

### AeroCube-7 Proximity Operations CONOPS



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# **RPO CONOPS Flow Chart**





# **RPO CONOPS Notional Diagram**



AEROSPACE

# **Initial Conditions**

- After deployment and during checkout, AeroCubes will drift due to initial deployment dispersions and checkout activities.
- Begin RPO with substantial (>10 km) in-track separation between AeroCubes, plus some radial and cross-track separation
  - Will use differential drag during checkout if available to minimize in-track separation as much as possible
- Assumptions for this analysis:
  - Orbit is 500 km circular, 65 deg inclination
  - Min drag area: 150 cm<sup>2</sup>
  - Max drag area: 500 cm<sup>2</sup>
  - *Mass: 2 kg*



# **Approach Staging Point**

- RPO process begins with approach to a "staging point" to ensure control of the spacecraft and efficacy of planned maneuver schemes.
- Use on-board propulsion to correct plane differences between AeroCubes:
  - Based on AeroCube-4 experience, as much as 0.002 deg of inclination change, costing 30 cm/s of  $\Delta V$ .
  - As much as 0.005 deg of RAAN change, costing 75 cm/s.
  - Budget 2 m/s total, performed in small amounts over many orbits.
  - Can perform plane change maneuvers at any time.
- Use differential drag with a bang-bang attitude-control scheme to approach the target AeroCube within 2 km in-track. This is the "staging point."
  - At end of differential drag process, mean motion of chaser and target are matched.



### Approach Staging Point, In-Plane Motion





# Approach Staging Point, Out-of-Plane Motion

At the staging point, the target has some cross-track motion.

The line of motion (the orbit velocity vector) of the target AeroCube goes into/out of the screen in this plot. The chaser AeroCube "orbits" around that line of motion

Risk mitigation: by following this trajectory, the chaser **never** crosses the path of the target, preventing collisions.



High-fidelity orbit propagation with TRACE



### **Tests at Staging Point: Eccentricity Reduction**



#### Tests at Staging Point: Cross-Track Reduction

Small burns (~10 mm/s) performed in the cross-track direction reduce the cross-track excursions of the Hill's orbit. These burns can occur as often as every half-orbit. In practice, will use longer lead times.







# Note on Risk from Staging-Point Tests

- In the event that a test burn is misaligned entirely in the in-track direction, the chaser will continue to corkscrew around the line of motion of the target due to cross-track motion. Closest approach to target's line of motion >50 m.
- If a test burn is misaligned in the cross-track direction, the change in cross-track motion is minimal.
- No individual burn is sufficient to put the chaser AeroCube on a collision course with the target.
  - Until we have high confidence in the performance of the thruster later in the mission, each burn will be followed by orbit determination and thruster-performance analysis to ensure that the new desired orbit was achieved.
  - In the event of an anomaly or undesired behavior, no further burns will be performed to ensure the safety of both spacecraft.



### Transfer: Staging Point to RPO Start Point





# **RPO:** Initiate In-Track Drift and Approach Target





### **RPO: Chaser-to-Target Range**

The duration of the RPO depends on the amount of in-track drift induced. For this  $\Delta V = 20$  mm/s (start and stop) example, the RPO takes about one day, with closest approach halfway.



In this example the closest approach is ~50 m. As confidence grows, the radius of the Hill's orbit will be reduced.





#### **RPO: Chaser-to-Target Relative Velocity**

In this example, the relative velocity between the chaser and target does not exceed 0.35 m/s (<1 mph).



Even is a collision did happen, these relative velocities are not high enough to cause fragmentation or a catastrophic breakup.



### ΔV Budget

Event	ΔV [m/s]	Comment
Plane corrections	2	
Staging-point test: eccentricity reduction	0.1	~10 mm/s per maneuver
Staging-point test: cross-track reduction	0.1	~10 mm/s per maneuver
Transfer from staging point to RPO start point	0.12	Maximum possible cost (for transfer in 1 orbit)
RPO: start and stop in-track drift	0.02	
RPO: modify radius of Hill's orbit	0.01	Changes radius by ~20 m

A full RPO cycle, including Hill's orbit modifications, costs ~30 mm/s.

Even with this conservative DV budget, the 10 m/s capacity of the AeroCube-7 system should permit dozens of RPO cycles with considerable propellant margin.



### Summary

- The AeroCube-7 RPO CONOPS has been designed to minimize risk to both vehicles and to build maximum confidence via incremental testing of maneuver schemes.
  - "Dress rehearsal" maneuvers at a staging point will characterize control authority on chaser AeroCube without risk to target.
  - Cross-track amplitude is always maintained to ensure that the chaser AeroCube never crosses the path of the target, preventing collision.
    - During RPO, chaser "corkscrews" around target
  - Radius of Hill's orbit will be reduced incrementally over several RPO cycles.
- ΔV budget has considerable margin
  - Each RPO cycle costs ~30 mm/s
  - Can perform dozens of RPO cycles

