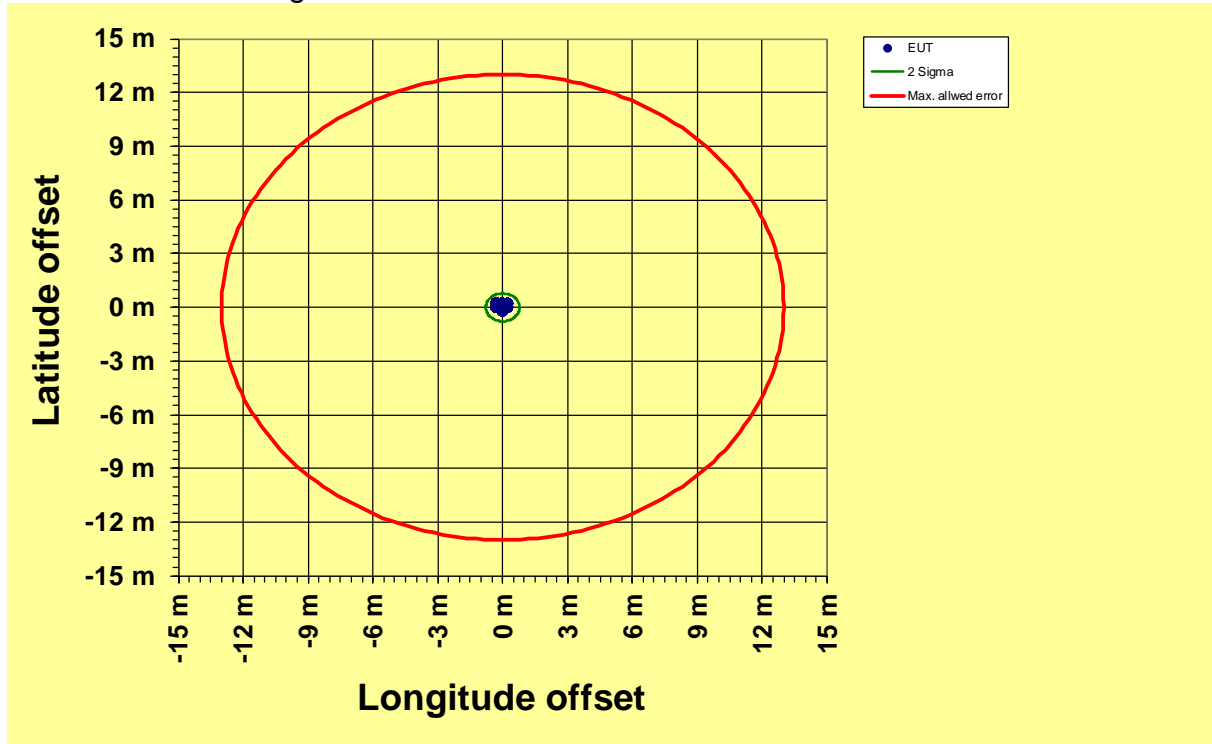
Longitude offset vs. time



Pass/ fail criteria:

95% of all position offsets to be within +/-13m with respect to reference position – **Passed**

Position offset during all interference scenarios



Pass/ fail criteria:

95% of all position offsets to be within +/-13m with respect to reference position – **Passed**

B.10 §5.7.2.5 Reacquisition test

Method of test

The reacquisition test is designed to simulate a temporary loss of signal, such as Passing under a bridge. To determine the re-acquisition pass/fail criteria, consider a single trial where the EUT provides a valid position fix that is within required accuracy at 30 s from restoration of the satellite signals, and maintains a tracking status for at least the next 60 s. This unit is considered to have passed one trial.

§ 5.7.2.5.1 Re-acquisition test procedures

§ 5.7.2.5.2 Interference conditions

The interference condition to be tested is shown below. This is a broadband noise Value centred at 1575.42 MHz.

Noise bandwidth (MHz)	Total RMS power (dBm)
1	-110.5

§ 5.7.2.6 Re-acquisition scenarios

§ 5.7.2.6.1 Test procedures

- a) The equipment under test is subjected to the broadband interference source.
- b) The simulator scenario shall be engaged and the satellite signals turned on.
- c) The equipment under test shall be powered and initialised.
- d) The EUT shall be allowed to reach steady-state accuracy before the satellites are to be switched off.
- e) The simulator RF output shall be removed for 30 s.
- f) The simulator RF output shall be restored to the EUT.
- g) After 30 s record a position and HDOP value as reported by the EUT. If after 30 s, no position report has been sent from the receiver, record a trial failure and go to step i).
- h) Ensure that the receiver continues position reporting for the next 60 s.
- i) Go to Step d) and repeat as required. (note that if the simulator scenario is reset, some receivers may require purging of all previous data to enable proper operation. This is due to the persistence of time data in the receiver and the inability of the receiver's software to deal with a backward transition in time).

B.10.1 § 5.7.2.6.2 Required results

Pass/fail determination

A failure by the EUT to provide a position output after 30 s, reporting a position with normalised error greater than 10 m, or failing to continue position reporting for 60 s after sampling indicates a failure mode, and results in declaring a trial failure. To determine the reacquisition time pass/fail criteria, the test disposition table shall be used.

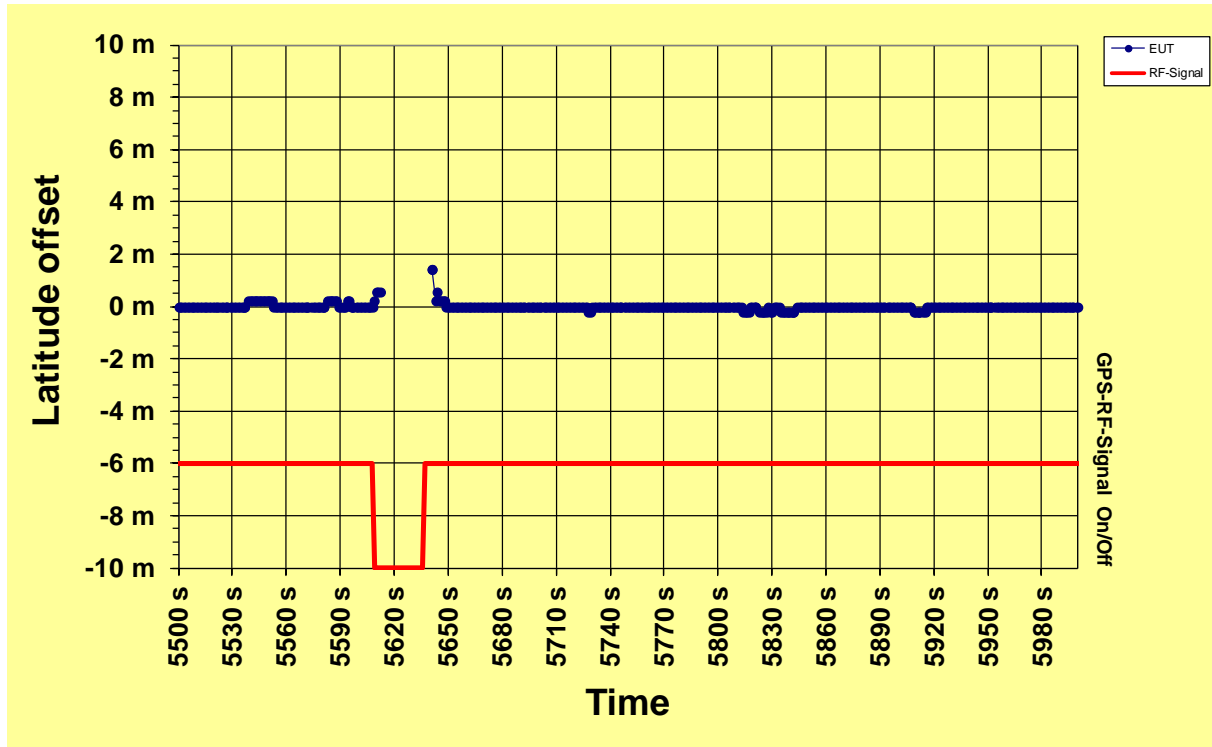
Conditions of tests performed – Simulated signal

2 nd March 10:55	Test start
11:21:30	Satellite signal shielded
11:22:00	shield removed
2 nd March 11:35	Test end

Test result: Passed

For details of validation of recorded data see the following pages.

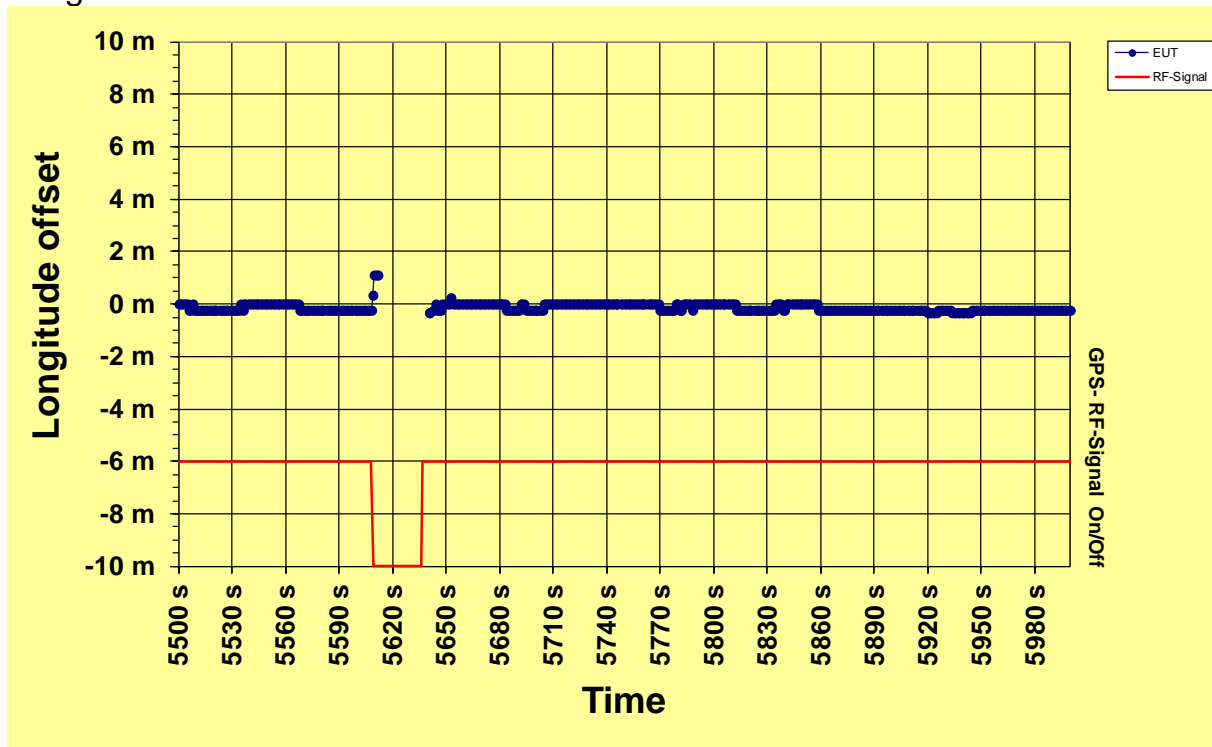
Latitude offset vs. time



Pass/ fail criteria:

95% of all position offsets to be within +/-13m with respect to reference, 30 Seconds after rf- signal has been restored – **Passed**

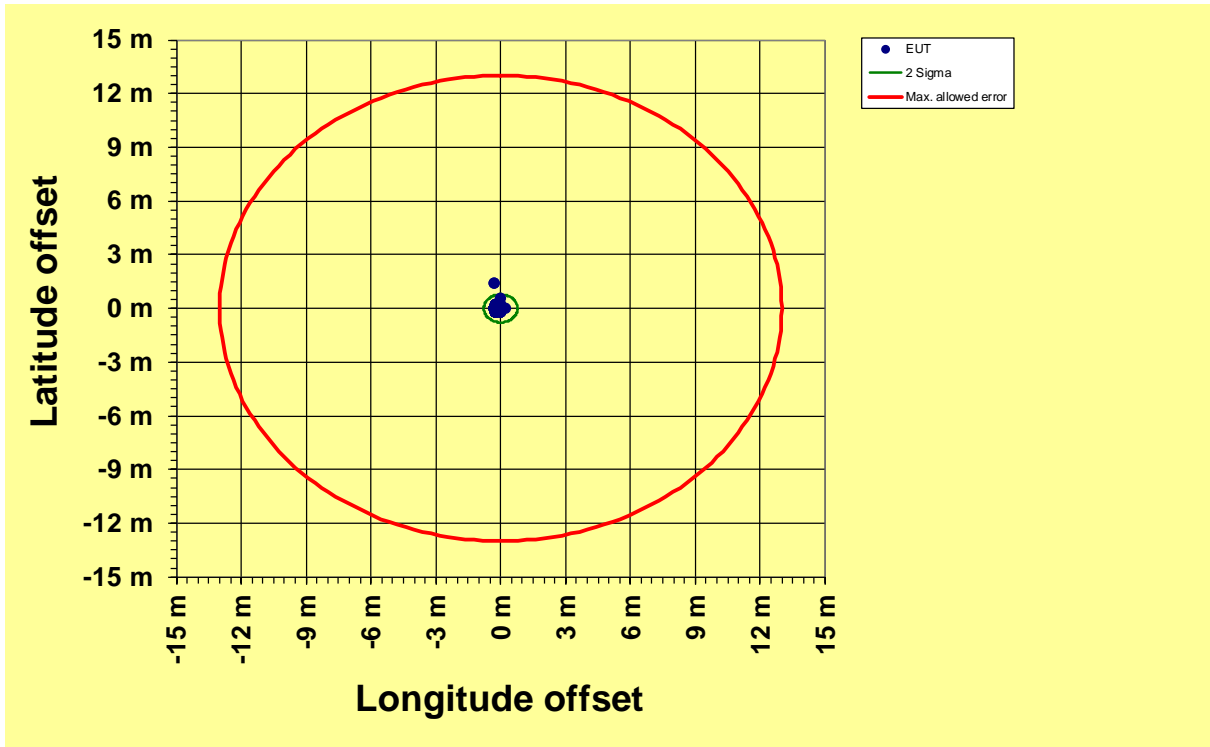
Longitude offset vs. time



Pass/ fail criteria:

95% of all position offsets to be within +/-13m with respect to reference, 30 Seconds after rf- signal has been restored – **Passed**

Position offset after time limit of 30seconds

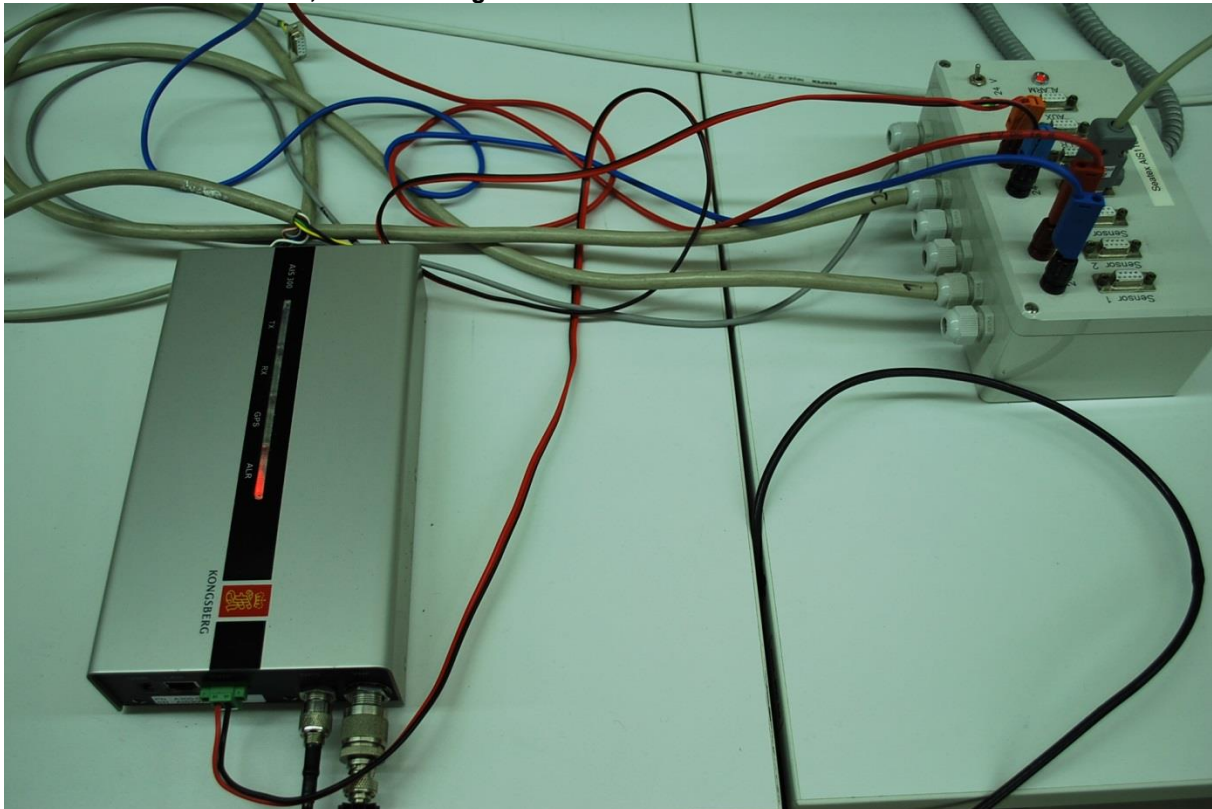


Pass/ fail criteria:

95% of all position offsets to be within +/-13m with respect to reference, 30 Seconds after rf- signal has been restored – **Passed**

Annex C Photos of equipment under test

EUT – Main unit at testside, BSH Hamburg



EUT- ID tag



EUT antenna inside RF-Chamber

