

APPENDIX B
PRELIMINARY INSTRUCTION BOOK
16

TS0046-2



INSTALLATION MANUAL 006-0191-00

REV. 0, DECEMBER, 1980

PRELIMINARY

section i general information

1.1 introduction

This manual contains information relative to the physical, mechanical, and electrical characteristics as well as installation procedures and operational characteristics for the KWX 56 Weather Radar System.

1.2 description of equipment

The KWX 56 TSO'd X-band Color Weather Radar System features a 160 nautical mile weather avoidance range plus the weather penetration advantages of X-band. The KWX 56 system is composed of the KI 244 Indicator and KA 126 Antenna/Receiver/Transmitter.

Digital storage of the weather picture and TV-like presentation give the KWX 56 the advantages of a bright, non-fading display with excellent contrast in all lighting conditions. Weather systems are displayed as 3 colors depicting rainfall intensity overlaid with range rings. Bearing marks at dead ahead and 20 degrees on either side aid the pilot in judging the bearing of storms and necessary heading changes.

The flat-plate antenna is fully stabilized to compensate for aircraft pitch and roll.

Display ranges 10, 20, 40, 80 and 160 nautical miles are selected by a rotary switch. The range of each range ring is annunciated with numeric characters.

The function switch provides the following modes of operation: OFF, STandBy, TEST, WX and MAP. In StandBy mode the display and transmitter circuits are disabled with the magnetron heater remaining on. The TEST mode activiates all circuitry including the transmitter and also puts a pattern on the display to verify all weather colors. The WX mode is the normal weather mode with green for light precipitation, yellow for moderate, and red for heavy. The MAP mode is used for terrain mapping. IN this mode the riable gain control is functional. The gain is adjustable from the automatic setting down to -16dB. display colors are changed in the MAP mode as follows green to blue, yellow stays the same, and red magenta.

Indicator brightness is adjustable over a wide range to accommodate varying ambient light conditions while automaticlly maintaining equal brightness between the three display colors.

A unique feature called Extended Sensitivity Time Control or Extended STC increases the displayed intensity of storms outside the normal STC range to more accurately represent the storm intensity. Extended STC relates the storm intensity to its distance and assigns a corresponding color. As a result, the display presents a more accurate picture of storm intensity.

The KWX 56 Weather Radar uses the KI 244 Indicator which features a 5-inch rectangular CRT, and uses 4.7 inches in the aircraft radio stack.

The KA 126 is equipped with a 10 inch or 12 inch phased array antenna. A full 160 nautical mile TSO'd performance is obtained with the 10 inch antenna. Increased performance can be obtained by using the 12 inch diameter array where radome space will allow. For installations that require operation above 25,000 feet, a pressurized antenna is available. This antenna requires a hose connection to the cabin atmosphere.

The KA 126 Antenna/Receiver/Transmitter uses only 2.5 amperes of 27.5VDC to power the total system. No 400Hz power is required.

The stabilization feature, although its utilization (and interconnect) is optional, accepts pitch and roll output signals from vertical gyros having DC, 400Hz or other AC pitch outputs. However, some vertical gyros were not designed with auxiliary outputs, and others may have operational restrictions which might limit stabilization usage to certain conditions. It is advisable to investigate interface possiblities prior to system installation.

technical characteristics

Power Requirement: Panel Lighting:

	WEATHER RADAR SYSTEM
TOTAL	CHARACTERISTICS
SPECIFICATIONS	Selectable 10nm, 20nm, 40nm, 80nm, 160nm
Display Ranges:	112.07dB 12 inches Antenna
Performance Index:	(Nominal per DO-134) 109_25dB 10 inches antenna (Nominal per DO-134)
Avoidance Range:	200nm 12 inch phased array 170nm 10 inch phased array Based on ARINC standards using measured digital radar performance and general aviation installation techniques.
- transit	27.5VDC <u>+</u> 10% 2.8 amperes maximum

KA 126 ANTENNA/RECEIVER/TRANSCEIVER

dimmer.

Optional 5VAC/DC or 13.75/27.5VDC light

TO ATTOMS	CHARACTERISTICS
SPECIFICATIONS	Zewy (Yehand)
Output Frequency:	9375 <u>+</u> 30MHz (X-band) 3.75 microseconds (nominal)
Pulsewidth:	109pps (nominal)
PRF:	8.OdB nominal
Noise Factor:	-109dBm minimum detectable signal
Receiver Sensitivity:	Effective to 25nm
STC:	7.5KW nominal, 6.0KW minimum
Peak Power Output:	30MHz
Intermediate Frequency:	90° (nominal)
Scan Sector:	13 looks per minute (nominal)
Scan Rate:	+ 12 degrees (nominal)
Beam Tilt:	+ 30° for all combinations of pitch, roll and
Stabilization	manual tilt.
	10 to 134VRMS 400-10KHz
yro Excitation	30 to 220 mv/degree
itch and roll inputs	

SPECIFICATION	CHARACTERISTIC						
Horizontal Beamwidth:	7.8° (nominal) for 12 inches antenna 9.0° (nominal) for 10 inches antenna						
Antenna Gain:	26.3dB (nominal) 12 inch antenna 25.3dB (nominal) 10 inch antenna						
Altitude:	25,000 feet unpressurized versions						
Temperature:	50,000 feet pressurized versions -55°C to +70°						
Cooling:	Natural Convection						
Size:	Base 10 inch diameter, see installation drawing for antenna clearances.						
Weight:	9.5 pounds						
Mounting:	Hard mounted on rear						
TSO:	C63b Class 7 B2AM∰XXXXXABABA						
	Env. Cat. of DO-160						
	pressurized antenna D2AMEXXXXXABABA						

KI 244 INDICATOR

SPECIFICATIONS	CHARACTERISTICS
Presentation:	TV-like PPI display with range rings and bearing marks.
Colors:	WX: Green, Yellow, Red MAP: Blue, Yellow, Magenta
Display Size:	4.25 x 3.20 (5 " diag. CRT)
Display Storage:	65,536 bit Digital Storage
Scan:	TV type raster
Range marks:	Distance annunciated on screen
Bearing marks:	Center and 20° either side.
Red Level (contour):	16dB above MDS (Nominal) within STC range
Extended STC:	Modifies targets from end of STC to maximum display range.
CONTROLS:	OFF/STBY/TEST/WX/MAP; TILT; 10/20/40/80/160nm; BRIGHT/MAPGAIN; STAB ON/OFF
Map Gain:	16dB nominal manual gain reduction
Altitude:	20,000 feet
Temperature:	-15°c to +55°c

SPECIFICATION	CHARACTERISTIC
Size:	4.7 inches high Silver Crown tray 11.5 inches deep
Weight:	8.3 pounds with mounting tray
TSO:	C63b class 7 A1B1/A/XS/XXXXXZBABA ENV. CAT. OF DO-160

1.4 units and accessories supplied

The KWX 56 Weather Radar System is comprised of the KI 244 Indicator (KPN 066-3065-00/03) and the KA 126 Antenna/Receiver/Transmitter (KPN 071-1220-00/02). The KI 244 Indicator is available with either 5V AC/DC lighting or 13.75/28.5VDC lighting configurations in black or gray face panels. Three antenna configurations are offered. Depending on the space limitations and performance of the air frame, a 12 inch pressurized or unpressurized 10 inch or 12 inch antenna is offered.

For installations in aircraft with ceilings above 25,000 feet a pressurized system must be used. Since the antenna is pressurized by venting humid cabin air into the wave guide, a means should be used to remove the moisture before it can condense on the cold inner antenna surfaces. It is therefore recommended that a desiccator be installed. King offers such a desiccator (KPN 071-1245-00) and refil crystals (KPN 016-1153-00).

Installation kits are included with each unit that include all required connectors and templates. Harness wire must be supplied by the installation agency.

The lowing assemblies are used in the KWX 56 system.

Discription	King Part Number
KA 126 Non-pressurized 12 inch Antenna KA 126 Non-pressurized 10 inch Antenna KA 126 Pressurized 12 inch Antenna	071-1220-00 071-1220-01 071-1220-02 071-1220-03
KA 126 Installation Kit: Connector Sub-min f37p Hood and Level Assembly Template	050-1853-00 030-2348-03 030-2351-02 047-5777-01
KI 244 Radar Display BLK W/14/28VDC lighting KI 244 Radar Display BLK W/5VDC lighting KI 244 Radar Display GRY W/14/28VDC lighting KI 244 Radar Display GRY W/5VDC lighting	066-3065-00 066-3065-01 066-3065-02 066-3065-03
KI 244 Installation Kit Connector 72 Contact Screws	050-1852-00 030-2410-00
ipment required; but not supplied	050-1853-01

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Desiccant Chamber (See Figure 2-6) KIT -971-1245-00 016-1158-00 Desiccant Refill 090-0236-02 TUBLIG

Installation Parts not supplied or available from King

in the same and the Radome Kit Shims

license requirements

The transmitter, as installed in the aircraft, requires an Aircraft Radio Station License. This license is obtained by filing FCC Form 404. The KXW 56 may be operated for up to 30 days without a station license, after filing the FCC Form 404 and while awaiting the receipt of the station license, providing a copy of the FCC Form 404 is kept in the aircraft.

This equipment has been type accepted by the FCC and entered on their list of type accepted equipment as King KA 126 and must be identified as King KA 126 on your FCC Form 404, Aircraft Radio Station License Application.

section ii installation

2.1 general

Suggestions and factors to consider before installing the KWX 56 Weather Radar System in an aircraft are given in this section. Refer to Figure 2.5 for the KWX 56 interconnect diagrams.

2.2 unpacking and inspecting equipment

Exercise extreme care when unpacking the equipment. Make a visual inspection of the unit for evidence of damage incurred during shipment. If a claim for damage is to be made, save the shipping container to substantiate the claim. The claim should be promptly filed with the transporation company. It would be advisable to retain the container when all equipment has been removed, in the event that equipment storage or reshipment should become necessary.

2.3 equipment Installation

2.3.1 HARNESS CONSIDERATIONS

- A. When fabricating the interconnect harness, care needs to be used in selecting the gyro input pin. Folow the notes on the harness drawing(Figure 2.5), especially for the gyro connection. If a series resistor needs to added do so with care. Avoid crushing the resistor with the wire bundle.
- Route the harness to avoid ADF or COMM antennas and bundles. Avoid sharp edges that could cut
 the insulation.
- C. Do not route harness in such a way that it could be struck by the antenna flat plate.

2.3.2 ANTENNA/RECEIVER/TRANSMITTER INSTALLATION

A. The KA 126 is designed to be rigidly mounted in the nose sections of the aircraft. If the nose section is not accessable, pod mounting is possible. The bulkhead or antenna mounting plate must be very close to perpendicular to the aircraft center line. The selected location must have adequate clearance for the full antenna sweep and tilt range (See Figure 2-1). The nose section need not be pressurized but if the aircraft is to be operated at altitudes greater than 25,000 feet a pressurized antenna must be used. The radome needs a minimum transmissivity of 90 percent. (See dot Advisory Circular No. 43-14 for radome considerations.)

NOTE

It is crutial to the preformance of the KWX 56 weather radar system that care be taken in alignment of the aircraft with respect to the KA 126 unit.

- B. Using the template (KPN 047-5772-01) horizontal \pm 1/2 to the flight line of the aircraft mark the center of the 4 mounting holes. Drill holes with 17/64" bit.
- C. Mount the KA 126 with 4-1/4 X 20 screws and lock washers.
- D. Using the leveling surfaces on the base of the KA 126 (See Figure 2-2) shim the unit until it is mounted dead ahead \pm 1/2°.

NOTE

If the unit is shimed exercise caution not to bind or distort the base casting.

- E. Check the antenna for clearance in all extreme positions.
- F. If used, mount desiccant chamber rigidly to air frame.

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:3.3 KI 244 INDICATOR INSTALLATION

- A. Carefully plan the location of the KI 244 before any panel alterations are made. 3e sure of adequate depth behind the panel for the mounting rack and harness. Avoid area of high heat sources such as heaters. Shaded or darker areas of the panel are preferred to locations of direct sunlight. The compass safe distance is 9 inchs. Don't route the harness with ADF Cables or antenna.
- B. After determining the location of the KI 244 indicator, refer to Figure 2-3 for the mounting hole dimensions.
- C. Install the tray using the 4 mounting holes in the front and attach the rear to a structual member by means of a rigid support.
- D. Secure the harness to the mounting rack.
- E. Slide the KI 244 into the mounting tray. Secure the indicator. Use a 3/32m hex wrench to tighten the locking device located on the lower front of the unit (See Figure 2-3).
- 2.4 post installation check

2.4.1 TEST EQUIPMENT REQUIRED

Tilt Table.

2.4.2 STABILITY ALIGNMENT PROCEDURE

NOTE

TO ASSURE GOOD QUALITY PERFORMANCE THE FOLLOWING ALIGNMENT IS IMPORTANT.

- A. Remove the gyro from the aircraft and place it on a tilt table.
- B. Remove, with a pin prick, the King logo from the KI 244 to expose the alignment adjustments. (See Figure 2-4 for Identification).
- C. Verify 4.0± 0.4VDC volts at TP 902. If incorrect verify that the gyro excitation voltage requirements are met. (See interconnect drawing Figure 2-5).
- D. Set tilt table (gyro) for Ω^0 roll and Ω^0 pitch adjust R901 and R903 until the LED corresponding to each is just turning on or flickering.
- E. Set the table for 20° roll and 0° pitch adjust R904 for the transition points on its LED.
- F. Adjust the table for 0° roll and 10° pitch and adjust R902 until its LED is just turning on or flickering.
- G. Repeat steps D, E, and F until all LEDS turn on at the correct roll or pitch angle.
- H. Clip the King logo into the KI 244.
- I. Re-Install the gyro in the aircraft. Verify that the gyro is aligned with the aircraft axis.

2.4.3 IN FLIGHT ADJUSTMENTS

Slight errors in the alignment of the KA 126 or gyro with the airframe can and should be eliminated. Large errors (3° or more) should be corrected on the ground. If a large error is detected check the excitation voltage and roll and pitch settings. Trying to correct large errors in flight can cause the KA 126 antenna to travel into its mechanical stops.

Errors and nonrepeatability of the alignment can be introduced by the gyro. Many factors have an effect on the gyro and therefore, its output accuracy can vary up to 10%. Vertical gyro precesion may be experienced during take offs and prolonged maneuvers. The antenna stabilization may have a 30 to 50 pr with alignment errors.

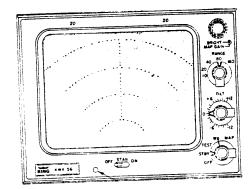
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-Mechanical Stigmmont Check

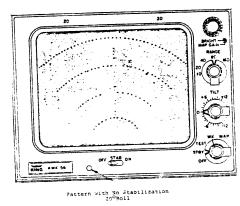
- an eligibitude in excess of 8,000 feet at 00 moll and an pitch.
- Select the of mile conge.
- Turn off the stabilization.
 - WW 20 NM Idjust the filt until the center of the terrain line is located on a range ring. Note the elevation reading.
- 5. If the terrain line is skeued, the mechanical alignment to the aircraft is in error (See Figure 2-4). The KA 126 is not aligned with the exis of the aircraft.
- Use the tilt central to ascertain the amount of error. Move the edge of the pattern in or out to align it with the range ring. Mote the tilt switch setting. The difference between this reading and the tilt reading, when the center of the terrain line is sligned to the range ring, is the amount of error. Errors of more than 2° must be corrected. Beturn to the ground and slightly rotate the KI 126 into alignment. Errors of less than Do can usually be tolerated.

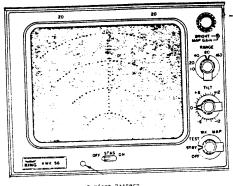
T. Automatic Stabilization Chack

- 1. Fly to an altitude in excess of 8,000 feet.
- Place the MI 344 in the Minite mode.
- 7. Turn on the automatic stabilization.
- 4. Phile flying level (60 roll), adjust the filt control until center of the terrain line is located on the Third range ring. Use Figure 2-4 to compare the KI 244 display.
- 5. Use the tilt control to find the amount of error (degrees of tilt required to adjust the ands of the terrain band to the range ring). If the error is greater than I degrees, recalibration on the ground is required. If the error is less than 3 degrees, adjust the rol' offset (3501) until the error is eliminated (Fig. 2-4).
- 6. Fly in $\sim 20^{\circ}$ right roll. Use Figure 2-4 for comparison. Note if the terrain line moves more than I degrees (using the filt as before). Errors of less than I degrees can be corrected by adjusting the roll gain (2004) until the error is corrected. Errors of more than T degrees, call for a recalibration on the ground. Trying to adjust for the large acros could cause the antenna to travel into its mechanical stops.

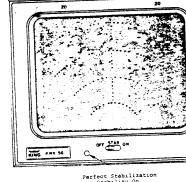


Pattern with Stability Error If occurs at 30Roll, problem is offset error. If occurs at 200Roll, problem is gain error.

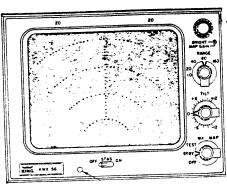




Perfect Pattern Stability Off 9 roll 9 pitch Mounting Surfaces Satisfactory



Perfect Stabilization
Stability On
No Change from Stability Off
and
OProll & OPpitch



MAX 24

HEATHER DADAR MITH MITH

FIGURE 2-2 KA 126 COMMECTOR P1261/J1261 NO KEYING PIN VIEW OF MALE OR SOLDER CUP VIEW OF FEMALE CONMECTOR

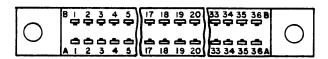


FIGURE 2-7 KI 244 COMMECTOR P2441 NO KEYING CABLE TERMINATION, SOLDER EYELET VIEW OF FEMALE CONNECTOR

Page 2-5

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FIGURE 2-4 KA 126 ANTENNA/RECEIVER/TRANSMITTER UNIT CUTLINE AND MOUNTING DRAWING

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FIGURE 2-5 KI 244 OUTLINE AND MOUNTING DRAWING

5-6 3-6

KING KMX 54 MESTHER PARAP

FIGURE 2-6 KWX 55 INTERCONMECT (Sheet 1 of 2)

Page 3-11

section iii operation

™.1 general

t is recommended that the aircraft engines be started before applying power to the Weather Radar. This procedure will increase the solid state circuitry reliability. The radar is turned on by rotating the mode control clockwise from the detented OFF position to STBY, TEST, WX, or MAP. If STBY is selected the display will not light up nor will the transmitter be operational. In TEST, WX or MAP the display will light up and the transmitter will become operational after 60 seconds. The 60 sec. warmup condition is annunicated in the lower left hand corner of the display. It is recommended that on initial turn-on the TEST mode be selected to adjust the BRIGHTness control for optimum viewing for the cockpit lighting conditions. After the warmup period the TEST pattern will appear and should be examined to verify that all colors are present. After examining, the TEST pattern it is recommended that the WX mode be momentarily selected to verify that ground targets are present, if the aircraft is on ten ramp. Finally the STBY mode should be selected until the radar is needed.

WARNING

NEVER OPERATE THE WEATHER RADAR ON THE GROUND WHEN PERSONNEL ARE FORWARD OF THE AIRCRAFT WINGS AND WITHIN 5 FEET OF THE AIRCRAFT NOSE. FAILURE TO OBSERVE THIS PRECAUTION MAY RESULT IN PERMANENT DAMAGE TO THE EYES AND OTHER BODY ORGANS.

The KI 244 has four modes of operation; STBY, TEST, WX and MAP. In STBY the display, antenna and transmitter are disabled but the magentron heater is kept on. IN TEST all circuity is activiated including the transmitter. A test pattern shows bands of all three colors. The WX mode is the normal weather mode with light precipitation depicted as green, moderate as yellow and heavy as red. The MAP mode is used for terrain mapping in conjunctin with the GAIN control. The colors are changed as follows green to blue, yellow stays the same, and red to magenta.

This enables the pilot to lect the displayed distance for most optimum viewing of the weather conditions that exist. There are rour range calibration rings on each range setting with numerical read-out of their range in nautical miles. In addition to the range rings, bearing markers are positioned dead ahead and 20 degrees to either side of the aircraft heading for use in judging the storm bearing and necessary heading changes. Since the KWX 56 is designed as a weather radar, it cannot be used as collision advoidance radar and should not be expected to.

Proper tilt adjustment is one of the most important factors in obtaining optimum value form a weather radar. Too high will pass the majority of the radar beam above the storm cell, particularly when the storm is a great distance. Too low an antenna tilt will clutter the indicator with ground returns. Optimum tilt is obtained by adjusting the antenna to obtain a slight amount of ground clutter. The maximum distance at which ground clutter can be obtained will depend greatly on the terrain and aircraft altitude.

Ground mapping by tilting the antenna down can be performed under certan conditions, particularly during over-water flights where coast lines and islands are apparent. When using the MAP mode, the Map contrast control is used to adjust the prominence of ground features.

Should the vertical gyro supplying pitch information to the Weather Radar fail, the display may become unusable because of eratic or unsuitable tilting of the antenna. If this sould happen the STAB ON/OFF switch should be turned off to disable the gyro inputs and allow normal use of the tilt control.

The pilot's guide should be consulted for a more thorough description of operating procedures for the weather radar and information on the interpretation of weather and ground returns.

WARNING

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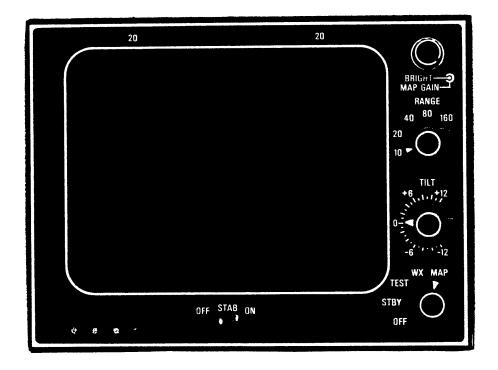
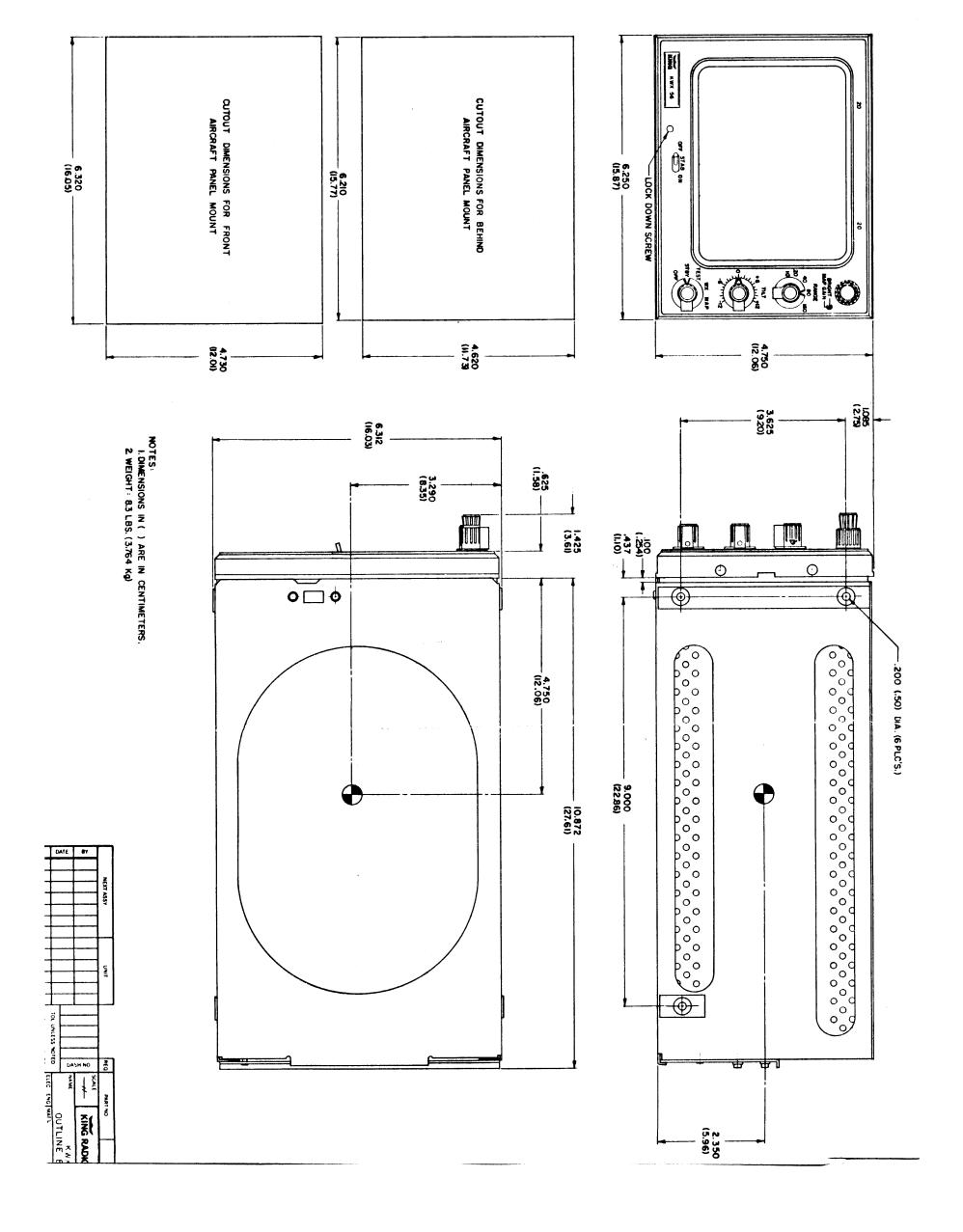
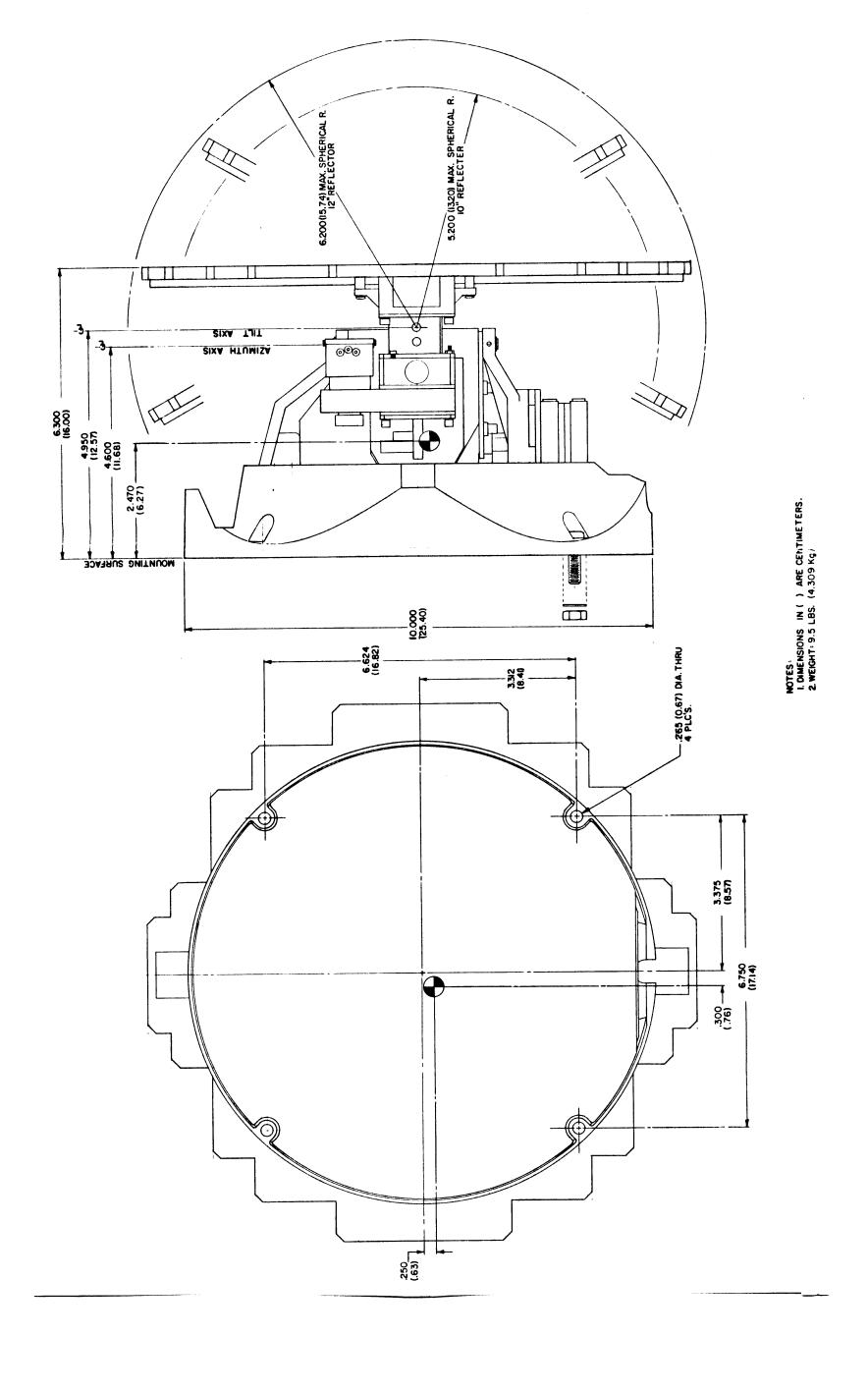


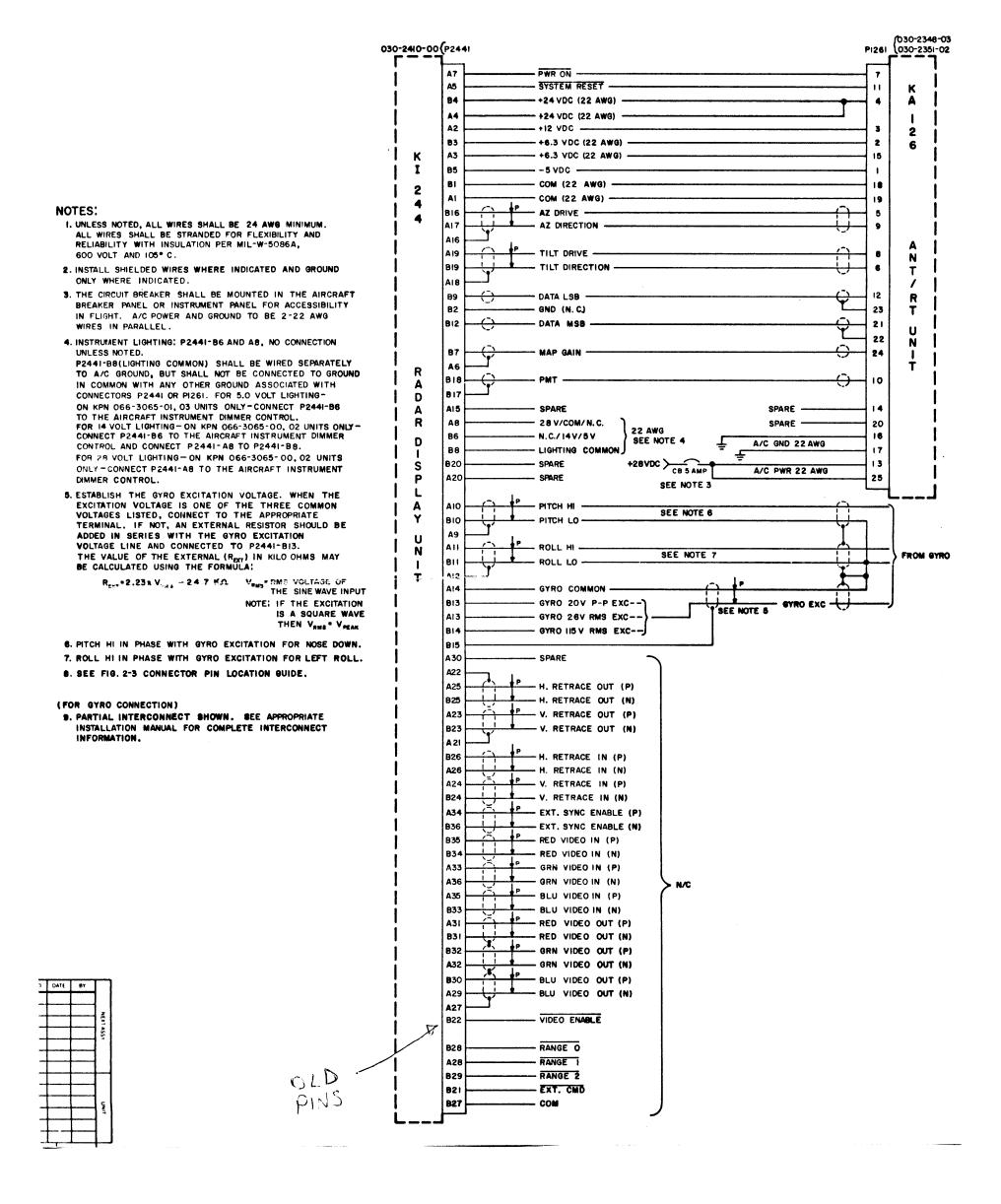
FIGURE T-1KX 244 THDICATOR UNIT

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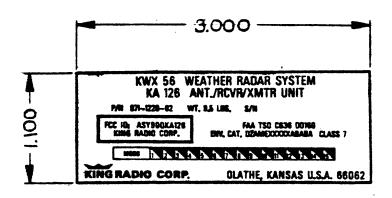
INC 1477-9







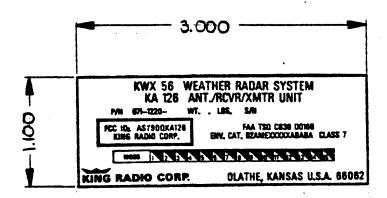
APPENDIX C IDENTIFICATION LABELS



NOTES:

- 1.) MATERIAL TO BE .003 TO .005 SUB-SURFACE PRINTED VINYL, COATED WITH PERMANENT PRESSURE SENSITIVE ADHESIVE; WITH PROTECTIVE LINER.
- 2.) TAGS TO BE LAMINATED WITH MARKEM CORP. # 2064 OR EQUAL.
- 3.) LETTERS AND MODE BLOCK TO BE SILVER, UNIT NAME, NUMBER & LOGO TO BE BLUE, PER SAMPLE PROVIDED BY KING RADIO CORPORATION.
- 4.) BACKGROUND TO BE SEMI-GLOSS BLACK.
- 5.) ARTWORK: 003-1674-01.
- 6.) KING PART NUMBER 057-2368-01 WITH SERIAL NUMBER STAMPED CONSECUTIVELY AS REQUESTED BY PURCHASE ORDER.
- 7.) CUTTING DIE NO. 2368.

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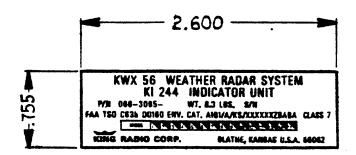


NOTES:

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- 1.) MATERIAL TO BE .003 TO .005 SUB-SURFACE PRINTED VINYL, COATED WITH PERMANENT PRESSURE SENSITIVE ADHESIVE; WITH PROTECTIVE LINER.
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- 4.) BACKGROUND TO BE SEMI-GLOSS BLACK.
- 5.) ARTWORK: 003-1674-00.
- 6.) KING PART NUMBER 057-2369-00 WITH SERIAL NUMBER STAMPED CONSECUTIVELY AS REQUESTED BY PURCHASE ORDER.
- &.) CUTTING DIE NO. 2368.

011-1220-00/02 KA 126							6							
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NOTES:

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- 6.) KING PART NUMBER 057-2369-00 WITH SERIAL NUMBER STAMPED CONSECUTIVELY AS REQUESTED BY PURCHASE ORDER.
- 7.) CUTTING DIE NO. 2369.

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APPENDIX D
PHOTOGRAPHS OF EQUIPMENT

