

09 01 2016; 10:26:33AM DAS Application Started  
09 01 2016; 10:26:33AM Opened Project C:\DAS 2.0\tomsat\  
09 01 2016; 10:27:30AM Mission Editor Changes Applied  
09 01 2016; 10:27:35AM Processing Requirement 4.3-1: Return  
Status : Not Run

=====  
No Project Data Available  
=====

=====  
End of Requirement 4.3-1  
09 01 2016; 10:27:38AM Processing Requirement 4.3-2: Return Status  
: Passed

=====  
No Project Data Available  
=====

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End of Requirement 4.3-2  
09 01 2016; 10:27:40AM Requirement 4.4-3: Compliant

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End of Requirement 4.4-3  
09 01 2016; 10:28:00AM Processing Requirement 4.5-1: Return  
Status : Passed

=====  
Run Data  
=====

\*\*INPUT\*\*

Space Structure Name = TOMSAT Complete  
Space Structure Type = Payload  
Perigee Altitude = 500.000000 (km)  
Apogee Altitude = 500.000000 (km)  
Inclination = 85.000000 (deg)  
RAAN = 0.000000 (deg)  
Argument of Perigee = 0.000000 (deg)  
Mean Anomaly = 0.000000 (deg)  
Final Area-To-Mass Ratio = 0.010750 (m<sup>2</sup>/kg)  
Start Year = 2017.000000 (yr)  
Initial Mass = 5.000000 (kg)  
Final Mass = 5.000000 (kg)  
Duration = 1.000000 (yr)  
Station-Kept = False  
Abandoned = True  
PMD Perigee Altitude = -1.000000 (km)  
PMD Apogee Altitude = -1.000000 (km)  
PMD Inclination = 0.000000 (deg)  
PMD RAAN = 0.000000 (deg)  
PMD Argument of Perigee = 0.000000 (deg)  
PMD Mean Anomaly = 0.000000 (deg)

\*\*OUTPUT\*\*

Collision Probability = 0.000000  
Returned Error Message: Normal Processing  
Date Range Error Message: Normal Date Range  
Status = Pass

=====

\*\*INPUT\*\*

Space Structure Name = Full Body Only  
Space Structure Type = Payload  
Perigee Altitude = 500.000000 (km)  
Apogee Altitude = 500.000000 (km)  
Inclination = 85.000000 (deg)  
RAAN = 0.000000 (deg)  
Argument of Perigee = 0.000000 (deg)  
Mean Anomaly = 0.000000 (deg)  
Final Area-To-Mass Ratio = 0.043712 (m<sup>2</sup>/kg)  
Start Year = 2017.000000 (yr)  
Initial Mass = 0.644200 (kg)  
Final Mass = 0.644200 (kg)  
Duration = 1.000000 (yr)  
Station-Kept = False  
Abandoned = True  
PMD Perigee Altitude = -1.000000 (km)  
PMD Apogee Altitude = -1.000000 (km)  
PMD Inclination = 0.000000 (deg)  
PMD RAAN = 0.000000 (deg)  
PMD Argument of Perigee = 0.000000 (deg)  
PMD Mean Anomaly = 0.000000 (deg)

\*\*OUTPUT\*\*

Collision Probability = 0.000000  
Returned Error Message: Normal Processing  
Date Range Error Message: Normal Date Range  
Status = Pass

=====

\*\*INPUT\*\*

Space Structure Name = Frame Part 1 (2001)  
Space Structure Type = Payload  
Perigee Altitude = 500.000000 (km)  
Apogee Altitude = 500.000000 (km)  
Inclination = 85.000000 (deg)  
RAAN = 0.000000 (deg)  
Argument of Perigee = 0.000000 (deg)

Mean Anomaly = 0.000000 (deg)  
Final Area-To-Mass Ratio = 0.043995 (m<sup>2</sup>/kg)  
Start Year = 2017.000000 (yr)  
Initial Mass = 0.298130 (kg)  
Final Mass = 0.298130 (kg)  
Duration = 1.000000 (yr)  
Station-Kept = False  
Abandoned = True  
PMD Perigee Altitude = -1.000000 (km)  
PMD Apogee Altitude = -1.000000 (km)  
PMD Inclination = 0.000000 (deg)  
PMD RAAN = 0.000000 (deg)  
PMD Argument of Perigee = 0.000000 (deg)  
PMD Mean Anomaly = 0.000000 (deg)

\*\*OUTPUT\*\*

Collision Probability = 0.000000  
Returned Error Message: Normal Processing  
Date Range Error Message: Normal Date Range  
Status = Pass

=====

\*\*INPUT\*\*

Space Structure Name = Frame Part 2 (4001)  
Space Structure Type = Payload  
Perigee Altitude = 500.000000 (km)  
Apogee Altitude = 500.000000 (km)  
Inclination = 85.000000 (deg)  
RAAN = 0.000000 (deg)  
Argument of Perigee = 0.000000 (deg)  
Mean Anomaly = 0.000000 (deg)  
Final Area-To-Mass Ratio = 0.054262 (m<sup>2</sup>/kg)  
Start Year = 2017.000000 (yr)  
Initial Mass = 0.346070 (kg)  
Final Mass = 0.346070 (kg)  
Duration = 1.000000 (yr)  
Station-Kept = False  
Abandoned = True  
PMD Perigee Altitude = -1.000000 (km)  
PMD Apogee Altitude = -1.000000 (km)  
PMD Inclination = 0.000000 (deg)  
PMD RAAN = 0.000000 (deg)  
PMD Argument of Perigee = 0.000000 (deg)  
PMD Mean Anomaly = 0.000000 (deg)

\*\*OUTPUT\*\*

Collision Probability = 0.000000  
Returned Error Message: Normal Processing

Date Range Error Message: Normal Date Range  
Status = Pass

=====

\*\*INPUT\*\*

Space Structure Name = Bus Electronics  
Space Structure Type = Payload  
Perigee Altitude = 500.000000 (km)  
Apogee Altitude = 500.000000 (km)  
Inclination = 85.000000 (deg)  
RAAN = 0.000000 (deg)  
Argument of Perigee = 0.000000 (deg)  
Mean Anomaly = 0.000000 (deg)  
Final Area-To-Mass Ratio = 0.012604 (m<sup>2</sup>/kg)  
Start Year = 2017.000000 (yr)  
Initial Mass = 0.627060 (kg)  
Final Mass = 0.627060 (kg)  
Duration = 1.000000 (yr)  
Station-Kept = False  
Abandoned = True  
PMD Perigee Altitude = -1.000000 (km)  
PMD Apogee Altitude = -1.000000 (km)  
PMD Inclination = 0.000000 (deg)  
PMD RAAN = 0.000000 (deg)  
PMD Argument of Perigee = 0.000000 (deg)  
PMD Mean Anomaly = 0.000000 (deg)

\*\*OUTPUT\*\*

Collision Probability = 0.000000  
Returned Error Message: Normal Processing  
Date Range Error Message: Normal Date Range  
Status = Pass

=====

\*\*INPUT\*\*

Space Structure Name = Reaction Wheels  
Space Structure Type = Payload  
Perigee Altitude = 500.000000 (km)  
Apogee Altitude = 500.000000 (km)  
Inclination = 85.000000 (deg)  
RAAN = 0.000000 (deg)  
Argument of Perigee = 0.000000 (deg)  
Mean Anomaly = 0.000000 (deg)  
Final Area-To-Mass Ratio = 0.012760 (m<sup>2</sup>/kg)  
Start Year = 2017.000000 (yr)  
Initial Mass = 0.225010 (kg)  
Final Mass = 0.225010 (kg)

Duration = 1.000000 (yr)  
Station-Kept = False  
Abandoned = True  
PMD Perigee Altitude = -1.000000 (km)  
PMD Apogee Altitude = -1.000000 (km)  
PMD Inclination = 0.000000 (deg)  
PMD RAAN = 0.000000 (deg)  
PMD Argument of Perigee = 0.000000 (deg)  
PMD Mean Anomaly = 0.000000 (deg)

\*\*OUTPUT\*\*

Collision Probability = 0.000000  
Returned Error Message: Normal Processing  
Date Range Error Message: Normal Date Range  
Status = Pass

=====

\*\*INPUT\*\*

Space Structure Name = Star Tracker  
Space Structure Type = Payload  
Perigee Altitude = 500.000000 (km)  
Apogee Altitude = 500.000000 (km)  
Inclination = 85.000000 (deg)  
RAAN = 0.000000 (deg)  
Argument of Perigee = 0.000000 (deg)  
Mean Anomaly = 0.000000 (deg)  
Final Area-To-Mass Ratio = 0.017369 (m<sup>2</sup>/kg)  
Start Year = 2017.000000 (yr)  
Initial Mass = 0.054800 (kg)  
Final Mass = 0.054800 (kg)  
Duration = 1.000000 (yr)  
Station-Kept = False  
Abandoned = True  
PMD Perigee Altitude = -1.000000 (km)  
PMD Apogee Altitude = -1.000000 (km)  
PMD Inclination = 0.000000 (deg)  
PMD RAAN = 0.000000 (deg)  
PMD Argument of Perigee = 0.000000 (deg)  
PMD Mean Anomaly = 0.000000 (deg)

\*\*OUTPUT\*\*

Collision Probability = 0.000000  
Returned Error Message: Normal Processing  
Date Range Error Message: Normal Date Range  
Status = Pass

=====

\*\*INPUT\*\*

Space Structure Name = Single Torque Rod  
Space Structure Type = Payload  
Perigee Altitude = 500.000000 (km)  
Apogee Altitude = 500.000000 (km)  
Inclination = 85.000000 (deg)  
RAAN = 0.000000 (deg)  
Argument of Perigee = 0.000000 (deg)  
Mean Anomaly = 0.000000 (deg)  
Final Area-To-Mass Ratio = 0.034161 (m<sup>2</sup>/kg)  
Start Year = 2017.000000 (yr)  
Initial Mass = 0.018440 (kg)  
Final Mass = 0.018440 (kg)  
Duration = 1.000000 (yr)  
Station-Kept = False  
Abandoned = True  
PMD Perigee Altitude = -1.000000 (km)  
PMD Apogee Altitude = -1.000000 (km)  
PMD Inclination = 0.000000 (deg)  
PMD RAAN = 0.000000 (deg)  
PMD Argument of Perigee = 0.000000 (deg)  
PMD Mean Anomaly = 0.000000 (deg)

\*\*OUTPUT\*\*

Collision Probability = 0.000000  
Returned Error Message: Normal Processing  
Date Range Error Message: Normal Date Range  
Status = Pass

=====

\*\*INPUT\*\*

Space Structure Name = Zenith Lid Assembly  
Space Structure Type = Payload  
Perigee Altitude = 500.000000 (km)  
Apogee Altitude = 500.000000 (km)  
Inclination = 85.000000 (deg)  
RAAN = 0.000000 (deg)  
Argument of Perigee = 0.000000 (deg)  
Mean Anomaly = 0.000000 (deg)  
Final Area-To-Mass Ratio = 0.035553 (m<sup>2</sup>/kg)  
Start Year = 2017.000000 (yr)  
Initial Mass = 0.141260 (kg)  
Final Mass = 0.141260 (kg)  
Duration = 1.000000 (yr)  
Station-Kept = False  
Abandoned = True  
PMD Perigee Altitude = -1.000000 (km)  
PMD Apogee Altitude = -1.000000 (km)

PMD Inclination = 0.000000 (deg)  
PMD RAAN = 0.000000 (deg)  
PMD Argument of Perigee = 0.000000 (deg)  
PMD Mean Anomaly = 0.000000 (deg)

\*\*OUTPUT\*\*

Collision Probability = 0.000000  
Returned Error Message: Normal Processing  
Date Range Error Message: Normal Date Range  
Status = Pass

=====

\*\*INPUT\*\*

Space Structure Name = Antenna Zenith  
Space Structure Type = Payload  
Perigee Altitude = 500.000000 (km)  
Apogee Altitude = 500.000000 (km)  
Inclination = 85.000000 (deg)  
RAAN = 0.000000 (deg)  
Argument of Perigee = 0.000000 (deg)  
Mean Anomaly = 0.000000 (deg)  
Final Area-To-Mass Ratio = 0.037208 (m<sup>2</sup>/kg)  
Start Year = 2017.000000 (yr)  
Initial Mass = 0.015900 (kg)  
Final Mass = 0.015900 (kg)  
Duration = 1.000000 (yr)  
Station-Kept = False  
Abandoned = True  
PMD Perigee Altitude = -1.000000 (km)  
PMD Apogee Altitude = -1.000000 (km)  
PMD Inclination = 0.000000 (deg)  
PMD RAAN = 0.000000 (deg)  
PMD Argument of Perigee = 0.000000 (deg)  
PMD Mean Anomaly = 0.000000 (deg)

\*\*OUTPUT\*\*

Collision Probability = 0.000000  
Returned Error Message: Normal Processing  
Date Range Error Message: Normal Date Range  
Status = Pass

=====

\*\*INPUT\*\*

Space Structure Name = Wing Assembly  
Space Structure Type = Payload  
Perigee Altitude = 500.000000 (km)

Apogee Altitude = 500.000000 (km)  
Inclination = 85.000000 (deg)  
RAAN = 0.000000 (deg)  
Argument of Perigee = 0.000000 (deg)  
Mean Anomaly = 0.000000 (deg)  
Final Area-To-Mass Ratio = 0.054297 (m<sup>2</sup>/kg)  
Start Year = 2017.000000 (yr)  
Initial Mass = 0.148130 (kg)  
Final Mass = 0.148130 (kg)  
Duration = 1.000000 (yr)  
Station-Kept = False  
Abandoned = True  
PMD Perigee Altitude = -1.000000 (km)  
PMD Apogee Altitude = -1.000000 (km)  
PMD Inclination = 0.000000 (deg)  
PMD RAAN = 0.000000 (deg)  
PMD Argument of Perigee = 0.000000 (deg)  
PMD Mean Anomaly = 0.000000 (deg)

\*\*OUTPUT\*\*

Collision Probability = 0.000000  
Returned Error Message: Normal Processing  
Date Range Error Message: Normal Date Range  
Status = Pass

=====

\*\*INPUT\*\*

Space Structure Name = Nadir Lid Assembly  
Space Structure Type = Payload  
Perigee Altitude = 500.000000 (km)  
Apogee Altitude = 500.000000 (km)  
Inclination = 85.000000 (deg)  
RAAN = 0.000000 (deg)  
Argument of Perigee = 0.000000 (deg)  
Mean Anomaly = 0.000000 (deg)  
Final Area-To-Mass Ratio = 0.044140 (m<sup>2</sup>/kg)  
Start Year = 2017.000000 (yr)  
Initial Mass = 0.099590 (kg)  
Final Mass = 0.099590 (kg)  
Duration = 1.000000 (yr)  
Station-Kept = False  
Abandoned = True  
PMD Perigee Altitude = -1.000000 (km)  
PMD Apogee Altitude = -1.000000 (km)  
PMD Inclination = 0.000000 (deg)  
PMD RAAN = 0.000000 (deg)  
PMD Argument of Perigee = 0.000000 (deg)  
PMD Mean Anomaly = 0.000000 (deg)



\*\*OUTPUT\*\*

Collision Probability = 0.000000  
Returned Error Message: Normal Processing  
Date Range Error Message: Normal Date Range  
Status = Pass

=====

\*\*INPUT\*\*

Space Structure Name = Antenna Nadir  
Space Structure Type = Payload  
Perigee Altitude = 500.000000 (km)  
Apogee Altitude = 500.000000 (km)  
Inclination = 85.000000 (deg)  
RAAN = 0.000000 (deg)  
Argument of Perigee = 0.000000 (deg)  
Mean Anomaly = 0.000000 (deg)  
Final Area-To-Mass Ratio = 0.037208 (m<sup>2</sup>/kg)  
Start Year = 2017.000000 (yr)  
Initial Mass = 0.015900 (kg)  
Final Mass = 0.015900 (kg)  
Duration = 1.000000 (yr)  
Station-Kept = False  
Abandoned = True  
PMD Perigee Altitude = -1.000000 (km)  
PMD Apogee Altitude = -1.000000 (km)  
PMD Inclination = 0.000000 (deg)  
PMD RAAN = 0.000000 (deg)  
PMD Argument of Perigee = 0.000000 (deg)  
PMD Mean Anomaly = 0.000000 (deg)

\*\*OUTPUT\*\*

Collision Probability = 0.000000  
Returned Error Message: Normal Processing  
Date Range Error Message: Normal Date Range  
Status = Pass

=====

\*\*INPUT\*\*

Space Structure Name = Uplink Receiver  
Space Structure Type = Payload  
Perigee Altitude = 500.000000 (km)  
Apogee Altitude = 500.000000 (km)  
Inclination = 85.000000 (deg)  
RAAN = 0.000000 (deg)  
Argument of Perigee = 0.000000 (deg)  
Mean Anomaly = 0.000000 (deg)

Final Area-To-Mass Ratio = 0.035037 (m<sup>2</sup>/kg)  
Start Year = 2017.000000 (yr)  
Initial Mass = 0.027960 (kg)  
Final Mass = 0.027960 (kg)  
Duration = 1.000000 (yr)  
Station-Kept = False  
Abandoned = True  
PMD Perigee Altitude = -1.000000 (km)  
PMD Apogee Altitude = -1.000000 (km)  
PMD Inclination = 0.000000 (deg)  
PMD RAAN = 0.000000 (deg)  
PMD Argument of Perigee = 0.000000 (deg)  
PMD Mean Anomaly = 0.000000 (deg)

\*\*OUTPUT\*\*

Collision Probability = 0.000000  
Returned Error Message: Normal Processing  
Date Range Error Message: Normal Date Range  
Status = Pass

=====

\*\*INPUT\*\*

Space Structure Name = Payload Assembly  
Space Structure Type = Payload  
Perigee Altitude = 500.000000 (km)  
Apogee Altitude = 500.000000 (km)  
Inclination = 85.000000 (deg)  
RAAN = 0.000000 (deg)  
Argument of Perigee = 0.000000 (deg)  
Mean Anomaly = 0.000000 (deg)  
Final Area-To-Mass Ratio = 0.006980 (m<sup>2</sup>/kg)  
Start Year = 2017.000000 (yr)  
Initial Mass = 1.311400 (kg)  
Final Mass = 1.311400 (kg)  
Duration = 1.000000 (yr)  
Station-Kept = False  
Abandoned = True  
PMD Perigee Altitude = -1.000000 (km)  
PMD Apogee Altitude = -1.000000 (km)  
PMD Inclination = 0.000000 (deg)  
PMD RAAN = 0.000000 (deg)  
PMD Argument of Perigee = 0.000000 (deg)  
PMD Mean Anomaly = 0.000000 (deg)

\*\*OUTPUT\*\*

Collision Probability = 0.000000  
Returned Error Message: Normal Processing  
Date Range Error Message: Normal Date Range

Status = Pass

=====

\*\*INPUT\*\*

Space Structure Name = 18267  
Space Structure Type = Payload  
Perigee Altitude = 500.000000 (km)  
Apogee Altitude = 500.000000 (km)  
Inclination = 85.000000 (deg)  
RAAN = 0.000000 (deg)  
Argument of Perigee = 0.000000 (deg)  
Mean Anomaly = 0.000000 (deg)  
Final Area-To-Mass Ratio = 0.071468 (m<sup>2</sup>/kg)  
Start Year = 2017.000000 (yr)  
Initial Mass = 0.016690 (kg)  
Final Mass = 0.016690 (kg)  
Duration = 1.000000 (yr)  
Station-Kept = False  
Abandoned = True  
PMD Perigee Altitude = -1.000000 (km)  
PMD Apogee Altitude = -1.000000 (km)  
PMD Inclination = 0.000000 (deg)  
PMD RAAN = 0.000000 (deg)  
PMD Argument of Perigee = 0.000000 (deg)  
PMD Mean Anomaly = 0.000000 (deg)

\*\*OUTPUT\*\*

Collision Probability = 0.000000  
Returned Error Message: Normal Processing  
Date Range Error Message: Normal Date Range  
Status = Pass

=====

\*\*INPUT\*\*

Space Structure Name = 18265  
Space Structure Type = Payload  
Perigee Altitude = 500.000000 (km)  
Apogee Altitude = 500.000000 (km)  
Inclination = 85.000000 (deg)  
RAAN = 0.000000 (deg)  
Argument of Perigee = 0.000000 (deg)  
Mean Anomaly = 0.000000 (deg)  
Final Area-To-Mass Ratio = 0.048369 (m<sup>2</sup>/kg)  
Start Year = 2017.000000 (yr)  
Initial Mass = 0.010920 (kg)  
Final Mass = 0.010920 (kg)  
Duration = 1.000000 (yr)

Station-Kept = False  
Abandoned = True  
PMD Perigee Altitude = -1.000000 (km)  
PMD Apogee Altitude = -1.000000 (km)  
PMD Inclination = 0.000000 (deg)  
PMD RAAN = 0.000000 (deg)  
PMD Argument of Perigee = 0.000000 (deg)  
PMD Mean Anomaly = 0.000000 (deg)

\*\*OUTPUT\*\*

Collision Probability = 0.000000  
Returned Error Message: Normal Processing  
Date Range Error Message: Normal Date Range  
Status = Pass

=====

\*\*INPUT\*\*

Space Structure Name = 18264  
Space Structure Type = Payload  
Perigee Altitude = 500.000000 (km)  
Apogee Altitude = 500.000000 (km)  
Inclination = 85.000000 (deg)  
RAAN = 0.000000 (deg)  
Argument of Perigee = 0.000000 (deg)  
Mean Anomaly = 0.000000 (deg)  
Final Area-To-Mass Ratio = 0.067530 (m<sup>2</sup>/kg)  
Start Year = 2017.000000 (yr)  
Initial Mass = 0.012320 (kg)  
Final Mass = 0.012320 (kg)  
Duration = 1.000000 (yr)  
Station-Kept = False  
Abandoned = True  
PMD Perigee Altitude = -1.000000 (km)  
PMD Apogee Altitude = -1.000000 (km)  
PMD Inclination = 0.000000 (deg)  
PMD RAAN = 0.000000 (deg)  
PMD Argument of Perigee = 0.000000 (deg)  
PMD Mean Anomaly = 0.000000 (deg)

\*\*OUTPUT\*\*

Collision Probability = 0.000000  
Returned Error Message: Normal Processing  
Date Range Error Message: Normal Date Range  
Status = Pass

=====

\*\*INPUT\*\*

Space Structure Name = 18268  
Space Structure Type = Payload  
Perigee Altitude = 500.000000 (km)  
Apogee Altitude = 500.000000 (km)  
Inclination = 85.000000 (deg)  
RAAN = 0.000000 (deg)  
Argument of Perigee = 0.000000 (deg)  
Mean Anomaly = 0.000000 (deg)  
Final Area-To-Mass Ratio = 0.024134 (m<sup>2</sup>/kg)  
Start Year = 2017.000000 (yr)  
Initial Mass = 0.004040 (kg)  
Final Mass = 0.004040 (kg)  
Duration = 1.000000 (yr)  
Station-Kept = False  
Abandoned = True  
PMD Perigee Altitude = -1.000000 (km)  
PMD Apogee Altitude = -1.000000 (km)  
PMD Inclination = 0.000000 (deg)  
PMD RAAN = 0.000000 (deg)  
PMD Argument of Perigee = 0.000000 (deg)  
PMD Mean Anomaly = 0.000000 (deg)

\*\*OUTPUT\*\*

Collision Probability = 0.000000  
Returned Error Message: Normal Processing  
Date Range Error Message: Normal Date Range  
Status = Pass

=====

\*\*INPUT\*\*

Space Structure Name = 18266  
Space Structure Type = Payload  
Perigee Altitude = 500.000000 (km)  
Apogee Altitude = 500.000000 (km)  
Inclination = 85.000000 (deg)  
RAAN = 0.000000 (deg)  
Argument of Perigee = 0.000000 (deg)  
Mean Anomaly = 0.000000 (deg)  
Final Area-To-Mass Ratio = 0.015816 (m<sup>2</sup>/kg)  
Start Year = 2017.000000 (yr)  
Initial Mass = 0.010280 (kg)  
Final Mass = 0.010280 (kg)  
Duration = 1.000000 (yr)  
Station-Kept = False  
Abandoned = True  
PMD Perigee Altitude = -1.000000 (km)  
PMD Apogee Altitude = -1.000000 (km)  
PMD Inclination = 0.000000 (deg)

PMD RAAN = 0.000000 (deg)  
PMD Argument of Perigee = 0.000000 (deg)  
PMD Mean Anomaly = 0.000000 (deg)

\*\*OUTPUT\*\*

Collision Probability = 0.000000  
Returned Error Message: Normal Processing  
Date Range Error Message: Normal Date Range  
Status = Pass

=====

\*\*INPUT\*\*

Space Structure Name = 18263  
Space Structure Type = Payload  
Perigee Altitude = 500.000000 (km)  
Apogee Altitude = 500.000000 (km)  
Inclination = 85.000000 (deg)  
RAAN = 0.000000 (deg)  
Argument of Perigee = 0.000000 (deg)  
Mean Anomaly = 0.000000 (deg)  
Final Area-To-Mass Ratio = 0.041225 (m<sup>2</sup>/kg)  
Start Year = 2017.000000 (yr)  
Initial Mass = 0.001280 (kg)  
Final Mass = 0.001280 (kg)  
Duration = 1.000000 (yr)  
Station-Kept = False  
Abandoned = True  
PMD Perigee Altitude = -1.000000 (km)  
PMD Apogee Altitude = -1.000000 (km)  
PMD Inclination = 0.000000 (deg)  
PMD RAAN = 0.000000 (deg)  
PMD Argument of Perigee = 0.000000 (deg)  
PMD Mean Anomaly = 0.000000 (deg)

\*\*OUTPUT\*\*

Collision Probability = 0.000000  
Returned Error Message: Normal Processing  
Date Range Error Message: Normal Date Range  
Status = Pass

=====

\*\*INPUT\*\*

Space Structure Name = 18270  
Space Structure Type = Payload  
Perigee Altitude = 500.000000 (km)  
Apogee Altitude = 500.000000 (km)

Inclination = 85.000000 (deg)  
RAAN = 0.000000 (deg)  
Argument of Perigee = 0.000000 (deg)  
Mean Anomaly = 0.000000 (deg)  
Final Area-To-Mass Ratio = 0.069232 (m<sup>2</sup>/kg)  
Start Year = 2017.000000 (yr)  
Initial Mass = 0.000740 (kg)  
Final Mass = 0.000740 (kg)  
Duration = 1.000000 (yr)  
Station-Kept = False  
Abandoned = True  
PMD Perigee Altitude = -1.000000 (km)  
PMD Apogee Altitude = -1.000000 (km)  
PMD Inclination = 0.000000 (deg)  
PMD RAAN = 0.000000 (deg)  
PMD Argument of Perigee = 0.000000 (deg)  
PMD Mean Anomaly = 0.000000 (deg)

\*\*OUTPUT\*\*

Collision Probability = 0.000000  
Returned Error Message: Normal Processing  
Date Range Error Message: Normal Date Range  
Status = Pass

=====

\*\*INPUT\*\*

Space Structure Name = Laser  
Space Structure Type = Payload  
Perigee Altitude = 500.000000 (km)  
Apogee Altitude = 500.000000 (km)  
Inclination = 85.000000 (deg)  
RAAN = 0.000000 (deg)  
Argument of Perigee = 0.000000 (deg)  
Mean Anomaly = 0.000000 (deg)  
Final Area-To-Mass Ratio = 0.013244 (m<sup>2</sup>/kg)  
Start Year = 2017.000000 (yr)  
Initial Mass = 0.352850 (kg)  
Final Mass = 0.352850 (kg)  
Duration = 1.000000 (yr)  
Station-Kept = False  
Abandoned = True  
PMD Perigee Altitude = -1.000000 (km)  
PMD Apogee Altitude = -1.000000 (km)  
PMD Inclination = 0.000000 (deg)  
PMD RAAN = 0.000000 (deg)  
PMD Argument of Perigee = 0.000000 (deg)  
PMD Mean Anomaly = 0.000000 (deg)

\*\*OUTPUT\*\*

Collision Probability = 0.000000  
Returned Error Message: Normal Processing  
Date Range Error Message: Normal Date Range  
Status = Pass

=====

\*\*INPUT\*\*

Space Structure Name = STIM 210 IMU  
Space Structure Type = Payload  
Perigee Altitude = 500.000000 (km)  
Apogee Altitude = 500.000000 (km)  
Inclination = 85.000000 (deg)  
RAAN = 0.000000 (deg)  
Argument of Perigee = 0.000000 (deg)  
Mean Anomaly = 0.000000 (deg)  
Final Area-To-Mass Ratio = 0.017272 (m<sup>2</sup>/kg)  
Start Year = 2017.000000 (yr)  
Initial Mass = 0.115730 (kg)  
Final Mass = 0.115730 (kg)  
Duration = 1.000000 (yr)  
Station-Kept = False  
Abandoned = True  
PMD Perigee Altitude = -1.000000 (km)  
PMD Apogee Altitude = -1.000000 (km)  
PMD Inclination = 0.000000 (deg)  
PMD RAAN = 0.000000 (deg)  
PMD Argument of Perigee = 0.000000 (deg)  
PMD Mean Anomaly = 0.000000 (deg)

\*\*OUTPUT\*\*

Collision Probability = 0.000000  
Returned Error Message: Normal Processing  
Date Range Error Message: Normal Date Range  
Status = Pass

=====

===== End of Requirement 4.5-1 =====  
09 01 2016; 10:28:12AM Requirement 4.5-2: Compliant  
09 01 2016; 10:28:13AM Processing Requirement 4.6 Return Status :  
Passed

=====

Project Data

=====

\*\*INPUT\*\*



Space Structure Name = TOMSAT Complete  
Space Structure Type = Payload

Perigee Altitude = 500.000000 (km)  
Apogee Altitude = 500.000000 (km)  
Inclination = 85.000000 (deg)  
RAAN = 0.000000 (deg)  
Argument of Perigee = 0.000000 (deg)  
Mean Anomaly = 0.000000 (deg)  
Area-To-Mass Ratio = 0.010750 (m<sup>2</sup>/kg)  
Start Year = 2017.000000 (yr)  
Initial Mass = 5.000000 (kg)  
Final Mass = 5.000000 (kg)  
Duration = 1.000000 (yr)  
Station Kept = False  
Abandoned = True  
PMD Perigee Altitude = 495.686931 (km)  
PMD Apogee Altitude = 495.686931 (km)  
PMD Inclination = 84.999772 (deg)  
PMD RAAN = 117.071516 (deg)  
PMD Argument of Perigee = 318.092471 (deg)  
PMD Mean Anomaly = 0.000000 (deg)

\*\*OUTPUT\*\*

Suggested Perigee Altitude = 495.686931 (km)  
Suggested Apogee Altitude = 495.686931 (km)  
Returned Error Message = Passes LEO reentry orbit criteria.

Released Year = 2022 (yr)  
Requirement = 61  
Compliance Status = Pass

=====

\*\*INPUT\*\*

Space Structure Name = Full Body Only  
Space Structure Type = Payload

Perigee Altitude = 500.000000 (km)  
Apogee Altitude = 500.000000 (km)  
Inclination = 85.000000 (deg)  
RAAN = 0.000000 (deg)  
Argument of Perigee = 0.000000 (deg)  
Mean Anomaly = 0.000000 (deg)  
Area-To-Mass Ratio = 0.043712 (m<sup>2</sup>/kg)  
Start Year = 2017.000000 (yr)  
Initial Mass = 0.644200 (kg)  
Final Mass = 0.644200 (kg)  
Duration = 1.000000 (yr)  
Station Kept = False

Abandoned = True  
PMD Perigee Altitude = 468.646763 (km)  
PMD Apogee Altitude = 489.939472 (km)  
PMD Inclination = 84.996926 (deg)  
PMD RAAN = 115.952401 (deg)  
PMD Argument of Perigee = 53.726904 (deg)  
PMD Mean Anomaly = 0.000000 (deg)

\*\*OUTPUT\*\*

Suggested Perigee Altitude = 468.646763 (km)  
Suggested Apogee Altitude = 489.939472 (km)  
Returned Error Message = Passes LEO reentry orbit criteria.

Released Year = 2021 (yr)  
Requirement = 61  
Compliance Status = Pass

=====

\*\*INPUT\*\*

Space Structure Name = Frame Part 1 (2001)  
Space Structure Type = Payload

Perigee Altitude = 500.000000 (km)  
Apogee Altitude = 500.000000 (km)  
Inclination = 85.000000 (deg)  
RAAN = 0.000000 (deg)  
Argument of Perigee = 0.000000 (deg)  
Mean Anomaly = 0.000000 (deg)  
Area-To-Mass Ratio = 0.043995 (m<sup>2</sup>/kg)  
Start Year = 2017.000000 (yr)  
Initial Mass = 0.298130 (kg)  
Final Mass = 0.298130 (kg)  
Duration = 1.000000 (yr)  
Station Kept = False  
Abandoned = True  
PMD Perigee Altitude = 468.577491 (km)  
PMD Apogee Altitude = 489.661822 (km)  
PMD Inclination = 84.996897 (deg)  
PMD RAAN = 115.941185 (deg)  
PMD Argument of Perigee = 54.198981 (deg)  
PMD Mean Anomaly = 0.000000 (deg)

\*\*OUTPUT\*\*

Suggested Perigee Altitude = 468.577491 (km)  
Suggested Apogee Altitude = 489.661822 (km)  
Returned Error Message = Passes LEO reentry orbit criteria.

Released Year = 2021 (yr)

Requirement = 61  
Compliance Status = Pass

=====

\*\*INPUT\*\*

Space Structure Name = Frame Part 2 (4001)  
Space Structure Type = Payload

Perigee Altitude = 500.000000 (km)  
Apogee Altitude = 500.000000 (km)  
Inclination = 85.000000 (deg)  
RAAN = 0.000000 (deg)  
Argument of Perigee = 0.000000 (deg)  
Mean Anomaly = 0.000000 (deg)  
Area-To-Mass Ratio = 0.054262 (m<sup>2</sup>/kg)  
Start Year = 2017.000000 (yr)  
Initial Mass = 0.346070 (kg)  
Final Mass = 0.346070 (kg)  
Duration = 1.000000 (yr)  
Station Kept = False  
Abandoned = True  
PMD Perigee Altitude = 461.762281 (km)  
PMD Apogee Altitude = 483.739555 (km)  
PMD Inclination = 84.995810 (deg)  
PMD RAAN = 115.536667 (deg)  
PMD Argument of Perigee = 55.535161 (deg)  
PMD Mean Anomaly = 0.000000 (deg)

\*\*OUTPUT\*\*

Suggested Perigee Altitude = 461.762281 (km)  
Suggested Apogee Altitude = 483.739555 (km)  
Returned Error Message = Passes LEO reentry orbit criteria.

Released Year = 2020 (yr)  
Requirement = 61  
Compliance Status = Pass

=====

\*\*INPUT\*\*

Space Structure Name = Bus Electronics  
Space Structure Type = Payload

Perigee Altitude = 500.000000 (km)  
Apogee Altitude = 500.000000 (km)  
Inclination = 85.000000 (deg)  
RAAN = 0.000000 (deg)  
Argument of Perigee = 0.000000 (deg)

Mean Anomaly = 0.000000 (deg)  
Area-To-Mass Ratio = 0.012604 (m<sup>2</sup>/kg)  
Start Year = 2017.000000 (yr)  
Initial Mass = 0.627060 (kg)  
Final Mass = 0.627060 (kg)  
Duration = 1.000000 (yr)  
Station Kept = False  
Abandoned = True  
PMD Perigee Altitude = 482.026152 (km)  
PMD Apogee Altitude = 507.678154 (km)  
PMD Inclination = 84.999622 (deg)  
PMD RAAN = 117.012477 (deg)  
PMD Argument of Perigee = 115.000663 (deg)  
PMD Mean Anomaly = 0.000000 (deg)

\*\*OUTPUT\*\*

Suggested Perigee Altitude = 482.026152 (km)  
Suggested Apogee Altitude = 507.678154 (km)  
Returned Error Message = Passes LEO reentry orbit criteria.

Released Year = 2022 (yr)  
Requirement = 61  
Compliance Status = Pass

=====

\*\*INPUT\*\*

Space Structure Name = Reaction Wheels  
Space Structure Type = Payload  
  
Perigee Altitude = 500.000000 (km)  
Apogee Altitude = 500.000000 (km)  
Inclination = 85.000000 (deg)  
RAAN = 0.000000 (deg)  
Argument of Perigee = 0.000000 (deg)  
Mean Anomaly = 0.000000 (deg)  
Area-To-Mass Ratio = 0.012760 (m<sup>2</sup>/kg)  
Start Year = 2017.000000 (yr)  
Initial Mass = 0.225010 (kg)  
Final Mass = 0.225010 (kg)  
Duration = 1.000000 (yr)  
Station Kept = False  
Abandoned = True  
PMD Perigee Altitude = 481.215873 (km)  
PMD Apogee Altitude = 508.350501 (km)  
PMD Inclination = 84.999609 (deg)  
PMD RAAN = 117.007563 (deg)  
PMD Argument of Perigee = 107.921015 (deg)  
PMD Mean Anomaly = 0.000000 (deg)

\*\*OUTPUT\*\*

Suggested Perigee Altitude = 481.215873 (km)  
Suggested Apogee Altitude = 508.350501 (km)  
Returned Error Message = Passes LEO reentry orbit criteria.

Released Year = 2022 (yr)  
Requirement = 61  
Compliance Status = Pass

=====

\*\*INPUT\*\*

Space Structure Name = Star Tracker  
Space Structure Type = Payload

Perigee Altitude = 500.000000 (km)  
Apogee Altitude = 500.000000 (km)  
Inclination = 85.000000 (deg)  
RAAN = 0.000000 (deg)  
Argument of Perigee = 0.000000 (deg)  
Mean Anomaly = 0.000000 (deg)  
Area-To-Mass Ratio = 0.017369 (m<sup>2</sup>/kg)  
Start Year = 2017.000000 (yr)  
Initial Mass = 0.054800 (kg)  
Final Mass = 0.054800 (kg)  
Duration = 1.000000 (yr)  
Station Kept = False  
Abandoned = True  
PMD Perigee Altitude = 478.731489 (km)  
PMD Apogee Altitude = 506.748387 (km)  
PMD Inclination = 84.999250 (deg)  
PMD RAAN = 116.862533 (deg)  
PMD Argument of Perigee = 79.455494 (deg)  
PMD Mean Anomaly = 0.000000 (deg)

\*\*OUTPUT\*\*

Suggested Perigee Altitude = 478.731489 (km)  
Suggested Apogee Altitude = 506.748387 (km)  
Returned Error Message = Passes LEO reentry orbit criteria.

Released Year = 2022 (yr)  
Requirement = 61  
Compliance Status = Pass

=====

\*\*INPUT\*\*

Space Structure Name = Single Torque Rod

Space Structure Type = Payload

Perigee Altitude = 500.000000 (km)  
Apogee Altitude = 500.000000 (km)  
Inclination = 85.000000 (deg)  
RAAN = 0.000000 (deg)  
Argument of Perigee = 0.000000 (deg)  
Mean Anomaly = 0.000000 (deg)  
Area-To-Mass Ratio = 0.034161 (m<sup>2</sup>/kg)  
Start Year = 2017.000000 (yr)  
Initial Mass = 0.018440 (kg)  
Final Mass = 0.018440 (kg)  
Duration = 1.000000 (yr)  
Station Kept = False  
Abandoned = True  
PMD Perigee Altitude = 472.350969 (km)  
PMD Apogee Altitude = 496.860116 (km)  
PMD Inclination = 84.997832 (deg)  
PMD RAAN = 116.302615 (deg)  
PMD Argument of Perigee = 59.246940 (deg)  
PMD Mean Anomaly = 0.000000 (deg)

\*\*OUTPUT\*\*

Suggested Perigee Altitude = 472.350969 (km)  
Suggested Apogee Altitude = 496.860116 (km)  
Returned Error Message = Passes LEO reentry orbit criteria.

Released Year = 2021 (yr)  
Requirement = 61  
Compliance Status = Pass

=====

\*\*INPUT\*\*

Space Structure Name = Zenith Lid Assembly  
Space Structure Type = Payload

Perigee Altitude = 500.000000 (km)  
Apogee Altitude = 500.000000 (km)  
Inclination = 85.000000 (deg)  
RAAN = 0.000000 (deg)  
Argument of Perigee = 0.000000 (deg)  
Mean Anomaly = 0.000000 (deg)  
Area-To-Mass Ratio = 0.035553 (m<sup>2</sup>/kg)  
Start Year = 2017.000000 (yr)  
Initial Mass = 0.141260 (kg)  
Final Mass = 0.141260 (kg)  
Duration = 1.000000 (yr)  
Station Kept = False  
Abandoned = True

PMD Perigee Altitude = 469.607548 (km)  
PMD Apogee Altitude = 498.182718 (km)  
PMD Inclination = 84.997699 (deg)  
PMD RAAN = 116.254658 (deg)  
PMD Argument of Perigee = 83.189185 (deg)  
PMD Mean Anomaly = 0.000000 (deg)

\*\*OUTPUT\*\*

Suggested Perigee Altitude = 469.607548 (km)  
Suggested Apogee Altitude = 498.182718 (km)  
Returned Error Message = Passes LEO reentry orbit criteria.

Released Year = 2021 (yr)  
Requirement = 61  
Compliance Status = Pass

=====

\*\*INPUT\*\*

Space Structure Name = Antenna Zenith  
Space Structure Type = Payload

Perigee Altitude = 500.000000 (km)  
Apogee Altitude = 500.000000 (km)  
Inclination = 85.000000 (deg)  
RAAN = 0.000000 (deg)  
Argument of Perigee = 0.000000 (deg)  
Mean Anomaly = 0.000000 (deg)  
Area-To-Mass Ratio = 0.037208 (m<sup>2</sup>/kg)  
Start Year = 2017.000000 (yr)  
Initial Mass = 0.015900 (kg)  
Final Mass = 0.015900 (kg)  
Duration = 1.000000 (yr)  
Station Kept = False  
Abandoned = True  
PMD Perigee Altitude = 470.864268 (km)  
PMD Apogee Altitude = 495.095091 (km)  
PMD Inclination = 84.997552 (deg)  
PMD RAAN = 116.193838 (deg)  
PMD Argument of Perigee = 57.988189 (deg)  
PMD Mean Anomaly = 0.000000 (deg)

\*\*OUTPUT\*\*

Suggested Perigee Altitude = 470.864268 (km)  
Suggested Apogee Altitude = 495.095091 (km)  
Returned Error Message = Passes LEO reentry orbit criteria.

Released Year = 2021 (yr)  
Requirement = 61

Compliance Status = Pass

=====

\*\*INPUT\*\*

Space Structure Name = Wing Assembly  
Space Structure Type = Payload

Perigee Altitude = 500.000000 (km)  
Apogee Altitude = 500.000000 (km)  
Inclination = 85.000000 (deg)  
RAAN = 0.000000 (deg)  
Argument of Perigee = 0.000000 (deg)  
Mean Anomaly = 0.000000 (deg)  
Area-To-Mass Ratio = 0.054297 (m<sup>2</sup>/kg)  
Start Year = 2017.000000 (yr)  
Initial Mass = 0.148130 (kg)  
Final Mass = 0.148130 (kg)  
Duration = 1.000000 (yr)  
Station Kept = False  
Abandoned = True  
PMD Perigee Altitude = 461.742688 (km)  
PMD Apogee Altitude = 483.712948 (km)  
PMD Inclination = 84.995806 (deg)  
PMD RAAN = 115.535235 (deg)  
PMD Argument of Perigee = 55.524546 (deg)  
PMD Mean Anomaly = 0.000000 (deg)

\*\*OUTPUT\*\*

Suggested Perigee Altitude = 461.742688 (km)  
Suggested Apogee Altitude = 483.712948 (km)  
Returned Error Message = Passes LEO reentry orbit criteria.  
  
Released Year = 2020 (yr)  
Requirement = 61  
Compliance Status = Pass

=====

\*\*INPUT\*\*

Space Structure Name = Nadir Lid Assembly  
Space Structure Type = Payload

Perigee Altitude = 500.000000 (km)  
Apogee Altitude = 500.000000 (km)  
Inclination = 85.000000 (deg)  
RAAN = 0.000000 (deg)  
Argument of Perigee = 0.000000 (deg)  
Mean Anomaly = 0.000000 (deg)



Area-To-Mass Ratio = 0.044140 (m<sup>2</sup>/kg)  
Start Year = 2017.000000 (yr)  
Initial Mass = 0.099590 (kg)  
Final Mass = 0.099590 (kg)  
Duration = 1.000000 (yr)  
Station Kept = False  
Abandoned = True  
PMD Perigee Altitude = 468.535275 (km)  
PMD Apogee Altitude = 489.525424 (km)  
PMD Inclination = 84.996882 (deg)  
PMD RAAN = 115.935407 (deg)  
PMD Argument of Perigee = 54.471031 (deg)  
PMD Mean Anomaly = 0.000000 (deg)

\*\*OUTPUT\*\*

Suggested Perigee Altitude = 468.535275 (km)  
Suggested Apogee Altitude = 489.525424 (km)  
Returned Error Message = Passes LEO reentry orbit criteria.

Released Year = 2021 (yr)  
Requirement = 61  
Compliance Status = Pass

=====

\*\*INPUT\*\*

Space Structure Name = Antenna Nadir  
Space Structure Type = Payload

Perigee Altitude = 500.000000 (km)  
Apogee Altitude = 500.000000 (km)  
Inclination = 85.000000 (deg)  
RAAN = 0.000000 (deg)  
Argument of Perigee = 0.000000 (deg)  
Mean Anomaly = 0.000000 (deg)  
Area-To-Mass Ratio = 0.037208 (m<sup>2</sup>/kg)  
Start Year = 2017.000000 (yr)  
Initial Mass = 0.015900 (kg)  
Final Mass = 0.015900 (kg)  
Duration = 1.000000 (yr)  
Station Kept = False  
Abandoned = True  
PMD Perigee Altitude = 470.864268 (km)  
PMD Apogee Altitude = 495.095091 (km)  
PMD Inclination = 84.997552 (deg)  
PMD RAAN = 116.193838 (deg)  
PMD Argument of Perigee = 57.988189 (deg)  
PMD Mean Anomaly = 0.000000 (deg)

\*\*OUTPUT\*\*

Suggested Perigee Altitude = 470.864268 (km)  
Suggested Apogee Altitude = 495.095091 (km)  
Returned Error Message = Passes LEO reentry orbit criteria.

Released Year = 2021 (yr)  
Requirement = 61  
Compliance Status = Pass

=====

\*\*INPUT\*\*

Space Structure Name = Uplink Receiver  
Space Structure Type = Payload

Perigee Altitude = 500.000000 (km)  
Apogee Altitude = 500.000000 (km)  
Inclination = 85.000000 (deg)  
RAAN = 0.000000 (deg)  
Argument of Perigee = 0.000000 (deg)  
Mean Anomaly = 0.000000 (deg)  
Area-To-Mass Ratio = 0.035037 (m<sup>2</sup>/kg)  
Start Year = 2017.000000 (yr)  
Initial Mass = 0.027960 (kg)  
Final Mass = 0.027960 (kg)  
Duration = 1.000000 (yr)  
Station Kept = False  
Abandoned = True  
PMD Perigee Altitude = 471.958304 (km)  
PMD Apogee Altitude = 496.329049 (km)  
PMD Inclination = 84.997753 (deg)  
PMD RAAN = 116.271669 (deg)  
PMD Argument of Perigee = 58.923553 (deg)  
PMD Mean Anomaly = 0.000000 (deg)

\*\*OUTPUT\*\*

Suggested Perigee Altitude = 471.958304 (km)  
Suggested Apogee Altitude = 496.329049 (km)  
Returned Error Message = Passes LEO reentry orbit criteria.

Released Year = 2021 (yr)  
Requirement = 61  
Compliance Status = Pass

=====

\*\*INPUT\*\*

Space Structure Name = Payload Assembly  
Space Structure Type = Payload

Perigee Altitude = 500.000000 (km)  
Apogee Altitude = 500.000000 (km)  
Inclination = 85.000000 (deg)  
RAAN = 0.000000 (deg)  
Argument of Perigee = 0.000000 (deg)  
Mean Anomaly = 0.000000 (deg)  
Area-To-Mass Ratio = 0.006980 (m<sup>2</sup>/kg)  
Start Year = 2017.000000 (yr)  
Initial Mass = 1.311400 (kg)  
Final Mass = 1.311400 (kg)  
Duration = 1.000000 (yr)  
Station Kept = False  
Abandoned = True  
PMD Perigee Altitude = 497.255163 (km)  
PMD Apogee Altitude = 497.255163 (km)  
PMD Inclination = 85.000051 (deg)  
PMD RAAN = 117.184596 (deg)  
PMD Argument of Perigee = 325.360616 (deg)  
PMD Mean Anomaly = 0.000000 (deg)

\*\*OUTPUT\*\*

Suggested Perigee Altitude = 497.255163 (km)  
Suggested Apogee Altitude = 497.255163 (km)  
Returned Error Message = Passes LEO reentry orbit criteria.  
  
Released Year = 2023 (yr)  
Requirement = 61  
Compliance Status = Pass

=====

\*\*INPUT\*\*

Space Structure Name = 18267  
Space Structure Type = Payload  
  
Perigee Altitude = 500.000000 (km)  
Apogee Altitude = 500.000000 (km)  
Inclination = 85.000000 (deg)  
RAAN = 0.000000 (deg)  
Argument of Perigee = 0.000000 (deg)  
Mean Anomaly = 0.000000 (deg)  
Area-To-Mass Ratio = 0.071468 (m<sup>2</sup>/kg)  
Start Year = 2017.000000 (yr)  
Initial Mass = 0.016690 (kg)  
Final Mass = 0.016690 (kg)  
Duration = 1.000000 (yr)  
Station Kept = False  
Abandoned = True  
PMD Perigee Altitude = 451.047016 (km)

PMD Apogee Altitude = 468.362329 (km)  
PMD Inclination = 84.993665 (deg)  
PMD RAAN = 114.765990 (deg)  
PMD Argument of Perigee = 52.545834 (deg)  
PMD Mean Anomaly = 0.000000 (deg)

\*\*OUTPUT\*\*

Suggested Perigee Altitude = 451.047016 (km)  
Suggested Apogee Altitude = 468.362329 (km)  
Returned Error Message = Passes LEO reentry orbit criteria.

Released Year = 2019 (yr)  
Requirement = 61  
Compliance Status = Pass

=====

\*\*INPUT\*\*

Space Structure Name = 18265  
Space Structure Type = Payload  
  
Perigee Altitude = 500.000000 (km)  
Apogee Altitude = 500.000000 (km)  
Inclination = 85.000000 (deg)  
RAAN = 0.000000 (deg)  
Argument of Perigee = 0.000000 (deg)  
Mean Anomaly = 0.000000 (deg)  
Area-To-Mass Ratio = 0.048369 (m<sup>2</sup>/kg)  
Start Year = 2017.000000 (yr)  
Initial Mass = 0.010920 (kg)  
Final Mass = 0.010920 (kg)  
Duration = 1.000000 (yr)  
Station Kept = False  
Abandoned = True  
PMD Perigee Altitude = 465.553037 (km)  
PMD Apogee Altitude = 487.465190 (km)  
PMD Inclination = 84.996451 (deg)  
PMD RAAN = 115.773650 (deg)  
PMD Argument of Perigee = 56.503349 (deg)  
PMD Mean Anomaly = 0.000000 (deg)

\*\*OUTPUT\*\*

Suggested Perigee Altitude = 465.553037 (km)  
Suggested Apogee Altitude = 487.465190 (km)  
Returned Error Message = Passes LEO reentry orbit criteria.

Released Year = 2021 (yr)  
Requirement = 61  
Compliance Status = Pass

=====

\*\*INPUT\*\*

Space Structure Name = 18264  
Space Structure Type = Payload  
  
Perigee Altitude = 500.000000 (km)  
Apogee Altitude = 500.000000 (km)  
Inclination = 85.000000 (deg)  
RAAN = 0.000000 (deg)  
Argument of Perigee = 0.000000 (deg)  
Mean Anomaly = 0.000000 (deg)  
Area-To-Mass Ratio = 0.067530 (m<sup>2</sup>/kg)  
Start Year = 2017.000000 (yr)  
Initial Mass = 0.012320 (kg)  
Final Mass = 0.012320 (kg)  
Duration = 1.000000 (yr)  
Station Kept = False  
Abandoned = True  
PMD Perigee Altitude = 453.844947 (km)  
PMD Apogee Altitude = 472.220930 (km)  
PMD Inclination = 84.994207 (deg)  
PMD RAAN = 114.955495 (deg)  
PMD Argument of Perigee = 51.292470 (deg)  
PMD Mean Anomaly = 0.000000 (deg)

\*\*OUTPUT\*\*

Suggested Perigee Altitude = 453.844947 (km)  
Suggested Apogee Altitude = 472.220930 (km)  
Returned Error Message = Passes LEO reentry orbit criteria.  
  
Released Year = 2019 (yr)  
Requirement = 61  
Compliance Status = Pass

=====

\*\*INPUT\*\*

Space Structure Name = 18268  
Space Structure Type = Payload  
  
Perigee Altitude = 500.000000 (km)  
Apogee Altitude = 500.000000 (km)  
Inclination = 85.000000 (deg)  
RAAN = 0.000000 (deg)  
Argument of Perigee = 0.000000 (deg)  
Mean Anomaly = 0.000000 (deg)  
Area-To-Mass Ratio = 0.024134 (m<sup>2</sup>/kg)

Start Year = 2017.000000 (yr)  
Initial Mass = 0.004040 (kg)  
Final Mass = 0.004040 (kg)  
Duration = 1.000000 (yr)  
Station Kept = False  
Abandoned = True  
PMD Perigee Altitude = 476.528340 (km)  
PMD Apogee Altitude = 502.715992 (km)  
PMD Inclination = 84.998701 (deg)  
PMD RAAN = 116.644339 (deg)  
PMD Argument of Perigee = 62.152522 (deg)  
PMD Mean Anomaly = 0.000000 (deg)

\*\*OUTPUT\*\*

Suggested Perigee Altitude = 476.528340 (km)  
Suggested Apogee Altitude = 502.715992 (km)  
Returned Error Message = Passes LEO reentry orbit criteria.

Released Year = 2021 (yr)  
Requirement = 61  
Compliance Status = Pass

=====

\*\*INPUT\*\*

Space Structure Name = 18266  
Space Structure Type = Payload  
  
Perigee Altitude = 500.000000 (km)  
Apogee Altitude = 500.000000 (km)  
Inclination = 85.000000 (deg)  
RAAN = 0.000000 (deg)  
Argument of Perigee = 0.000000 (deg)  
Mean Anomaly = 0.000000 (deg)  
Area-To-Mass Ratio = 0.015816 (m<sup>2</sup>/kg)  
Start Year = 2017.000000 (yr)  
Initial Mass = 0.010280 (kg)  
Final Mass = 0.010280 (kg)  
Duration = 1.000000 (yr)  
Station Kept = False  
Abandoned = True  
PMD Perigee Altitude = 479.508900 (km)  
PMD Apogee Altitude = 507.372265 (km)  
PMD Inclination = 84.999372 (deg)  
PMD RAAN = 116.911912 (deg)  
PMD Argument of Perigee = 100.598747 (deg)  
PMD Mean Anomaly = 0.000000 (deg)

\*\*OUTPUT\*\*

Suggested Perigee Altitude = 479.508900 (km)  
Suggested Apogee Altitude = 507.372265 (km)  
Returned Error Message = Passes LEO reentry orbit criteria.

Released Year = 2022 (yr)  
Requirement = 61  
Compliance Status = Pass

=====

\*\*INPUT\*\*

Space Structure Name = 18263  
Space Structure Type = Payload

Perigee Altitude = 500.000000 (km)  
Apogee Altitude = 500.000000 (km)  
Inclination = 85.000000 (deg)  
RAAN = 0.000000 (deg)  
Argument of Perigee = 0.000000 (deg)  
Mean Anomaly = 0.000000 (deg)  
Area-To-Mass Ratio = 0.041225 (m<sup>2</sup>/kg)  
Start Year = 2017.000000 (yr)  
Initial Mass = 0.001280 (kg)  
Final Mass = 0.001280 (kg)  
Duration = 1.000000 (yr)  
Station Kept = False  
Abandoned = True  
PMD Perigee Altitude = 467.893971 (km)  
PMD Apogee Altitude = 493.627601 (km)  
PMD Inclination = 84.997169 (deg)  
PMD RAAN = 116.046948 (deg)  
PMD Argument of Perigee = 67.020697 (deg)  
PMD Mean Anomaly = 0.000000 (deg)

\*\*OUTPUT\*\*

Suggested Perigee Altitude = 467.893971 (km)  
Suggested Apogee Altitude = 493.627601 (km)  
Returned Error Message = Passes LEO reentry orbit criteria.

Released Year = 2021 (yr)  
Requirement = 61  
Compliance Status = Pass

=====

\*\*INPUT\*\*

Space Structure Name = 18270  
Space Structure Type = Payload

Perigee Altitude = 500.000000 (km)  
Apogee Altitude = 500.000000 (km)  
Inclination = 85.000000 (deg)  
RAAN = 0.000000 (deg)  
Argument of Perigee = 0.000000 (deg)  
Mean Anomaly = 0.000000 (deg)  
Area-To-Mass Ratio = 0.069232 (m<sup>2</sup>/kg)  
Start Year = 2017.000000 (yr)  
Initial Mass = 0.000740 (kg)  
Final Mass = 0.000740 (kg)  
Duration = 1.000000 (yr)  
Station Kept = False  
Abandoned = True  
PMD Perigee Altitude = 452.678558 (km)  
PMD Apogee Altitude = 470.563794 (km)  
PMD Inclination = 84.993977 (deg)  
PMD RAAN = 114.874519 (deg)  
PMD Argument of Perigee = 51.703181 (deg)  
PMD Mean Anomaly = 0.000000 (deg)

\*\*OUTPUT\*\*

Suggested Perigee Altitude = 452.678558 (km)  
Suggested Apogee Altitude = 470.563794 (km)  
Returned Error Message = Passes LEO reentry orbit criteria.

Released Year = 2019 (yr)  
Requirement = 61  
Compliance Status = Pass

=====

\*\*INPUT\*\*

Space Structure Name = Laser  
Space Structure Type = Payload  
  
Perigee Altitude = 500.000000 (km)  
Apogee Altitude = 500.000000 (km)  
Inclination = 85.000000 (deg)  
RAAN = 0.000000 (deg)  
Argument of Perigee = 0.000000 (deg)  
Mean Anomaly = 0.000000 (deg)  
Area-To-Mass Ratio = 0.013244 (m<sup>2</sup>/kg)  
Start Year = 2017.000000 (yr)  
Initial Mass = 0.352850 (kg)  
Final Mass = 0.352850 (kg)  
Duration = 1.000000 (yr)  
Station Kept = False  
Abandoned = True  
PMD Perigee Altitude = 481.060215 (km)  
PMD Apogee Altitude = 508.086133 (km)



PMD Inclination = 84.999572 (deg)  
PMD RAAN = 116.992558 (deg)  
PMD Argument of Perigee = 107.985415 (deg)  
PMD Mean Anomaly = 0.000000 (deg)

\*\*OUTPUT\*\*

Suggested Perigee Altitude = 481.060215 (km)  
Suggested Apogee Altitude = 508.086133 (km)  
Returned Error Message = Passes LEO reentry orbit criteria.

Released Year = 2022 (yr)  
Requirement = 61  
Compliance Status = Pass

=====

\*\*INPUT\*\*

Space Structure Name = STIM 210 IMU  
Space Structure Type = Payload

Perigee Altitude = 500.000000 (km)  
Apogee Altitude = 500.000000 (km)  
Inclination = 85.000000 (deg)  
RAAN = 0.000000 (deg)  
Argument of Perigee = 0.000000 (deg)  
Mean Anomaly = 0.000000 (deg)  
Area-To-Mass Ratio = 0.017272 (m<sup>2</sup>/kg)  
Start Year = 2017.000000 (yr)  
Initial Mass = 0.115730 (kg)  
Final Mass = 0.115730 (kg)  
Duration = 1.000000 (yr)  
Station Kept = False  
Abandoned = True  
PMD Perigee Altitude = 478.769001 (km)  
PMD Apogee Altitude = 506.797825 (km)  
PMD Inclination = 84.999257 (deg)  
PMD RAAN = 116.865604 (deg)  
PMD Argument of Perigee = 79.625801 (deg)  
PMD Mean Anomaly = 0.000000 (deg)

\*\*OUTPUT\*\*

Suggested Perigee Altitude = 478.769001 (km)  
Suggested Apogee Altitude = 506.797825 (km)  
Returned Error Message = Passes LEO reentry orbit criteria.

Released Year = 2022 (yr)  
Requirement = 61  
Compliance Status = Pass

=====

===== End of Requirement 4.6 =====

09 01 2016; 10:28:23AM \*\*\*\*\*Processing Requirement 4.7-1

Return Status : Passed

\*\*\*\*\*INPUT\*\*\*\*

Item Number = 1

name = TOMSAT Complete  
quantity = 1  
parent = 0  
materialID = 8  
type = Box  
Aero Mass = 5.000000  
Thermal Mass = 5.000000  
Diameter/Width = 0.100000  
Length = 0.330000  
Height = 0.100000

name = TOMSAT Complete  
quantity = 1  
parent = 1  
materialID = 8  
type = Box  
Aero Mass = 5.000000  
Thermal Mass = 5.000000  
Diameter/Width = 0.100000  
Length = 0.330000  
Height = 0.100000

\*\*\*\*\*OUTPUT\*\*\*\*

Item Number = 1

name = TOMSAT Complete  
Demise Altitude = 77.998191  
Debris Casualty Area = 0.000000  
Impact Kinetic Energy = 0.000000

\*\*\*\*\*

name = TOMSAT Complete  
Demise Altitude = 64.785566  
Debris Casualty Area = 0.000000  
Impact Kinetic Energy = 0.000000

\*\*\*\*\*

\*\*\*\*\*INPUT\*\*\*\*

Item Number = 2

name = Full Body Only  
quantity = 1

parent = 0  
materialID = 8  
type = Box  
Aero Mass = 0.644200  
Thermal Mass = 0.644200  
Diameter/Width = 0.110900  
Length = 0.340500  
Height = 0.103494

name = Full Body Only  
quantity = 1  
parent = 1  
materialID = 8  
type = Box  
Aero Mass = 0.644200  
Thermal Mass = 0.644200  
Diameter/Width = 0.110900  
Length = 0.340500  
Height = 0.103494

\*\*\*\*\*OUTPUT\*\*\*\*

Item Number = 2

name = Full Body Only  
Demise Altitude = 77.997058  
Debris Casualty Area = 0.000000  
Impact Kinetic Energy = 0.000000

\*\*\*\*\*

name = Full Body Only  
Demise Altitude = 71.855667  
Debris Casualty Area = 0.000000  
Impact Kinetic Energy = 0.000000

\*\*\*\*\*

\*\*\*\*\*INPUT\*\*\*\*

Item Number = 3

name = Frame Part 1 (2001)  
quantity = 1  
parent = 0  
materialID = 8  
type = Box  
Aero Mass = 0.298130  
Thermal Mass = 0.298130  
Diameter/Width = 0.110900  
Length = 0.130000  
Height = 0.103494

name = Frame Part 1 (2001)  
quantity = 1

parent = 1  
materialID = 8  
type = Box  
Aero Mass = 0.298130  
Thermal Mass = 0.298130  
Diameter/Width = 0.110900  
Length = 0.130000  
Height = 0.103494

\*\*\*\*\*OUTPUT\*\*\*\*

Item Number = 3

name = Frame Part 1 (2001)  
Demise Altitude = 77.988871  
Debris Casualty Area = 0.000000  
Impact Kinetic Energy = 0.000000

\*\*\*\*\*

name = Frame Part 1 (2001)  
Demise Altitude = 71.208589  
Debris Casualty Area = 0.000000  
Impact Kinetic Energy = 0.000000

\*\*\*\*\*

\*\*\*\*\*INPUT\*\*\*\*

Item Number = 4

name = Frame Part 2 (4001)  
quantity = 1  
parent = 0  
materialID = 8  
type = Box  
Aero Mass = 0.346070  
Thermal Mass = 0.346070  
Diameter/Width = 0.110900  
Length = 0.209230  
Height = 0.103494

name = Frame Part 2 (4001)  
quantity = 1  
parent = 1  
materialID = 8  
type = Box  
Aero Mass = 0.346070  
Thermal Mass = 0.346070  
Diameter/Width = 0.110900  
Length = 0.209230  
Height = 0.103494

\*\*\*\*\*OUTPUT\*\*\*\*

Item Number = 4

name = Frame Part 2 (4001)  
Demise Altitude = 77.989230  
Debris Casualty Area = 0.000000  
Impact Kinetic Energy = 0.000000

\*\*\*\*\*  
name = Frame Part 2 (4001)  
Demise Altitude = 72.054792  
Debris Casualty Area = 0.000000  
Impact Kinetic Energy = 0.000000

\*\*\*\*\*  
\*\*\*\*\*INPUT\*\*\*\*\*  
Item Number = 5

name = Bus Electronics  
quantity = 1  
parent = 0  
materialID = 23  
type = Box  
Aero Mass = 0.627060  
Thermal Mass = 0.627060  
Diameter/Width = 0.085046  
Length = 0.097050  
Height = 0.084882

name = Bus Electronics  
quantity = 1  
parent = 1  
materialID = 23  
type = Box  
Aero Mass = 0.627060  
Thermal Mass = 0.627060  
Diameter/Width = 0.085046  
Length = 0.097050  
Height = 0.084882

\*\*\*\*\*OUTPUT\*\*\*\*\*  
Item Number = 5

name = Bus Electronics  
Demise Altitude = 77.994972  
Debris Casualty Area = 0.000000  
Impact Kinetic Energy = 0.000000

\*\*\*\*\*  
name = Bus Electronics  
Demise Altitude = 71.933527  
Debris Casualty Area = 0.000000  
Impact Kinetic Energy = 0.000000

\*\*\*\*\*

\*\*\*\*\*INPUT\*\*\*\*

Item Number = 6

name = Reaction Wheels  
quantity = 1  
parent = 0  
materialID = 54  
type = Box  
Aero Mass = 0.225010  
Thermal Mass = 0.225010  
Diameter/Width = 0.052832  
Length = 0.071102  
Height = 0.039192

name = Reaction Wheels  
quantity = 1  
parent = 1  
materialID = 54  
type = Box  
Aero Mass = 0.225010  
Thermal Mass = 0.225010  
Diameter/Width = 0.052832  
Length = 0.071102  
Height = 0.039192

\*\*\*\*\*OUTPUT\*\*\*\*

Item Number = 6

name = Reaction Wheels  
Demise Altitude = 77.990707  
Debris Casualty Area = 0.000000  
Impact Kinetic Energy = 0.000000

\*\*\*\*\*

name = Reaction Wheels  
Demise Altitude = 0.000000  
Debris Casualty Area = 0.431908  
Impact Kinetic Energy = 149.537155

\*\*\*\*\*

\*\*\*\*\*INPUT\*\*\*\*

Item Number = 7

name = Star Tracker  
quantity = 2  
parent = 0  
materialID = 8  
type = Box

Aero Mass = 0.054800  
Thermal Mass = 0.054800  
Diameter/Width = 0.026670  
Length = 0.041829  
Height = 0.025400

name = Star Tracker  
quantity = 2  
parent = 1  
materialID = 8  
type = Box  
Aero Mass = 0.054800  
Thermal Mass = 0.054800  
Diameter/Width = 0.026670  
Length = 0.041829  
Height = 0.025400

\*\*\*\*\*OUTPUT\*\*\*\*

Item Number = 7

name = Star Tracker  
Demise Altitude = 77.995871  
Debris Casualty Area = 0.000000  
Impact Kinetic Energy = 0.000000

\*\*\*\*\*

name = Star Tracker  
Demise Altitude = 72.862183  
Debris Casualty Area = 0.000000  
Impact Kinetic Energy = 0.000000

\*\*\*\*\*

\*\*\*\*\*INPUT\*\*\*\*

Item Number = 8

name = Single Torque Rod  
quantity = 9  
parent = 0  
materialID = -1  
type = Cylinder  
Aero Mass = 0.018440  
Thermal Mass = 0.018440  
Diameter/Width = 0.008000

name = Single Torque Rod  
quantity = 9  
parent = 1  
materialID = -1  
type = Cylinder  
Aero Mass = 0.018440  
Thermal Mass = 0.018440

Diameter/Width = 0.008000  
Length = 0.078740

\*\*\*\*\*OUTPUT\*\*\*\*

Item Number = 8

name = Single Torque Rod  
Demise Altitude = 77.989894  
Debris Casualty Area = 0.000000  
Impact Kinetic Energy = 0.000000

\*\*\*\*\*

name = Single Torque Rod  
Demise Altitude = 74.688566  
Debris Casualty Area = 0.000000  
Impact Kinetic Energy = 0.000000

\*\*\*\*\*

\*\*\*\*\*INPUT\*\*\*\*

Item Number = 9

name = Zenith Lid Assembly  
quantity = 1  
parent = 0  
materialID = 8  
type = Box  
Aero Mass = 0.141260  
Thermal Mass = 0.141260  
Diameter/Width = 0.103494  
Length = 0.110900  
Height = 0.016740

name = Antenna Zenith sub  
quantity = 1  
parent = 1  
materialID = 40  
type = Box  
Aero Mass = 0.015900  
Thermal Mass = 0.015900  
Diameter/Width = 0.038099  
Length = 0.039878  
Height = 0.005000

\*\*\*\*\*OUTPUT\*\*\*\*

Item Number = 9

name = Zenith Lid Assembly  
Demise Altitude = 77.991676  
Debris Casualty Area = 0.000000  
Impact Kinetic Energy = 0.000000



\*\*\*\*\*

name = Antenna Zenith sub  
Demise Altitude = 71.860183  
Debris Casualty Area = 0.000000  
Impact Kinetic Energy = 0.000000

\*\*\*\*\*

\*\*\*\*\*INPUT\*\*\*\*

Item Number = 10

name = Antenna Zenith  
quantity = 1  
parent = 0  
materialID = 40  
type = Box  
Aero Mass = 0.015900  
Thermal Mass = 0.015900  
Diameter/Width = 0.038099  
Length = 0.039878  
Height = 0.005000

name = Antenna Zenith  
quantity = 1  
parent = 1  
materialID = 40  
type = Box  
Aero Mass = 0.015900  
Thermal Mass = 0.015900  
Diameter/Width = 0.038099  
Length = 0.039878  
Height = 0.005000

\*\*\*\*\*OUTPUT\*\*\*\*

Item Number = 10

name = Antenna Zenith  
Demise Altitude = 77.986887  
Debris Casualty Area = 0.000000  
Impact Kinetic Energy = 0.000000

\*\*\*\*\*

name = Antenna Zenith  
Demise Altitude = 68.422925  
Debris Casualty Area = 0.000000  
Impact Kinetic Energy = 0.000000

\*\*\*\*\*

\*\*\*\*\*INPUT\*\*\*\*

Item Number = 11

name = Wing Assembly  
quantity = 2  
parent = 0  
materialID = 8  
type = Flat Plate  
Aero Mass = 0.148130  
Thermal Mass = 0.148130  
Diameter/Width = 0.073660  
Length = 0.313500

name = Wing Assembly  
quantity = 2  
parent = 1  
materialID = 8  
type = Flat Plate  
Aero Mass = 0.148130  
Thermal Mass = 0.148130  
Diameter/Width = 0.073660  
Length = 0.313500

\*\*\*\*\*OUTPUT\*\*\*\*  
Item Number = 11

name = Wing Assembly  
Demise Altitude = 77.986988  
Debris Casualty Area = 0.000000  
Impact Kinetic Energy = 0.000000

\*\*\*\*\*  
name = Wing Assembly  
Demise Altitude = 66.885363  
Debris Casualty Area = 0.000000  
Impact Kinetic Energy = 0.000000

\*\*\*\*\*

\*\*\*\*\*INPUT\*\*\*\*  
Item Number = 12

name = Nadir Lid Assembly  
quantity = 1  
parent = 0  
materialID = 8  
type = Box  
Aero Mass = 0.099590  
Thermal Mass = 0.099590  
Diameter/Width = 0.103494  
Length = 0.110900  
Height = 0.007977

name = Antenna Nadir sub  
quantity = 1

parent = 1  
materialID = 40  
type = Box  
Aero Mass = 0.015900  
Thermal Mass = 0.015900  
Diameter/Width = 0.038099  
Length = 0.039878  
Height = 0.005000

\*\*\*\*\*OUTPUT\*\*\*\*

Item Number = 12

name = Nadir Lid Assembly  
Demise Altitude = 77.990613  
Debris Casualty Area = 0.000000  
Impact Kinetic Energy = 0.000000

\*\*\*\*\*

name = Antenna Nadir sub  
Demise Altitude = 68.977816  
Debris Casualty Area = 0.000000  
Impact Kinetic Energy = 0.000000

\*\*\*\*\*

\*\*\*\*\*INPUT\*\*\*\*

Item Number = 13

name = Antenna Nadir  
quantity = 1  
parent = 0  
materialID = 40  
type = Box  
Aero Mass = 0.015900  
Thermal Mass = 0.015900  
Diameter/Width = 0.038099  
Length = 0.039878  
Height = 0.005000

name = Antenna Nadir  
quantity = 1  
parent = 1  
materialID = 40  
type = Box  
Aero Mass = 0.015900  
Thermal Mass = 0.015900  
Diameter/Width = 0.038099  
Length = 0.039878  
Height = 0.005000

\*\*\*\*\*OUTPUT\*\*\*\*

Item Number = 13

name = Antenna Nadir  
Demise Altitude = 77.986887  
Debris Casualty Area = 0.000000  
Impact Kinetic Energy = 0.000000

\*\*\*\*\*  
name = Antenna Nadir  
Demise Altitude = 68.422925  
Debris Casualty Area = 0.000000  
Impact Kinetic Energy = 0.000000

\*\*\*\*\*  
\*\*\*\*\*INPUT\*\*\*\*\*  
Item Number = 14

name = Uplink Receiver  
quantity = 1  
parent = 0  
materialID = 8  
type = Cylinder  
Aero Mass = 0.027960  
Thermal Mass = 0.027960  
Diameter/Width = 0.031000

name = Uplink Receiver  
quantity = 1  
parent = 1  
materialID = 8  
type = Cylinder  
Aero Mass = 0.027960  
Thermal Mass = 0.027960  
Diameter/Width = 0.031000  
Length = 0.031601

\*\*\*\*\*OUTPUT\*\*\*\*\*  
Item Number = 14

name = Uplink Receiver  
Demise Altitude = 77.996660  
Debris Casualty Area = 0.000000  
Impact Kinetic Energy = 0.000000

\*\*\*\*\*  
name = Uplink Receiver  
Demise Altitude = 73.961207  
Debris Casualty Area = 0.000000  
Impact Kinetic Energy = 0.000000

\*\*\*\*\*

\*\*\*\*\*INPUT\*\*\*\*

Item Number = 15

name = Payload Assembly  
quantity = 1  
parent = 0  
materialID = 8  
type = Box  
Aero Mass = 1.311400  
Thermal Mass = 1.311400  
Diameter/Width = 0.082444  
Length = 0.164580  
Height = 0.056238

name = 18267 sub  
quantity = 1  
parent = 1  
materialID = 66  
type = Cylinder  
Aero Mass = 0.016690  
Thermal Mass = 0.016690  
Diameter/Width = 0.032000  
Length = 0.037275

name = 18265 sub  
quantity = 1  
parent = 1  
materialID = 66  
type = Cylinder  
Aero Mass = 0.010920  
Thermal Mass = 0.010920  
Diameter/Width = 0.016506  
Length = 0.032000

name = 18264 sub  
quantity = 1  
parent = 1  
materialID = 66  
type = Cylinder  
Aero Mass = 0.012320  
Thermal Mass = 0.012320  
Diameter/Width = 0.025999  
Length = 0.032000

name = 18268 sub  
quantity = 1  
parent = 1  
materialID = 66  
type = Cylinder  
Aero Mass = 0.004040  
Thermal Mass = 0.004040  
Diameter/Width = 0.025000

Length = 0.039000

name = 18266 sub  
quantity = 1  
parent = 1  
materialID = 66  
type = Cylinder  
Aero Mass = 0.010280  
Thermal Mass = 0.010280  
Diameter/Width = 0.050810  
Length = 0.032000

name = 18263 sub  
quantity = 1  
parent = 1  
materialID = 66  
type = Cylinder  
Aero Mass = 0.001280  
Thermal Mass = 0.001280  
Diameter/Width = 0.016490  
Length = 0.032000

name = 18270 sub  
quantity = 1  
parent = 1  
materialID = 66  
type = Cylinder  
Aero Mass = 0.000740  
Thermal Mass = 0.000740  
Diameter/Width = 0.016010  
Length = 0.032000

\*\*\*\*\*OUTPUT\*\*\*\*\*

Item Number = 15

name = Payload Assembly  
Demise Altitude = 77.999894  
Debris Casualty Area = 0.000000  
Impact Kinetic Energy = 0.000000

\*\*\*\*\*

name = 18267 sub  
Demise Altitude = 0.000000  
Debris Casualty Area = 0.402637  
Impact Kinetic Energy = 2.189265

\*\*\*\*\*

name = 18265 sub  
Demise Altitude = 0.000000  
Debris Casualty Area = 0.388107  
Impact Kinetic Energy = 2.261483

\*\*\*\*\*

name = 18264 sub  
Demise Altitude = 0.000000  
Debris Casualty Area = 0.395445  
Impact Kinetic Energy = 1.725148

\*\*\*\*\*

name = 18268 sub  
Demise Altitude = 0.000000  
Debris Casualty Area = 0.398445  
Impact Kinetic Energy = 0.163474

\*\*\*\*\*

name = 18266 sub  
Demise Altitude = 0.000000  
Debris Casualty Area = 0.410013  
Impact Kinetic Energy = 0.535933

\*\*\*\*\*

name = 18263 sub  
Demise Altitude = 0.000000  
Debris Casualty Area = 0.388093  
Impact Kinetic Energy = 0.031079

\*\*\*\*\*

name = 18270 sub  
Demise Altitude = 0.000000  
Debris Casualty Area = 0.387674  
Impact Kinetic Energy = 0.010730

\*\*\*\*\*

\*\*\*\*\*INPUT\*\*\*\*

Item Number = 16

name = 18267  
quantity = 1  
parent = 0  
materialID = 66  
type = Cylinder  
Aero Mass = 0.016690  
Thermal Mass = 0.016690  
Diameter/Width = 0.032000

name = 18267  
quantity = 1  
parent = 1  
materialID = 66  
type = Cylinder  
Aero Mass = 0.016690  
Thermal Mass = 0.016690  
Diameter/Width = 0.032000

Length = 0.037275

\*\*\*\*\*OUTPUT\*\*\*\*

Item Number = 16

name = 18267

Demise Altitude = 77.987199

Debris Casualty Area = 0.000000

Impact Kinetic Energy = 0.000000

\*\*\*\*\*

name = 18267

Demise Altitude = 0.000000

Debris Casualty Area = 0.402637

Impact Kinetic Energy = 2.189214

\*\*\*\*\*

\*\*\*\*\*INPUT\*\*\*\*

Item Number = 17

name = 18265

quantity = 1

parent = 0

materialID = 66

type = Cylinder

Aero Mass = 0.010920

Thermal Mass = 0.010920

Diameter/Width = 0.016506

name = 18265

quantity = 1

parent = 1

materialID = 66

type = Cylinder

Aero Mass = 0.010920

Thermal Mass = 0.010920

Diameter/Width = 0.016506

Length = 0.032000

\*\*\*\*\*OUTPUT\*\*\*\*

Item Number = 17

name = 18265

Demise Altitude = 77.992527

Debris Casualty Area = 0.000000

Impact Kinetic Energy = 0.000000

\*\*\*\*\*

name = 18265

Demise Altitude = 0.000000

Debris Casualty Area = 0.388107



Impact Kinetic Energy = 2.261433

\*\*\*\*\*

\*\*\*\*\*INPUT\*\*\*\*

Item Number = 18

name = 18264  
quantity = 1  
parent = 0  
materialID = 66  
type = Cylinder  
Aero Mass = 0.012320  
Thermal Mass = 0.012320  
Diameter/Width = 0.025999

name = 18264  
quantity = 1  
parent = 1  
materialID = 66  
type = Cylinder  
Aero Mass = 0.012320  
Thermal Mass = 0.012320  
Diameter/Width = 0.025999  
Length = 0.032000

\*\*\*\*\*OUTPUT\*\*\*\*

Item Number = 18

name = 18264  
Demise Altitude = 77.988996  
Debris Casualty Area = 0.000000  
Impact Kinetic Energy = 0.000000

\*\*\*\*\*

name = 18264  
Demise Altitude = 0.000000  
Debris Casualty Area = 0.395445  
Impact Kinetic Energy = 1.725225

\*\*\*\*\*

\*\*\*\*\*INPUT\*\*\*\*

Item Number = 19

name = 18268  
quantity = 1  
parent = 0  
materialID = 66  
type = Cylinder  
Aero Mass = 0.004040  
Thermal Mass = 0.004040

Diameter/Width = 0.025000

name = 18268  
quantity = 1  
parent = 1  
materialID = 66  
type = Cylinder  
Aero Mass = 0.004040  
Thermal Mass = 0.004040  
Diameter/Width = 0.025000  
Length = 0.039000

\*\*\*\*\*OUTPUT\*\*\*\*

Item Number = 19

name = 18268  
Demise Altitude = 77.990418  
Debris Casualty Area = 0.000000  
Impact Kinetic Energy = 0.000000

\*\*\*\*\*

name = 18268  
Demise Altitude = 0.000000  
Debris Casualty Area = 0.398445  
Impact Kinetic Energy = 0.163468

\*\*\*\*\*

\*\*\*\*\*INPUT\*\*\*\*

Item Number = 20

name = 18266  
quantity = 1  
parent = 0  
materialID = 66  
type = Cylinder  
Aero Mass = 0.010280  
Thermal Mass = 0.010280  
Diameter/Width = 0.050810

name = 18266  
quantity = 1  
parent = 1  
materialID = 66  
type = Cylinder  
Aero Mass = 0.010280  
Thermal Mass = 0.010280  
Diameter/Width = 0.050810  
Length = 0.032000

\*\*\*\*\*OUTPUT\*\*\*\*

Item Number = 20

name = 18266  
Demise Altitude = 77.987793  
Debris Casualty Area = 0.000000  
Impact Kinetic Energy = 0.000000

\*\*\*\*\*  
name = 18266  
Demise Altitude = 0.000000  
Debris Casualty Area = 0.410013  
Impact Kinetic Energy = 0.535938

\*\*\*\*\*  
\*\*\*\*\*INPUT\*\*\*\*\*  
Item Number = 21

name = 18263  
quantity = 1  
parent = 0  
materialID = 66  
type = Cylinder  
Aero Mass = 0.001280  
Thermal Mass = 0.001280  
Diameter/Width = 0.016490

name = 18263  
quantity = 1  
parent = 1  
materialID = 66  
type = Cylinder  
Aero Mass = 0.001280  
Thermal Mass = 0.001280  
Diameter/Width = 0.016490  
Length = 0.032000

\*\*\*\*\*OUTPUT\*\*\*\*\*  
Item Number = 21

name = 18263  
Demise Altitude = 77.998957  
Debris Casualty Area = 0.000000  
Impact Kinetic Energy = 0.000000

\*\*\*\*\*  
name = 18263  
Demise Altitude = 0.000000  
Debris Casualty Area = 0.388093  
Impact Kinetic Energy = 0.031079

\*\*\*\*\*

\*\*\*\*\*INPUT\*\*\*\*

Item Number = 22

name = 18270  
quantity = 1  
parent = 0  
materialID = 66  
type = Cylinder  
Aero Mass = 0.000740  
Thermal Mass = 0.000740  
Diameter/Width = 0.016010

name = 18270  
quantity = 1  
parent = 1  
materialID = 66  
type = Cylinder  
Aero Mass = 0.000740  
Thermal Mass = 0.000740  
Diameter/Width = 0.016010  
Length = 0.032000

\*\*\*\*\*OUTPUT\*\*\*\*

Item Number = 22

name = 18270  
Demise Altitude = 77.990207  
Debris Casualty Area = 0.000000  
Impact Kinetic Energy = 0.000000

\*\*\*\*\*

name = 18270  
Demise Altitude = 0.000000  
Debris Casualty Area = 0.387674  
Impact Kinetic Energy = 0.010731

\*\*\*\*\*

\*\*\*\*\*INPUT\*\*\*\*

Item Number = 23

name = Laser  
quantity = 1  
parent = 0  
materialID = 8  
type = Box  
Aero Mass = 0.352850  
Thermal Mass = 0.352850  
Diameter/Width = 0.082245  
Length = 0.120650  
Height = 0.020192

name = Laser  
quantity = 1  
parent = 1  
materialID = 8  
type = Box  
Aero Mass = 0.352850  
Thermal Mass = 0.352850  
Diameter/Width = 0.082245  
Length = 0.120650  
Height = 0.020192

\*\*\*\*\*OUTPUT\*\*\*\*

Item Number = 23

name = Laser  
Demise Altitude = 77.993863  
Debris Casualty Area = 0.000000  
Impact Kinetic Energy = 0.000000

\*\*\*\*\*

name = Laser  
Demise Altitude = 69.520269  
Debris Casualty Area = 0.000000  
Impact Kinetic Energy = 0.000000

\*\*\*\*\*

\*\*\*\*\*INPUT\*\*\*\*

Item Number = 24

name = STIM 210 IMU  
quantity = 1  
parent = 0  
materialID = 8  
type = Box  
Aero Mass = 0.115730  
Thermal Mass = 0.115730  
Diameter/Width = 0.048260  
Length = 0.064770  
Height = 0.025400

name = STIM 210 IMU  
quantity = 1  
parent = 1  
materialID = 8  
type = Box  
Aero Mass = 0.115730  
Thermal Mass = 0.115730  
Diameter/Width = 0.048260  
Length = 0.064770  
Height = 0.025400

\*\*\*\*\*OUTPUT\*\*\*\*

Item Number = 24

name = STIM 210 IMU  
Demise Altitude = 77.998129  
Debris Casualty Area = 0.000000  
Impact Kinetic Energy = 0.000000

\*\*\*\*\*

name = STIM 210 IMU  
Demise Altitude = 71.667113  
Debris Casualty Area = 0.000000  
Impact Kinetic Energy = 0.000000

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===== End of Requirement 4.7-1 =====